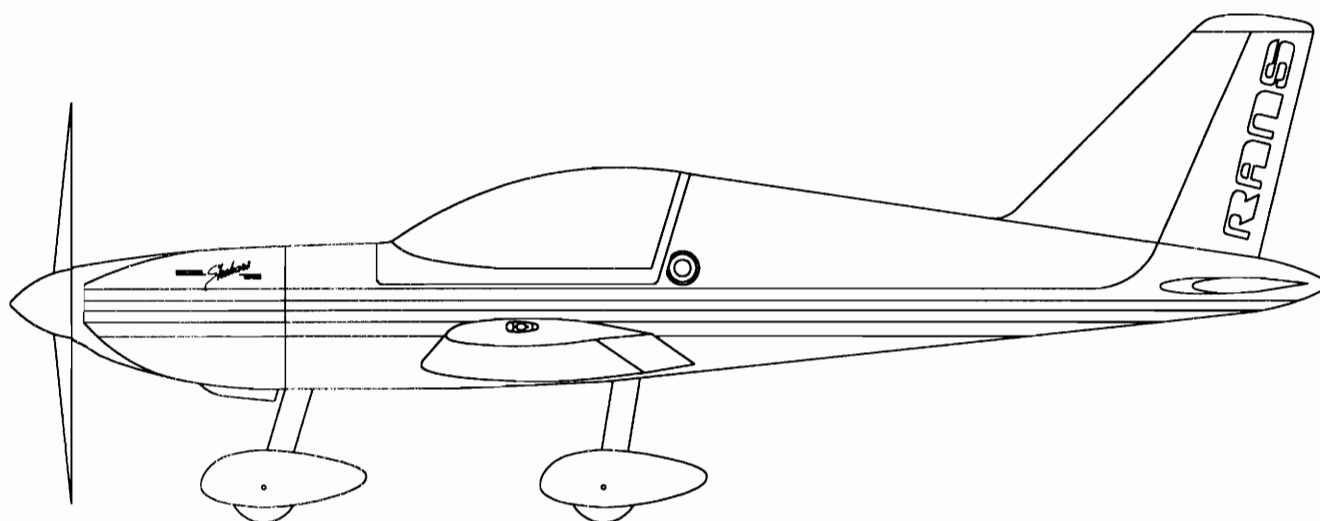


RANS

Shekari

S-16



RANS S-16 SHEKARI  
FIGURE DRAWINGS

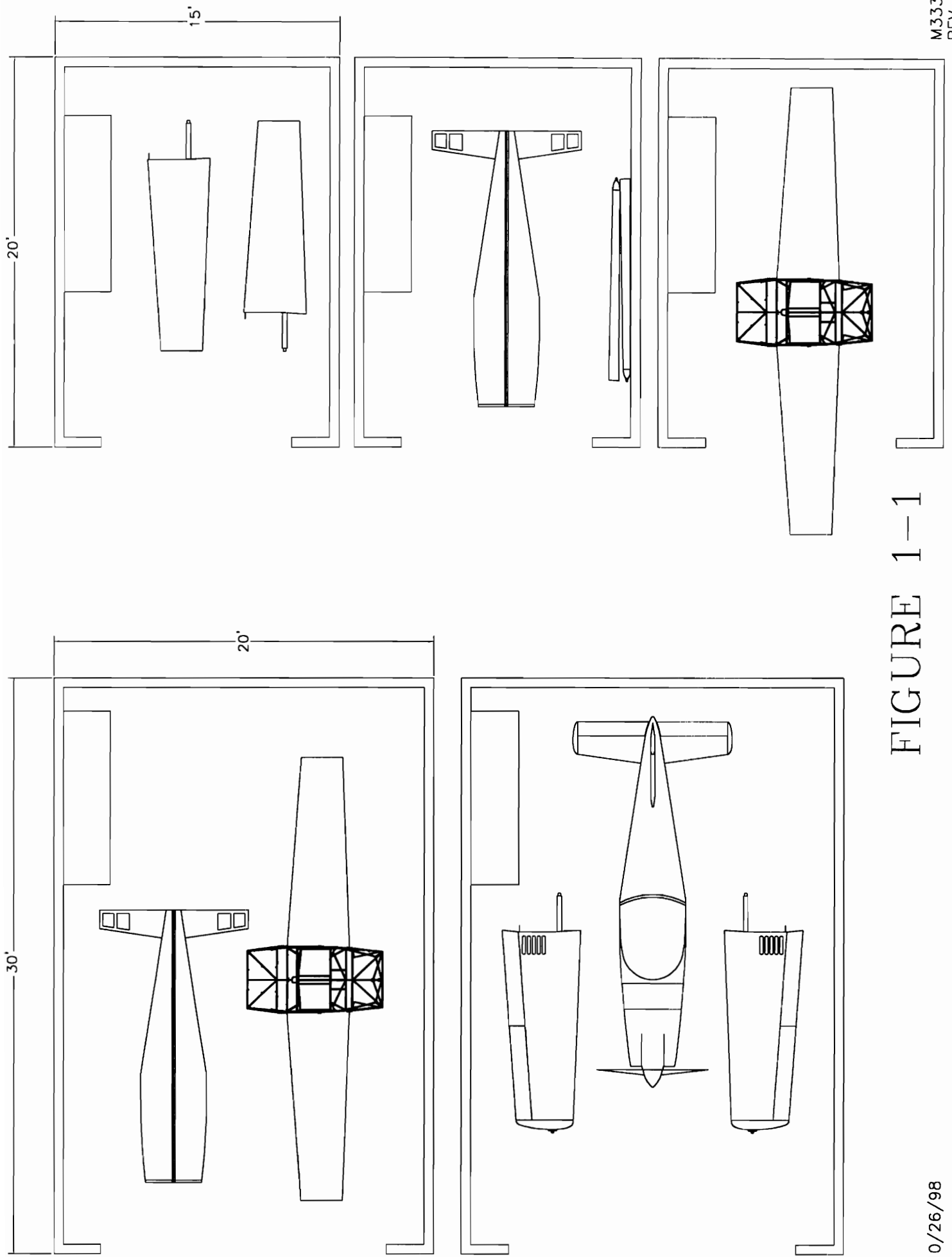


FIGURE 1-1

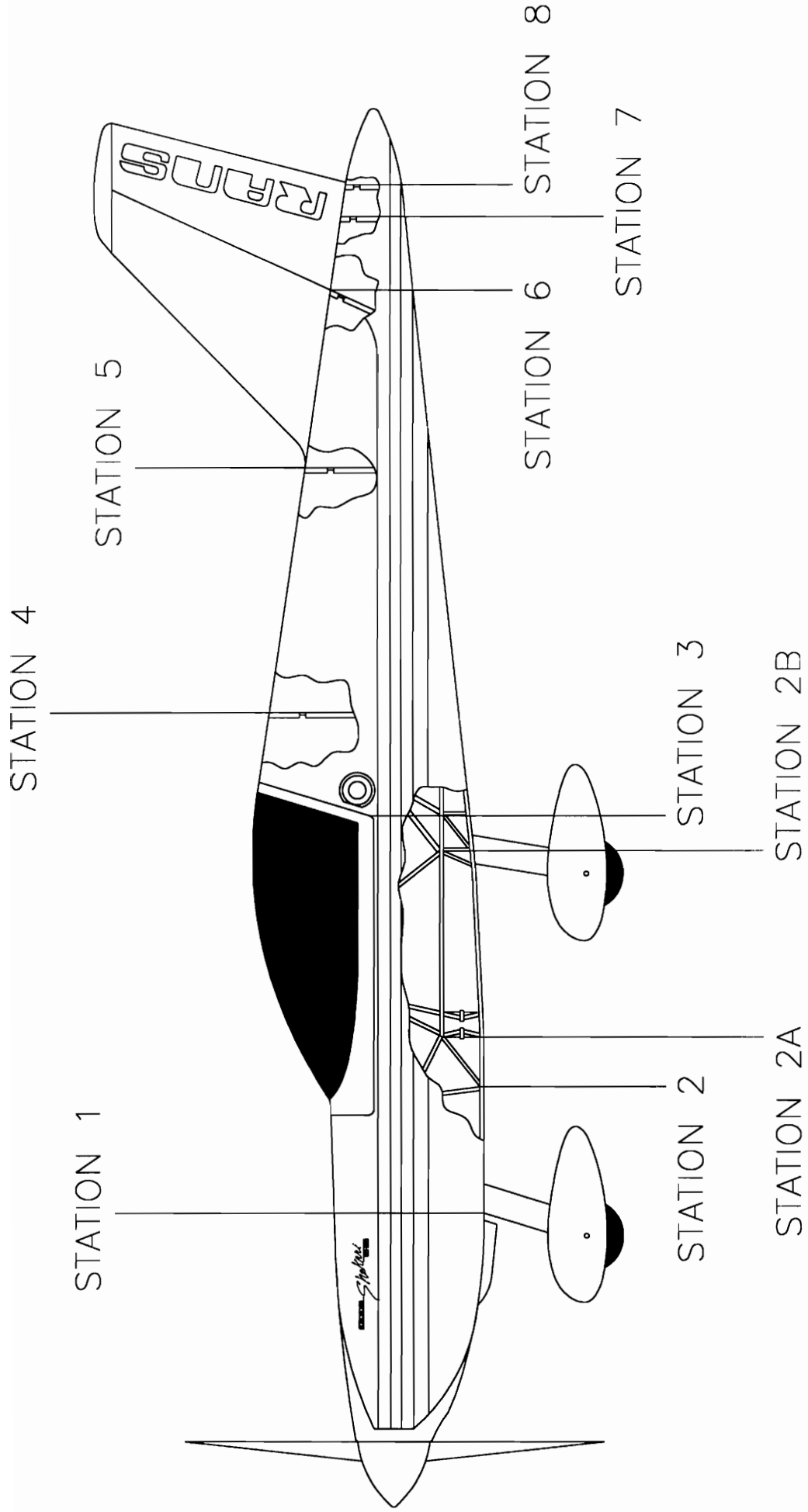


FIGURE 1-2

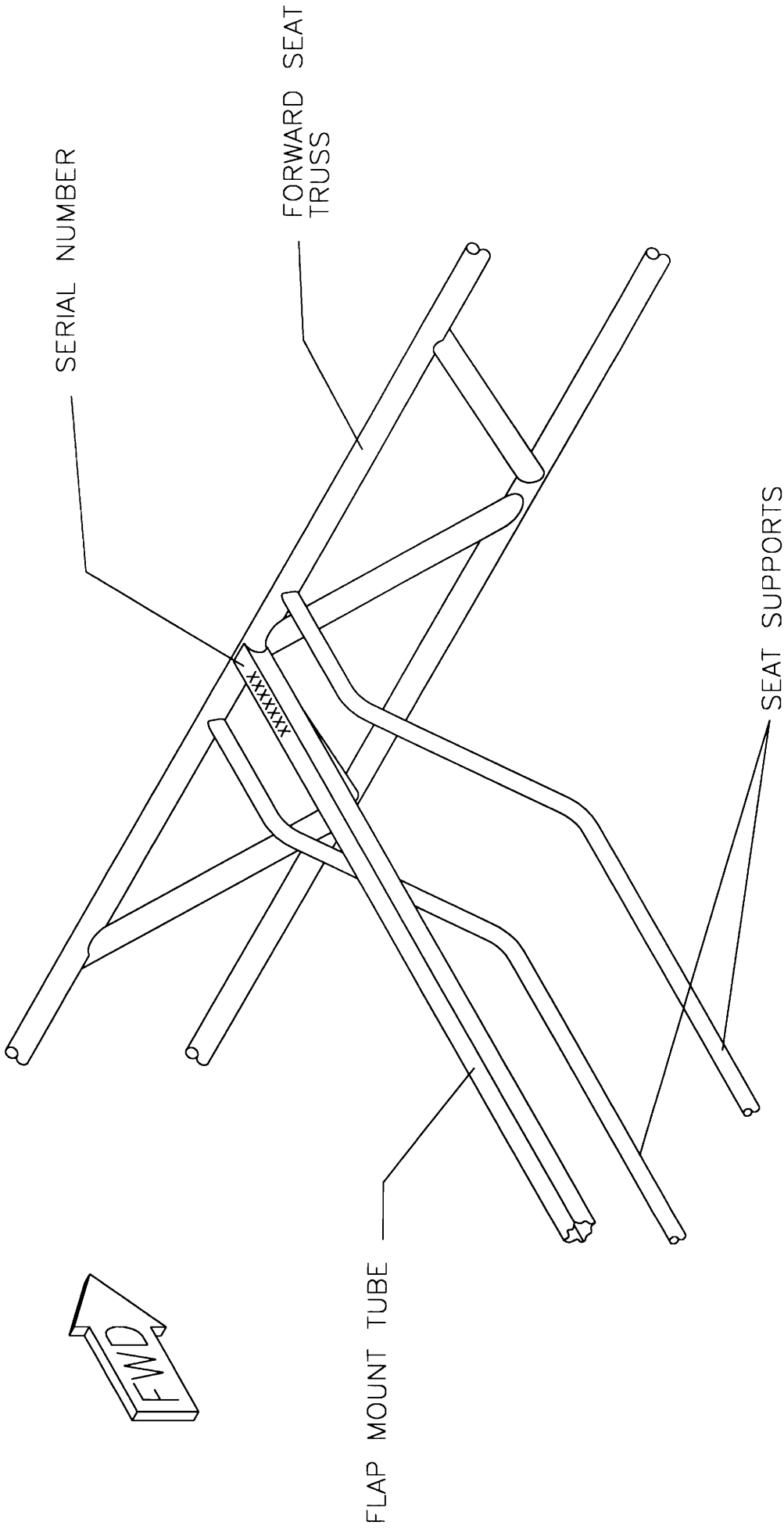
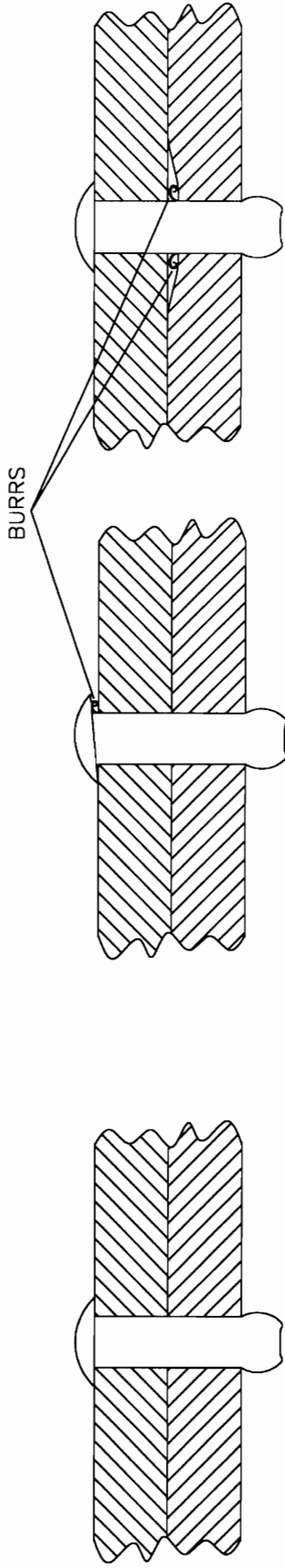
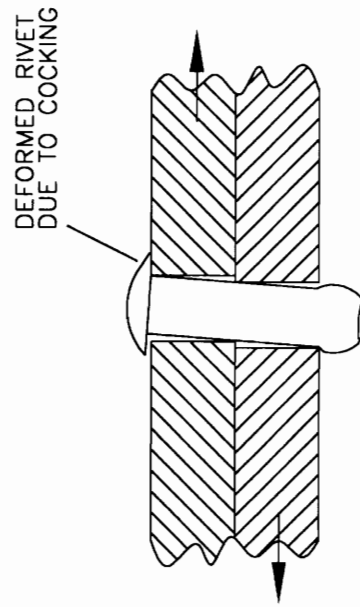


FIGURE 1-3

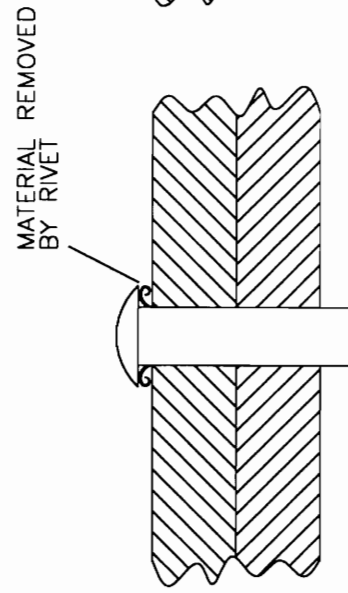


CORRECTLY INSTALLED RIVET

THE EFFECTS OF IMPROPER DE-BURRING



THE EFFECTS OF RIVETING THROUGH AN OVERSIZED HOLE



THE EFFECTS OF FORCING A RIVET THROUGH AN UNDERSIZED HOLE

FIGURE 1-4

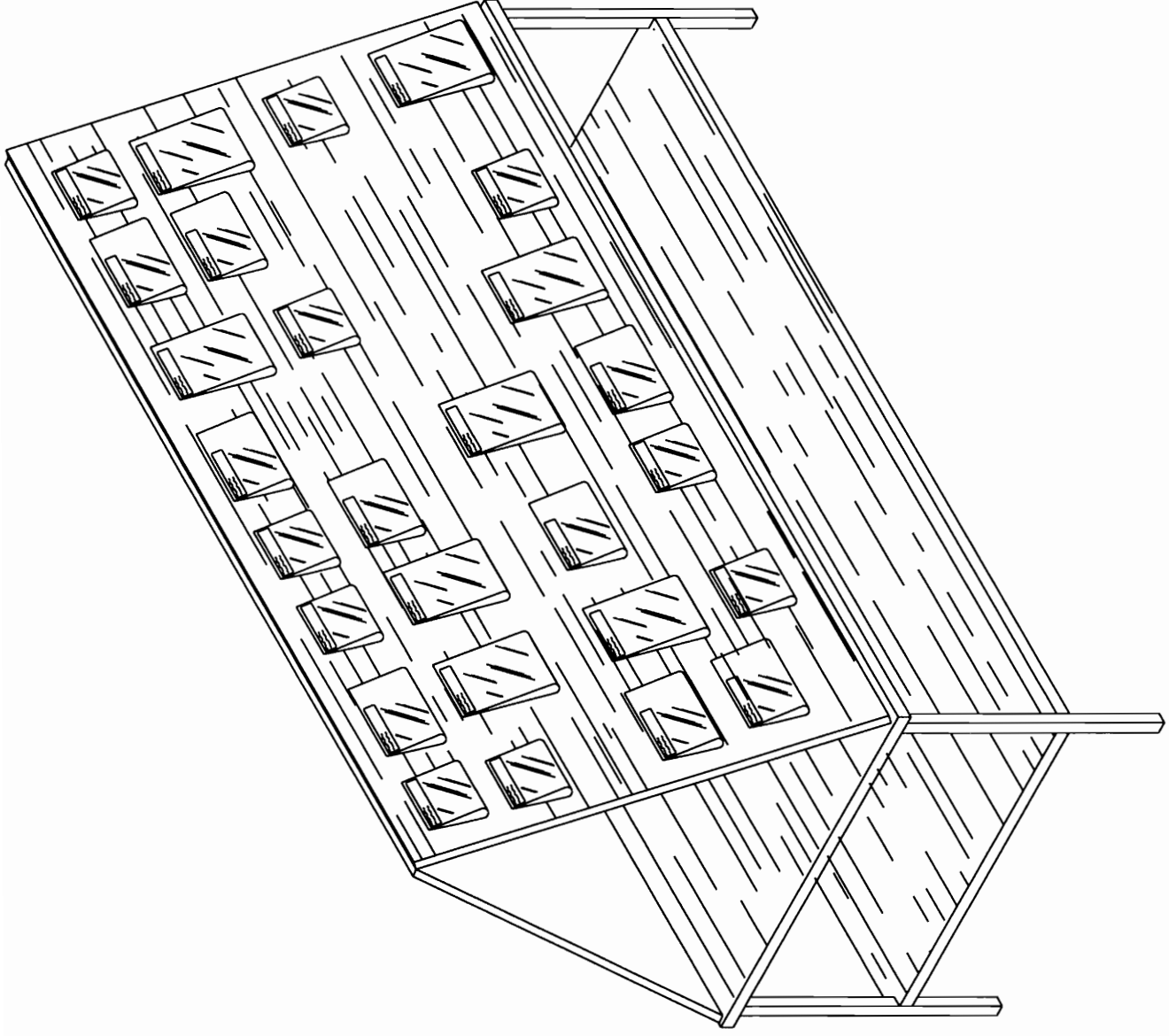


FIGURE 1-5

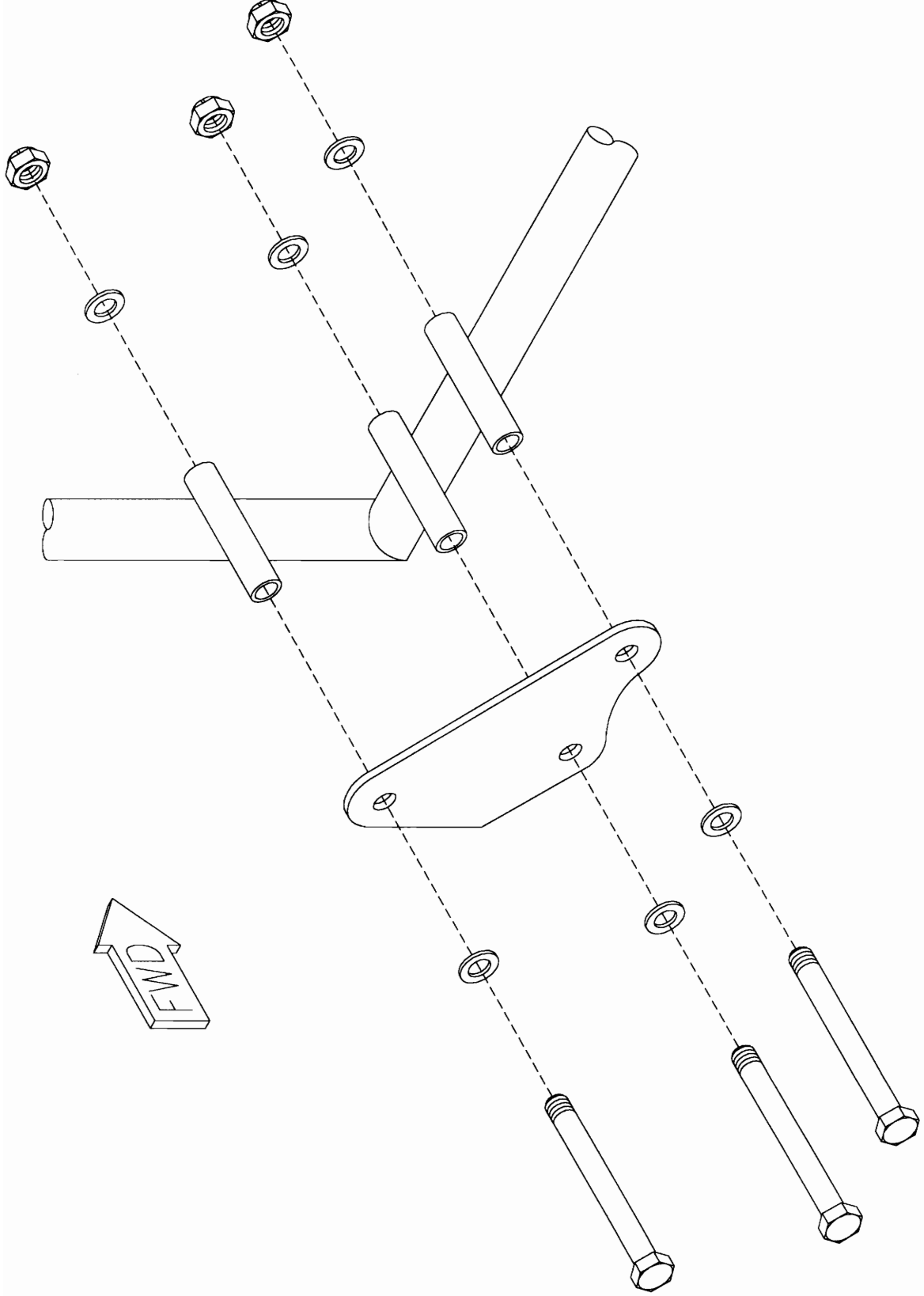


FIGURE 2-1

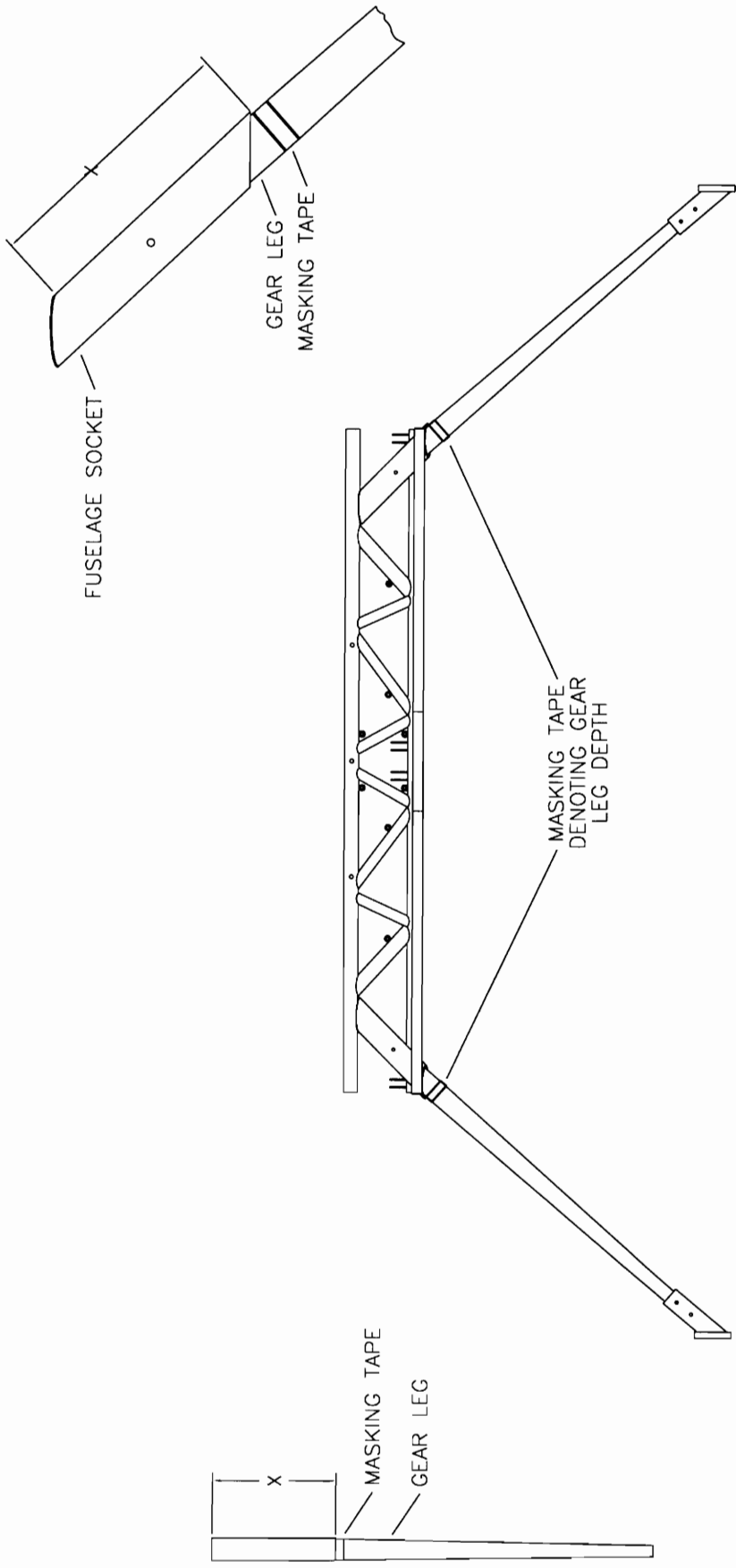
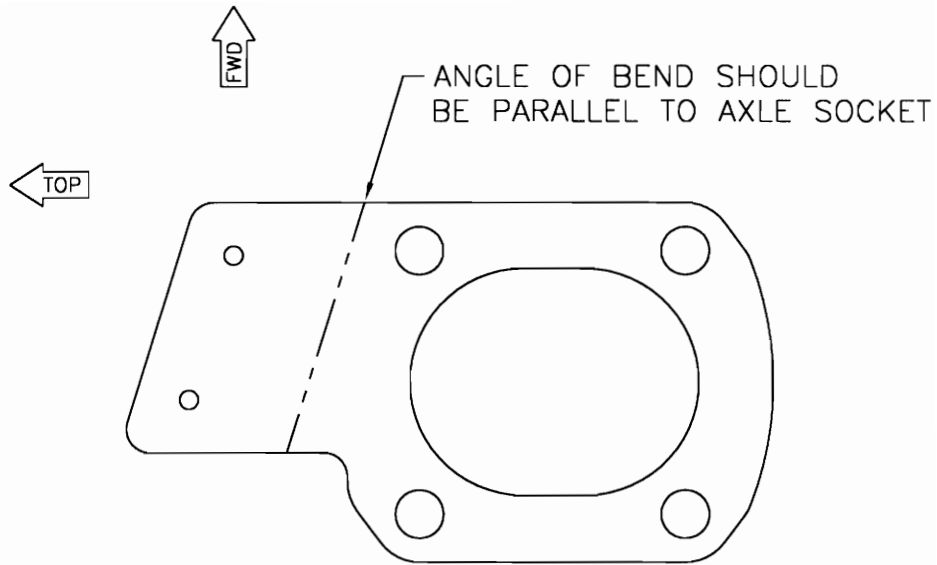


FIGURE 2-2

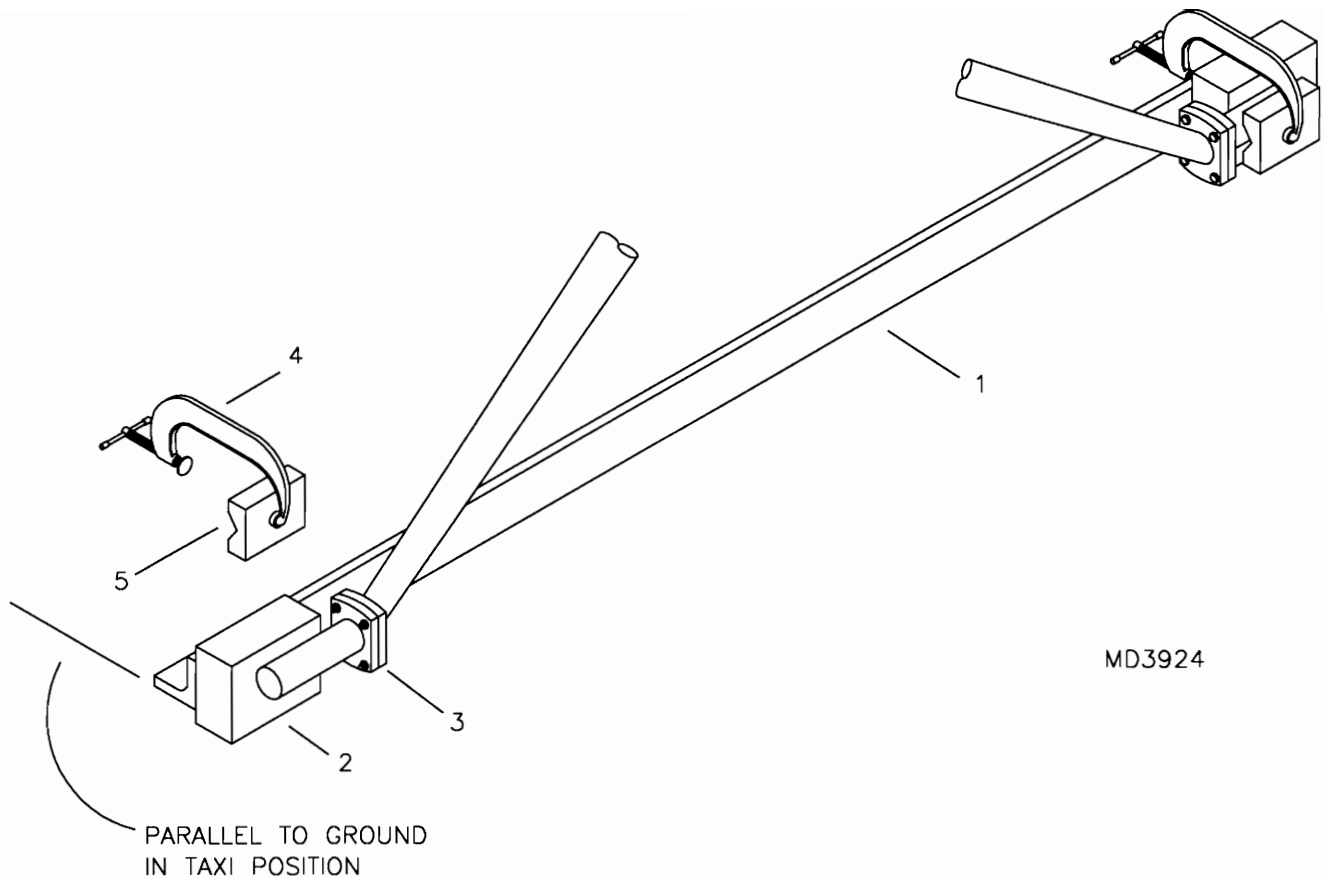




BEND LEFT AND RIGHT

MD3920

FIGURE 2-2a



1. STRAIGHT ANGLE OR BOARD (MAKE SURE IT IS STRAIGHT!)
2. SPACER BLOCK
3. BOLT AXLES TO GEAR LEGS
4. CLAMPS
5. VEE-BLOCK FOR CLAMPING AXLE

FIGURE 2-3  
S-16 SHEKARI

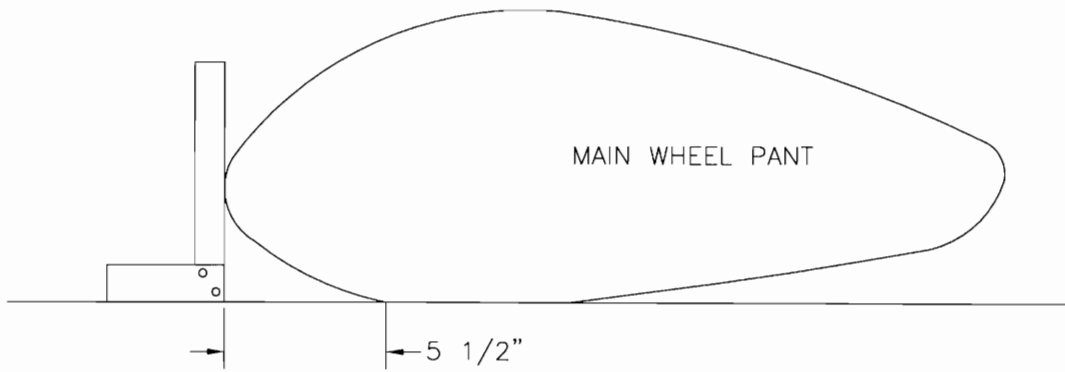
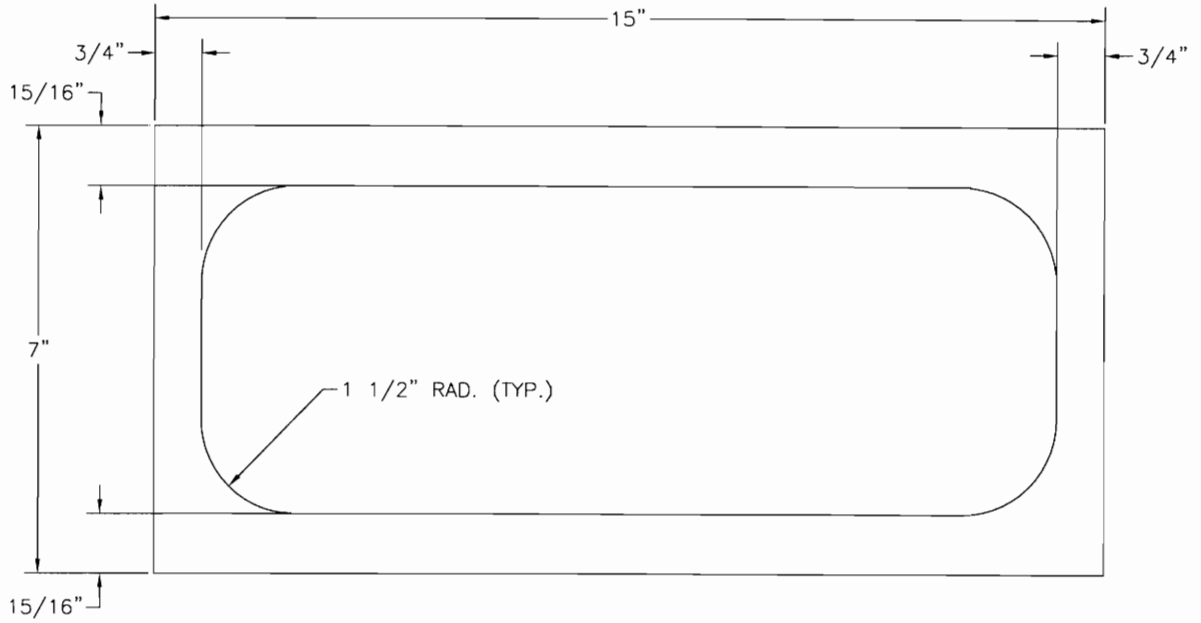
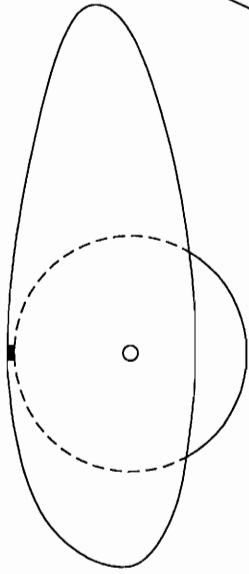


FIGURE 2-4

NOTE: PLACE 1/2" SPACER  
BETWEEN TOP OF WHEEL  
AND WHEEL PANT

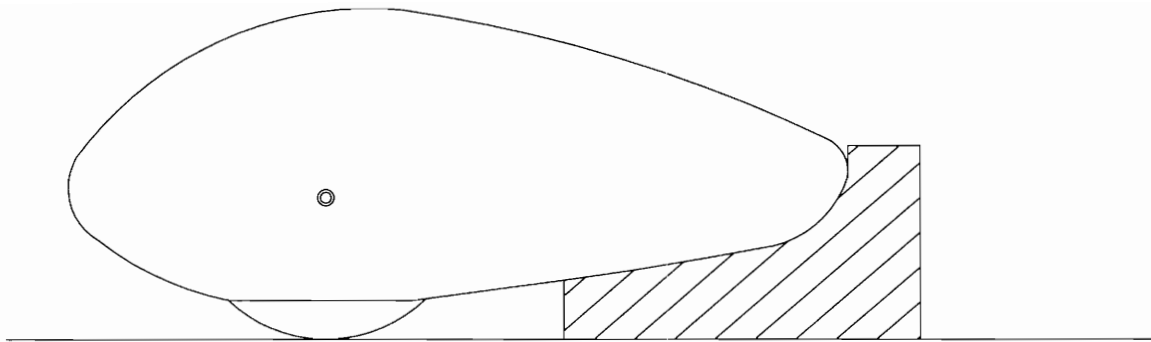


+

LOCATE DIMPLE ON CENTER  
OF AXLE AS INITIAL REF. USE  
1/2" BLOCK ON TOP OF TIRE  
FOR FINAL FIT.

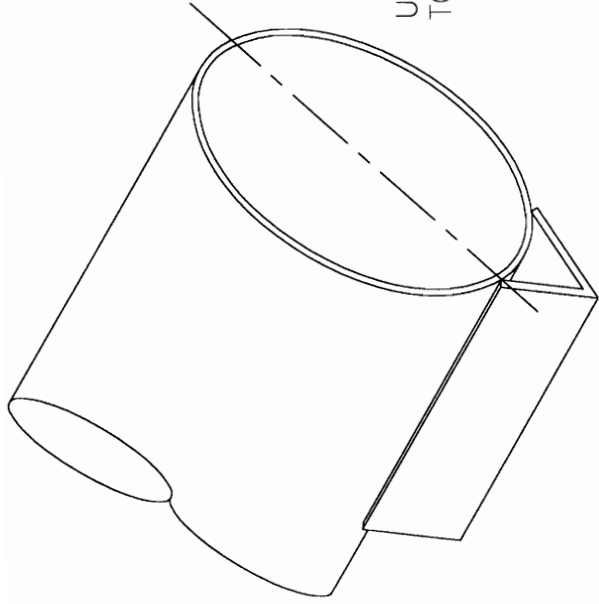
ALIGN WITH BOTTOM OF  
WHEEL PANTS.

FIGURE 2-5  
S-16 SHEKARI



CARDBOARD OR WOOD CUT OUT,  
PLACE UNDER MAIN WHEEL PANTS  
TO SET PARALLEL WITH NOSE PANT.

FIGURE 2-7



USE ANGLE TO DRAW LINE PARALLEL  
TO CENTER LINE OF TUBE



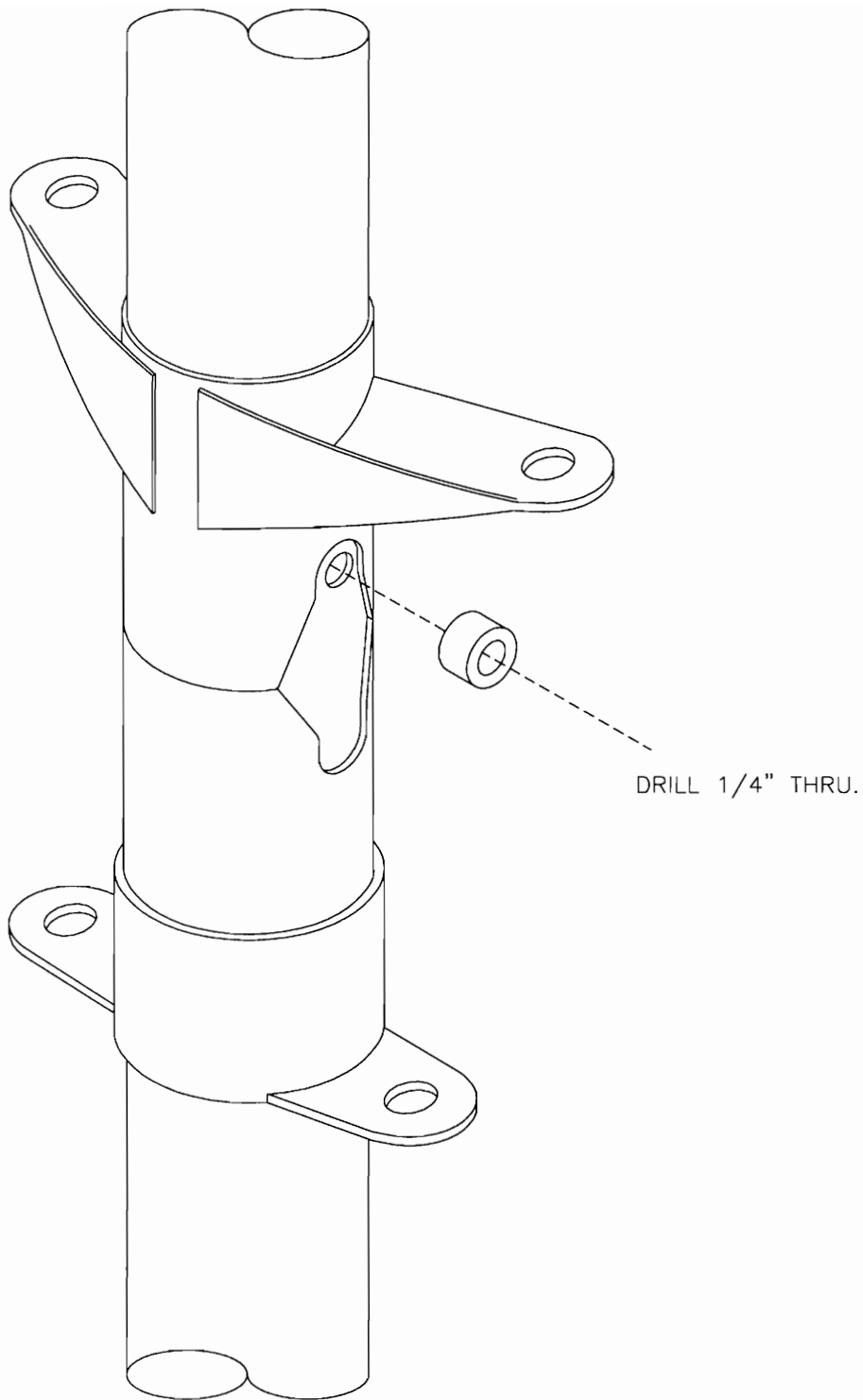


FIGURE 2-8  
S-16 SHEKARI



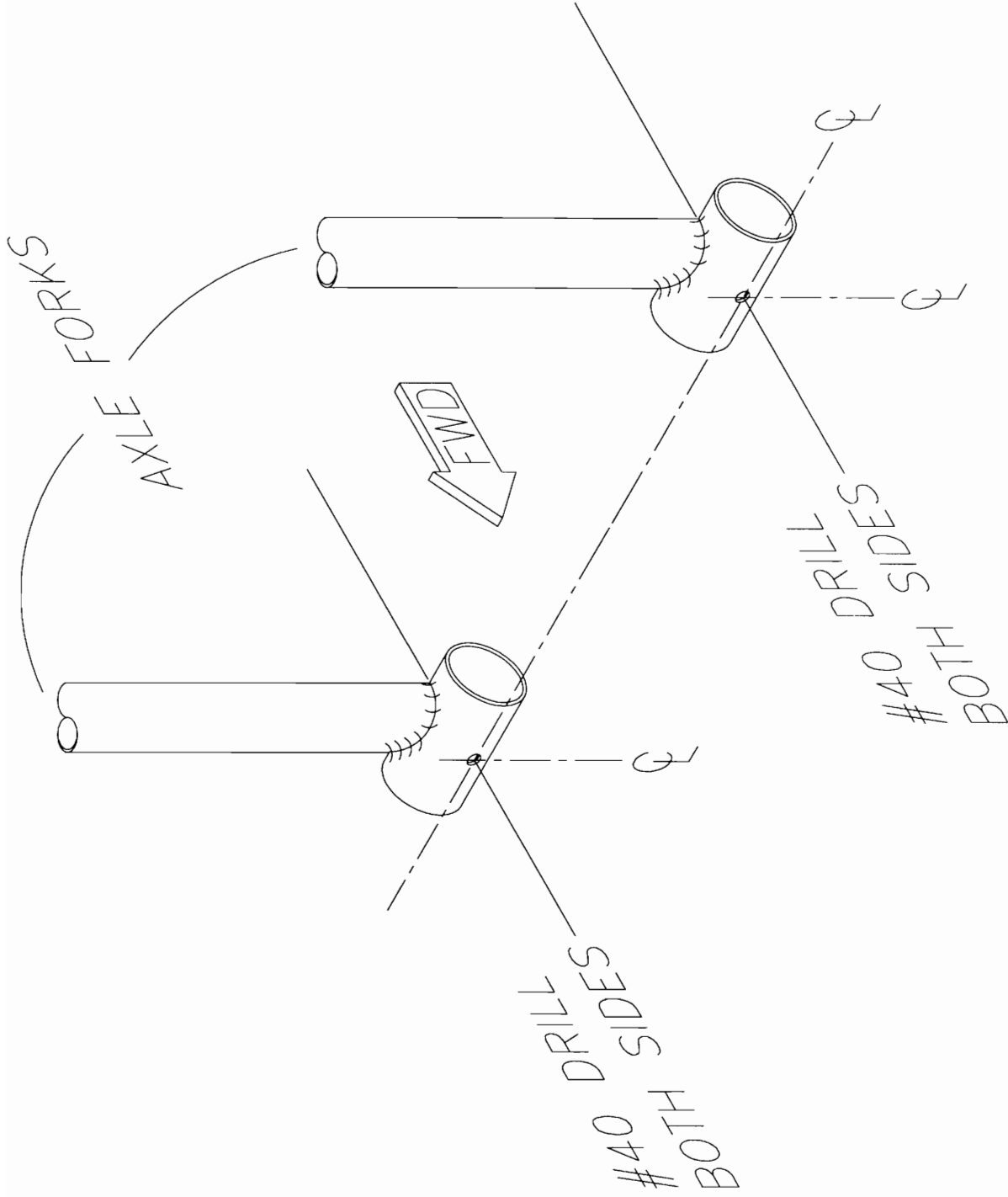


FIGURE 2-9

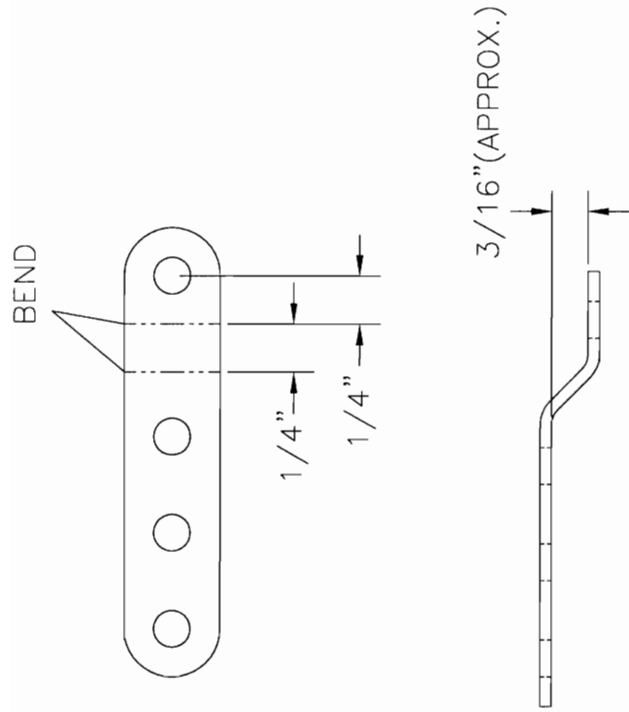


FIGURE 2-10

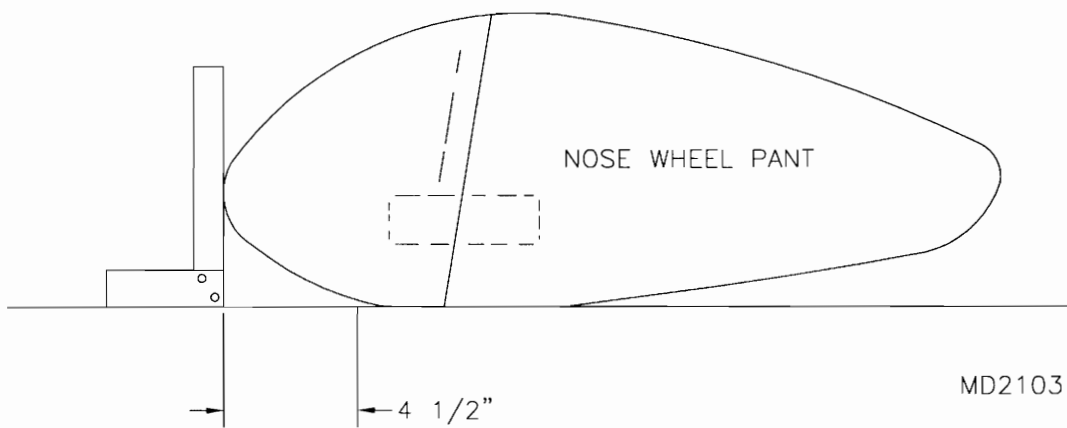
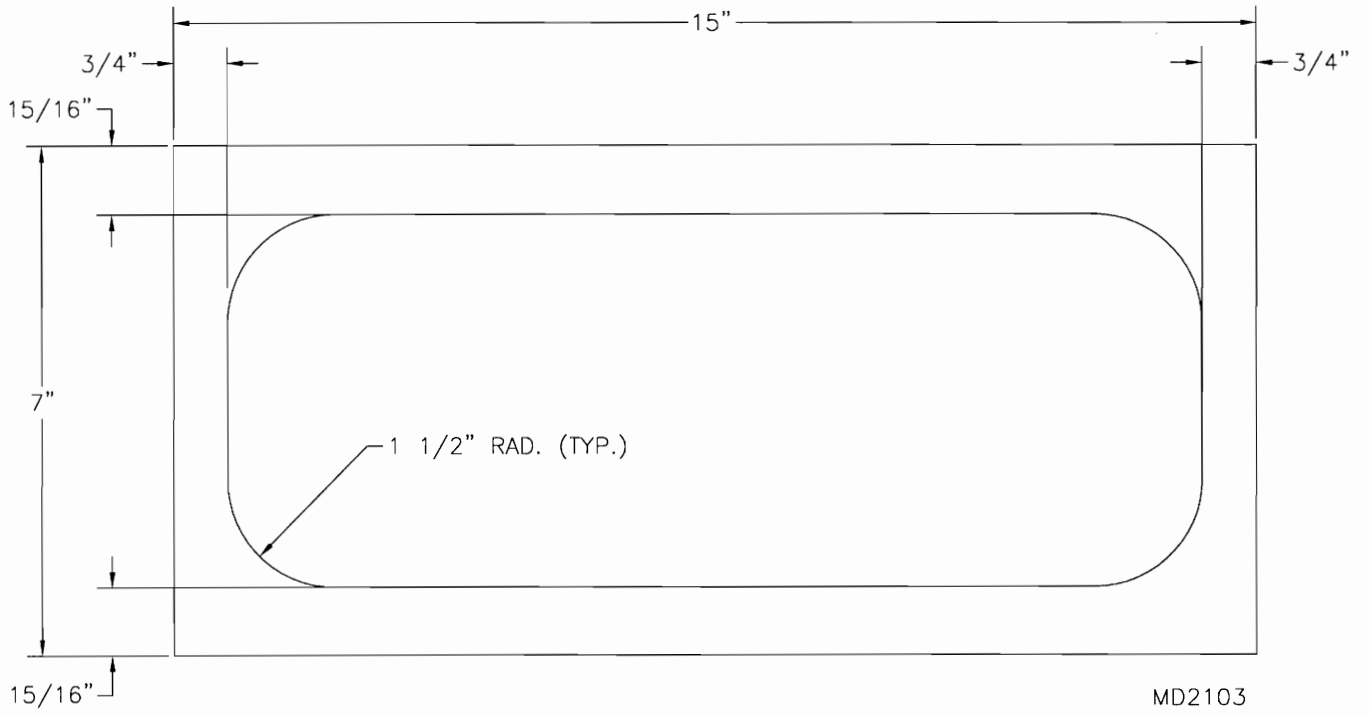


FIGURE 2-11

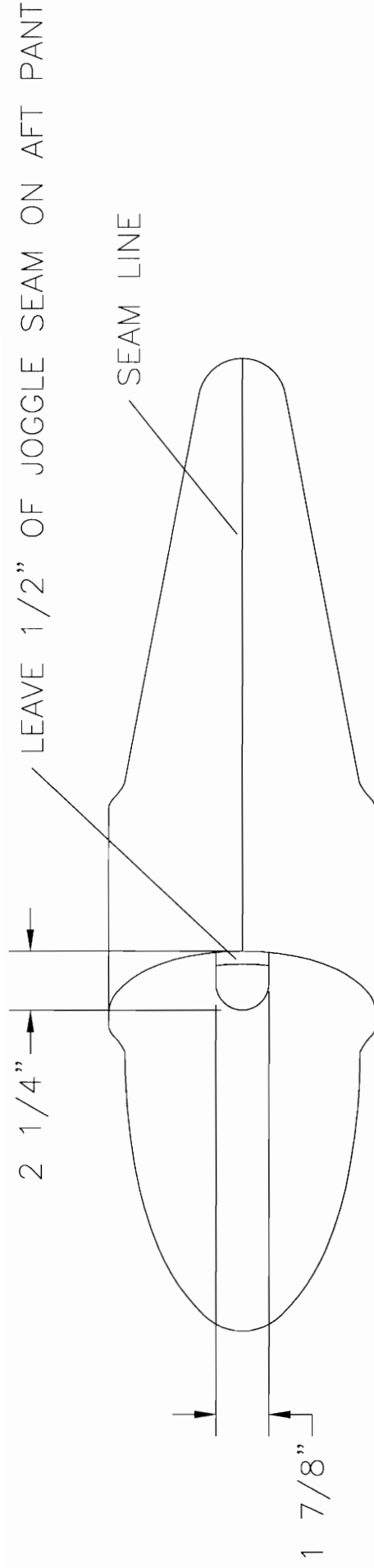


FIGURE 2-12

5" OFF CENTER LINE, 3/8" E.D.,  
USE #11 DRILL

USE NOSE GEAR AXLE TO LOCATE,  
MAINTAIN 1/2" E.D., USE #11 DRILL

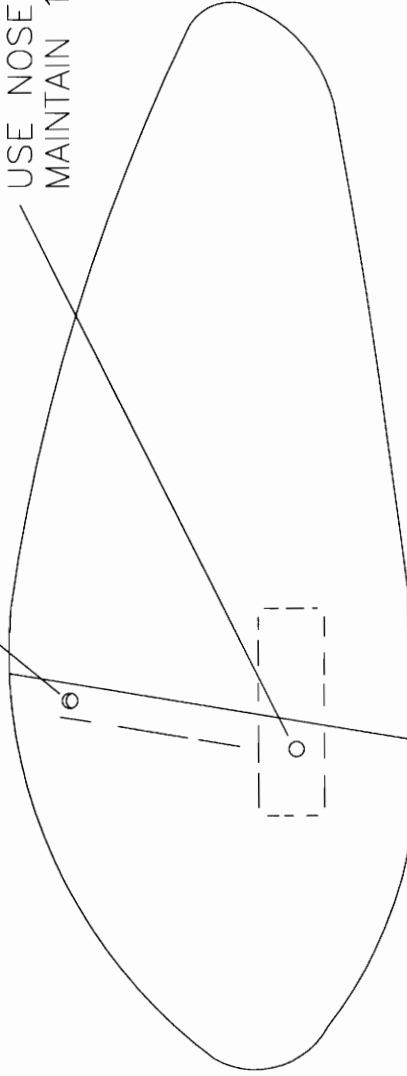


FIGURE 2-13

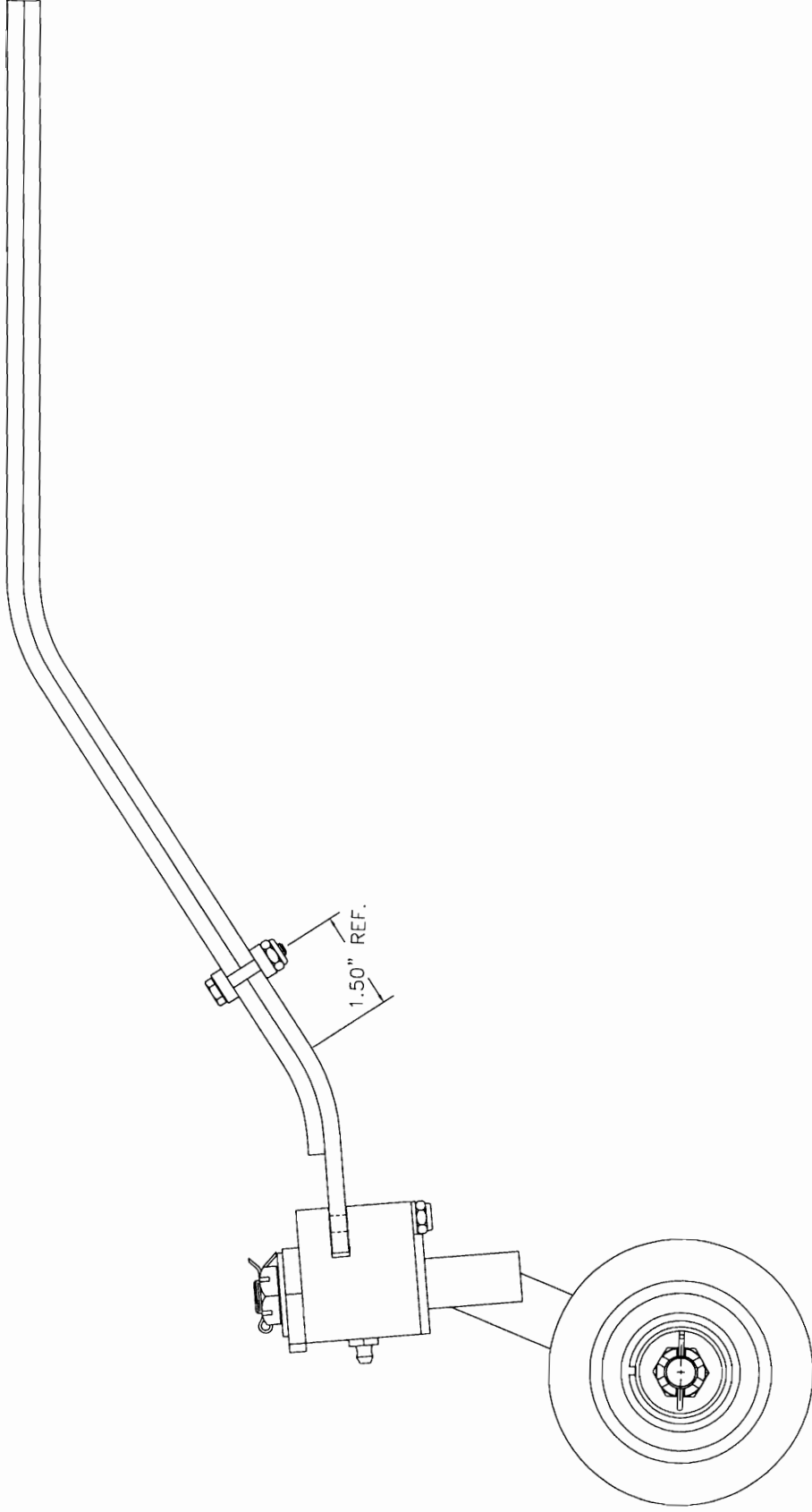


FIGURE 2-14

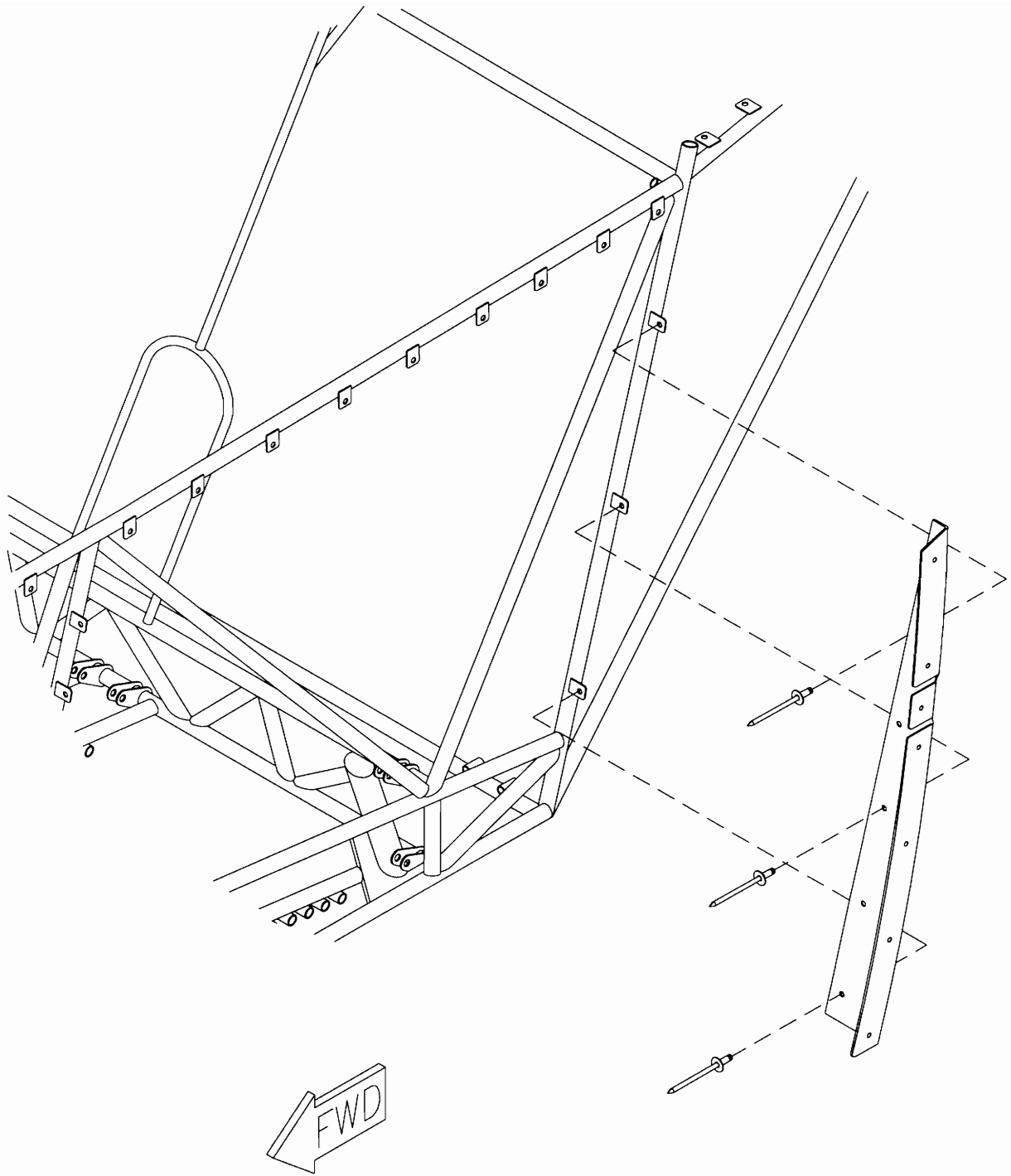


FIGURE 2-15

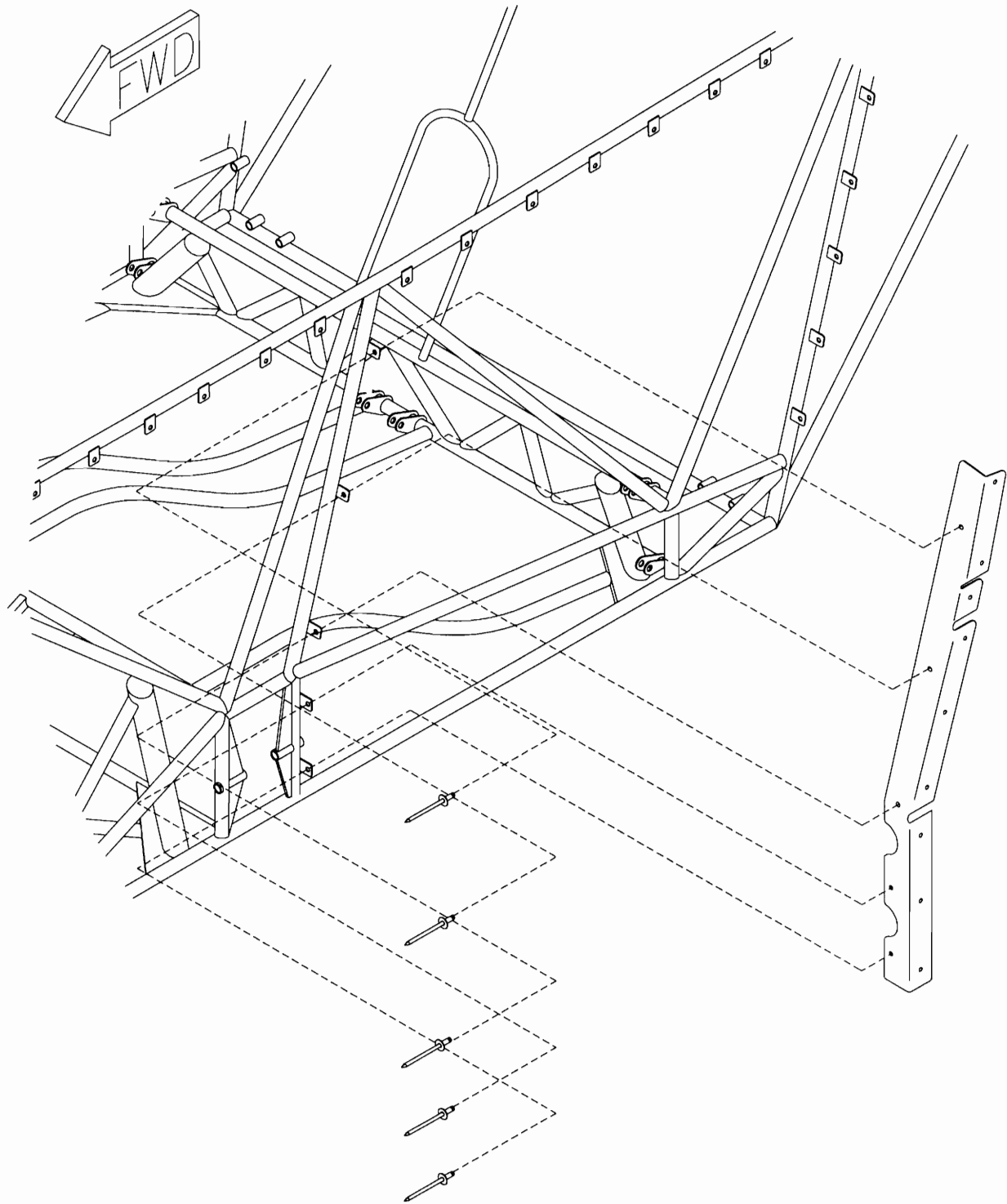


FIGURE 2-16



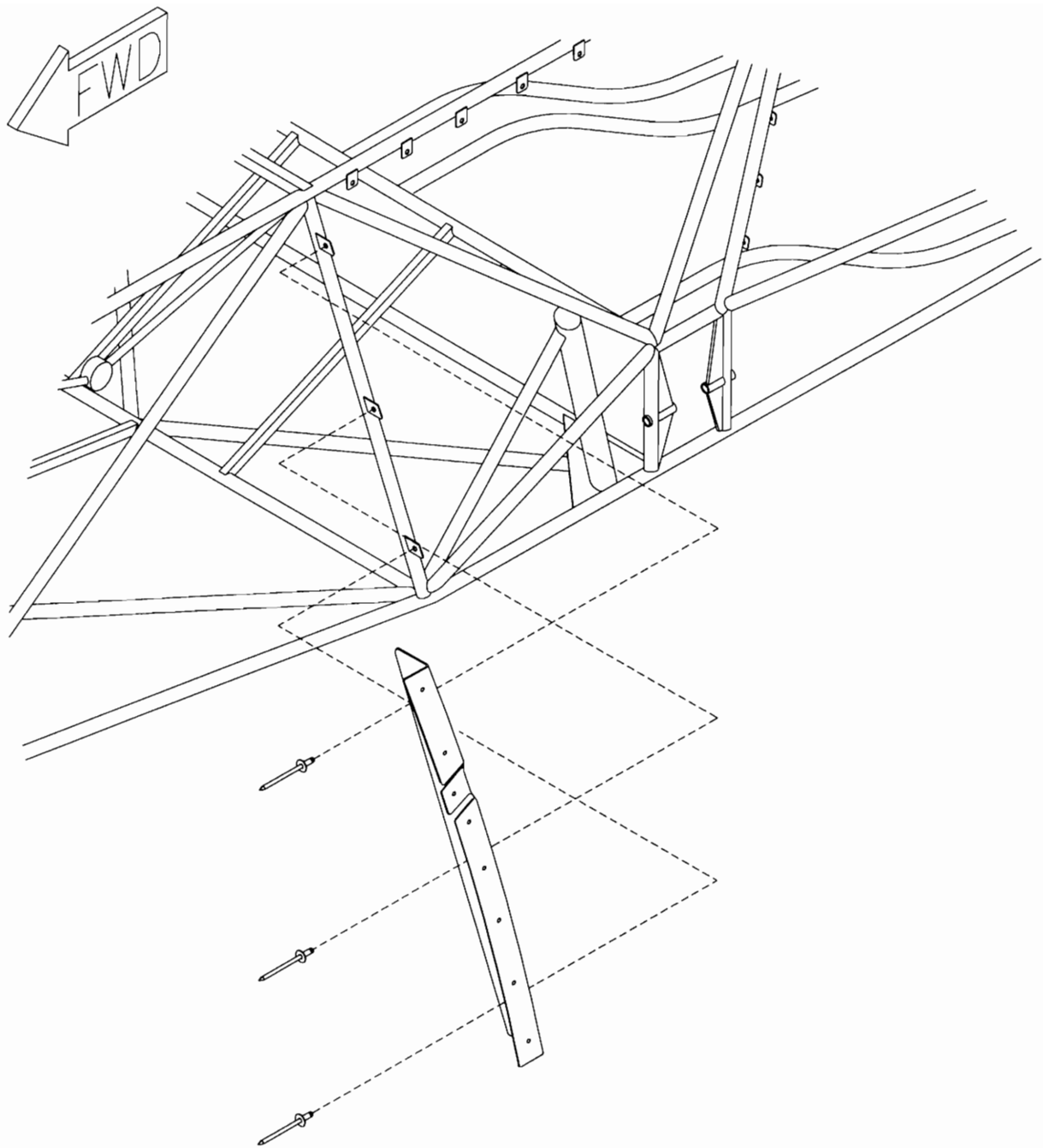
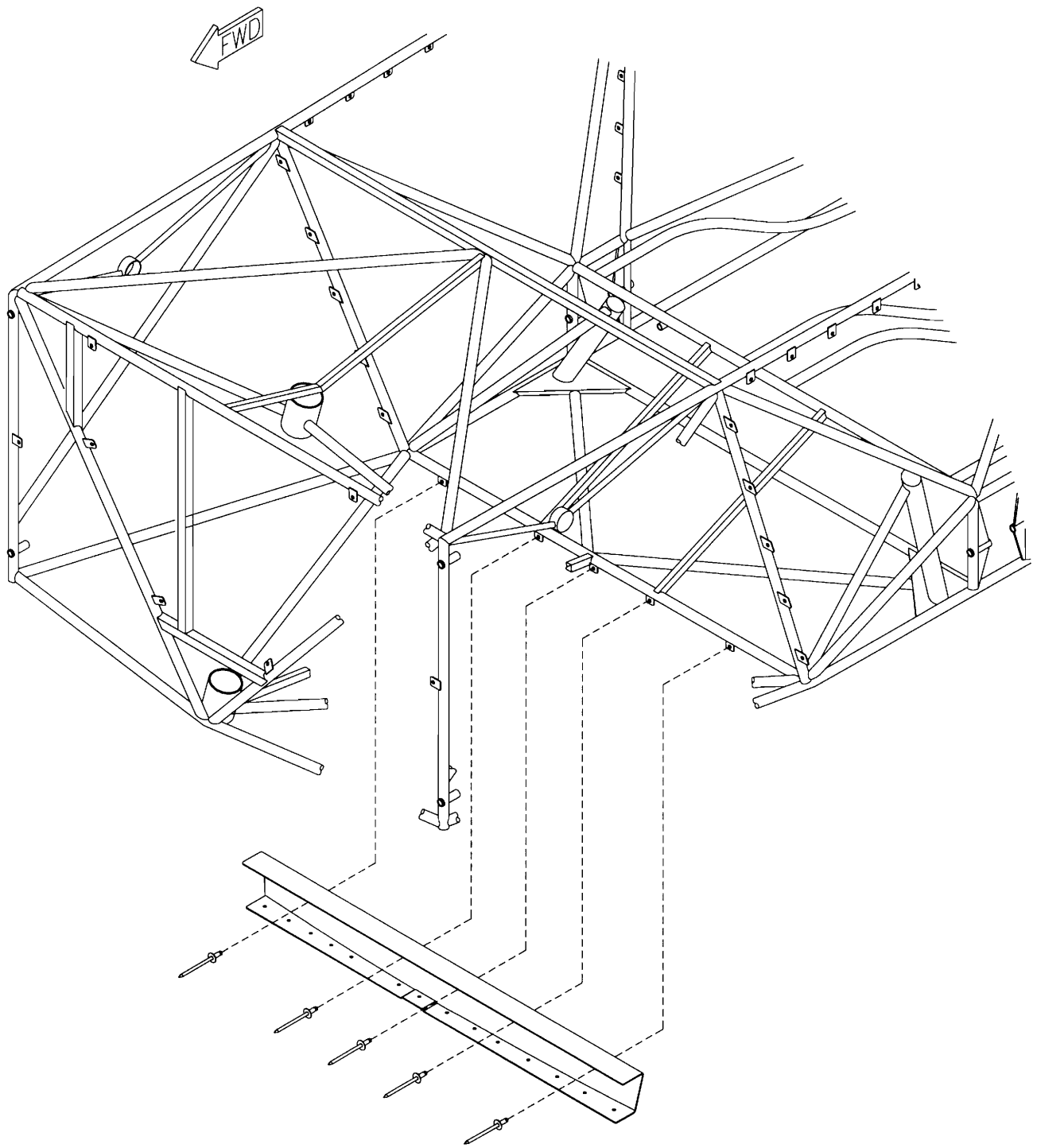
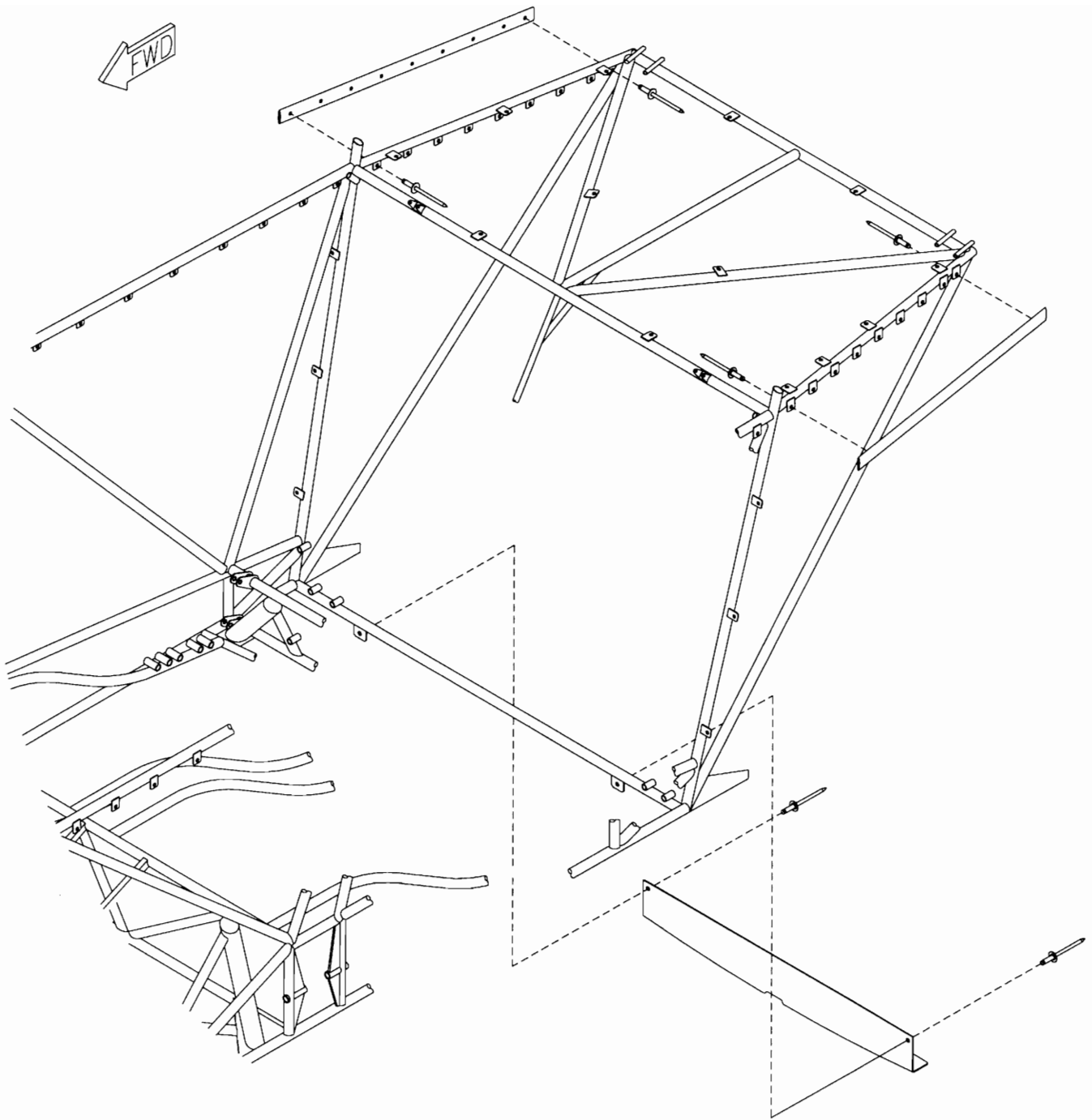


FIGURE 2-17





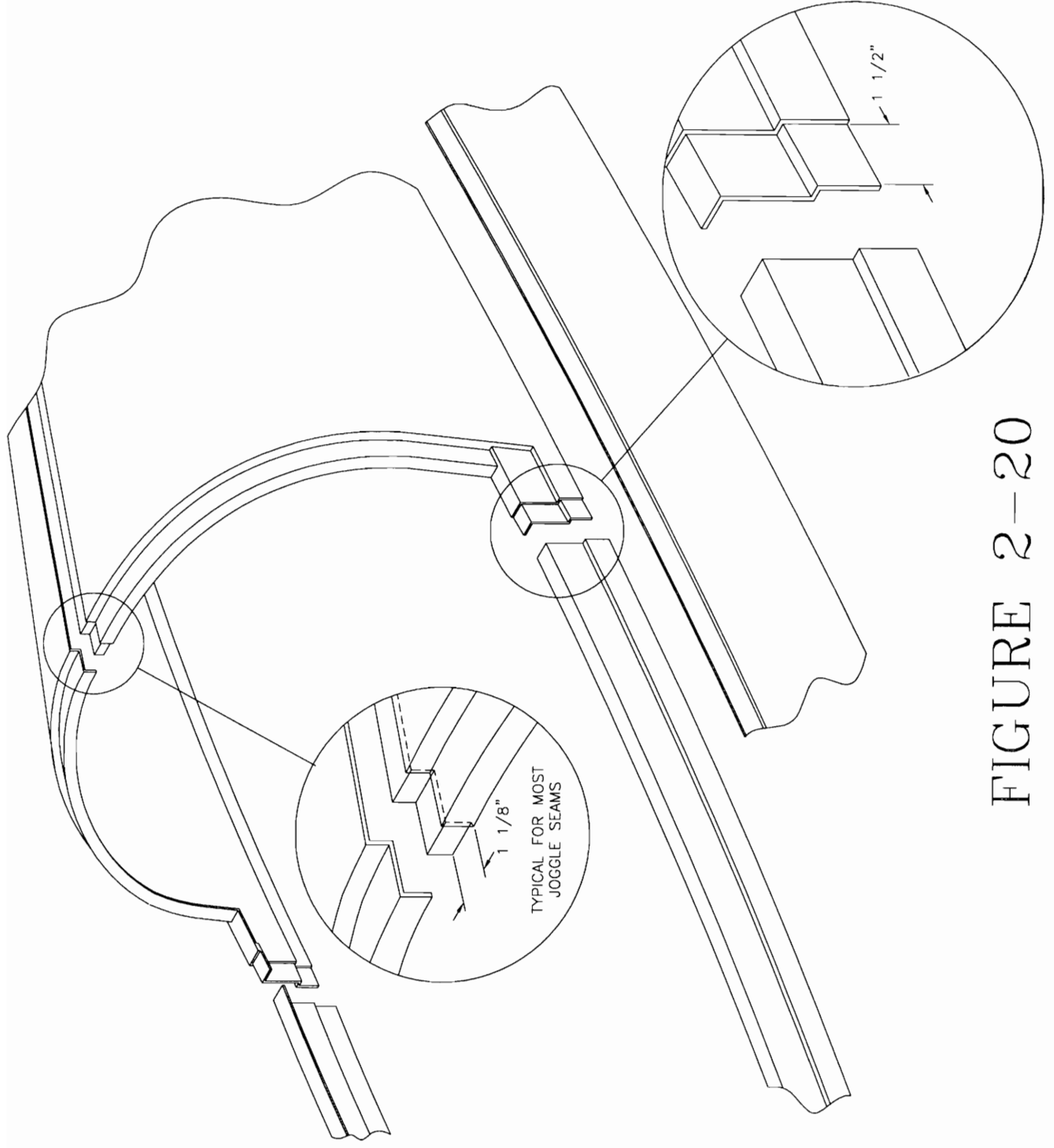


FIGURE 2-20

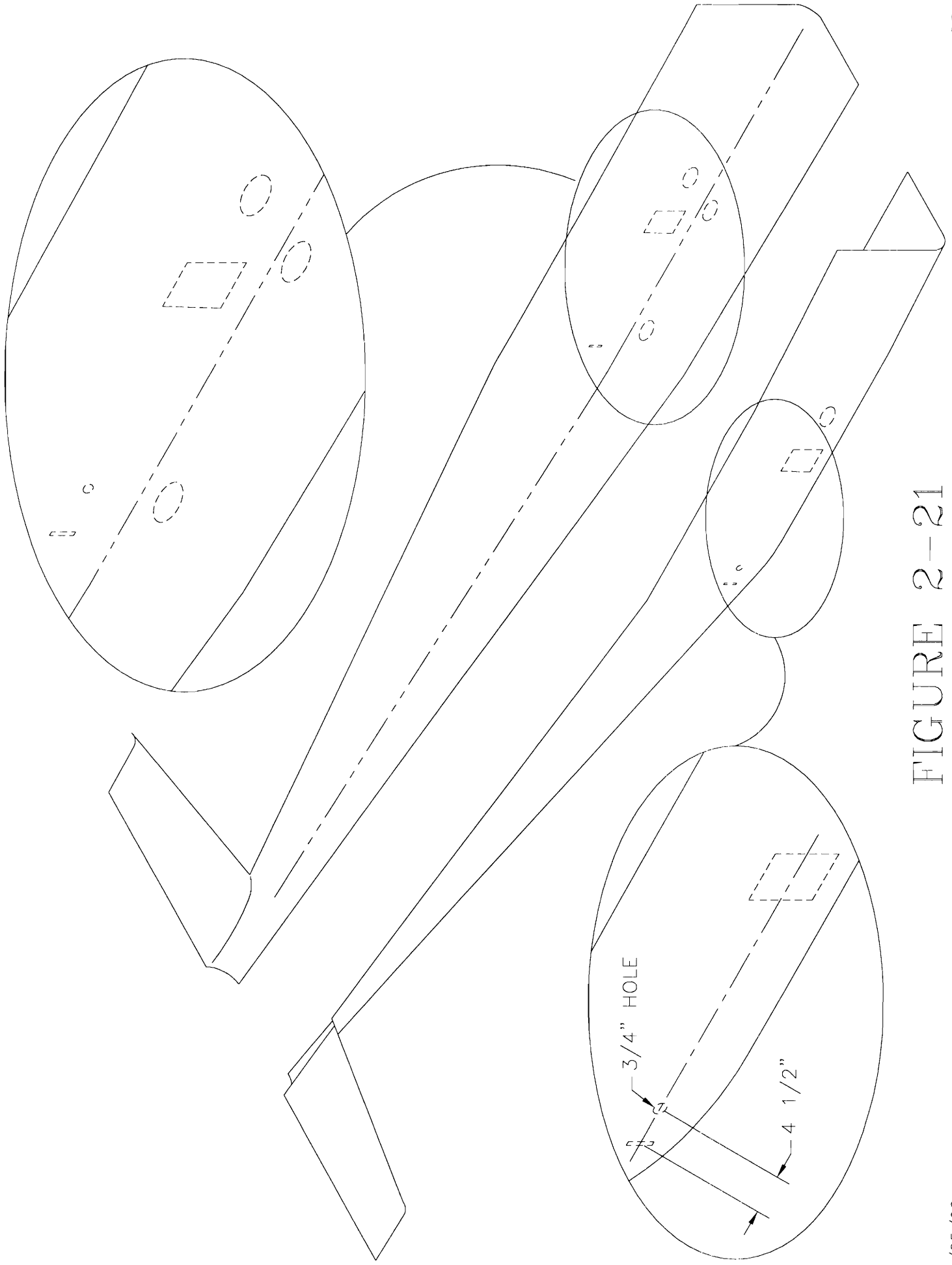


FIGURE 2-21

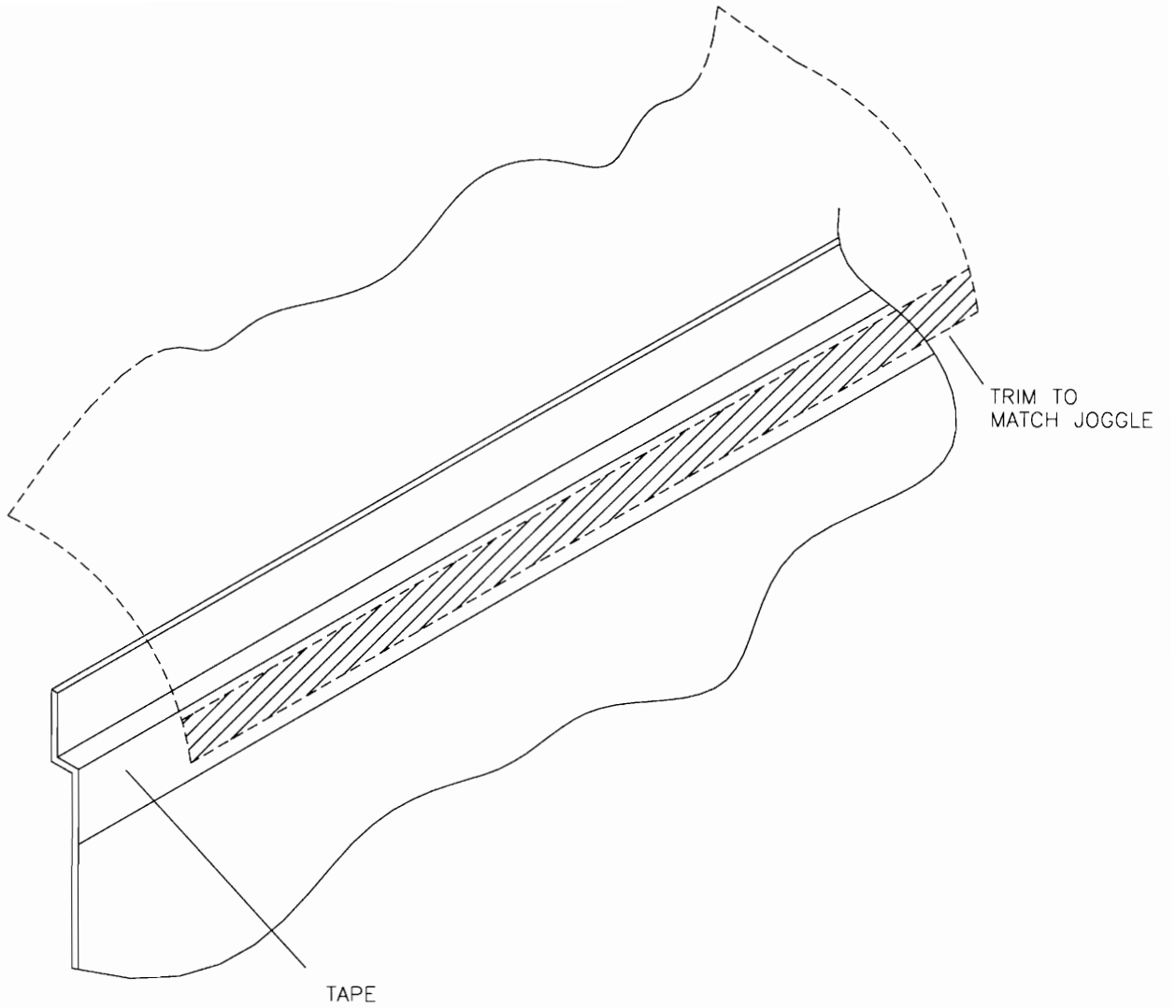


FIGURE 2-21A

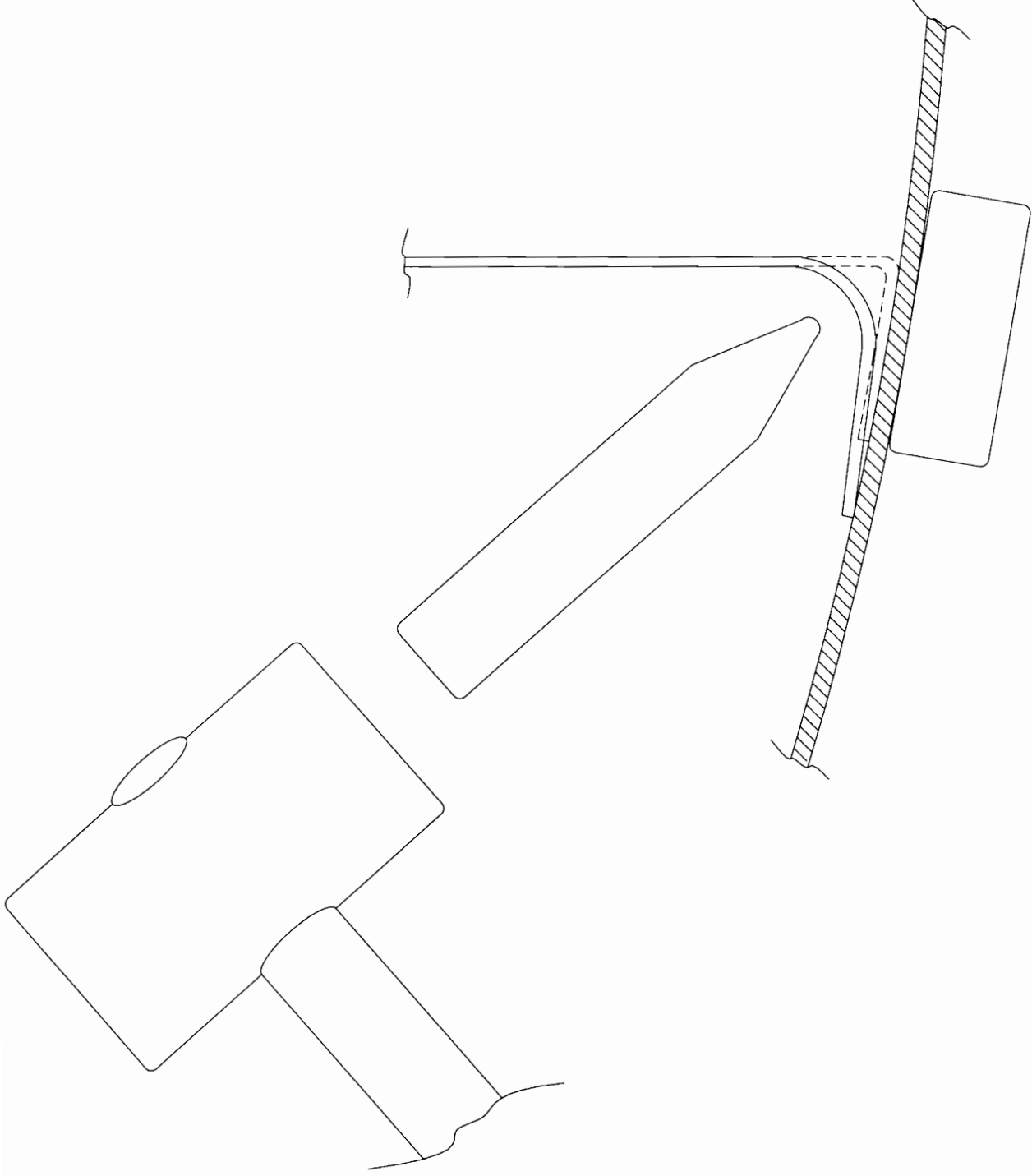
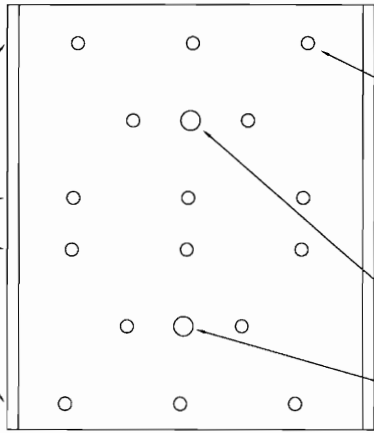


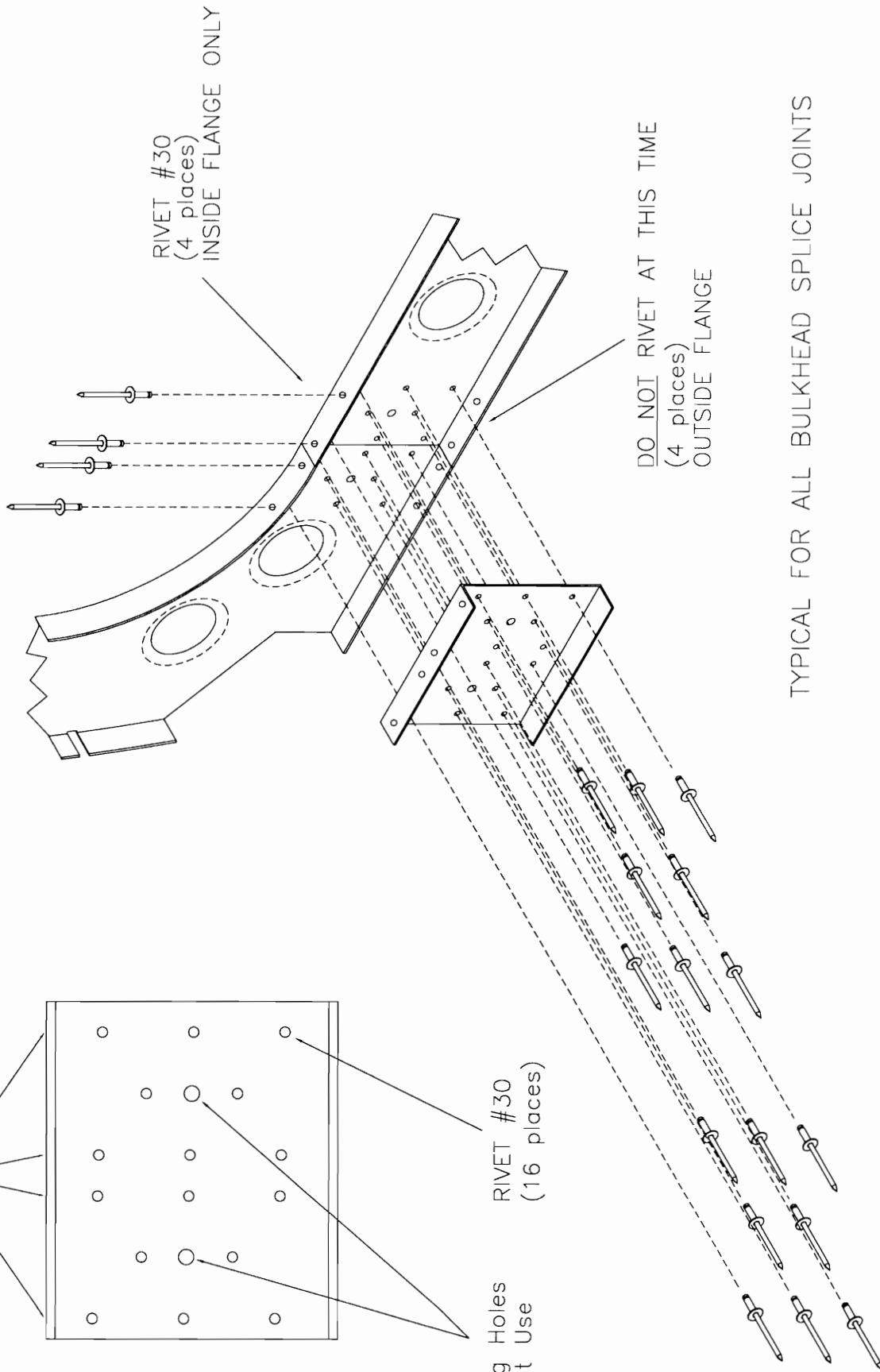
FIGURE 2-22

RIVET #30  
(4 places)



Tooling Holes  
Do not Use

RIVET #30  
(16 places)



TYPICAL FOR ALL BULKHEAD SPLICE JOINTS

DO NOT RIVET AT THIS TIME  
(4 places)  
OUTSIDE FLANGE

RIVET #30  
(4 places)  
INSIDE FLANGE ONLY

FIGURE 2-23



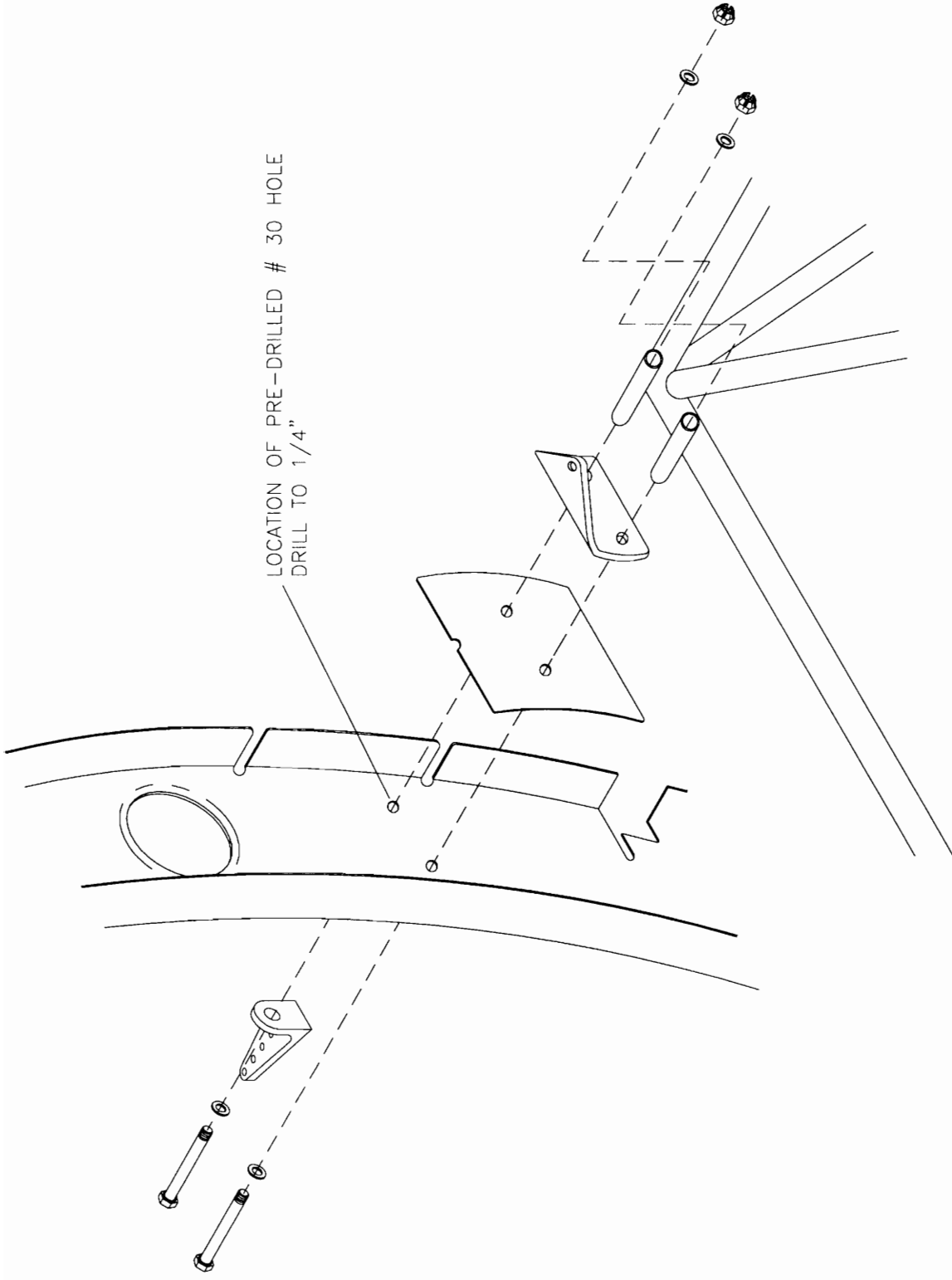


FIGURE 2-24

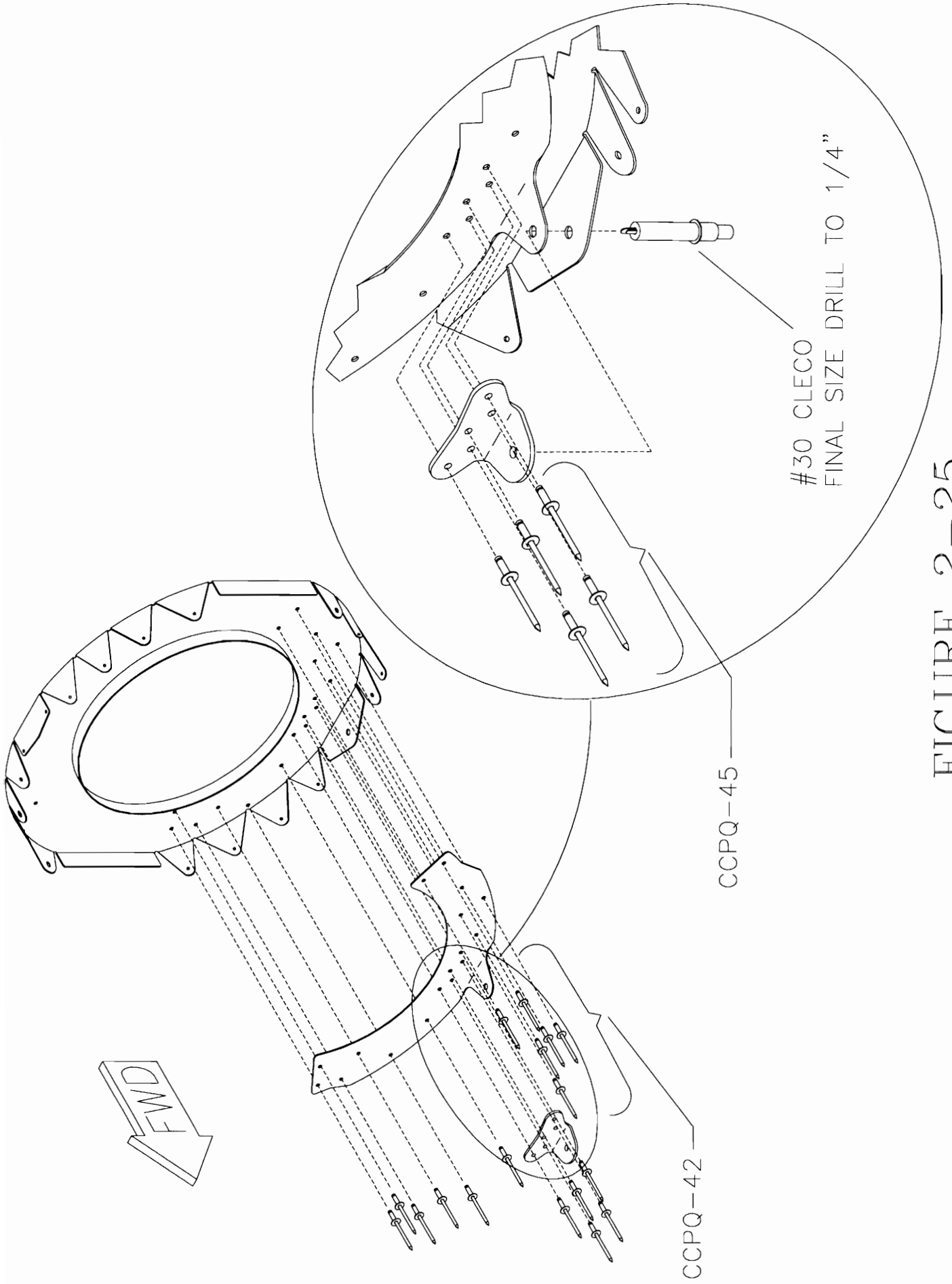


FIGURE 2-25

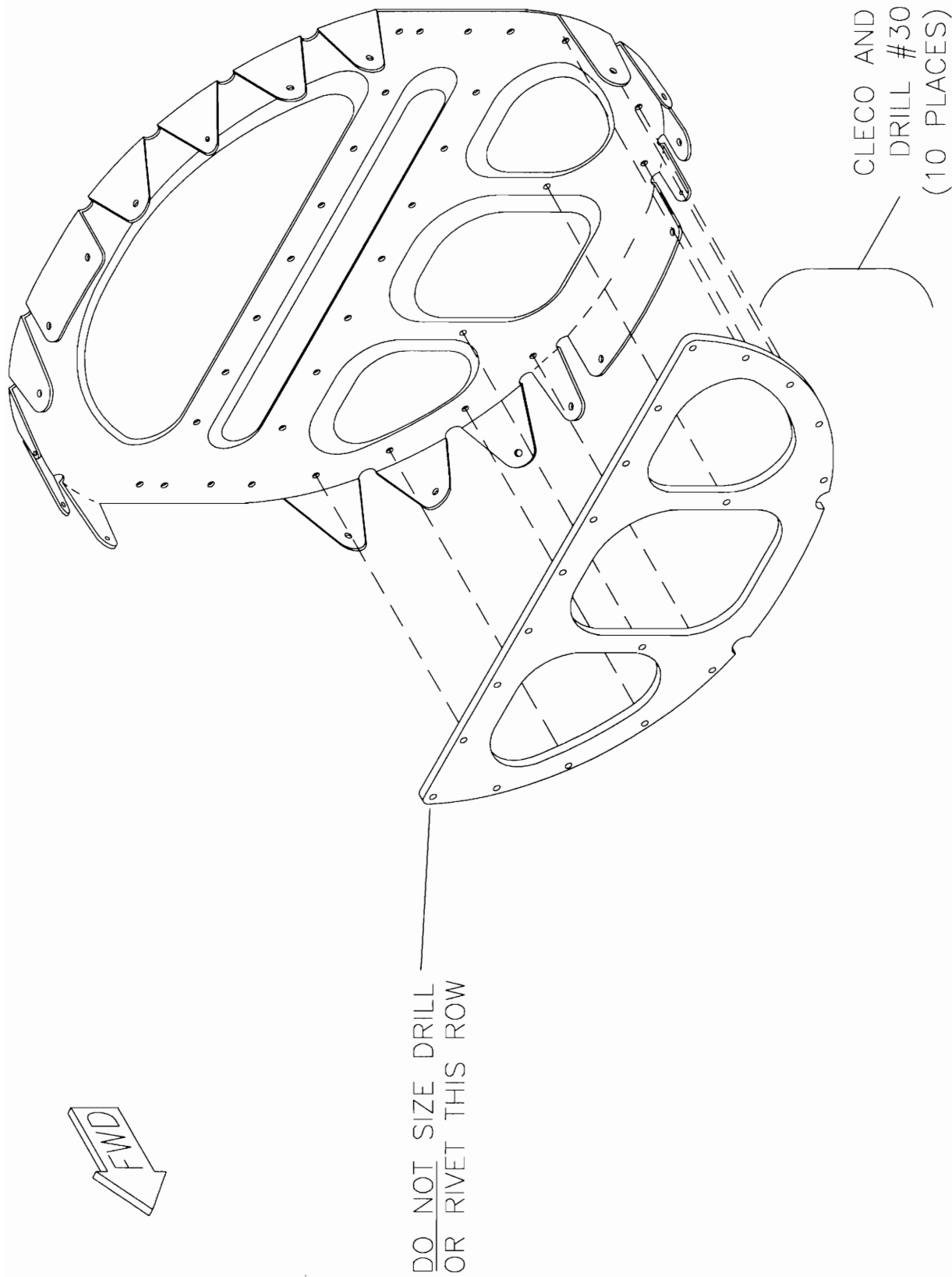


FIGURE 2-26

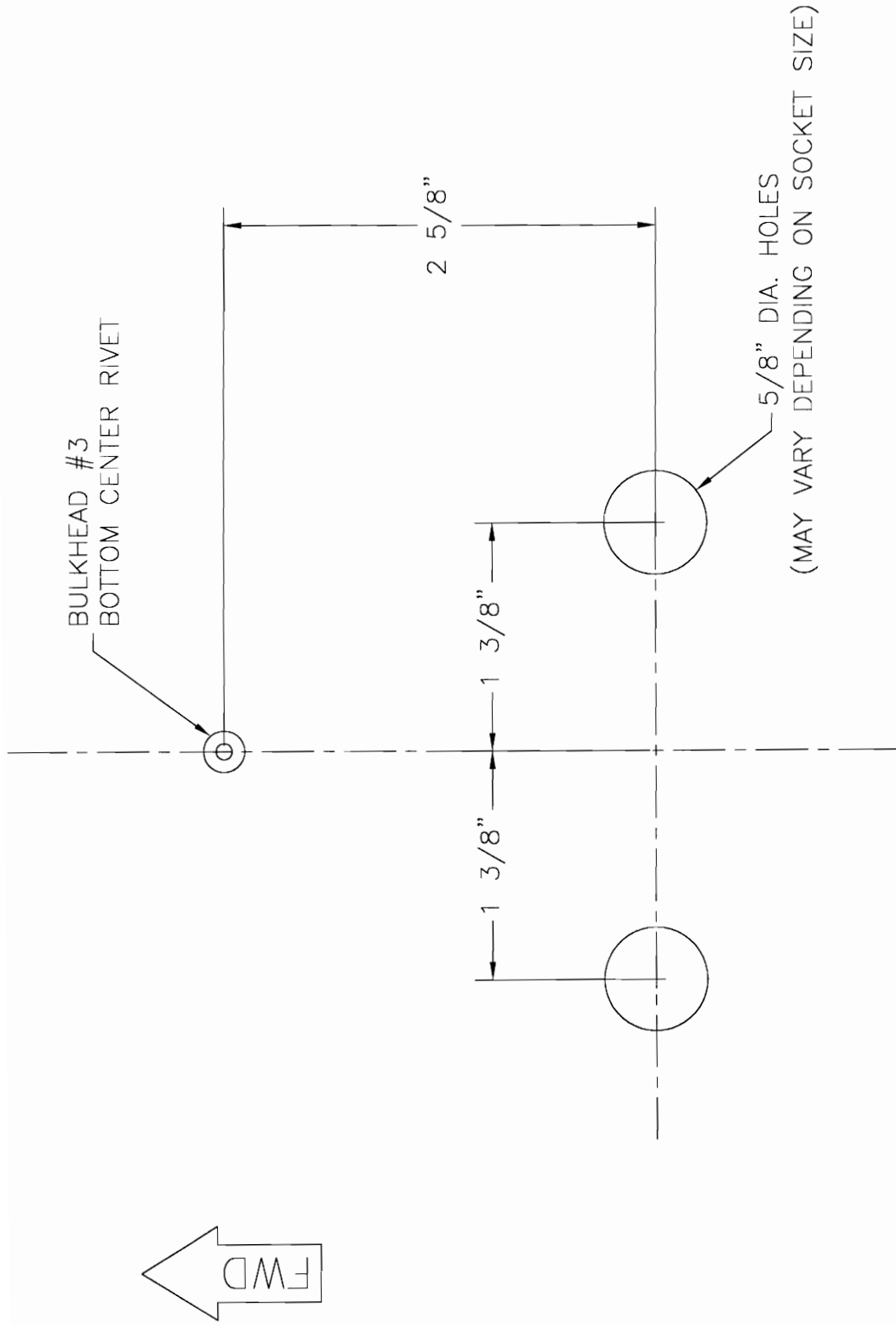
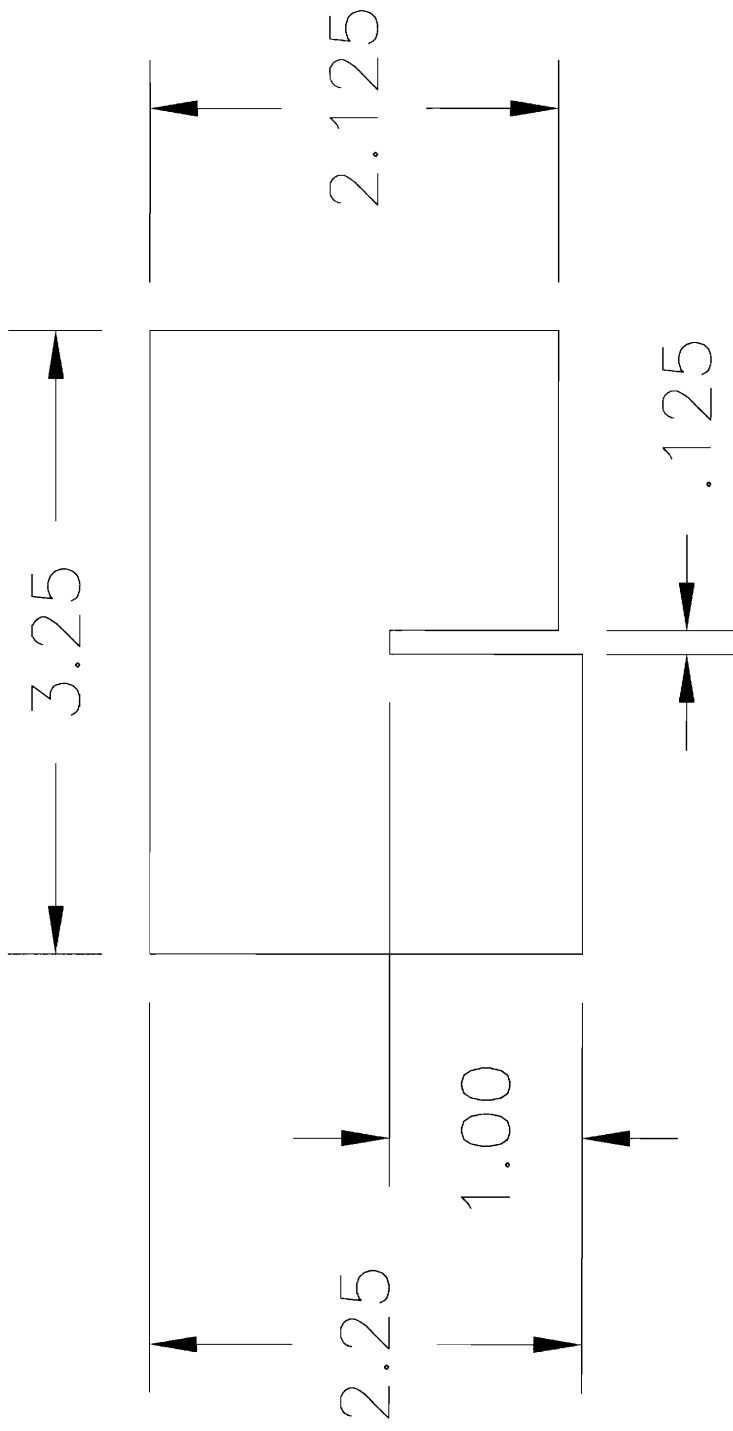
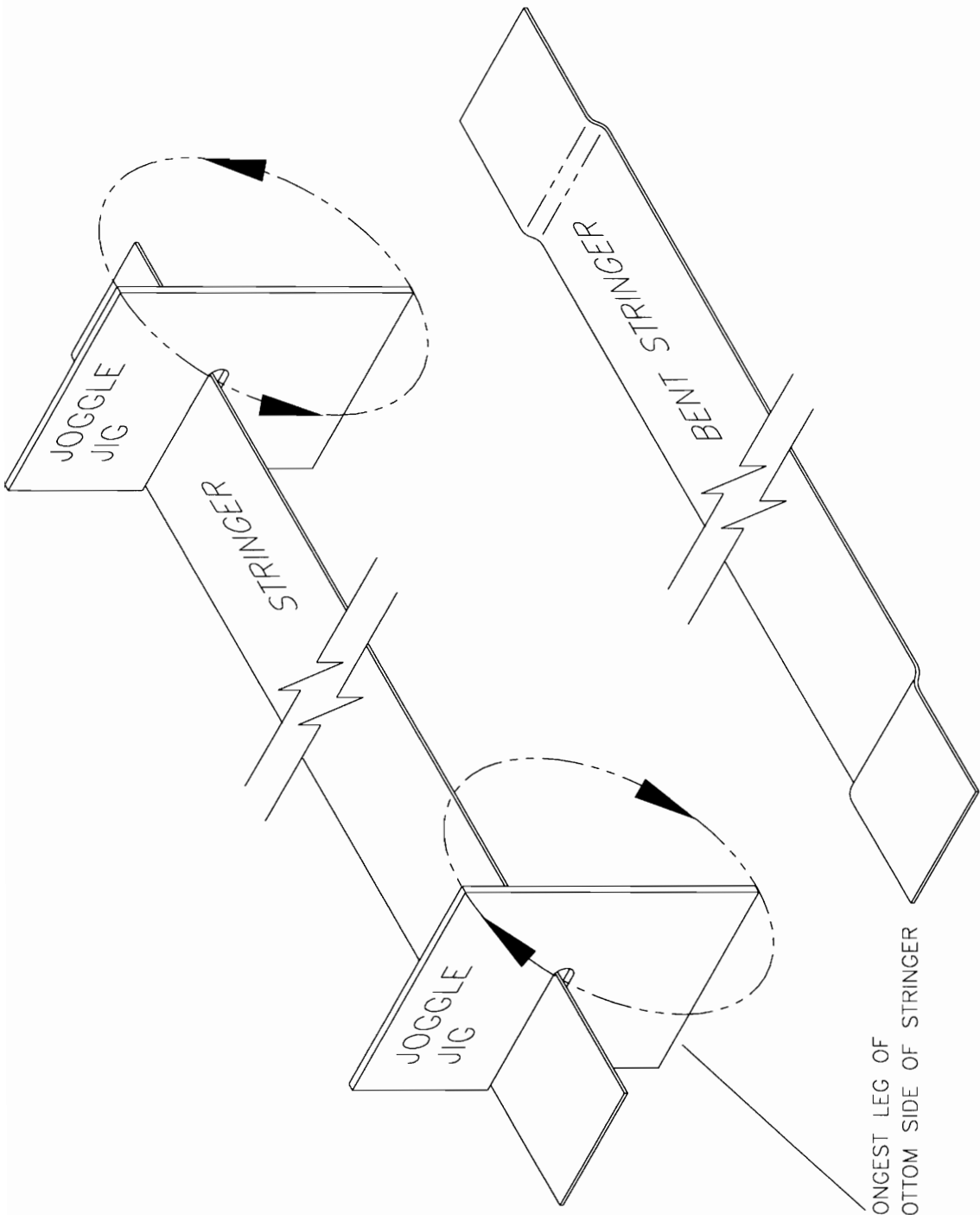


FIGURE 2-26A



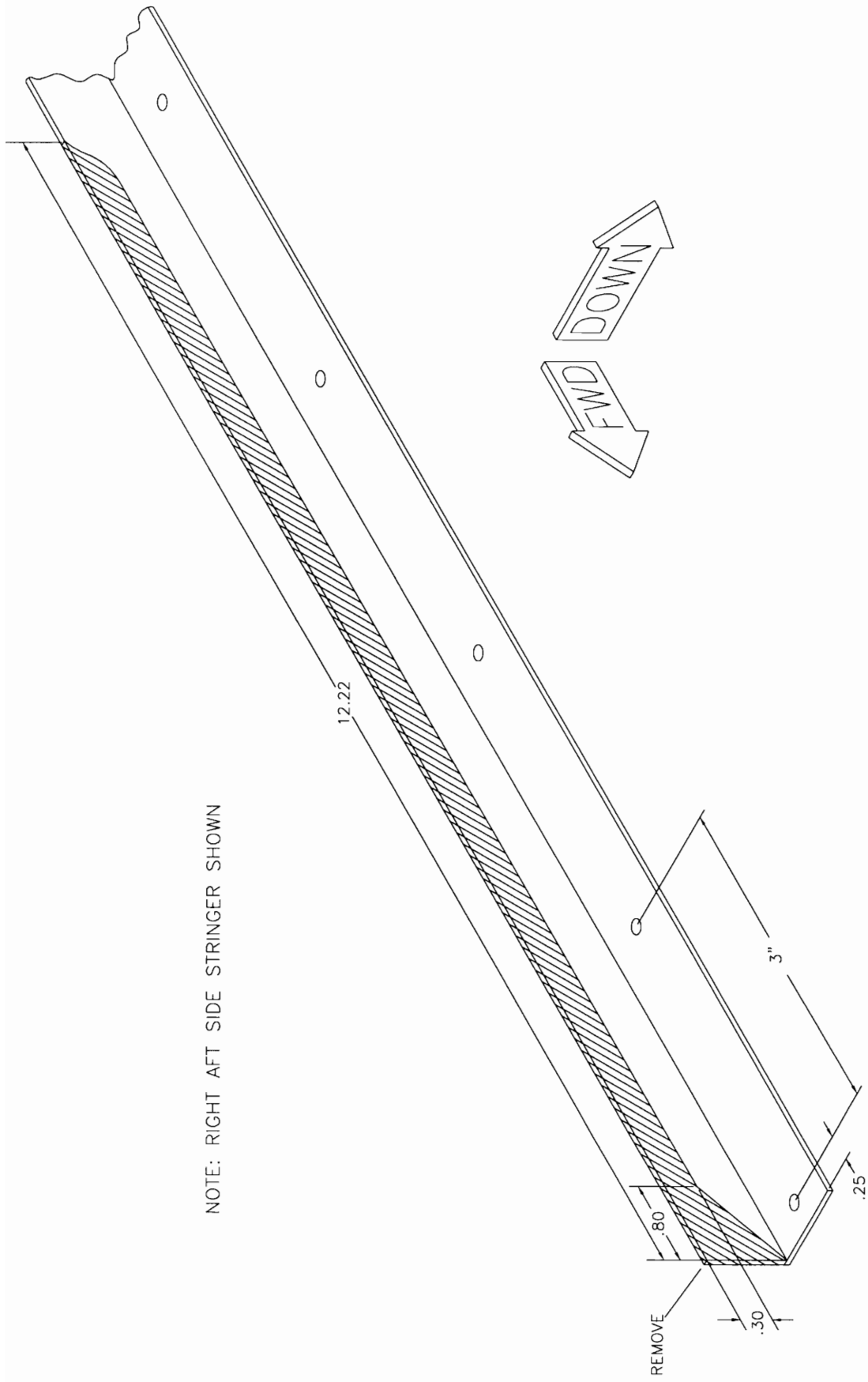
FABRICATE FROM .080 ALUMINUM RAW STOCK

FIGURE 2-27



NOTE: POSITION LONGEST LEG OF  
 JOGGLE JIG ON BOTTOM SIDE OF STRINGER

FIGURE 2-28



NOTE: RIGHT AFT SIDE STRINGER SHOWN

FIGURE 2-29

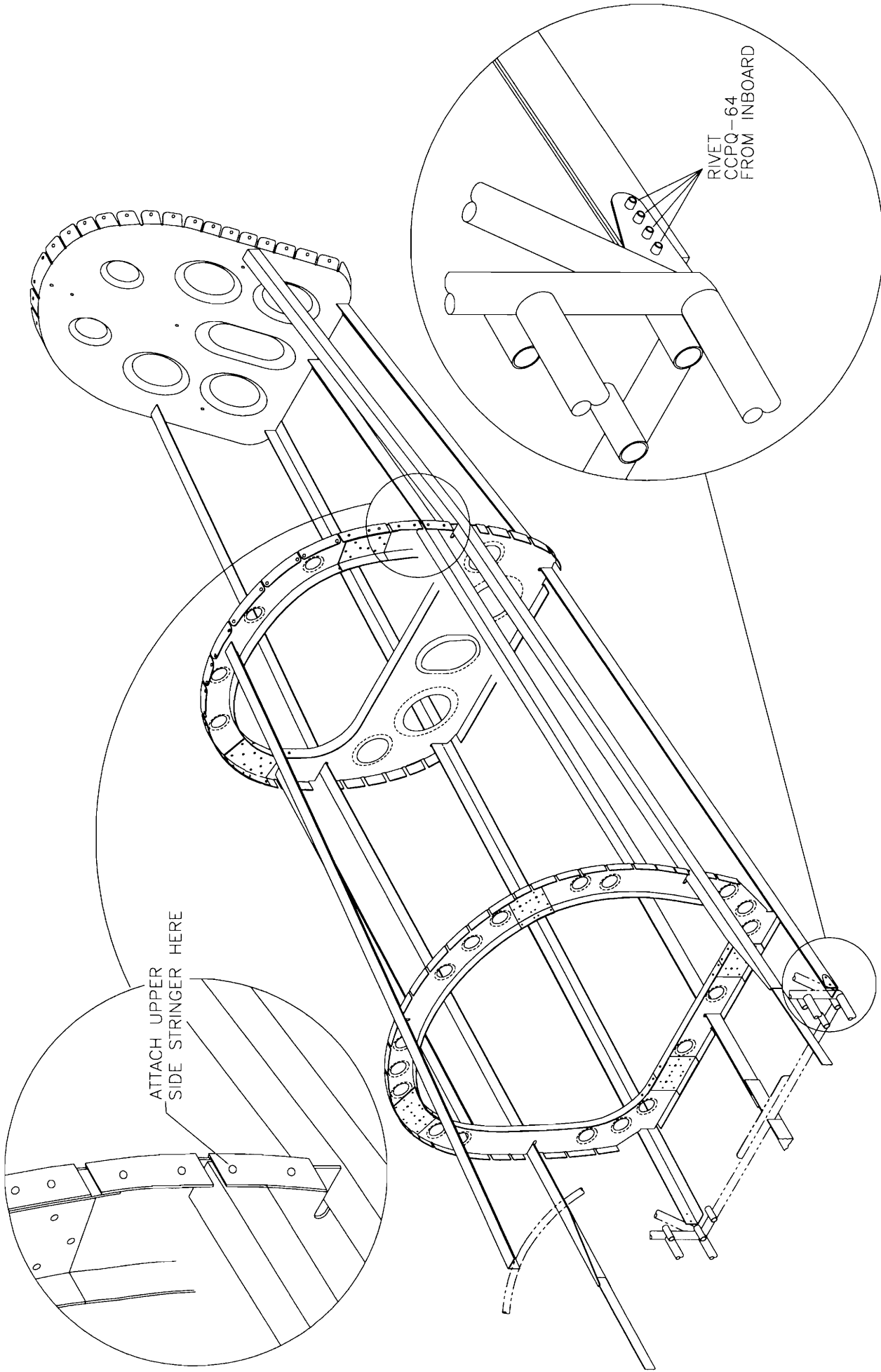
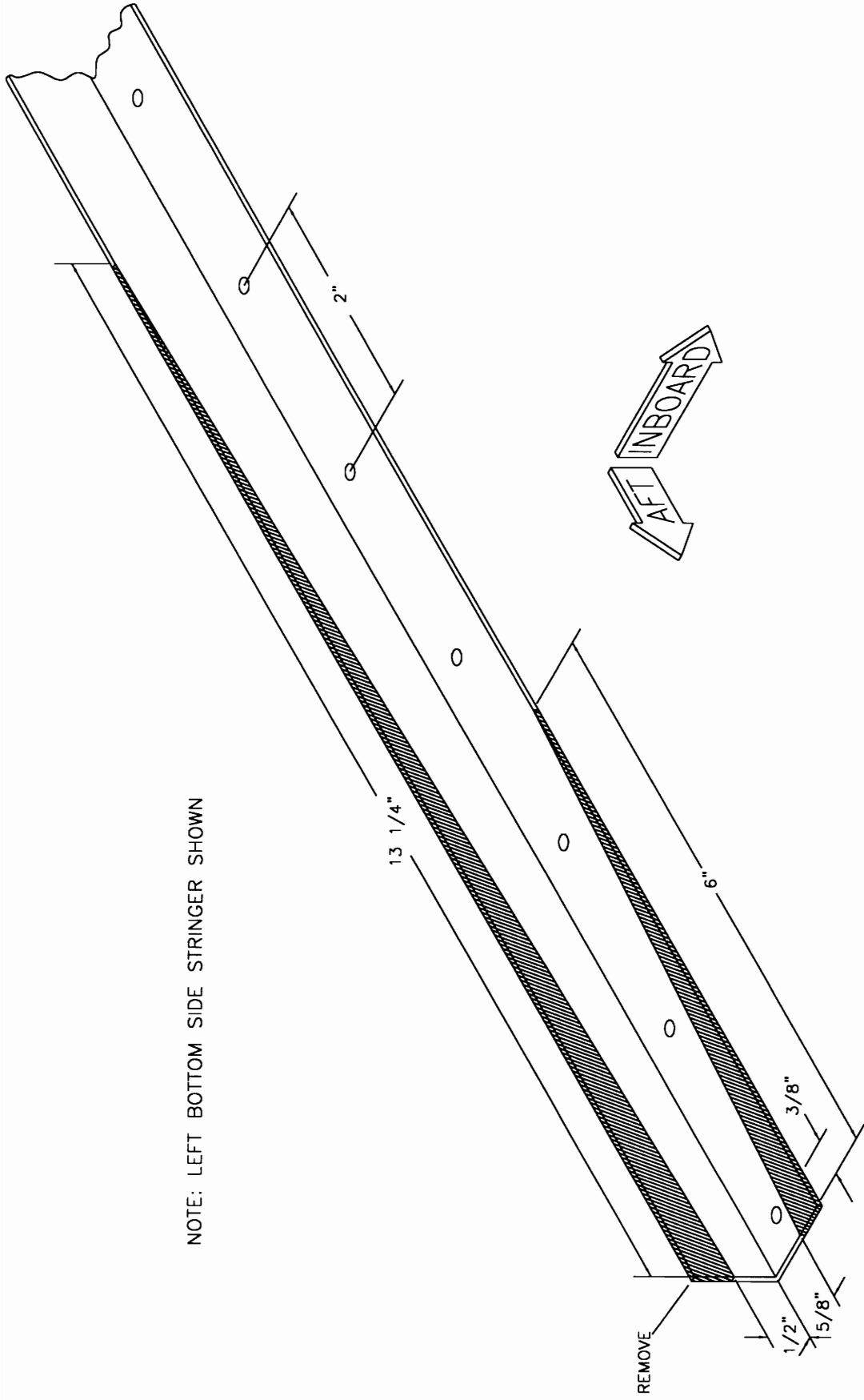


FIGURE 2-30





NOTE: LEFT BOTTOM SIDE STRINGER SHOWN

FIGURE 2-31

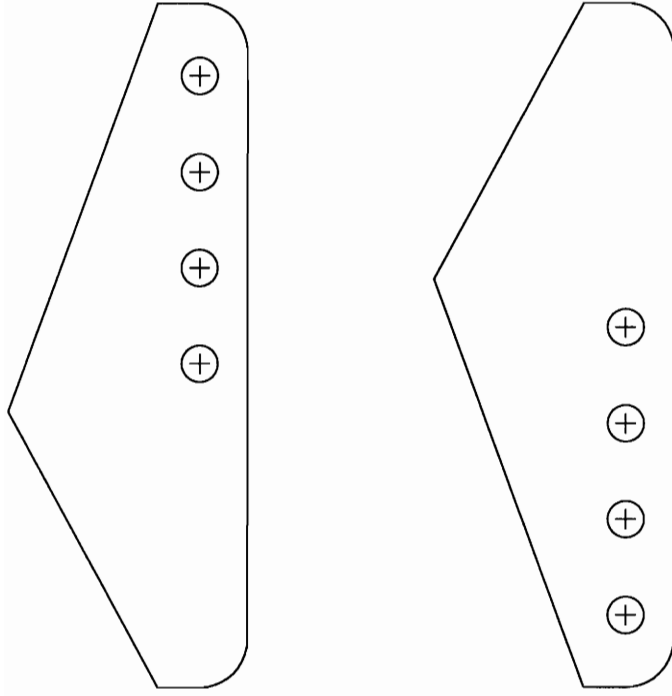
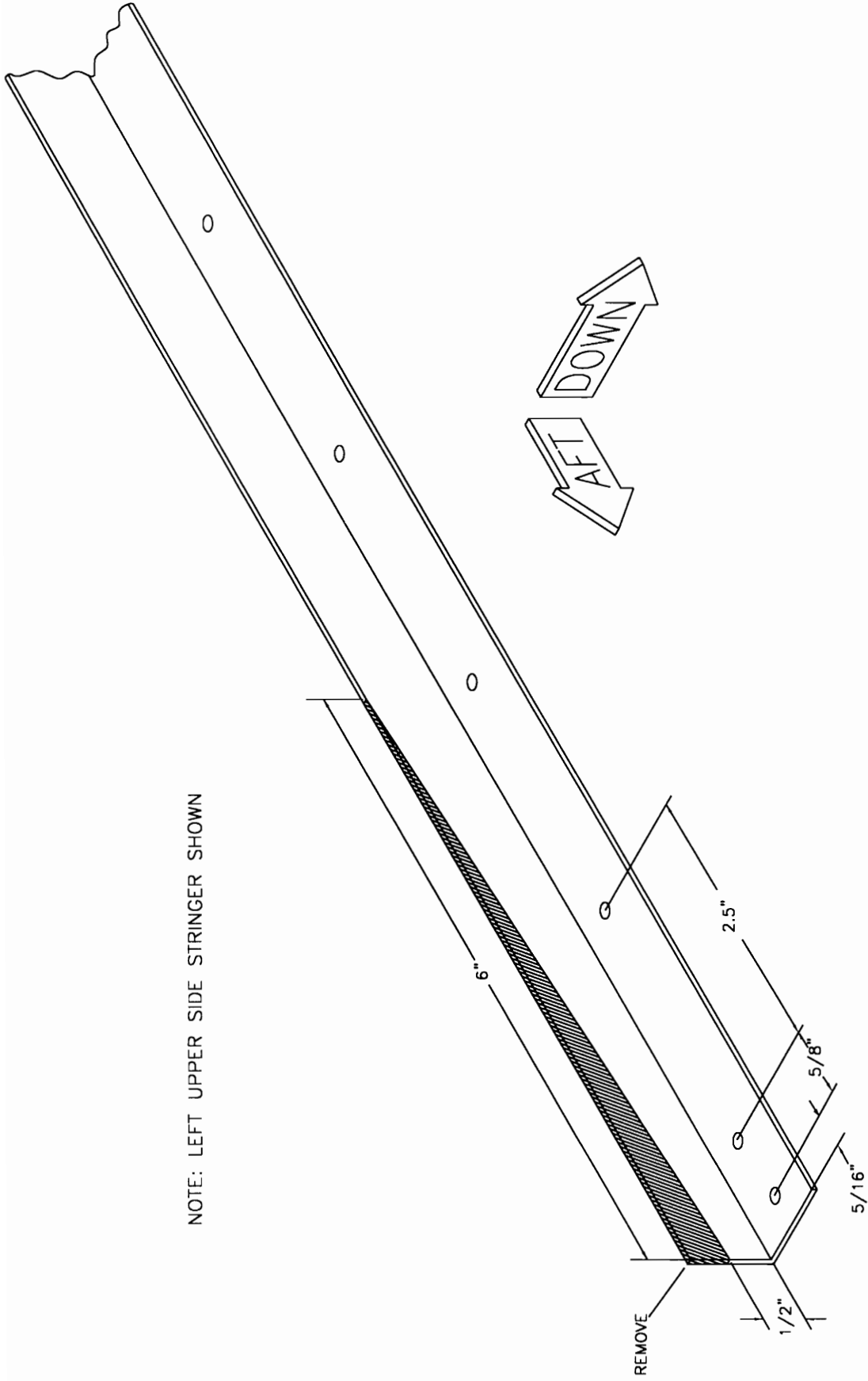


FIGURE 2-32



NOTE: LEFT UPPER SIDE STRINGER SHOWN

FIGURE 2-33

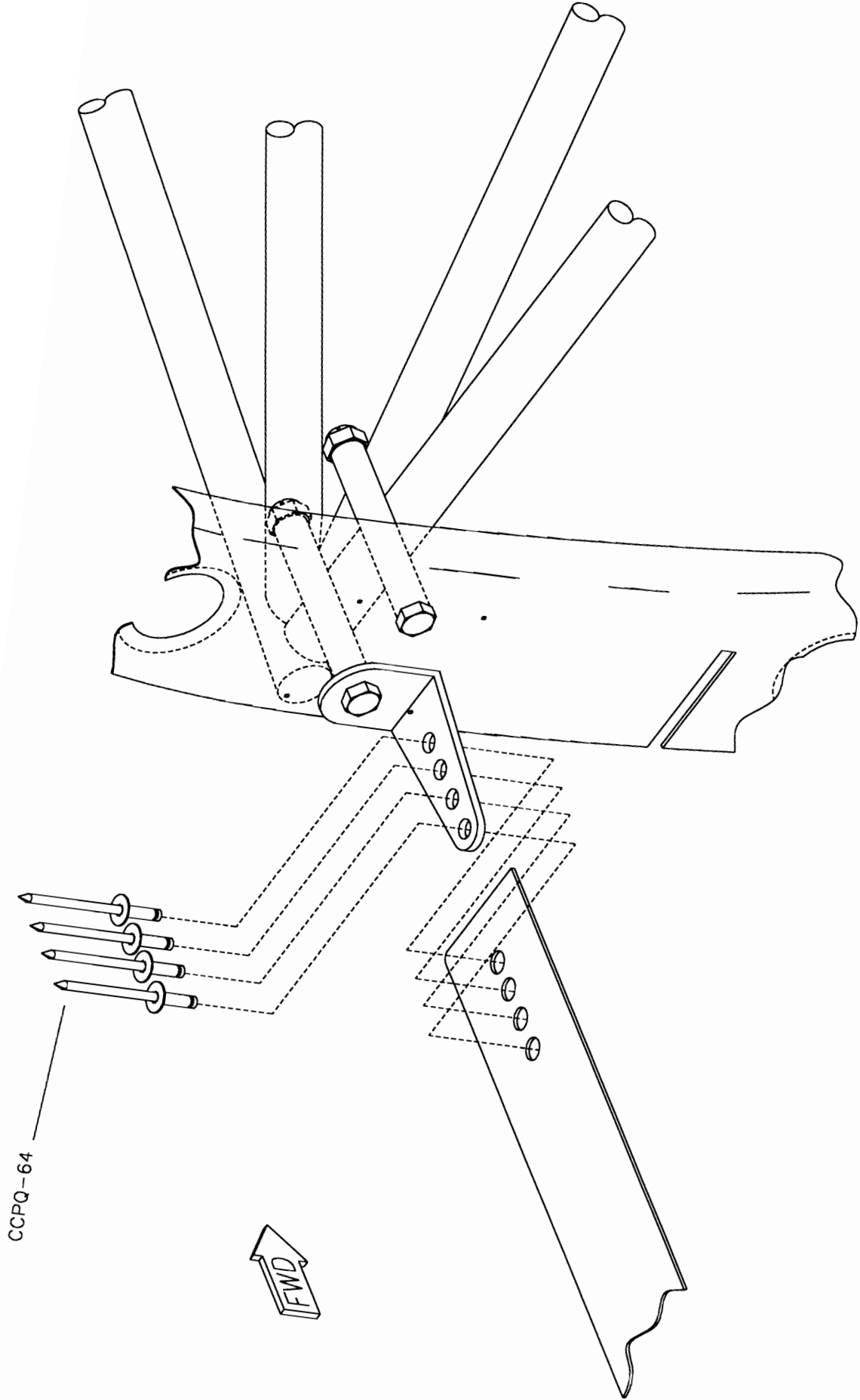
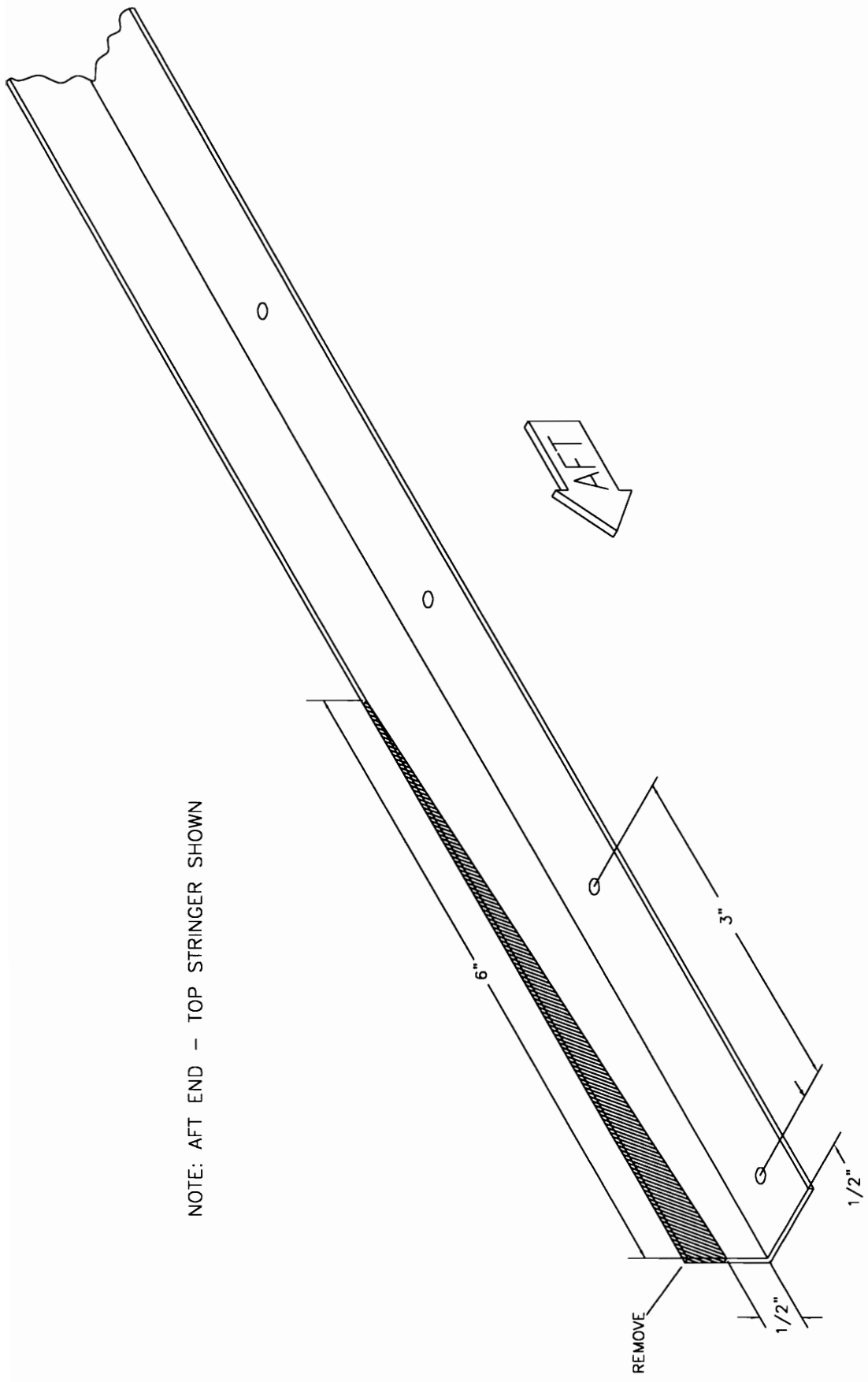


FIGURE 2-34



NOTE: AFT END - TOP STRINGER SHOWN

FIGURE 2-35

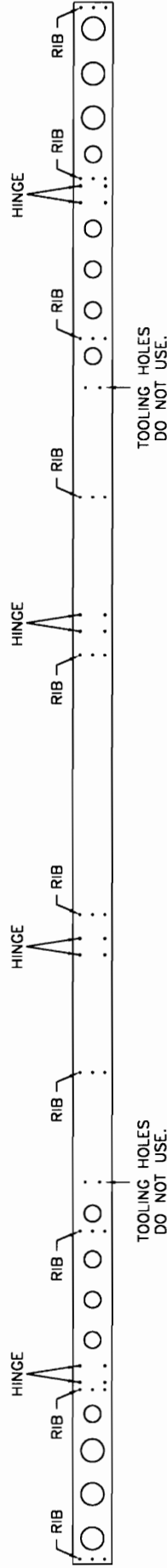


FIGURE 2-36

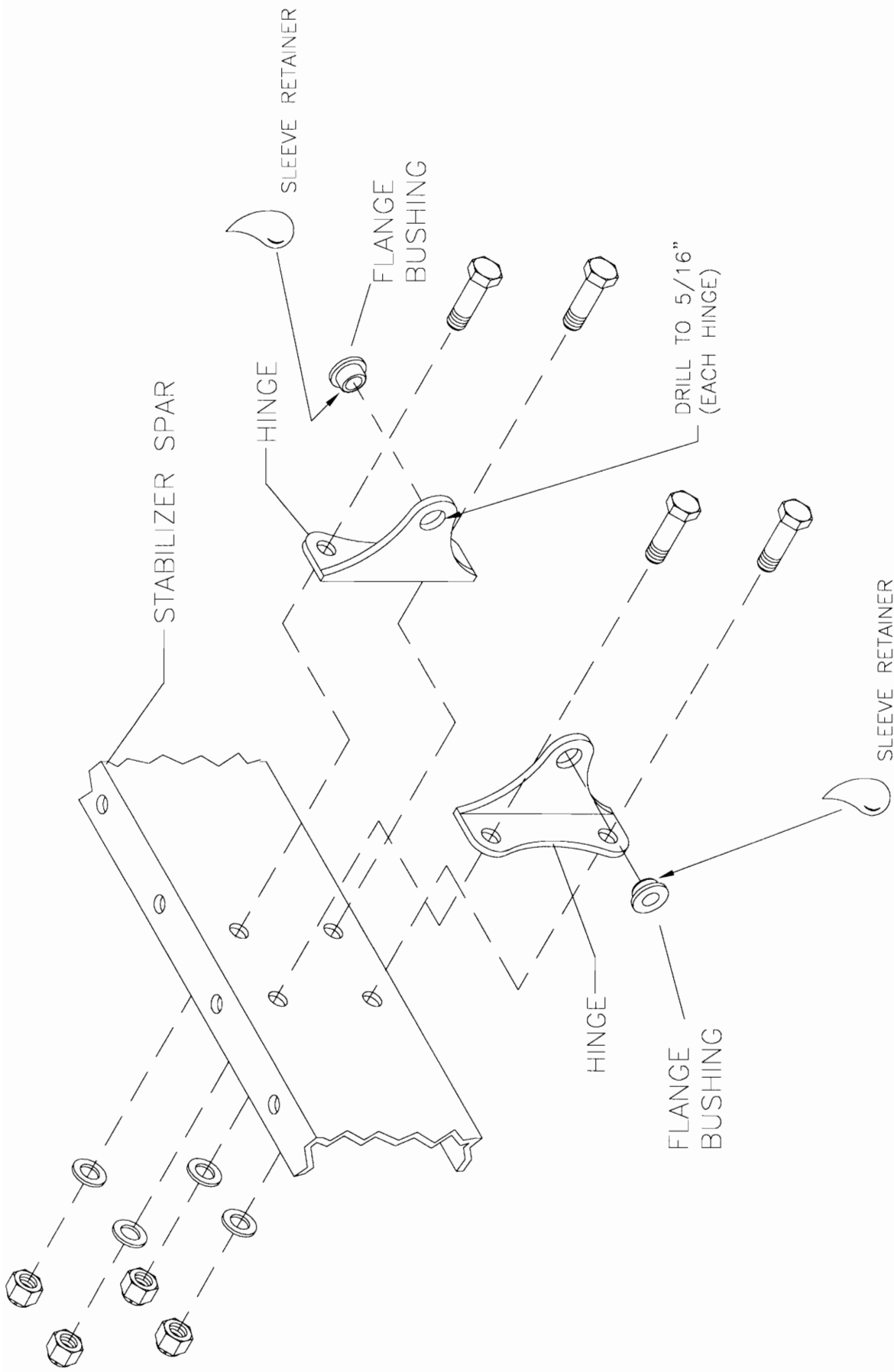


FIGURE 2-37

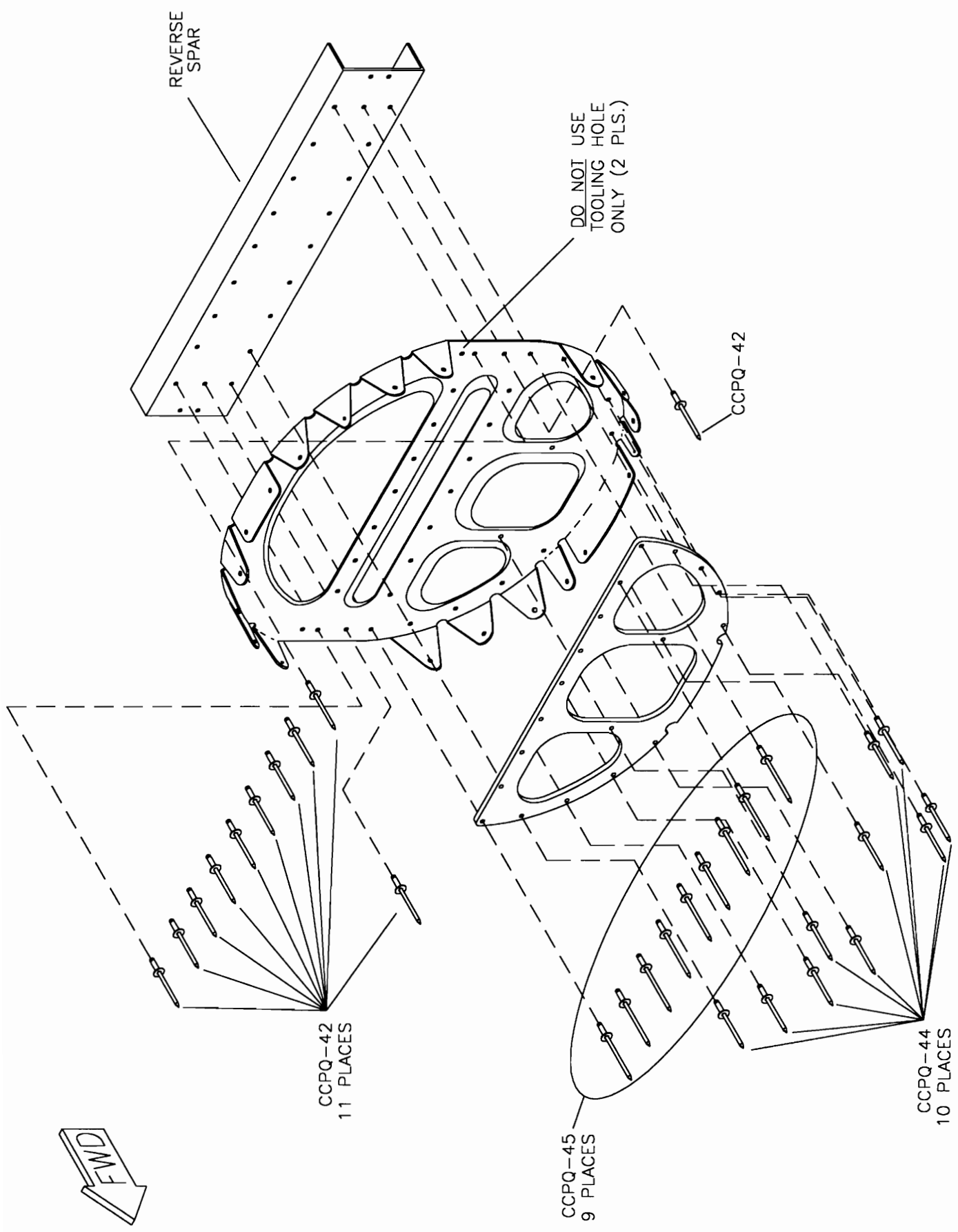


FIGURE 2-38



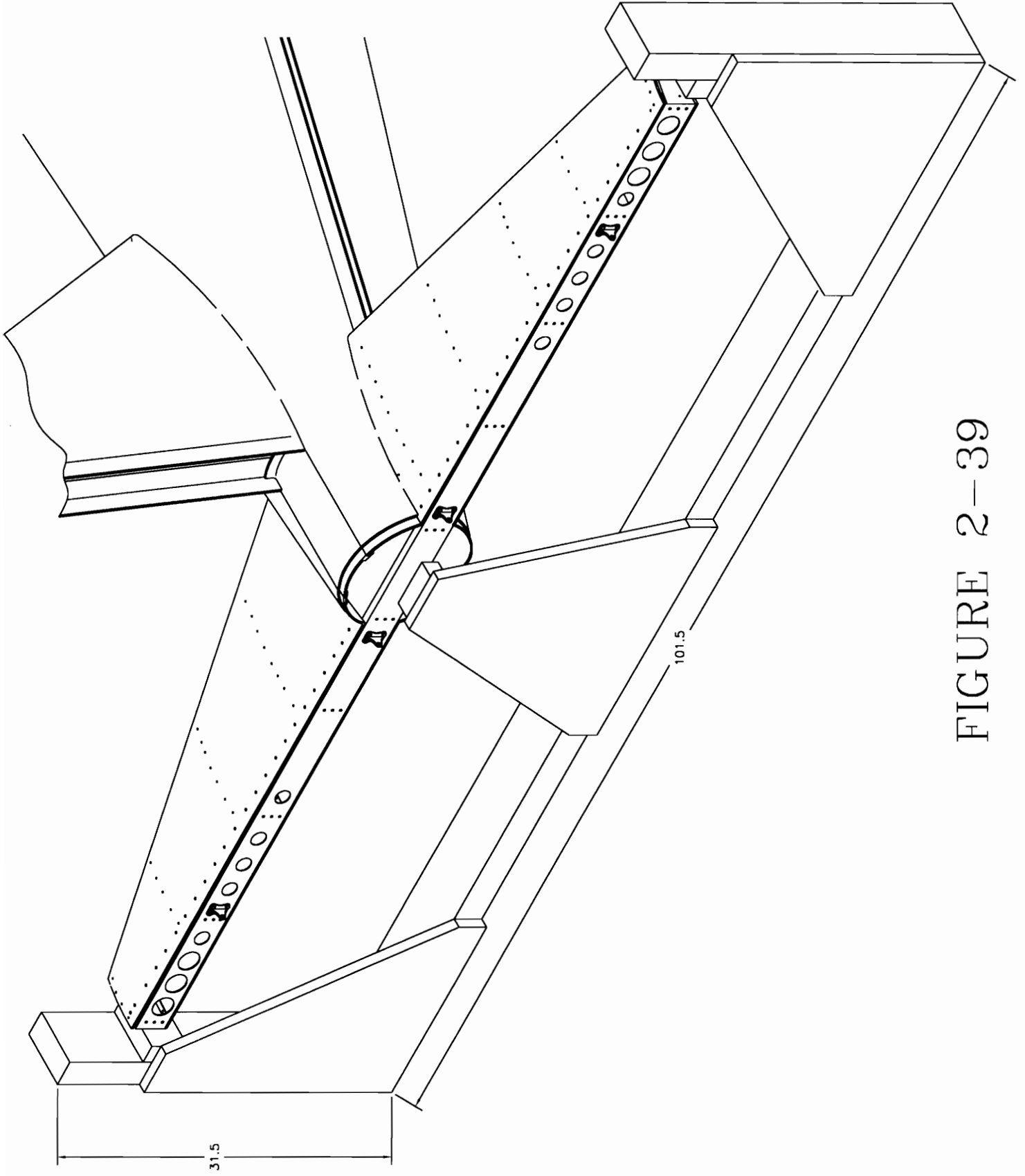


FIGURE 2-39

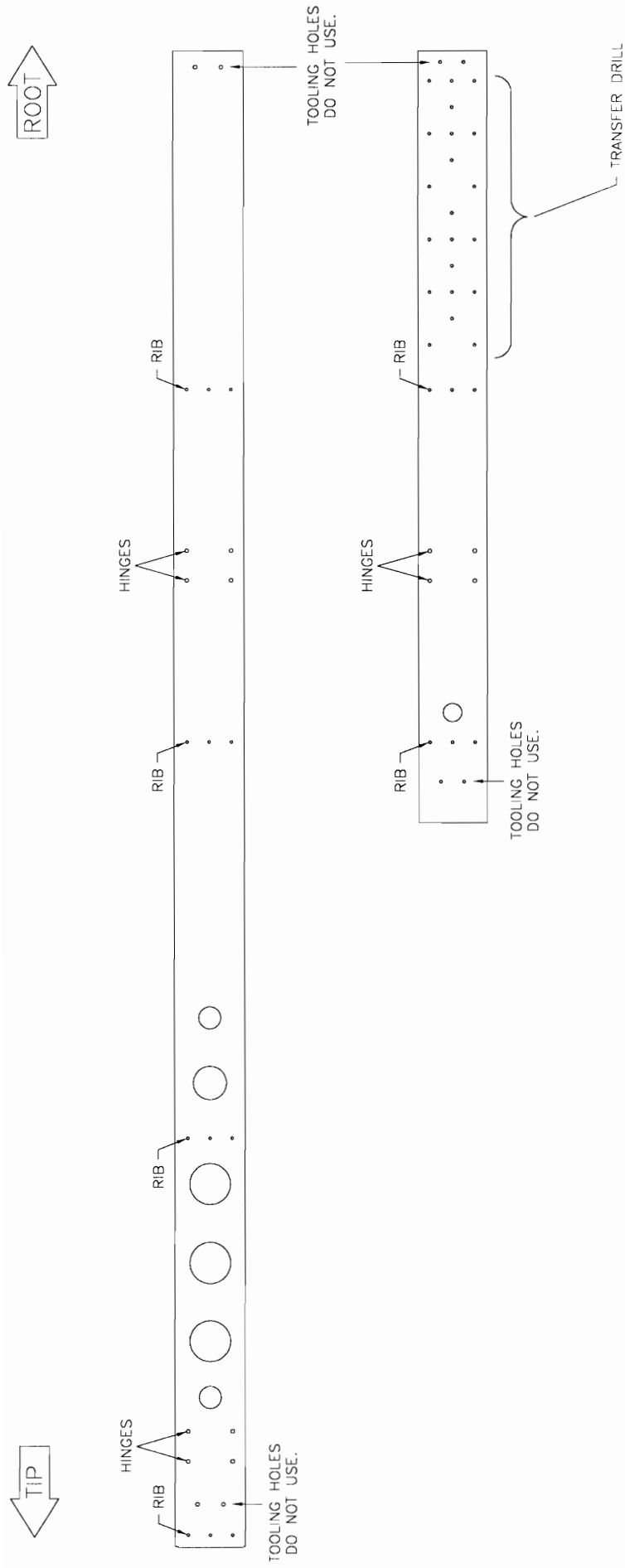


FIGURE 2-40

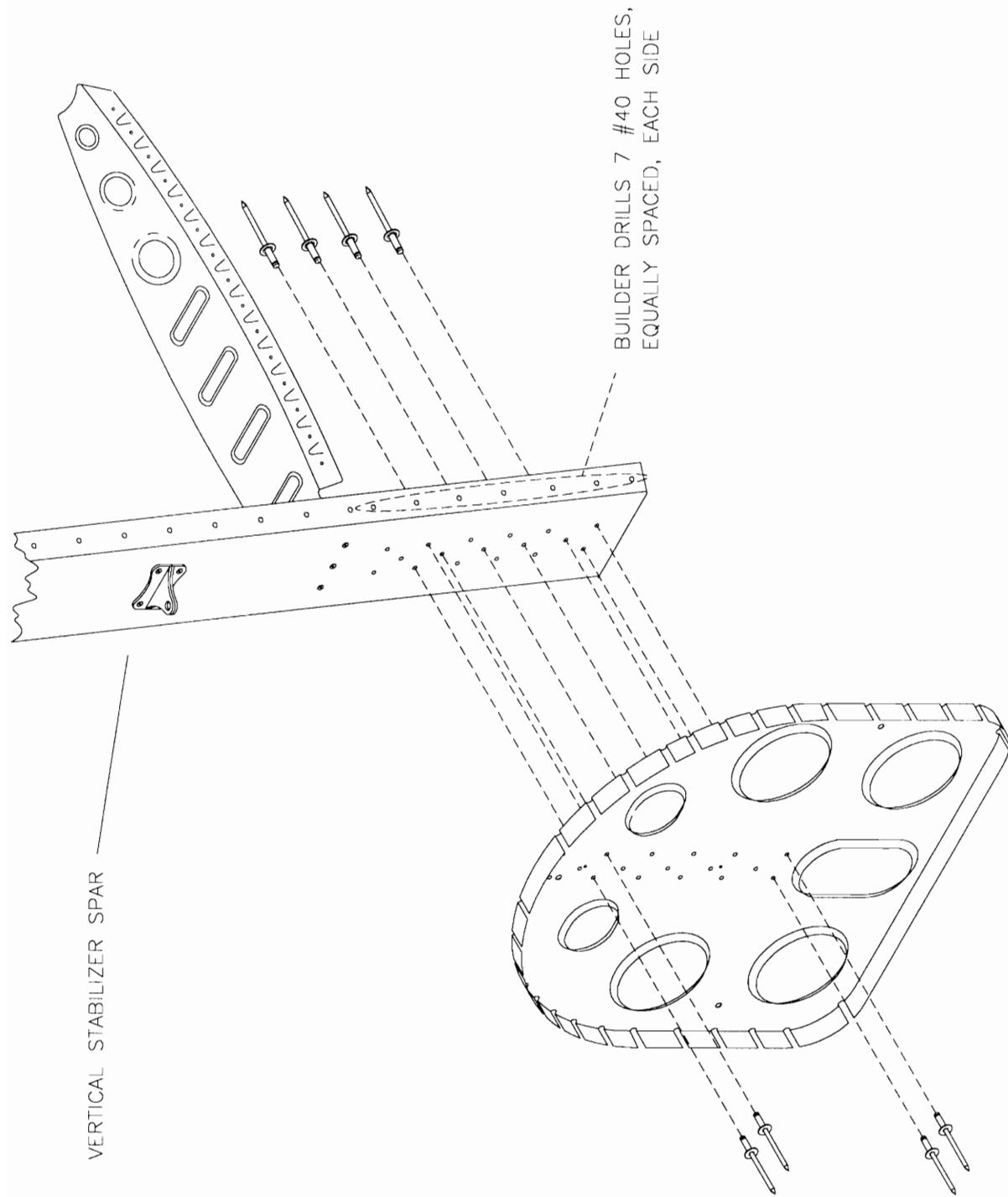


FIGURE 2-41

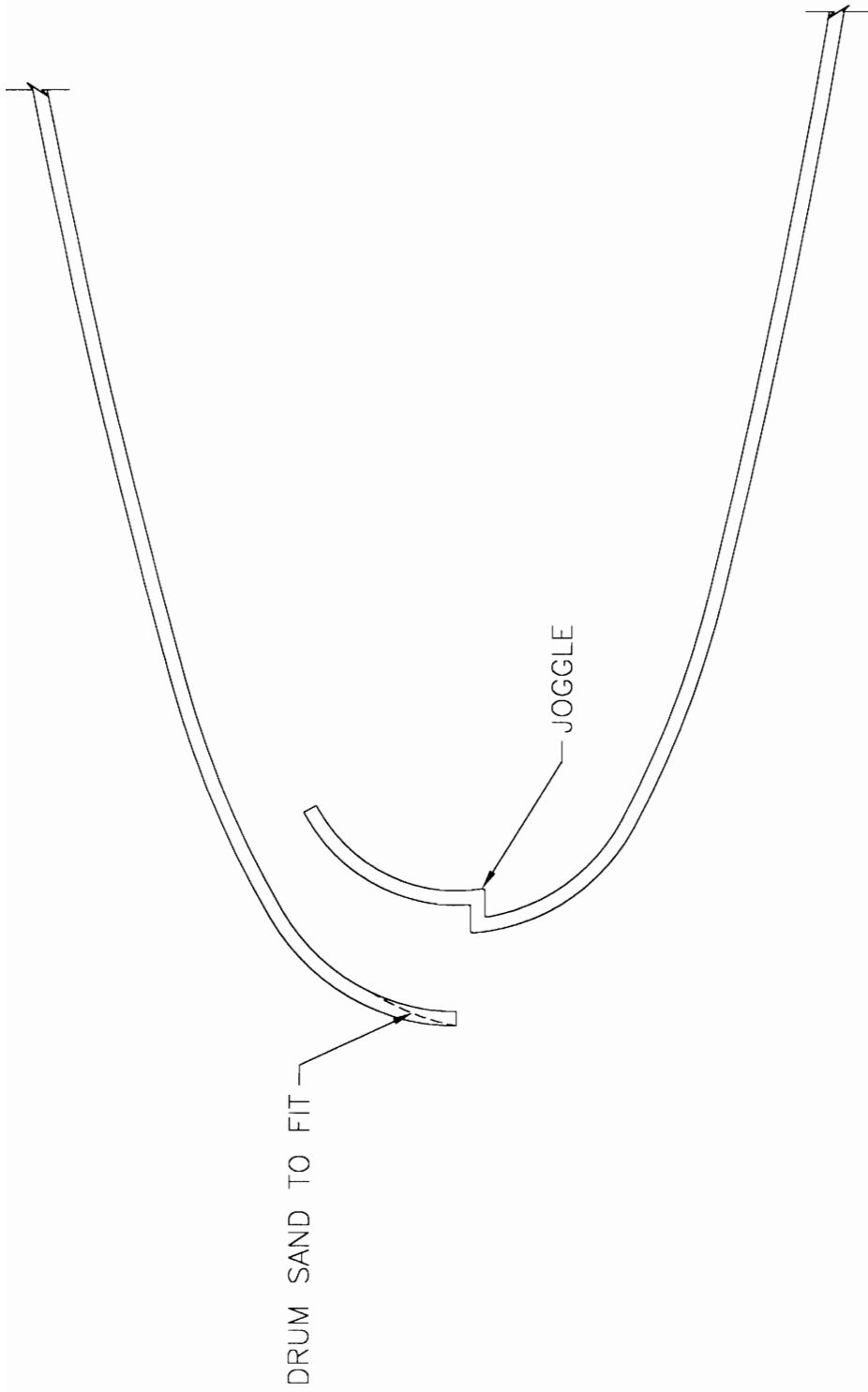


FIGURE 2-42

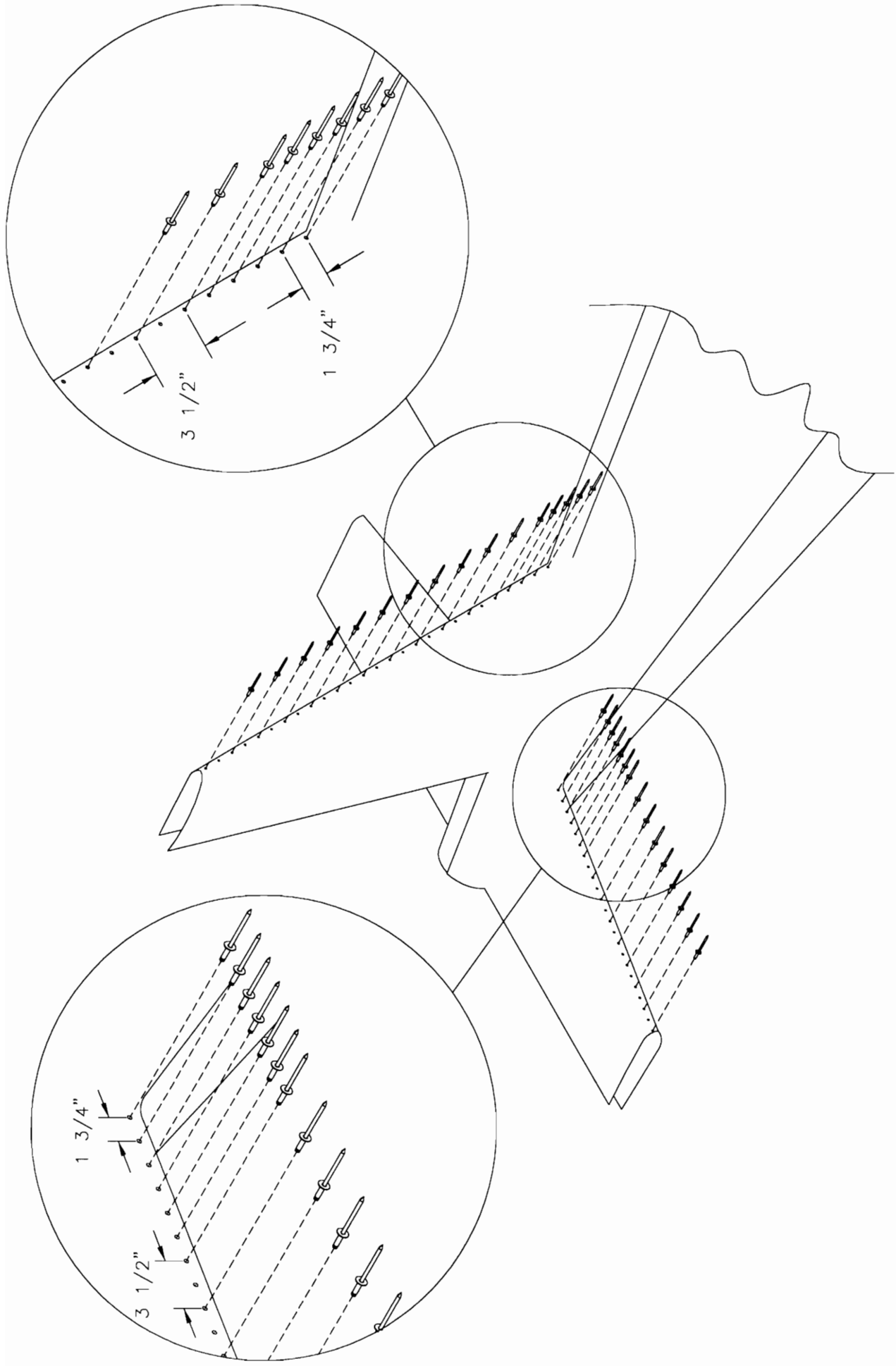


FIGURE 2-43

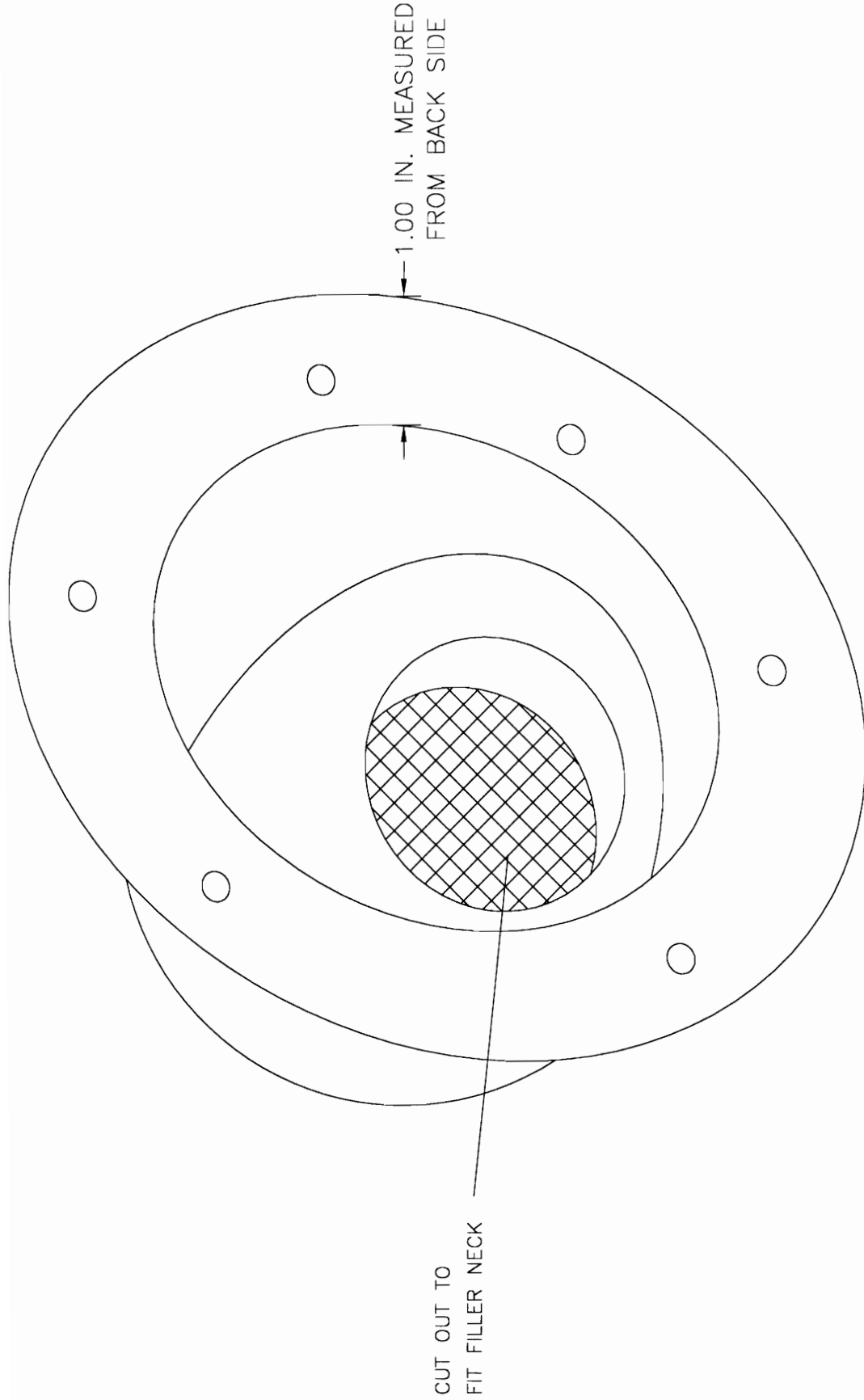


FIGURE 2-44

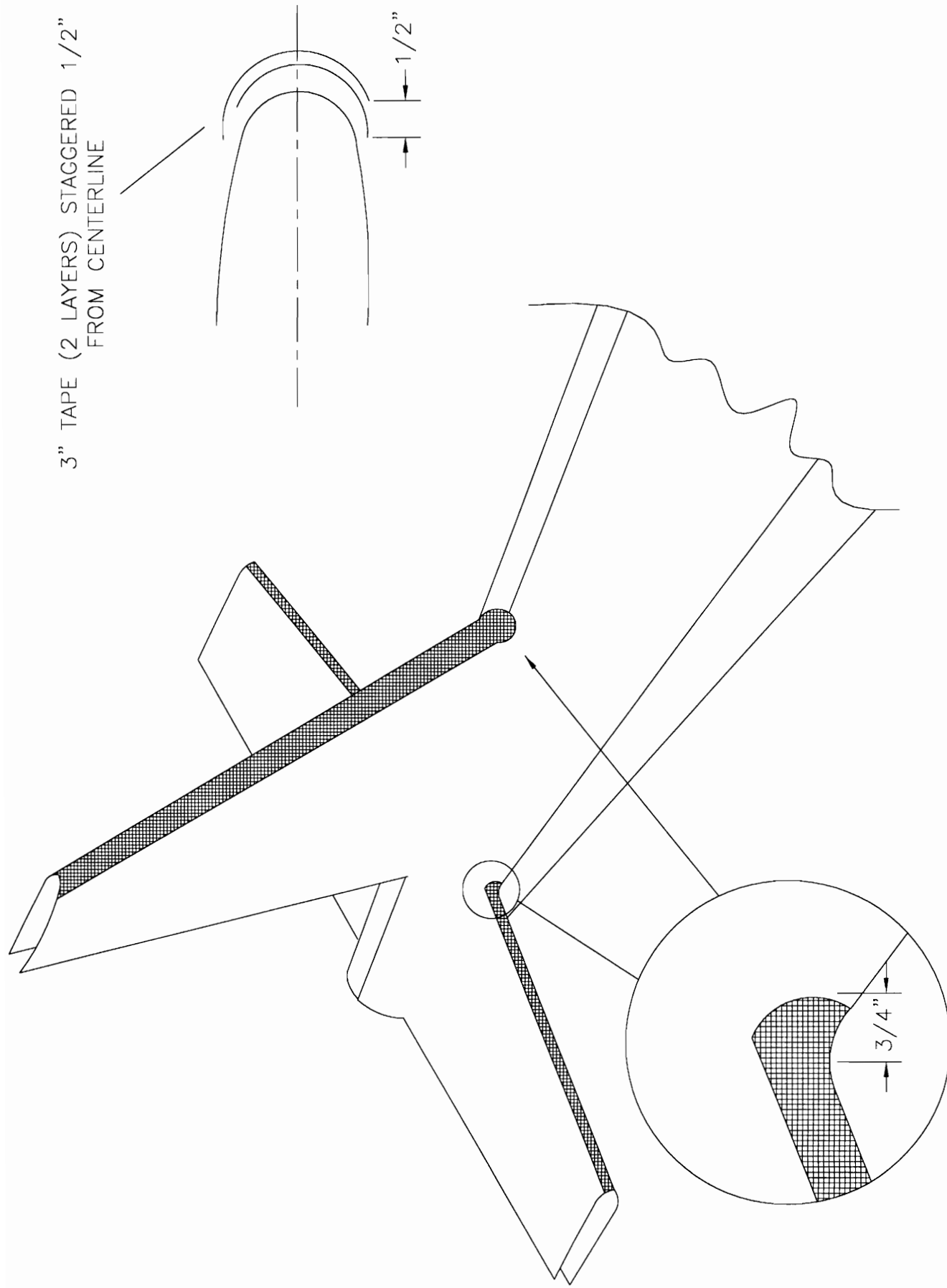


FIGURE 2-45

3/4" AROUND CORNER

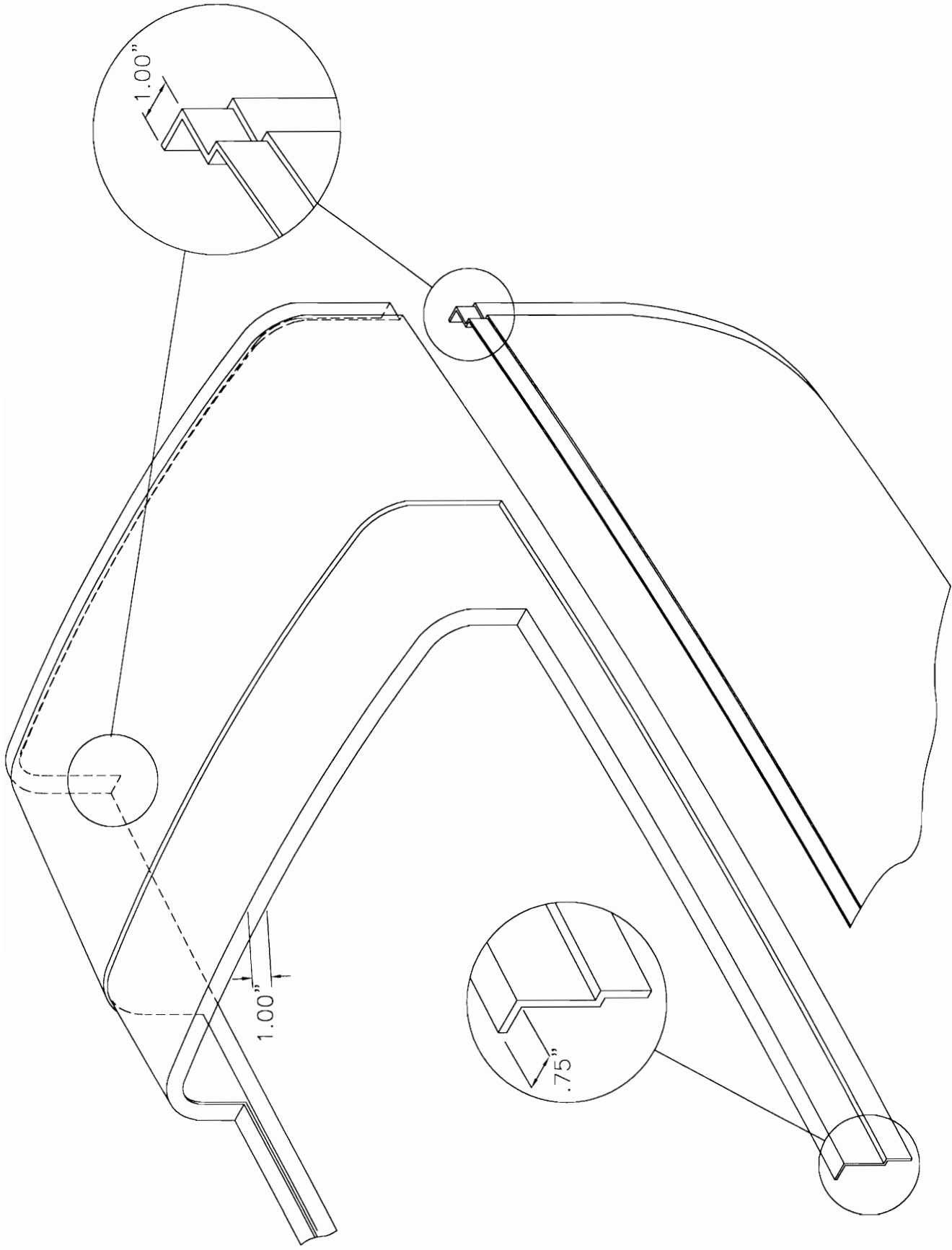


FIGURE 2-46



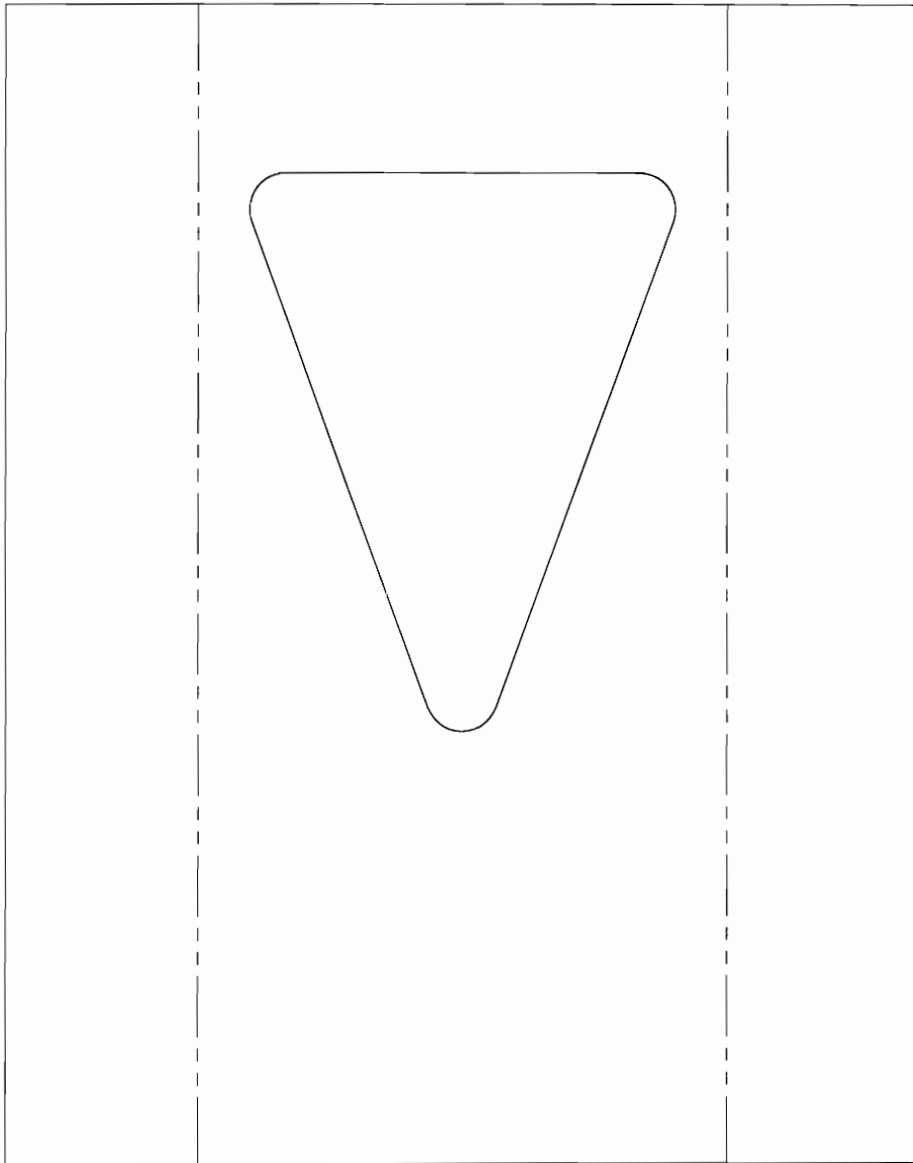
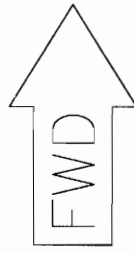


FIGURE 2-47

USE A TAILOR'S TAPE TO EVENLY SPACE 2 HOLES

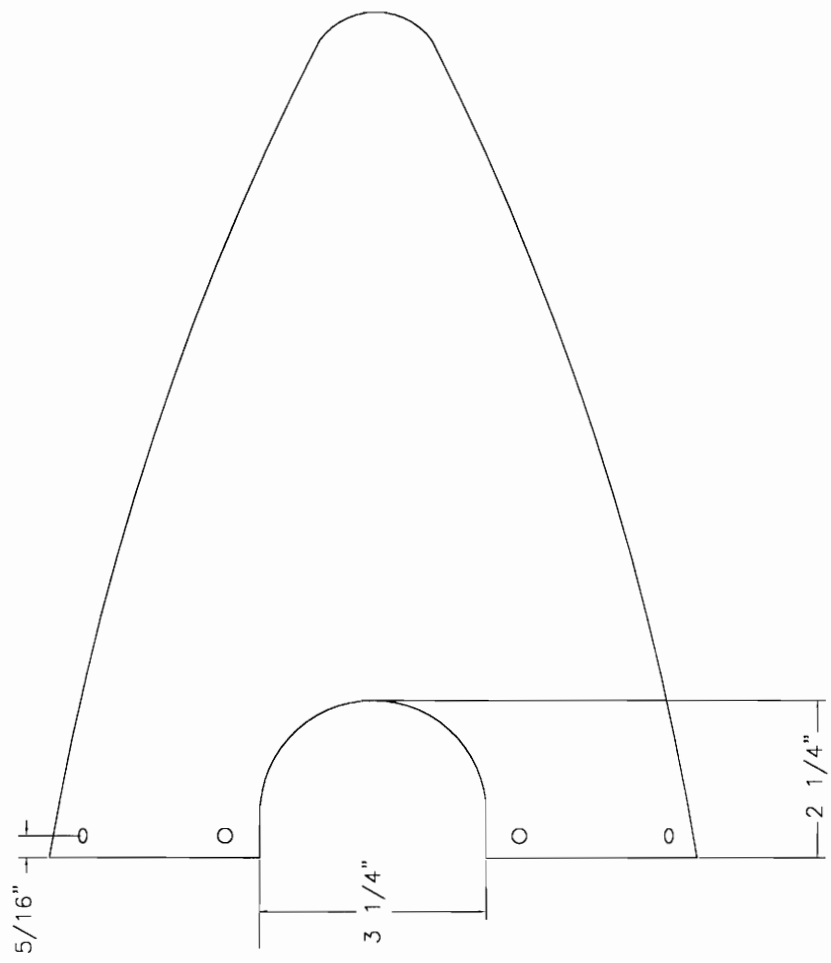
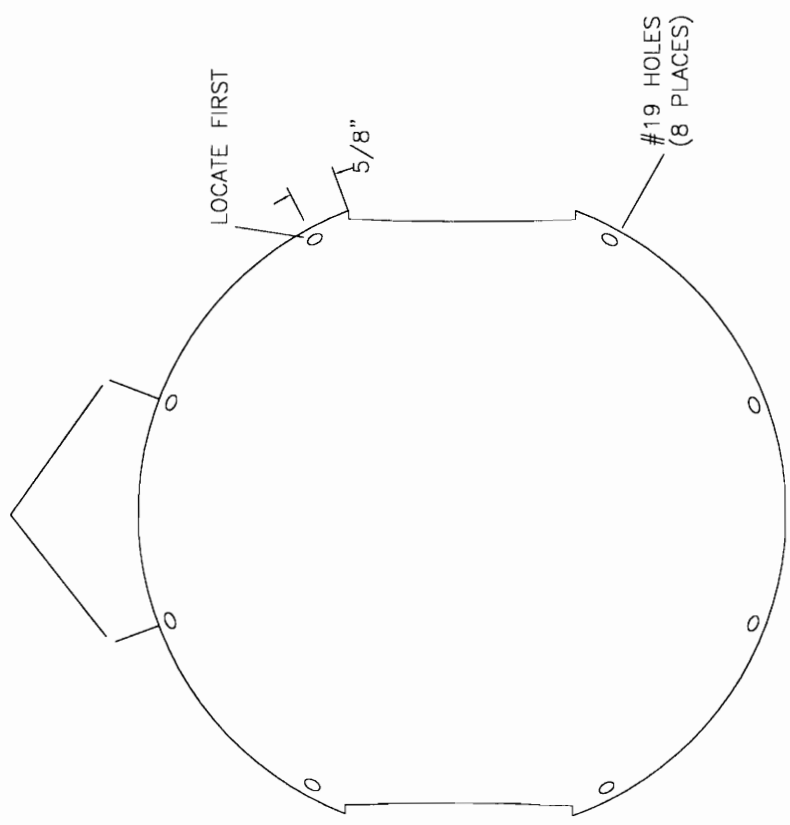


FIGURE 2-48

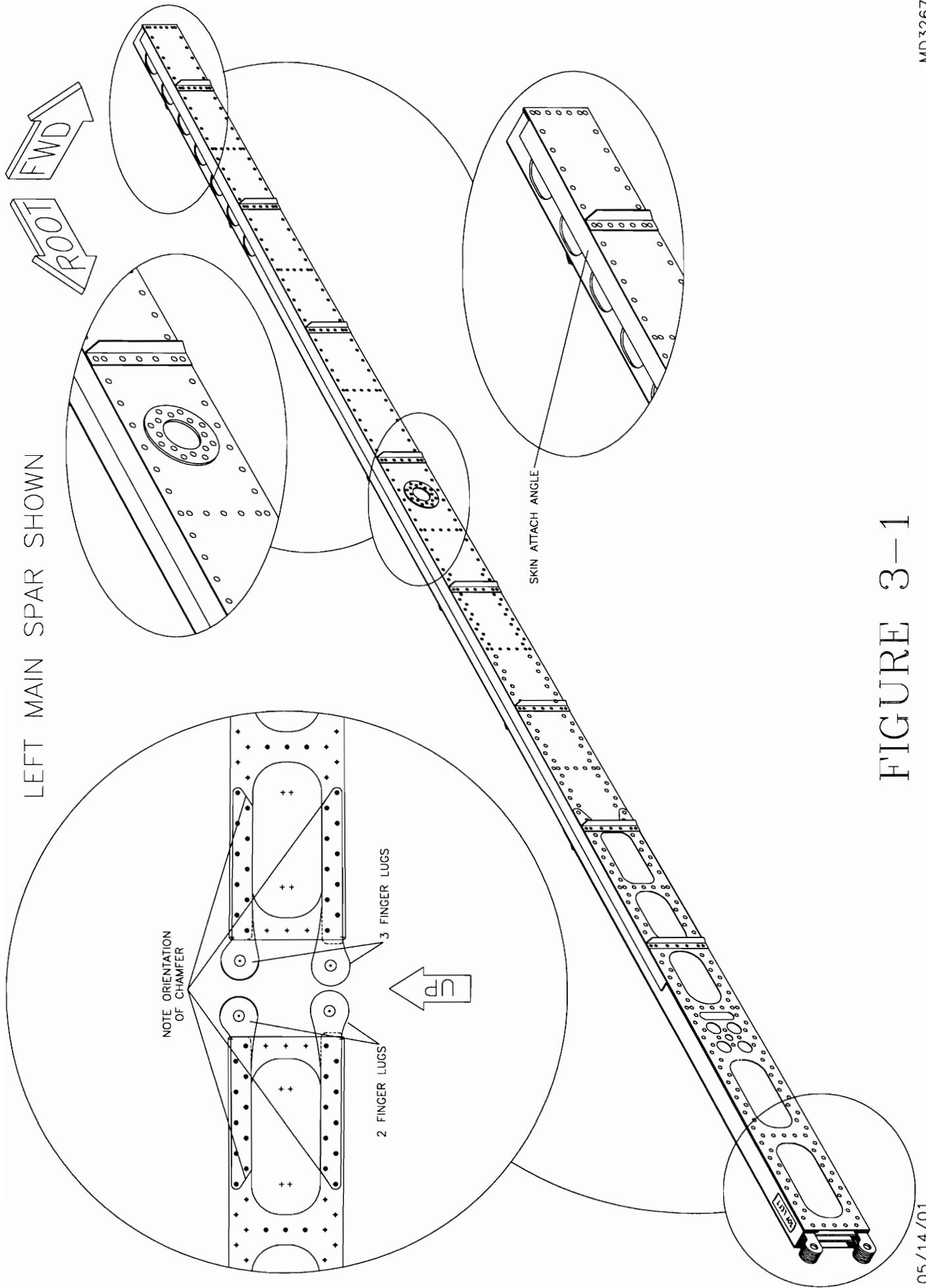
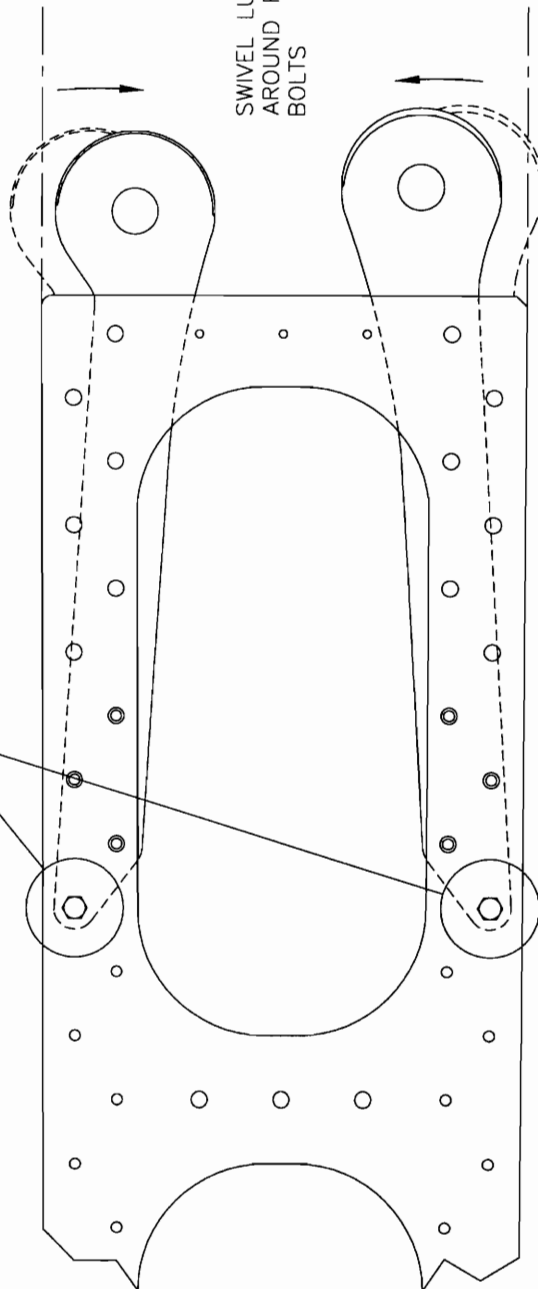


FIGURE 3-1

LEAVE ONLY THESE  
TWO BOLTS REMAINING



SWIVEL LUGS  
AROUND REAR  
BOLTS

REMOVE STIFFENER PLATES,  
BUT MARK THEIR POSITION  
FOR EASE AT TIME OF  
RE-ASSEMBLY.

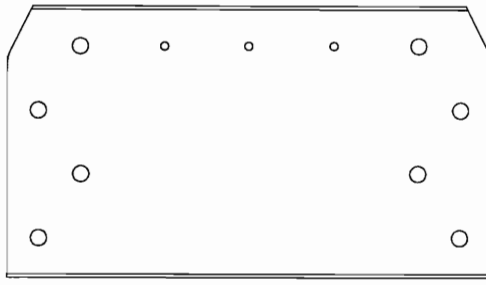
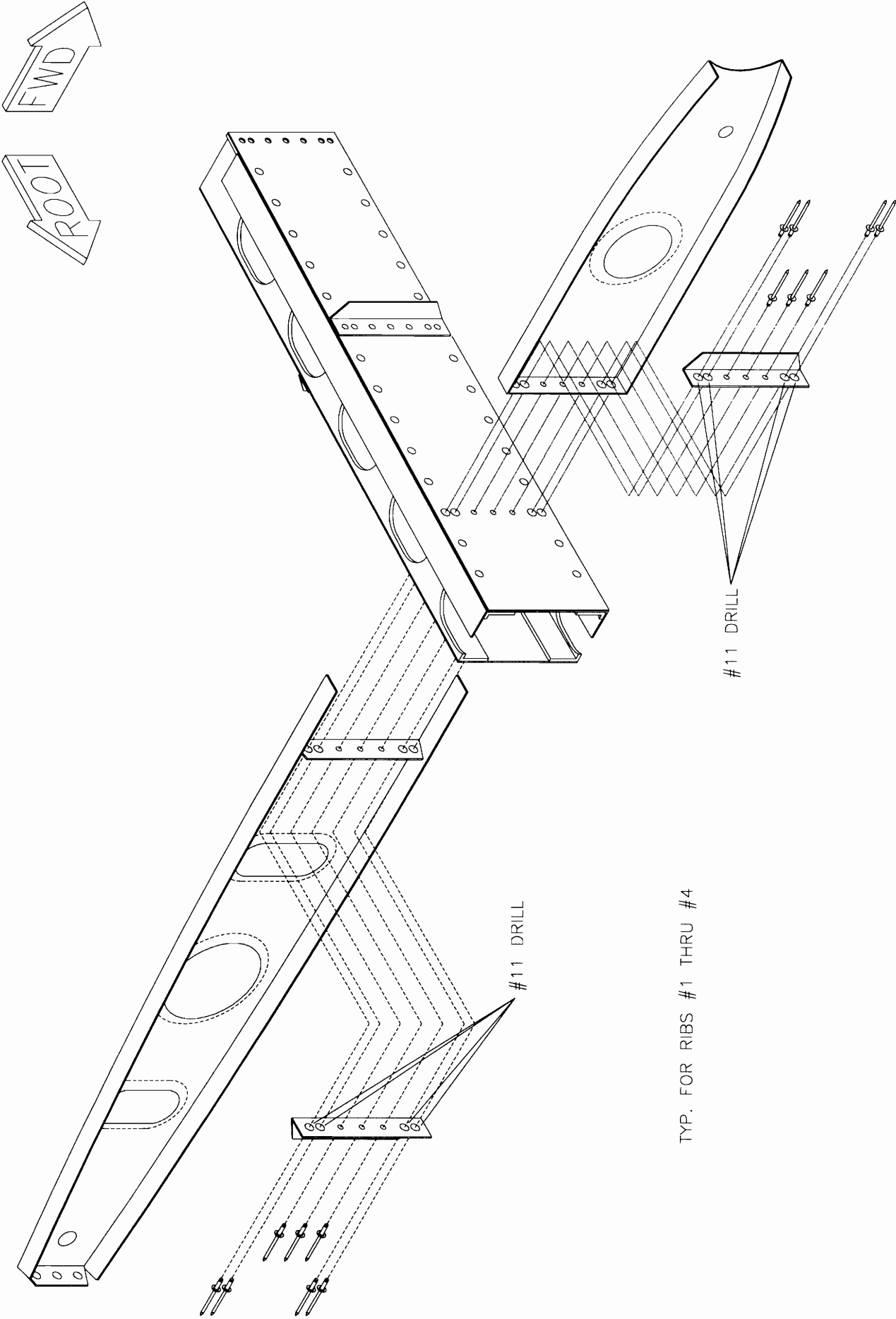


FIGURE 3-2



TYP. FOR RIBS #1 THRU #4

FIGURE 3-3

EXTRA HOLE  
DENOTES TOP CONTOUR

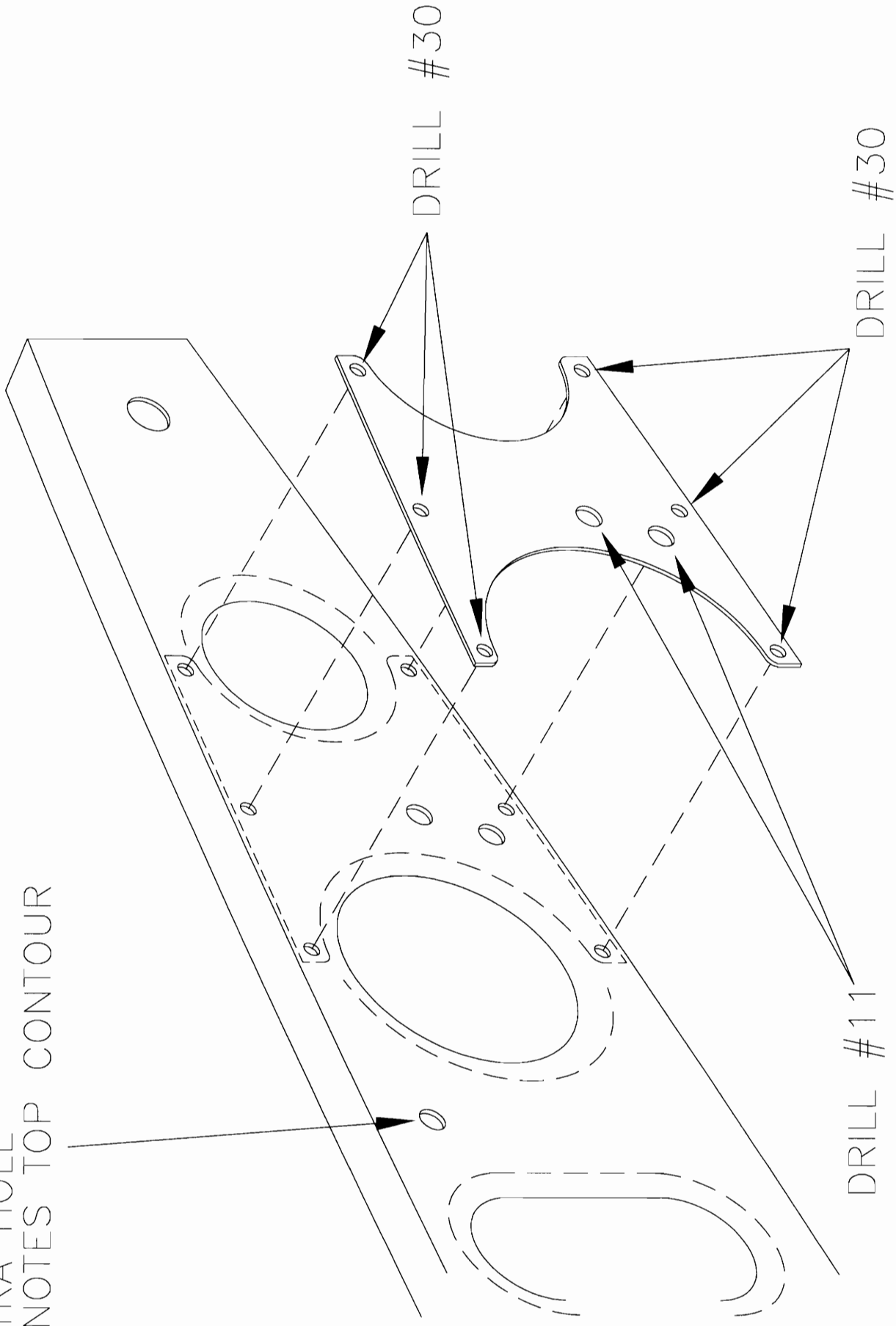


FIGURE 3-4

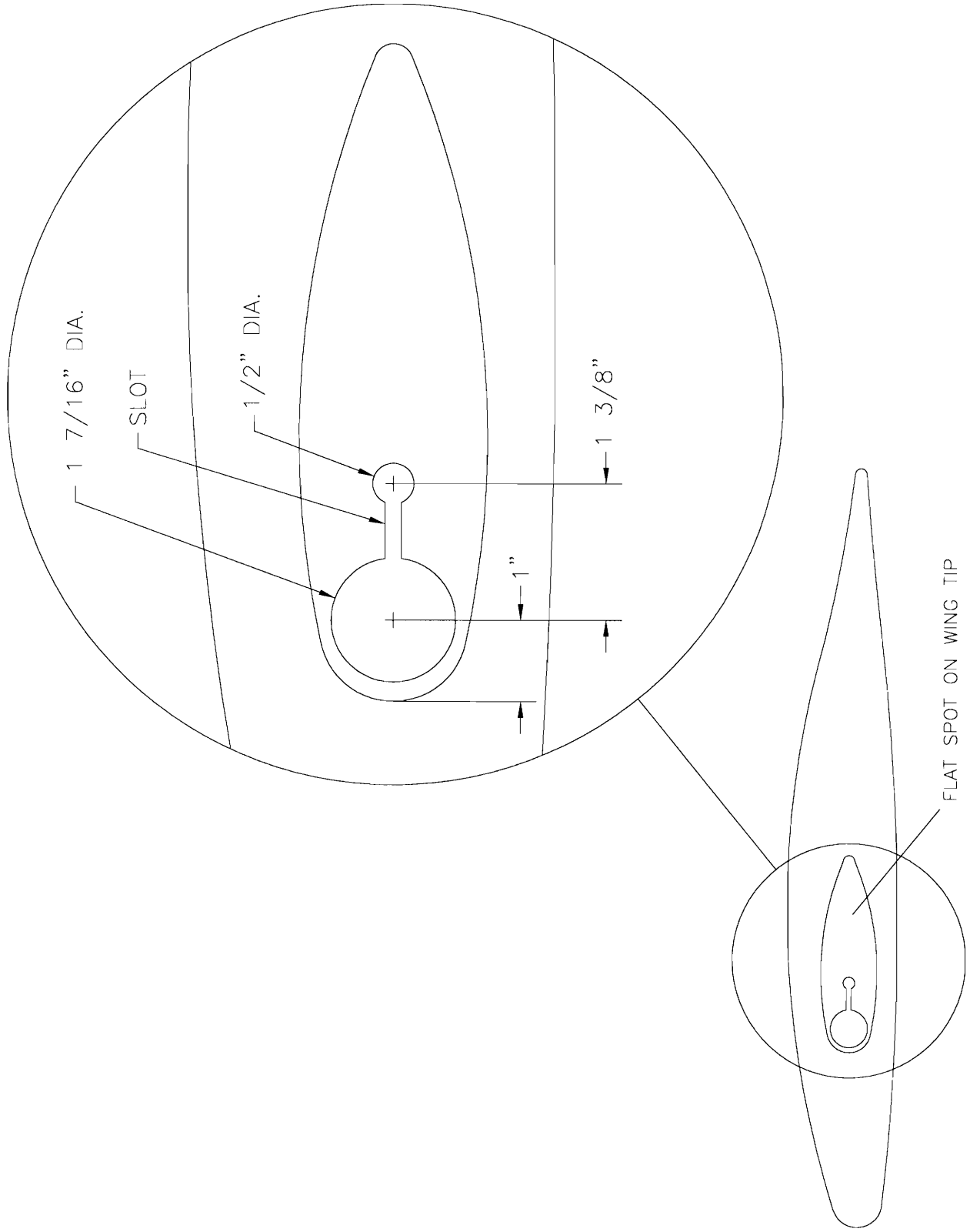


FIGURE 3-4A

LEFT TRAILING EDGE SPAR  
AFT. VIEW

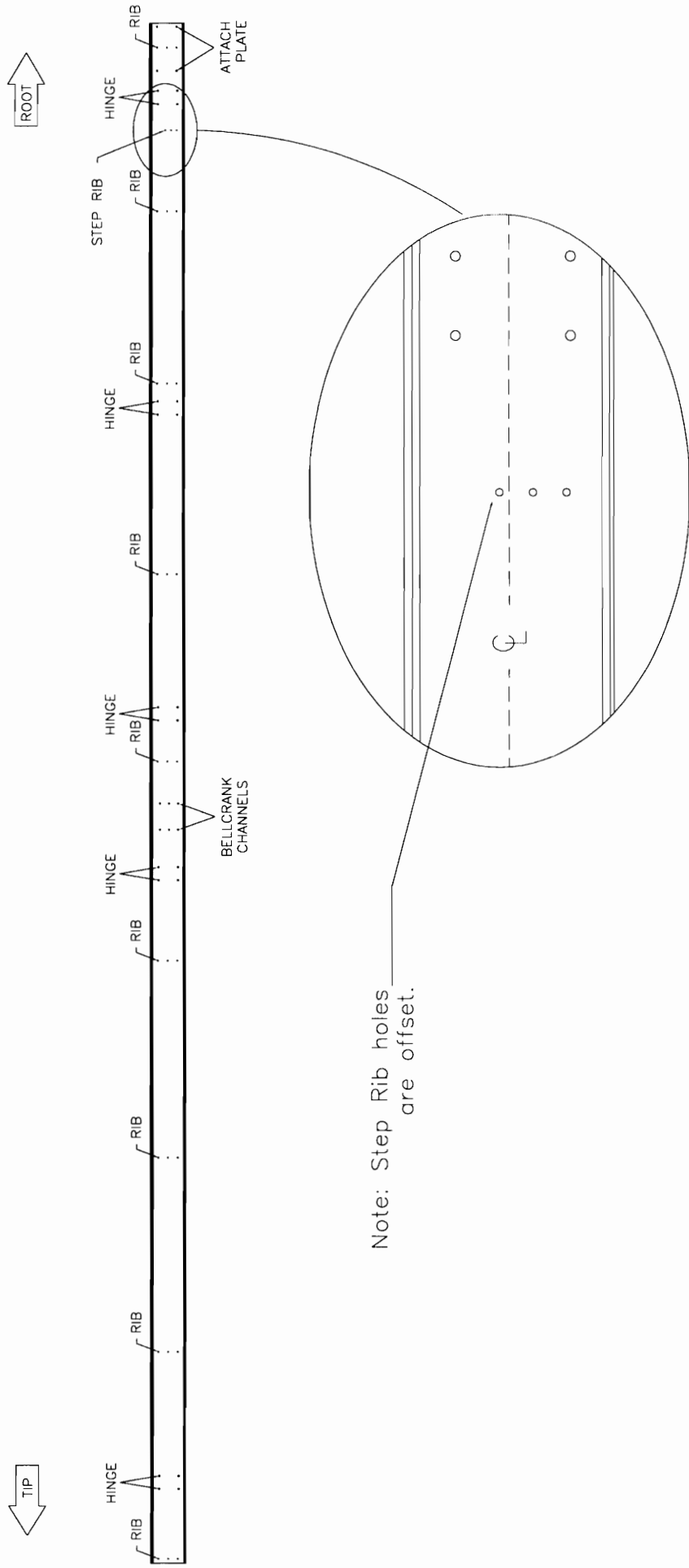


FIGURE 3-5



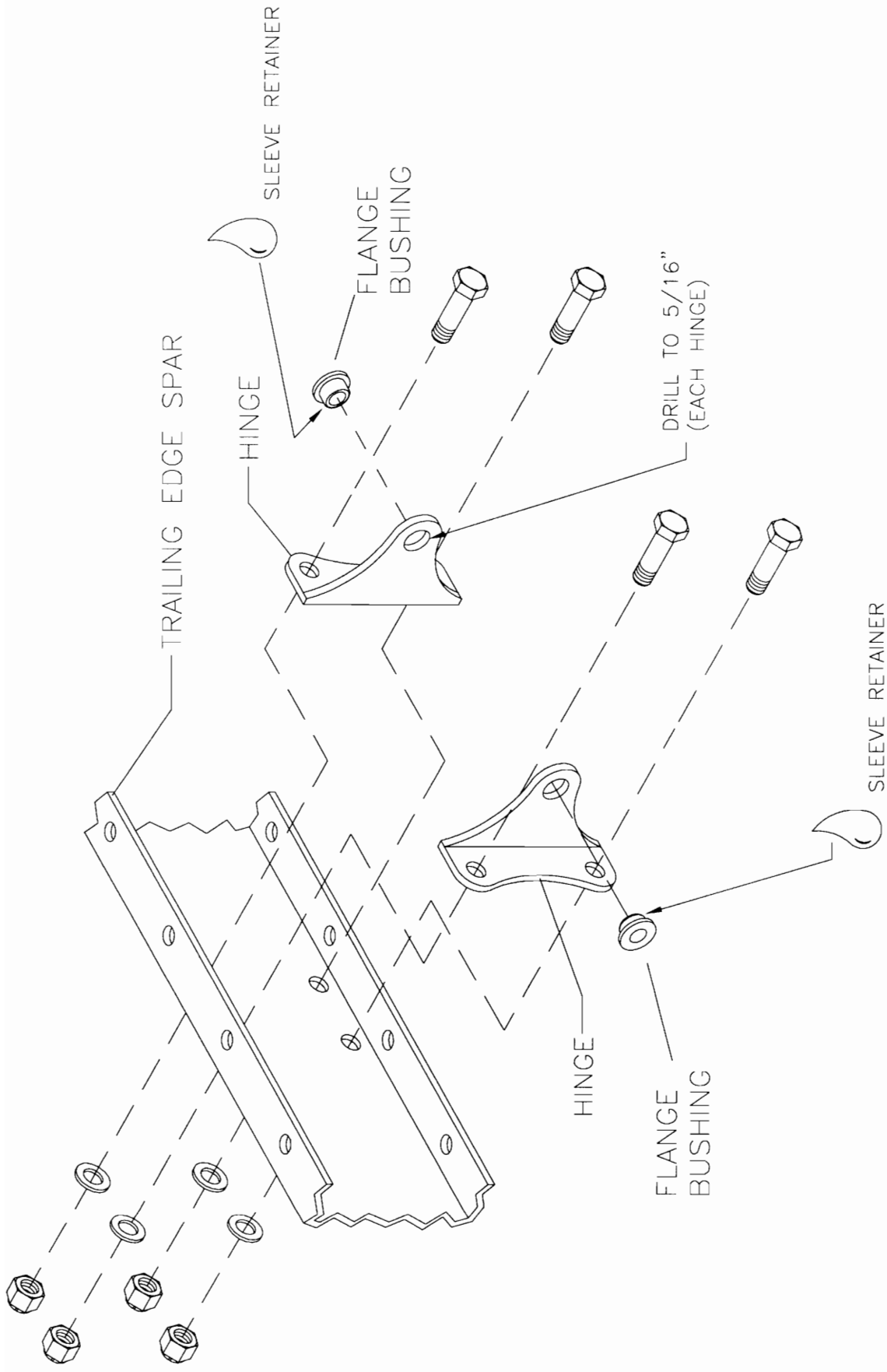


FIGURE 3-6

LEFT TRAILING EDGE SPAR SHOWN

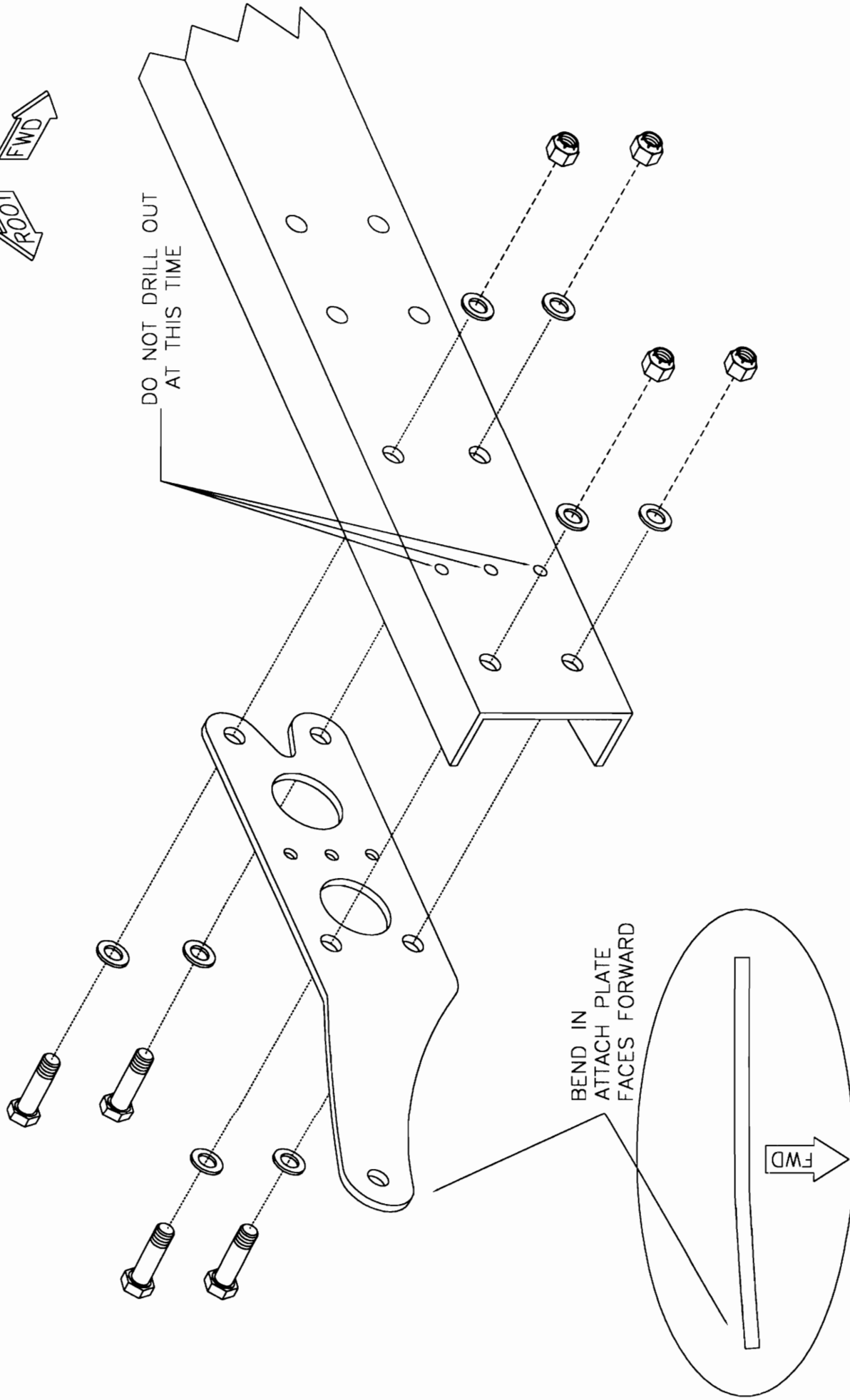
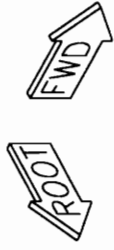
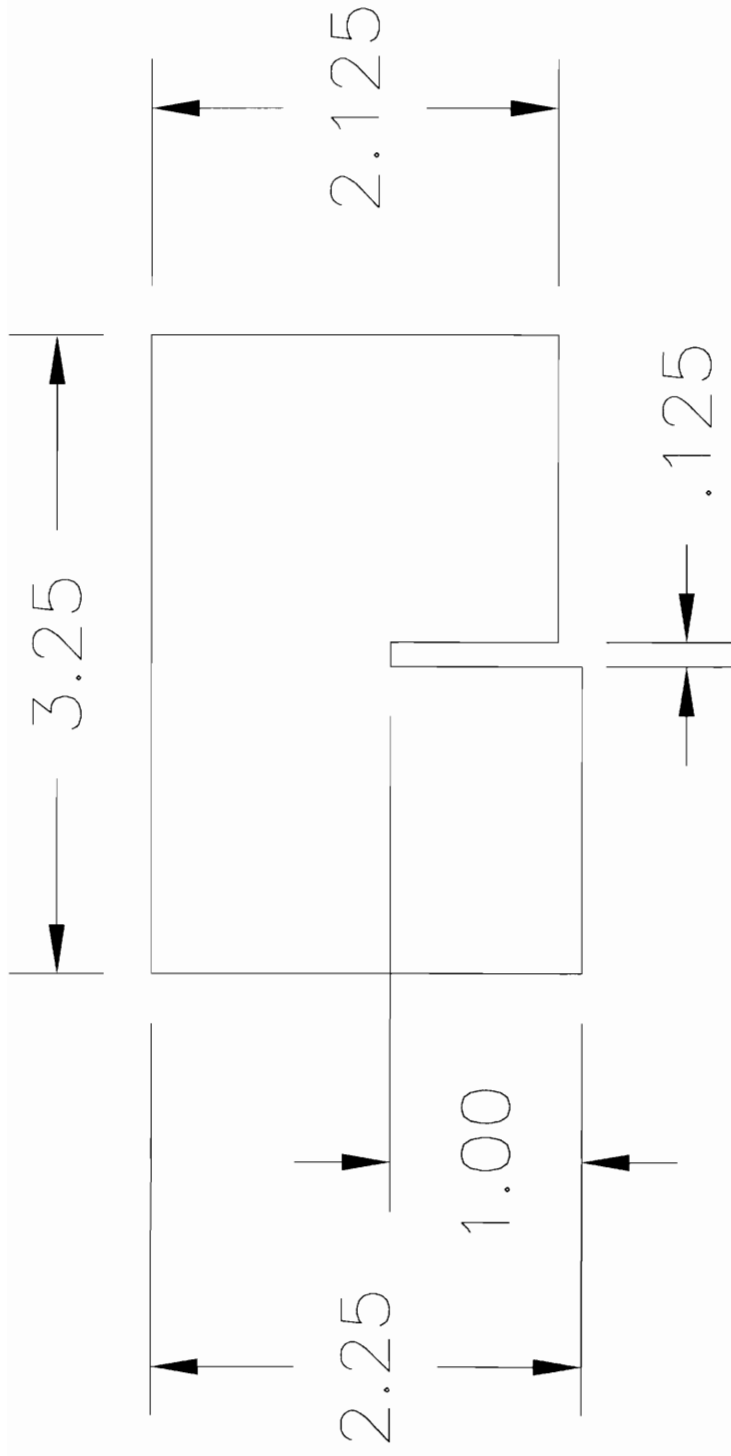


FIGURE 3-7



FABRICATE FROM .080 ALUMINUM RAW STOCK

FIGURE 3-8

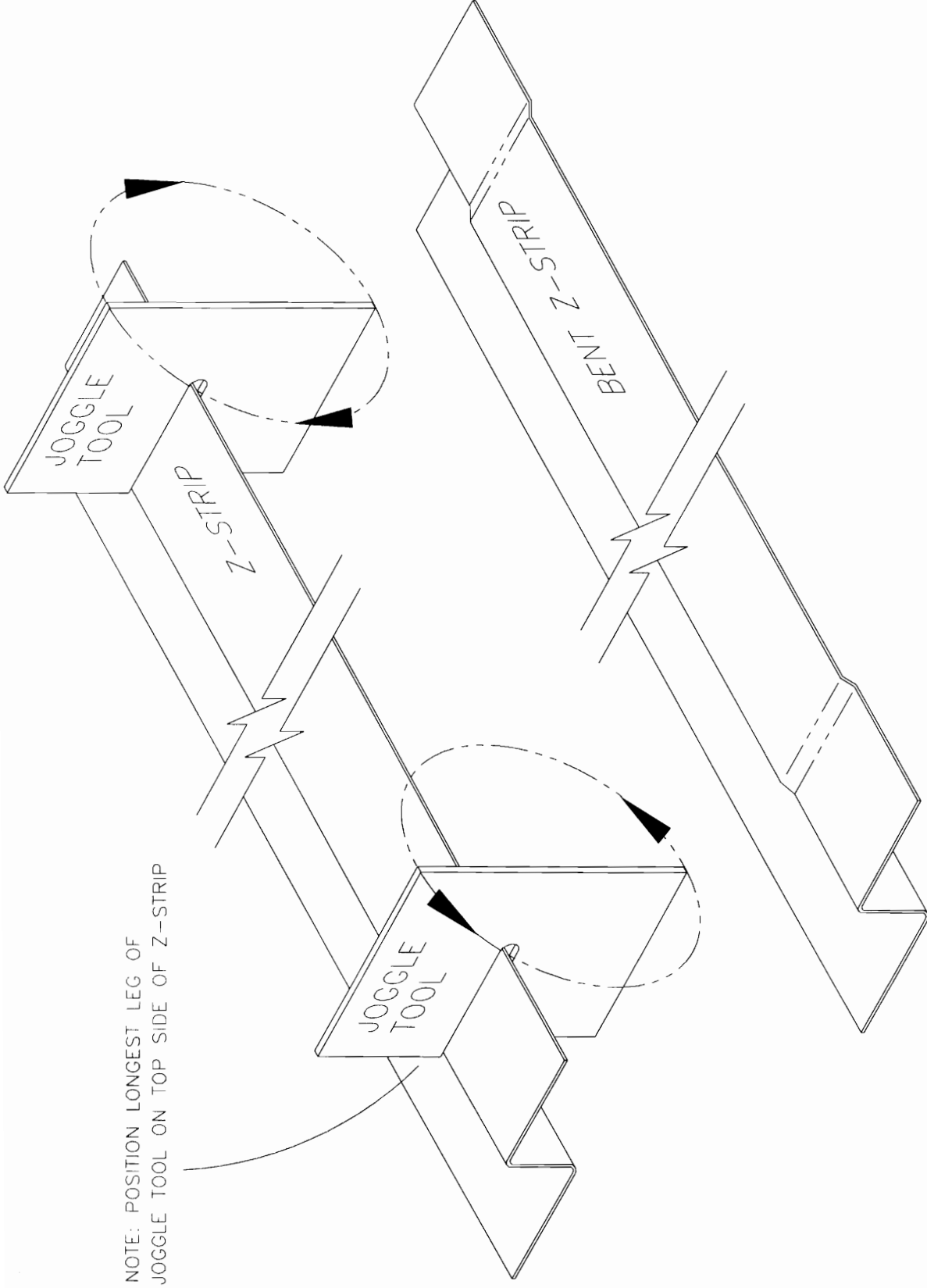
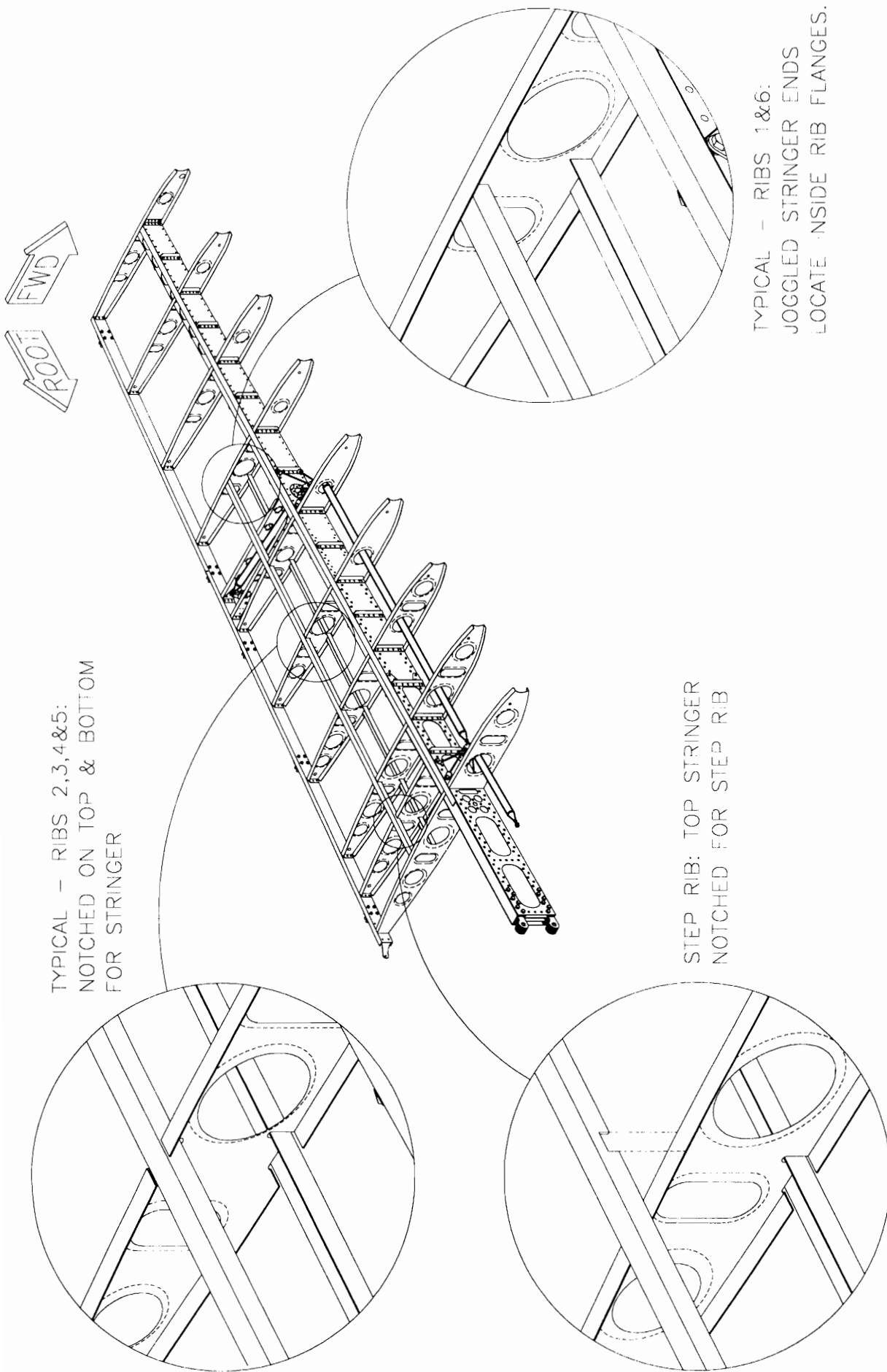


FIGURE 3-9

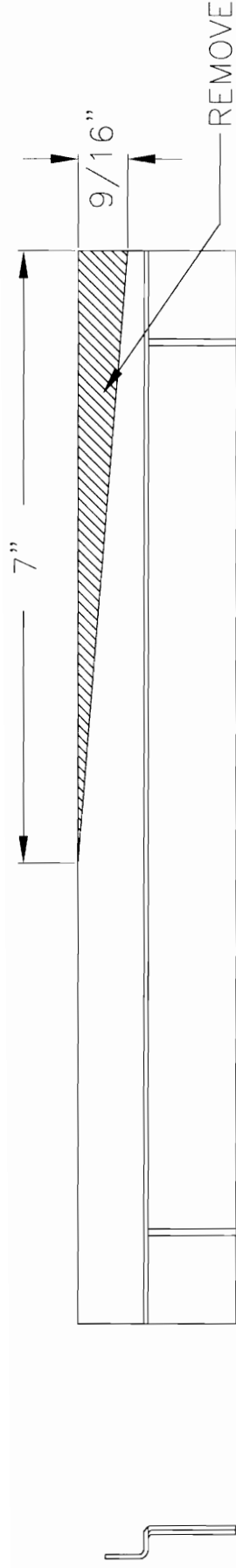


TYPICAL - RIBS 2,3,4&5:  
 NOTCHED ON TOP & BOTTOM  
 FOR STRINGER

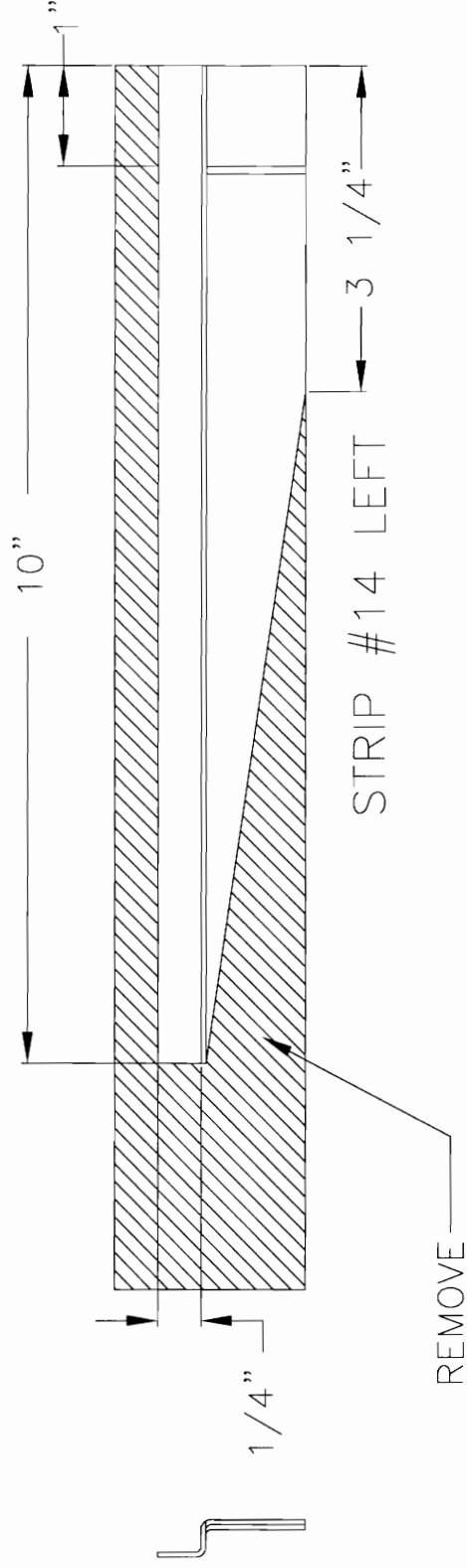
TYPICAL - RIBS 1&6:  
 JOGGED STRINGER ENDS  
 LOCATE INSIDE RIB FLANGES.

STEP RIB: TOP STRINGER  
 NOTCHED FOR STEP RIB

FIGURE 3-10

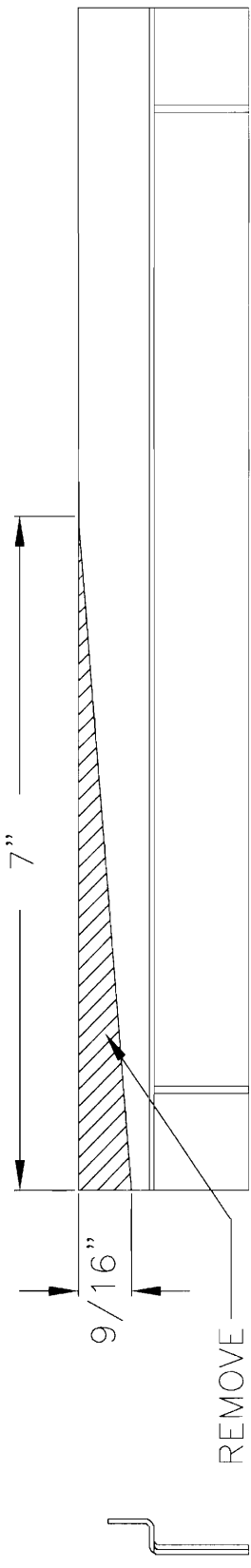


STRIP #13 LEFT

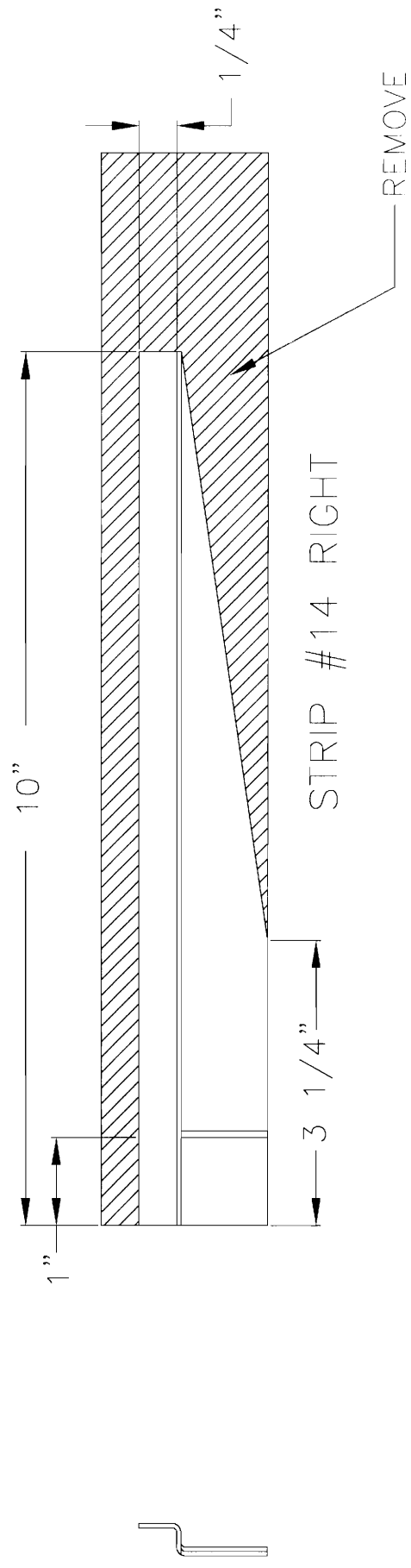


STRIP #14 LEFT

FIGURE 3-11



STRIP #13 RIGHT



STRIP #14 RIGHT

FIGURE 3-12

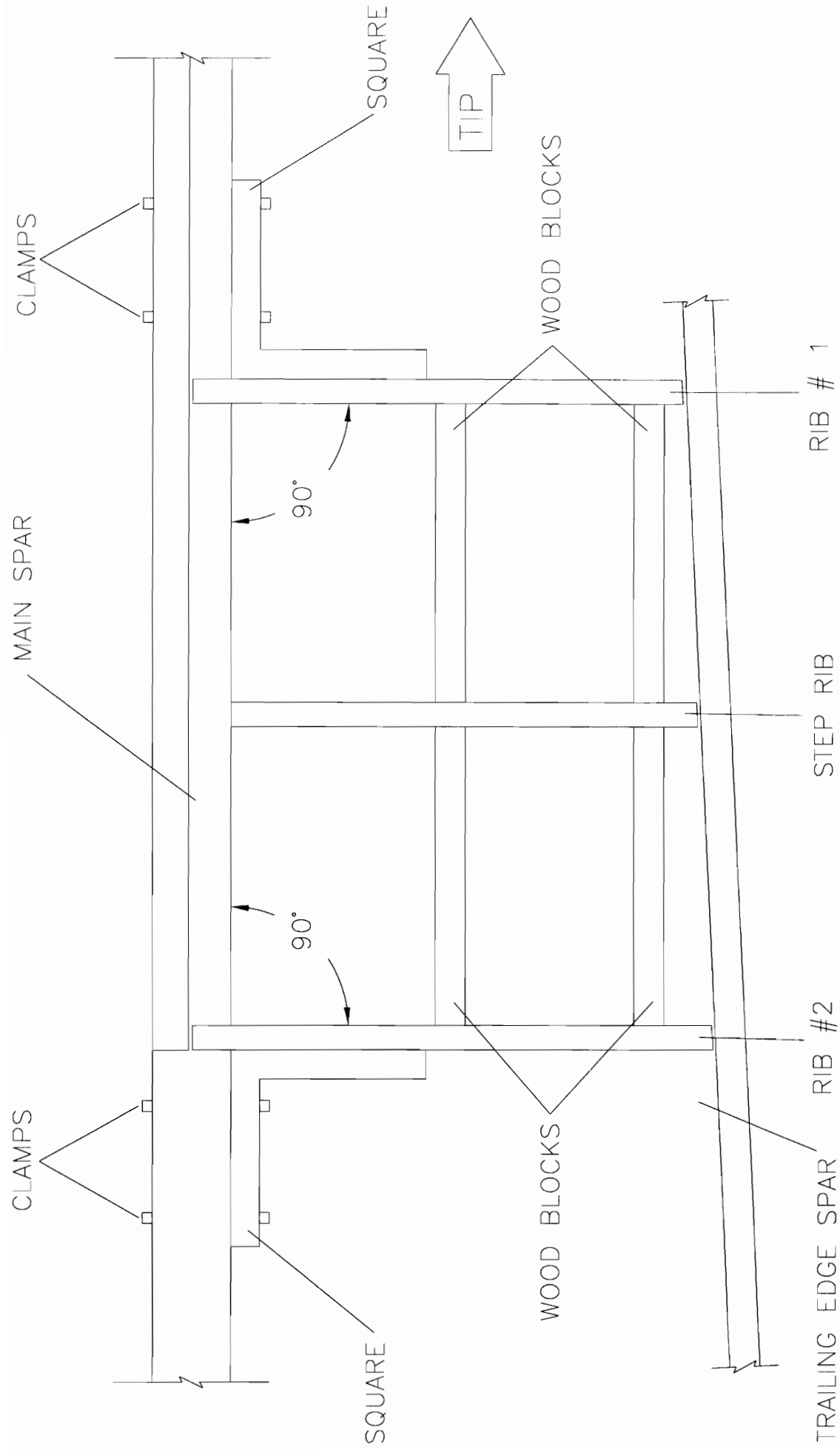


FIGURE 3-13



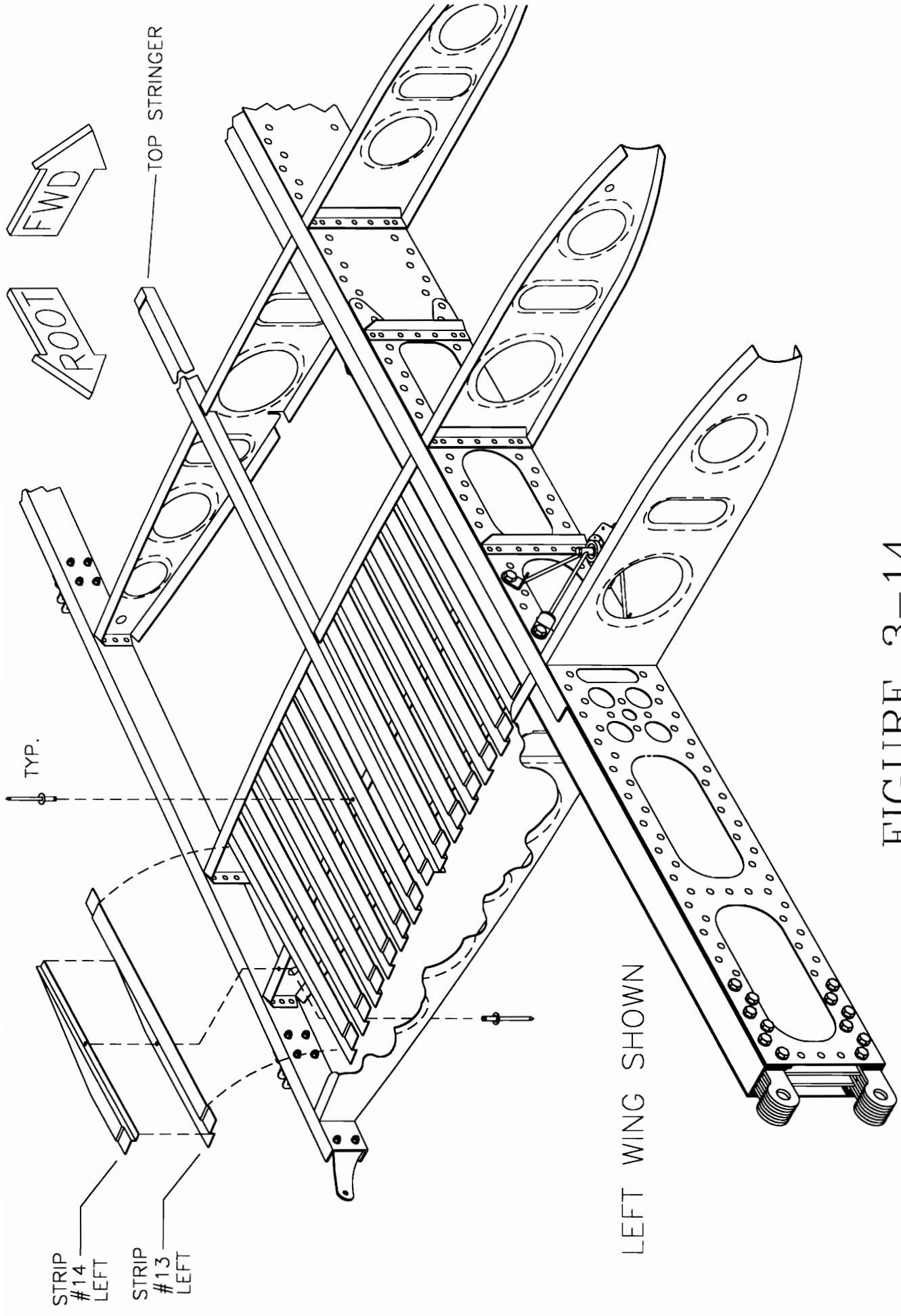


FIGURE 3-14

TYPICAL FOR ALL  
BELLCRANK  
ASSEMBLIES

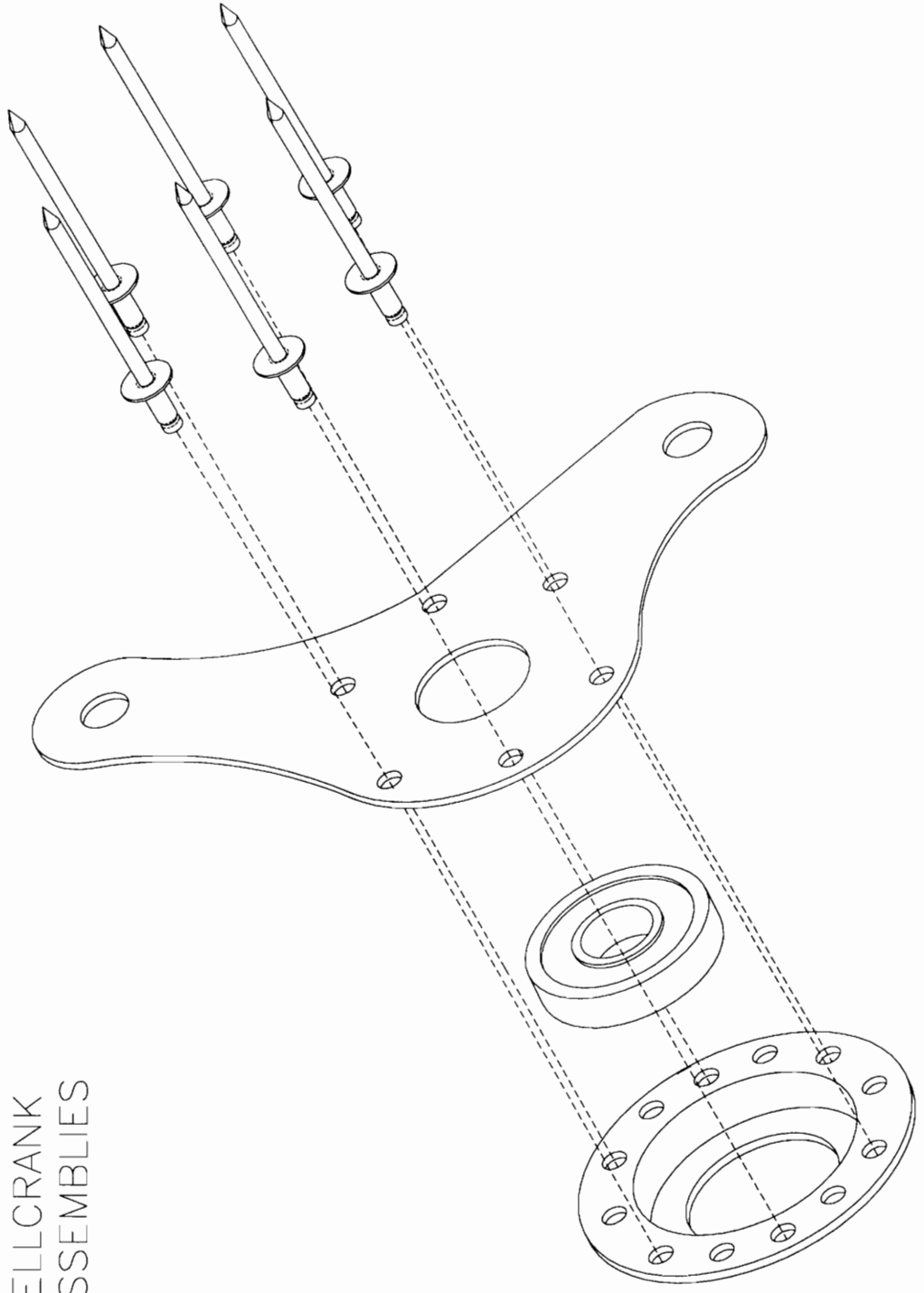


FIGURE 3-15

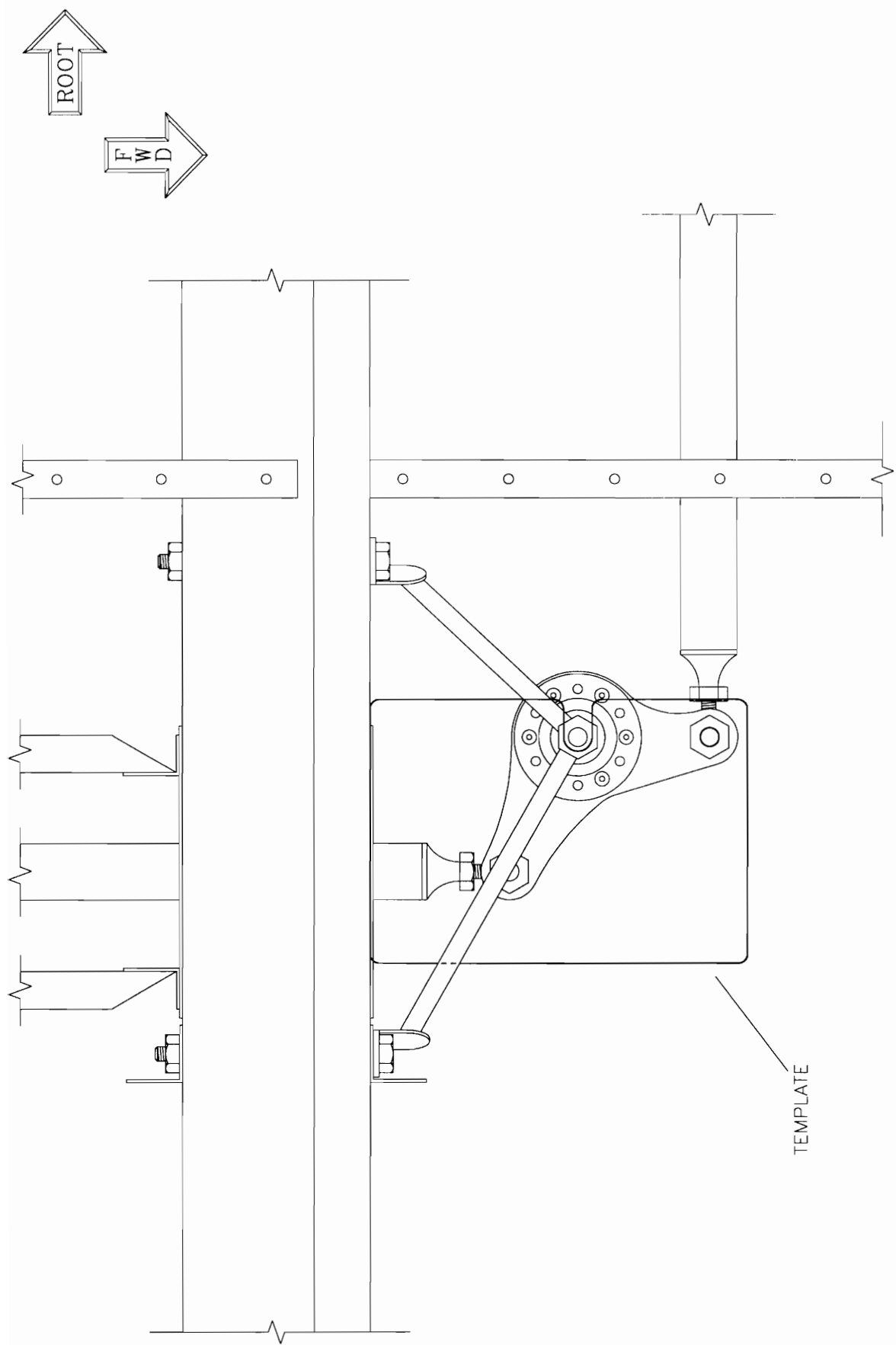


FIGURE 3-16

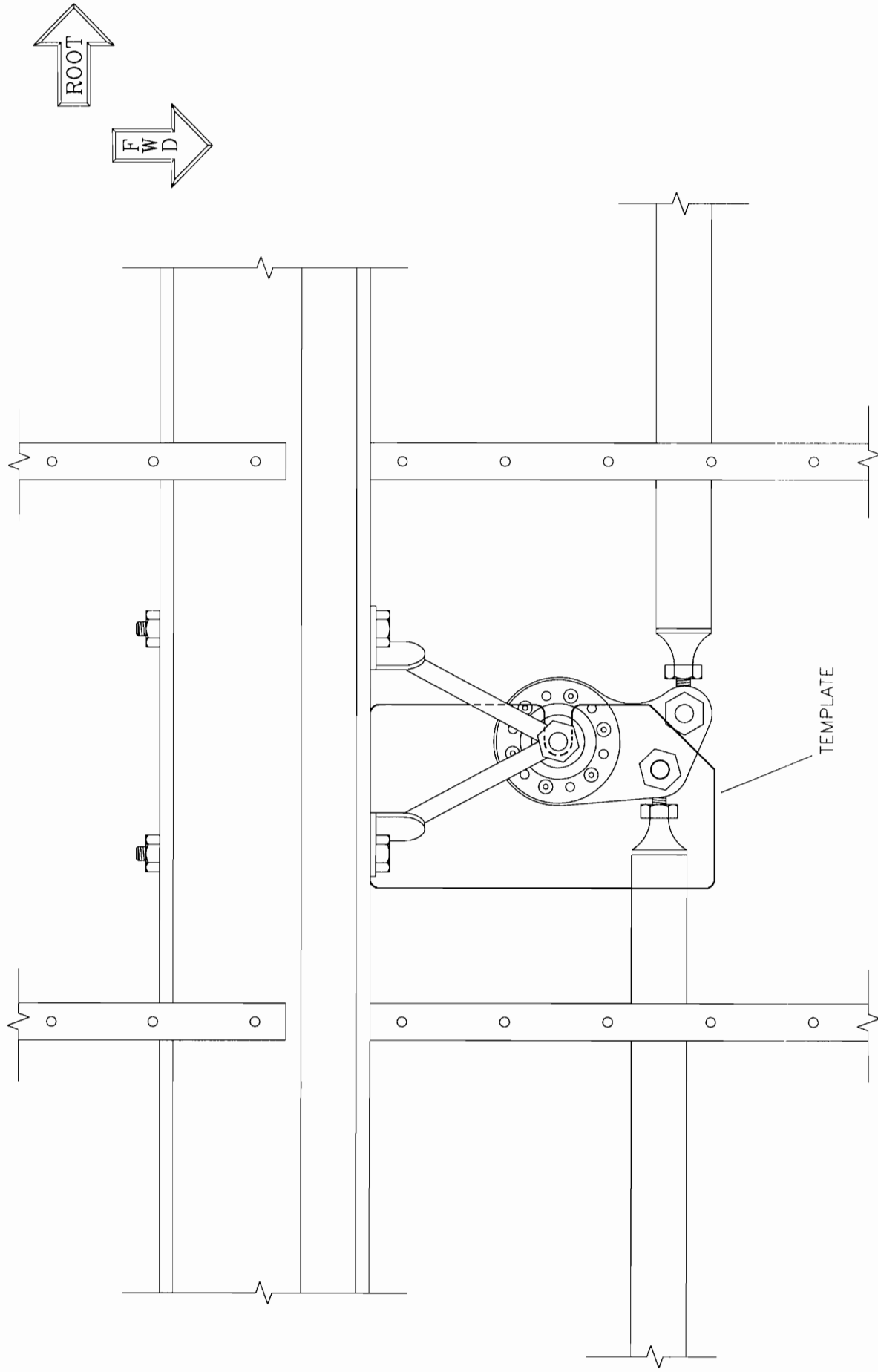


FIGURE 3-17

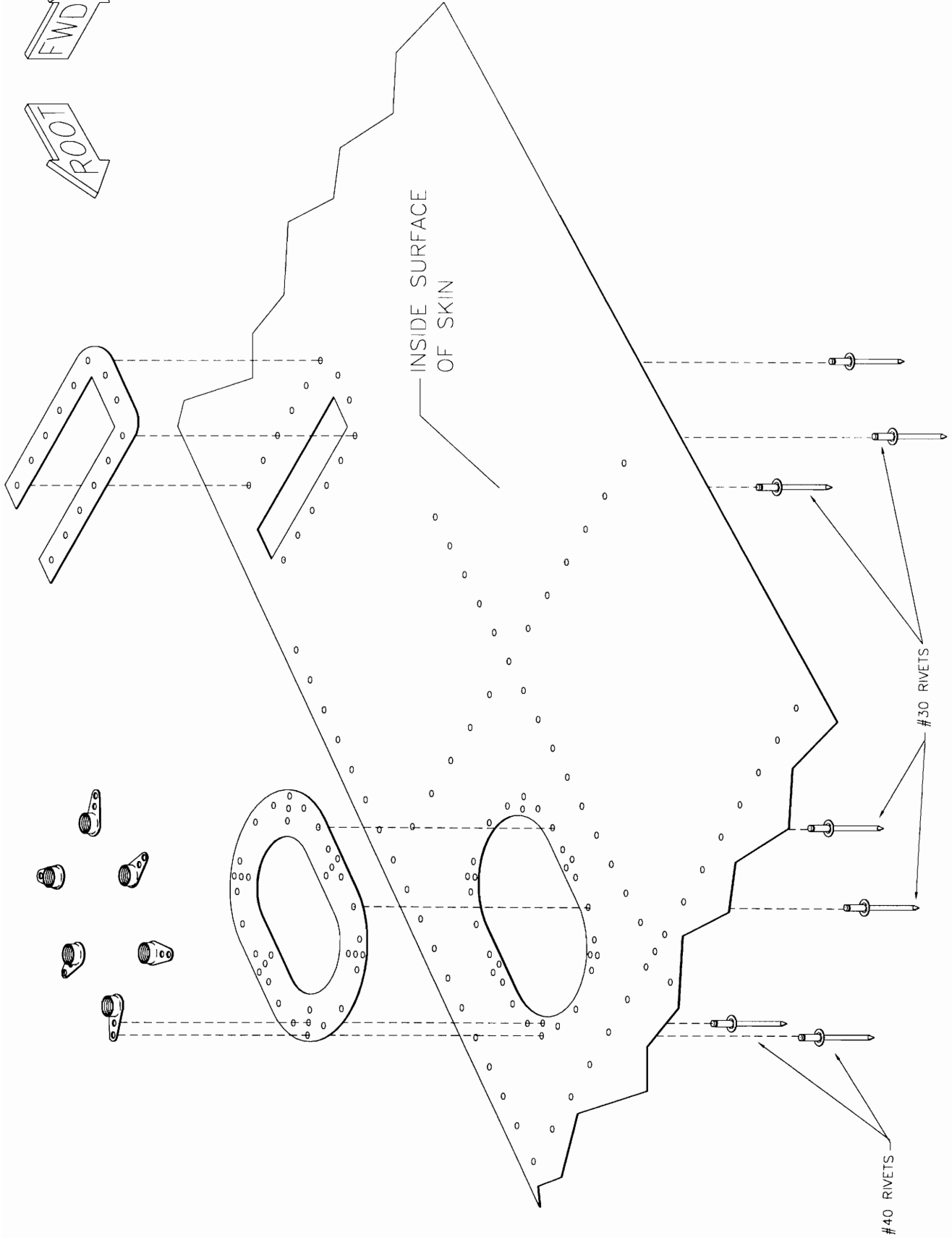
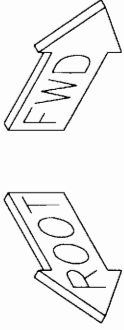


FIGURE 3-18

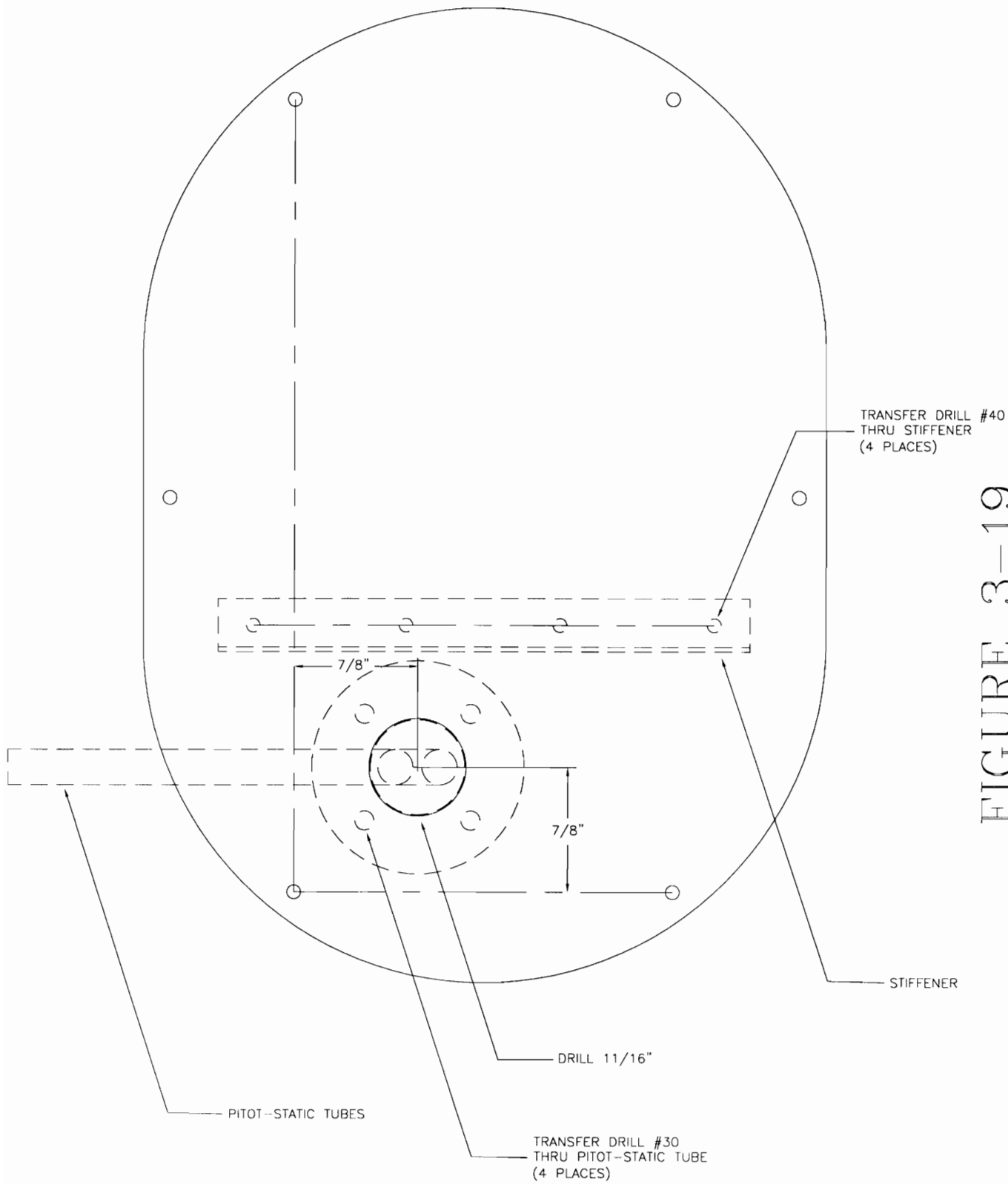


FIGURE 3-19

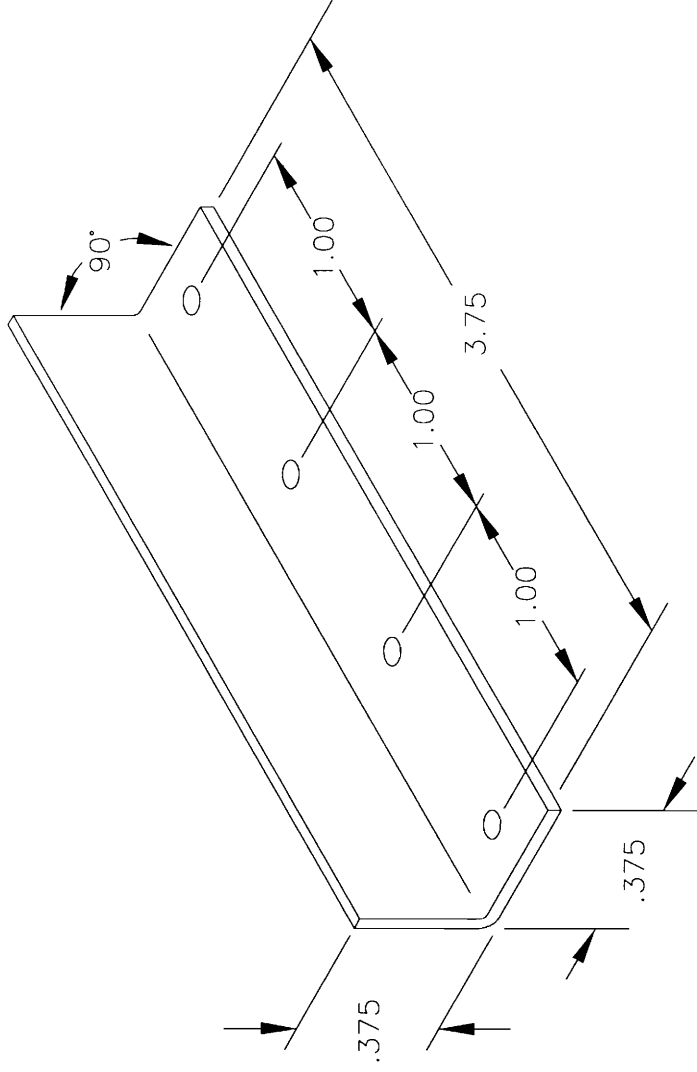


FIGURE 3-20

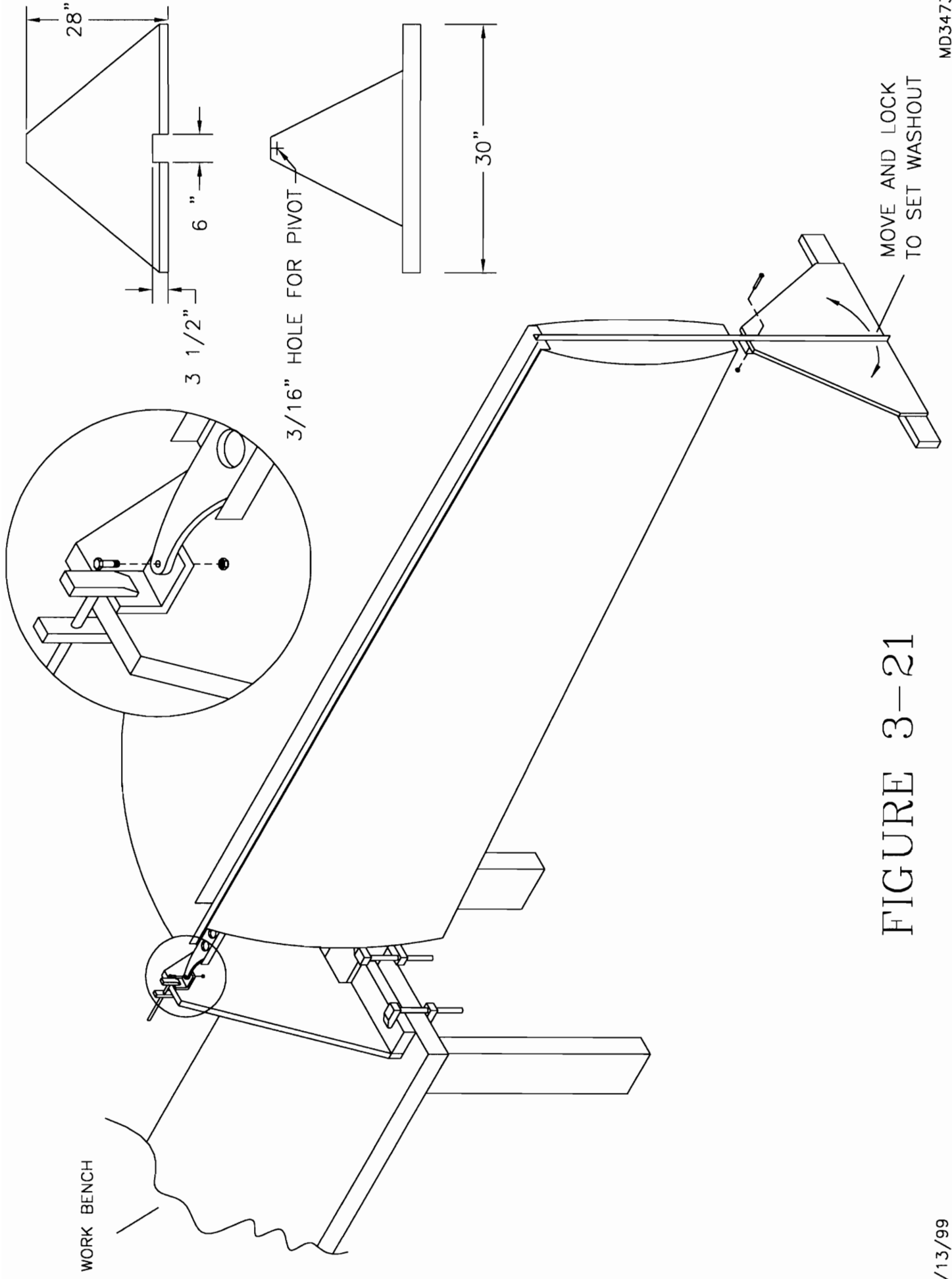
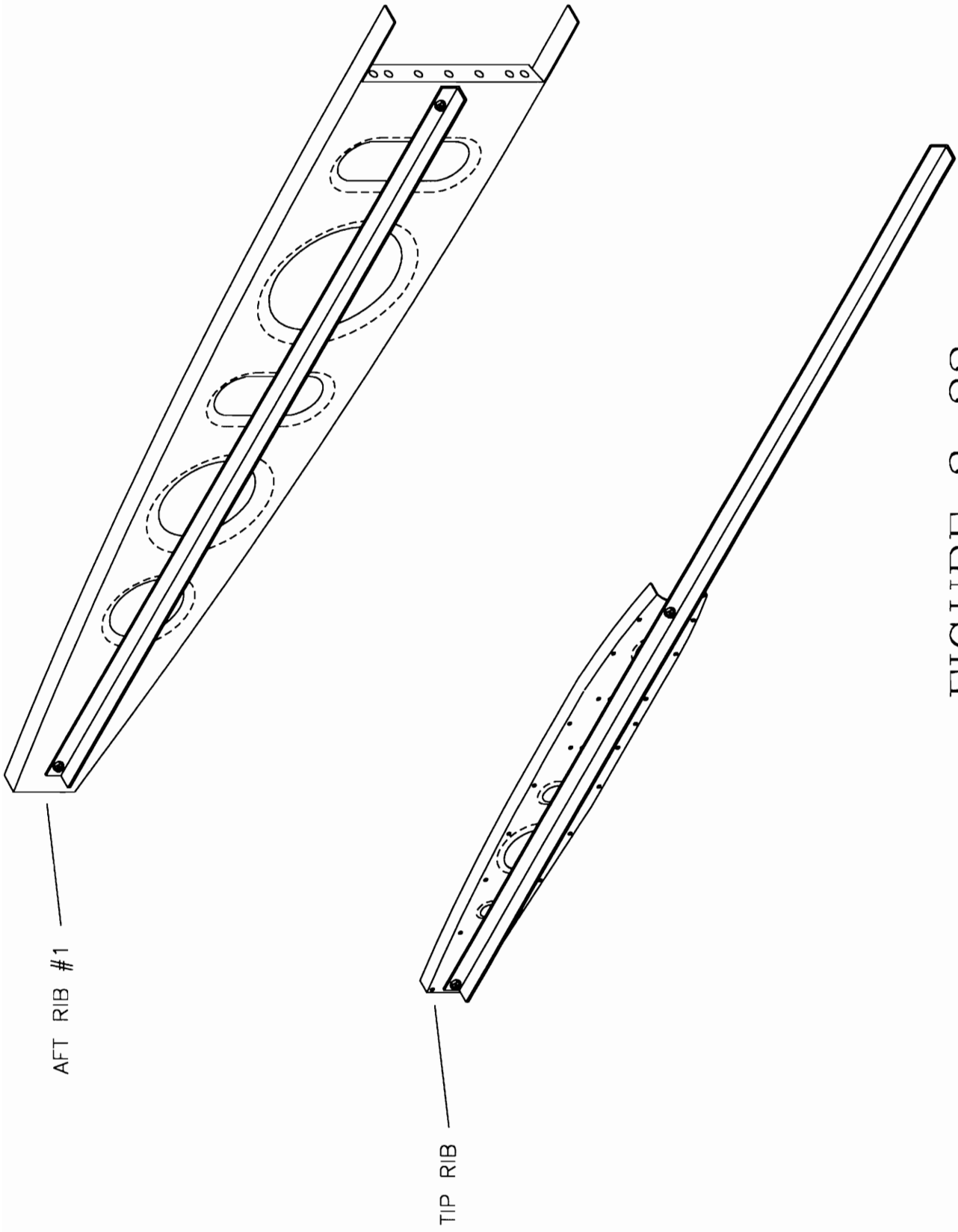


FIGURE 3-21

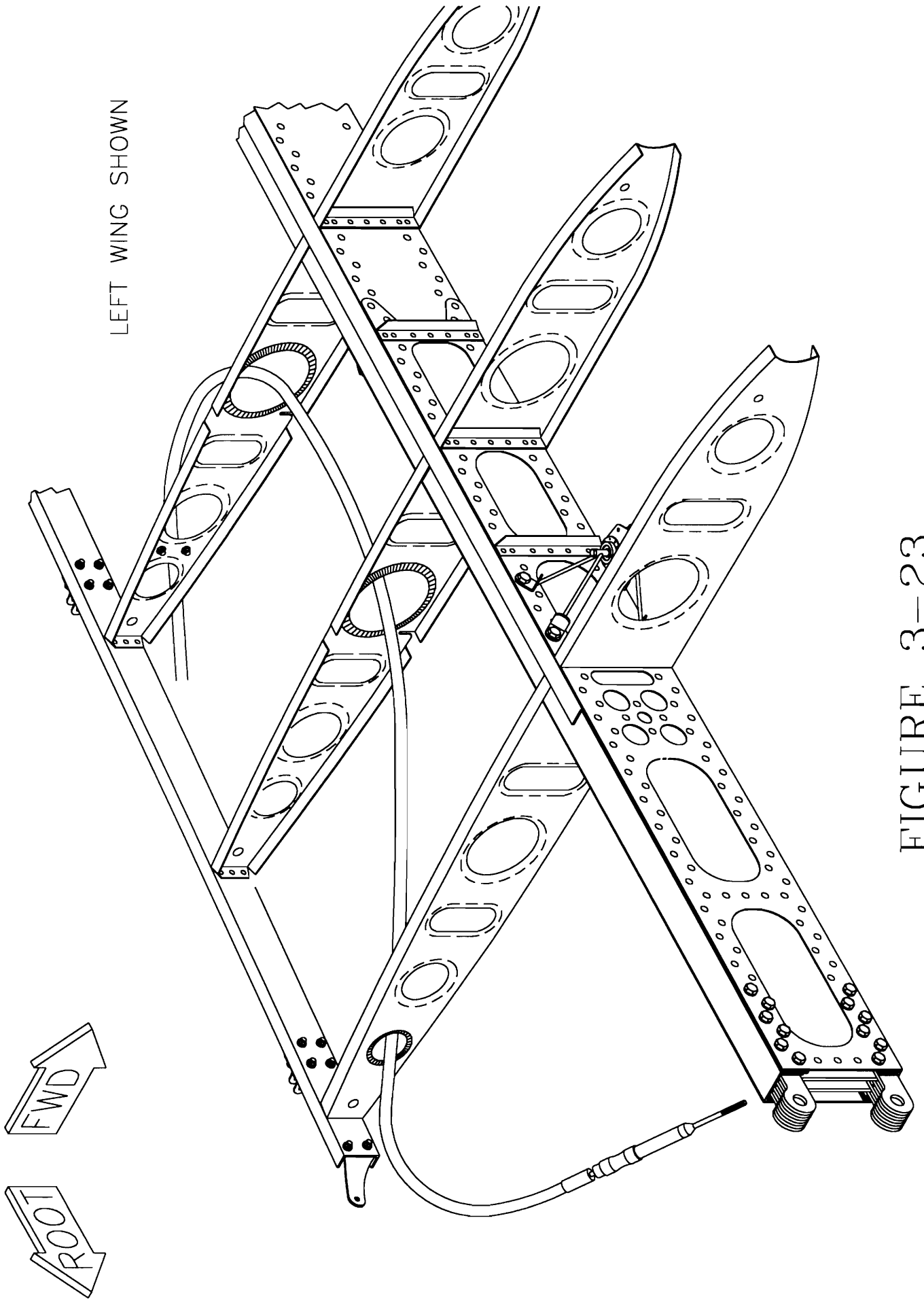




AFT RIB #1

TIP RIB

FIGURE 3-22

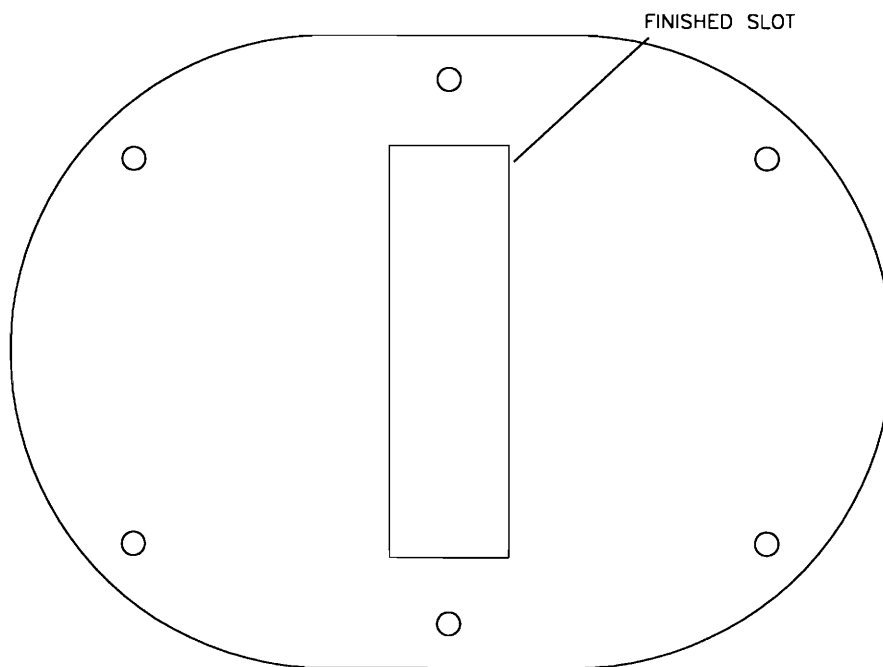
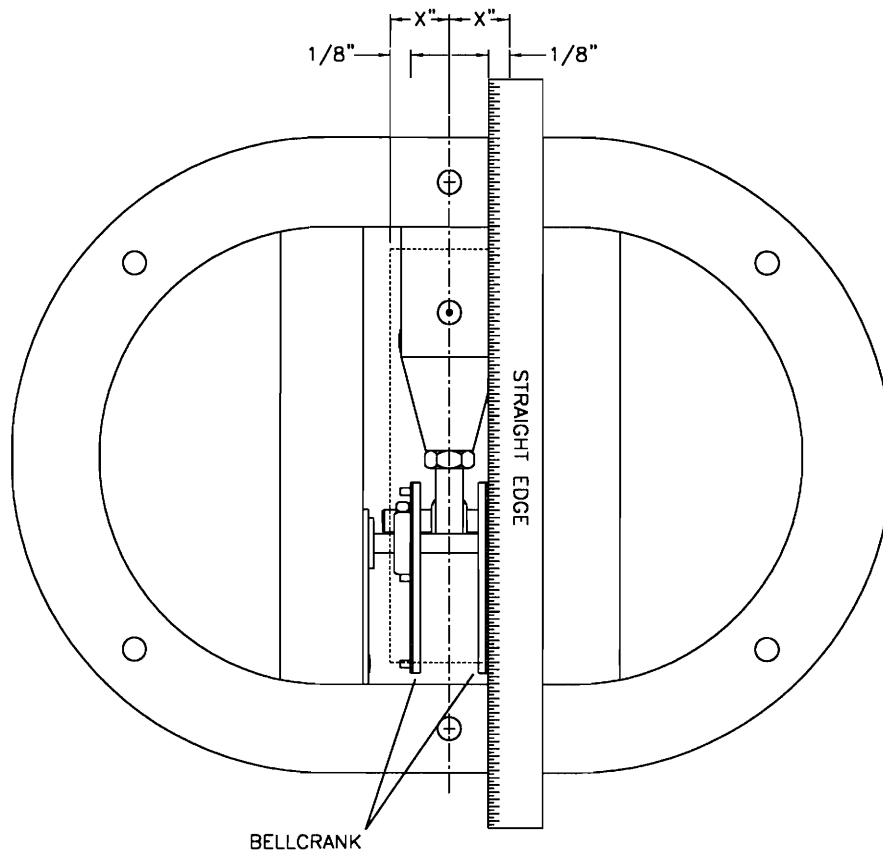


LEFT WING SHOWN

FWD

ROOT

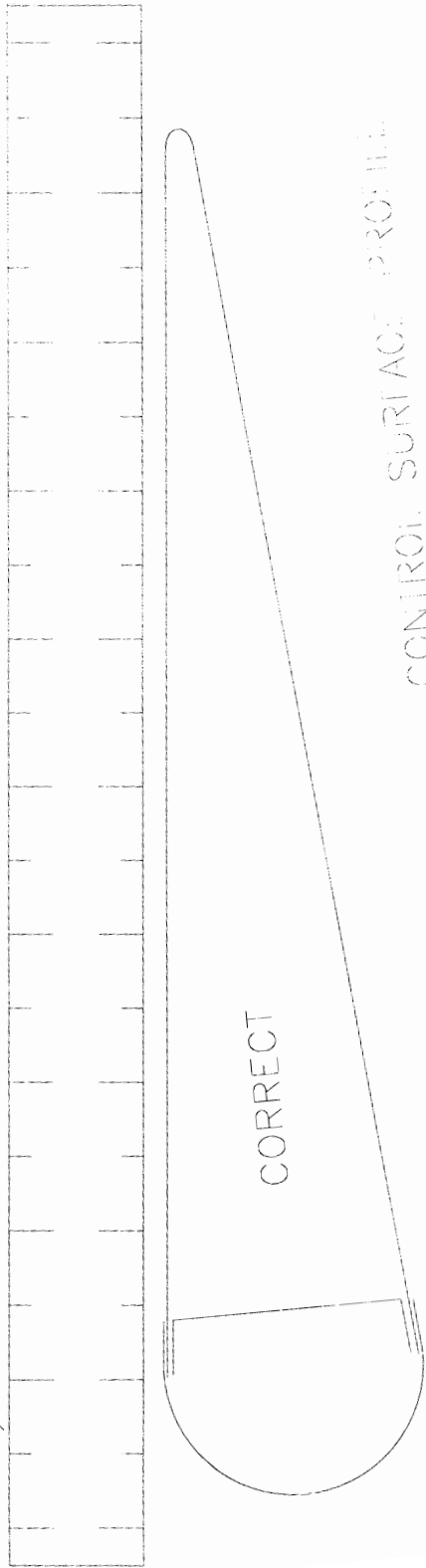
FIGURE 3-23



TRANSFER MEASUREMENTS TO  
ACCESS PANEL COVER

FIGURE 3-24

STRAIGHT EDGE OR RULER



CONTROL SURFACE PROFILE

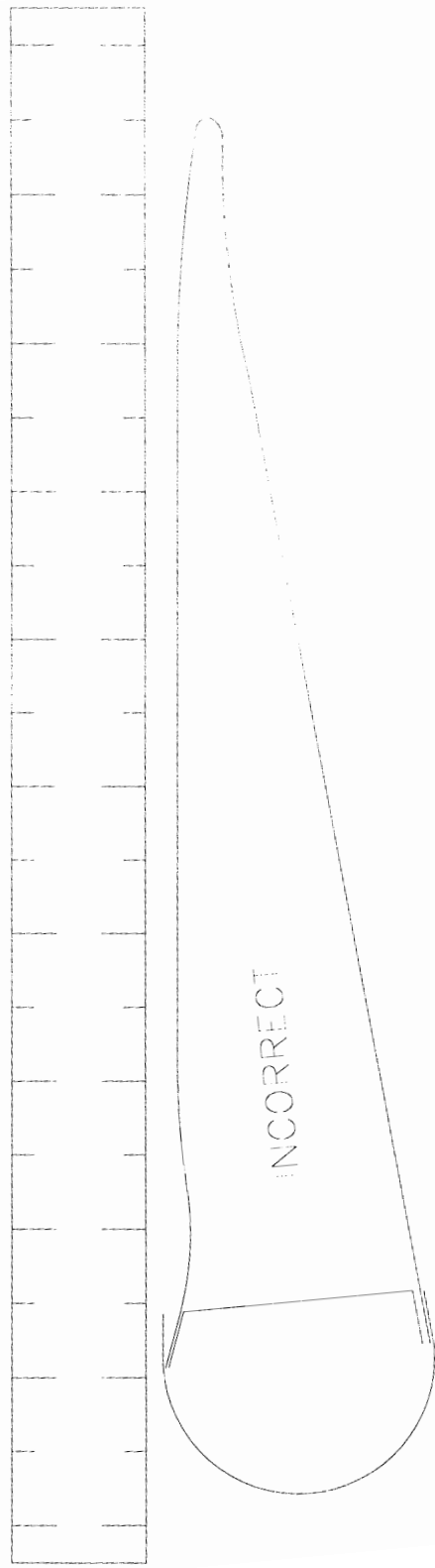
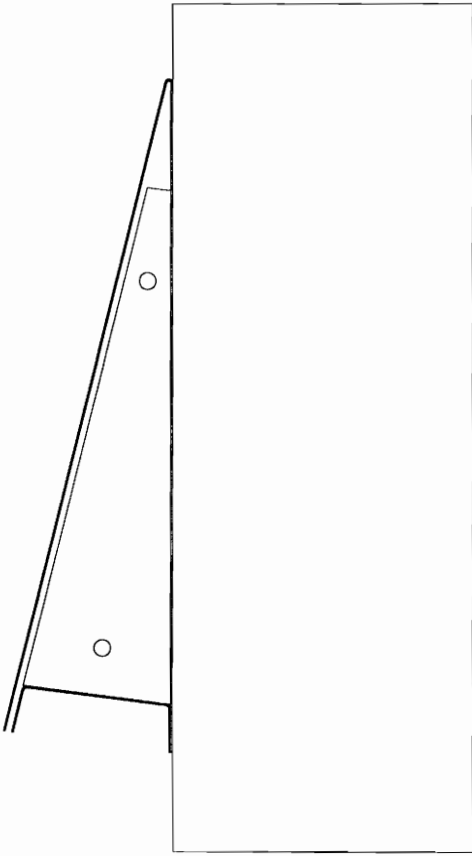
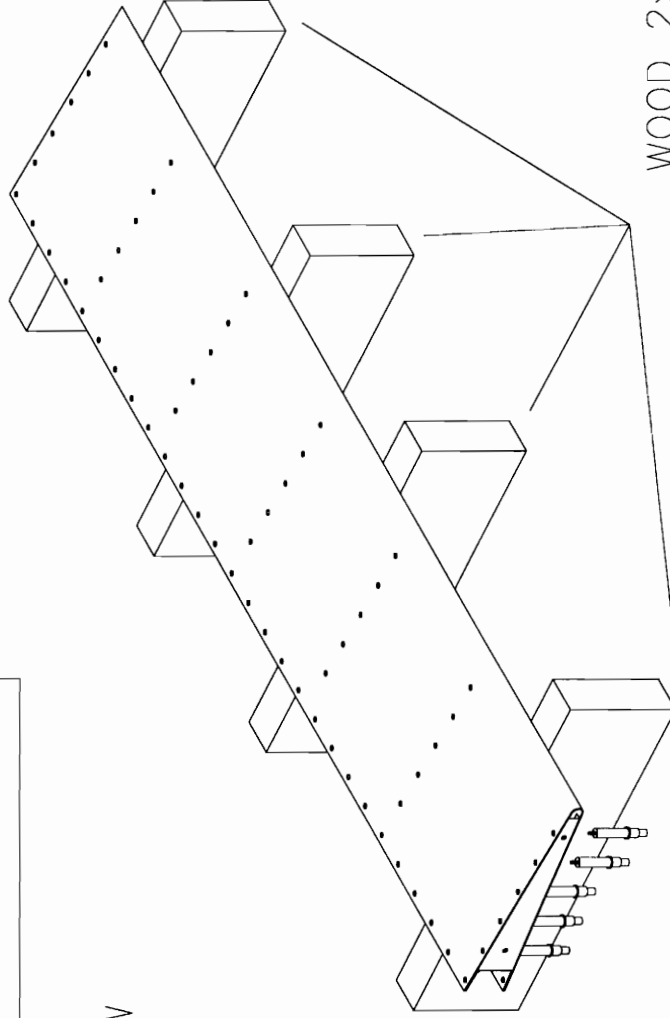


FIGURE 4 0

THIS  
PAGE  
INTENTIONALLY  
LEFT  
BLANK.



END VIEW



WOOD 2x6 BLOCKS

NOTE: LUMBER DIMENSIONS ARE NOT CRITICAL

FIGURE 4-2

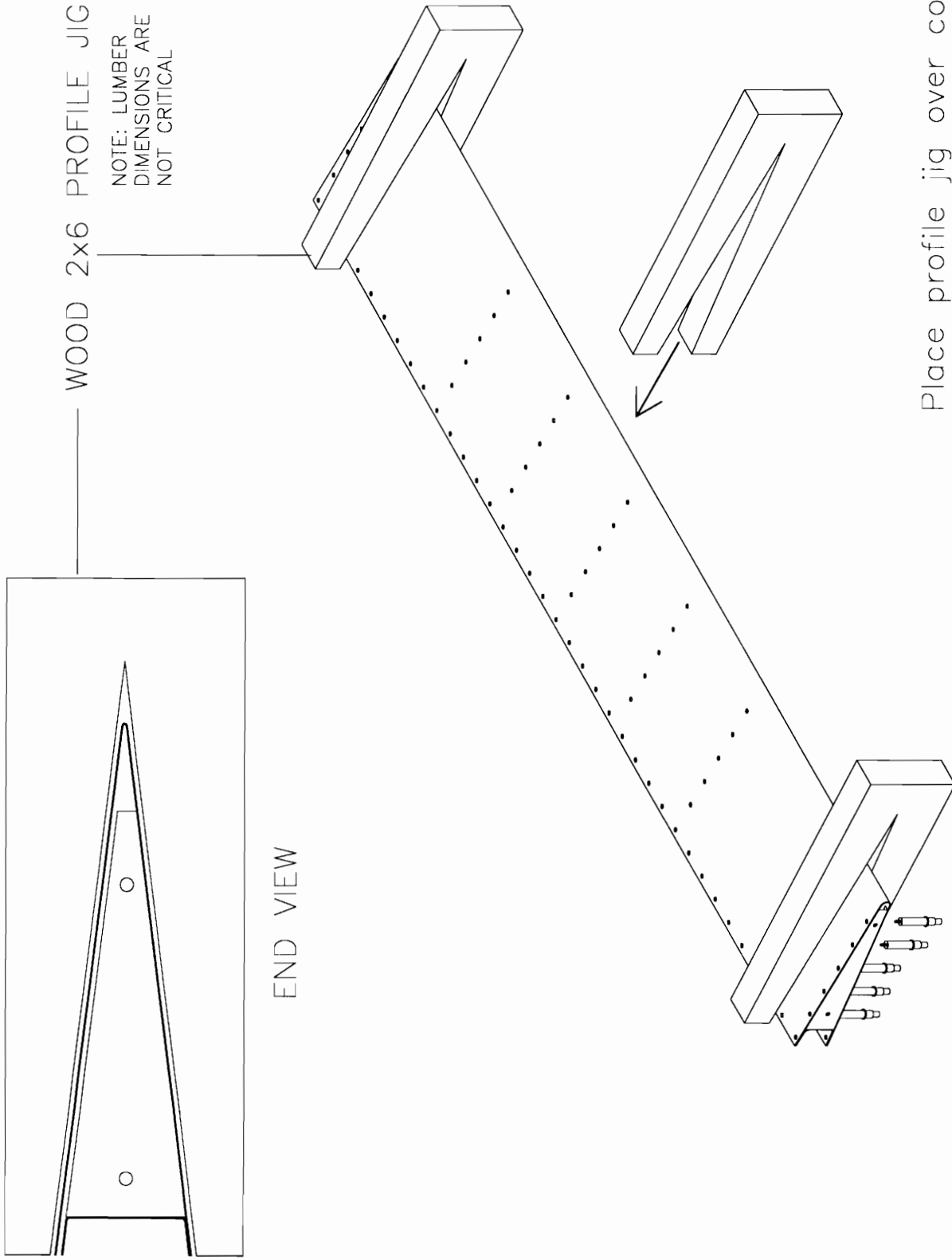
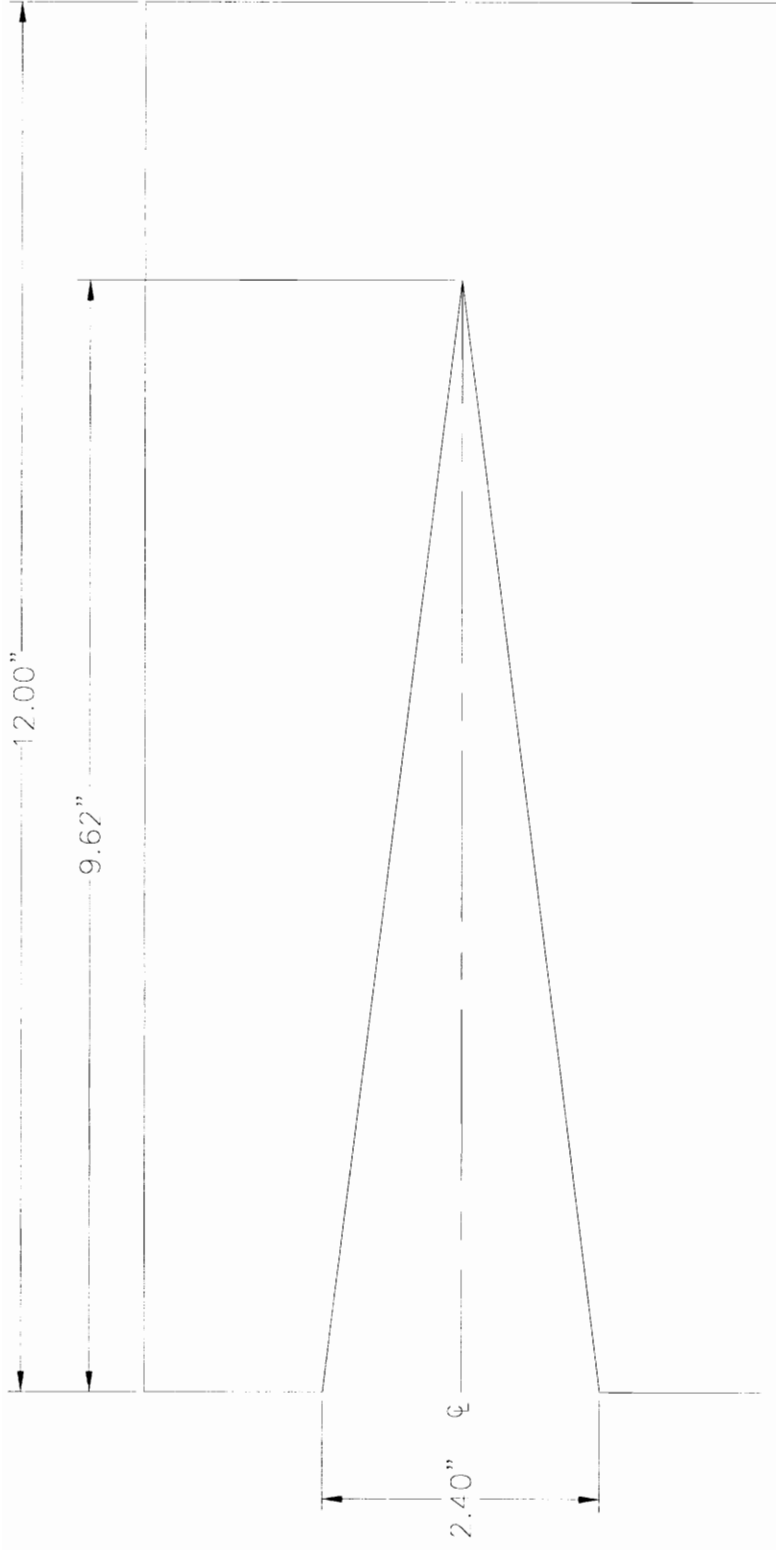


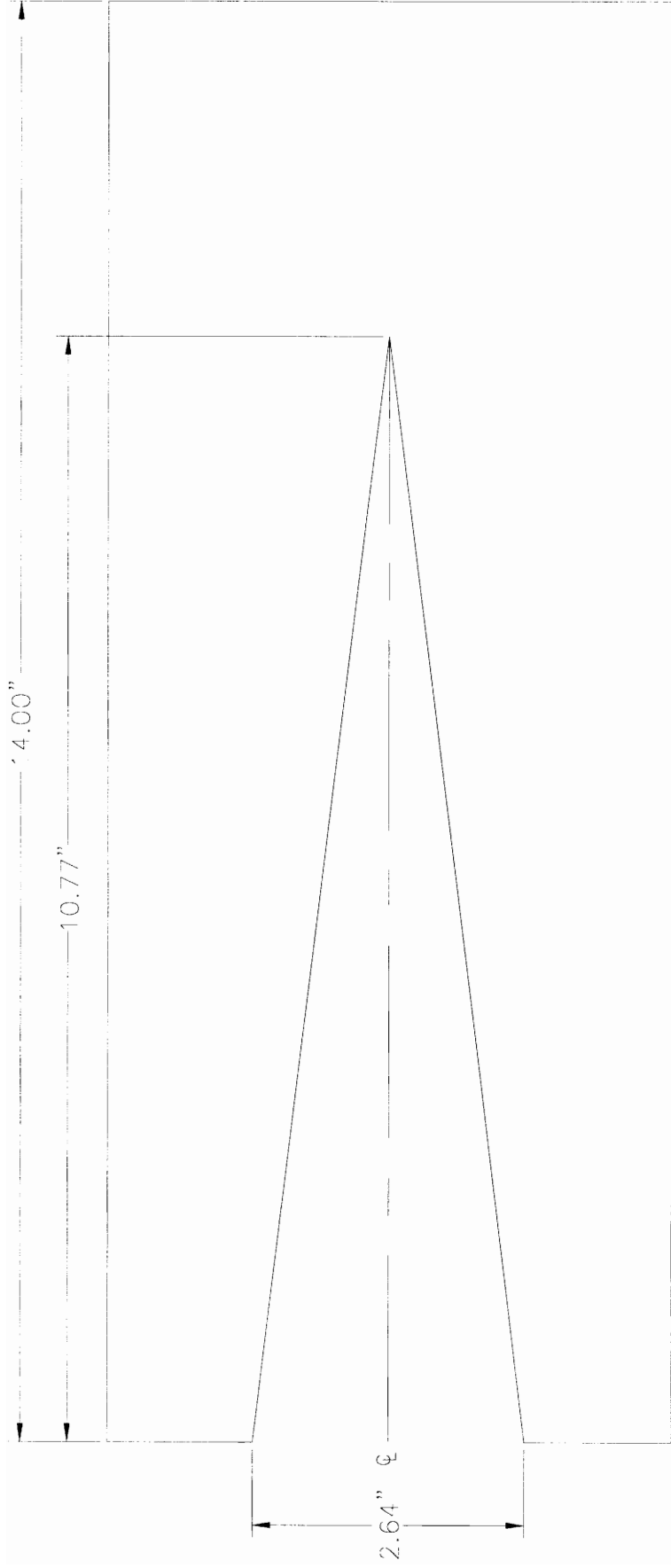
FIGURE 4-3



CONTROL SURFACE PROFILE JIG -- ALLERON

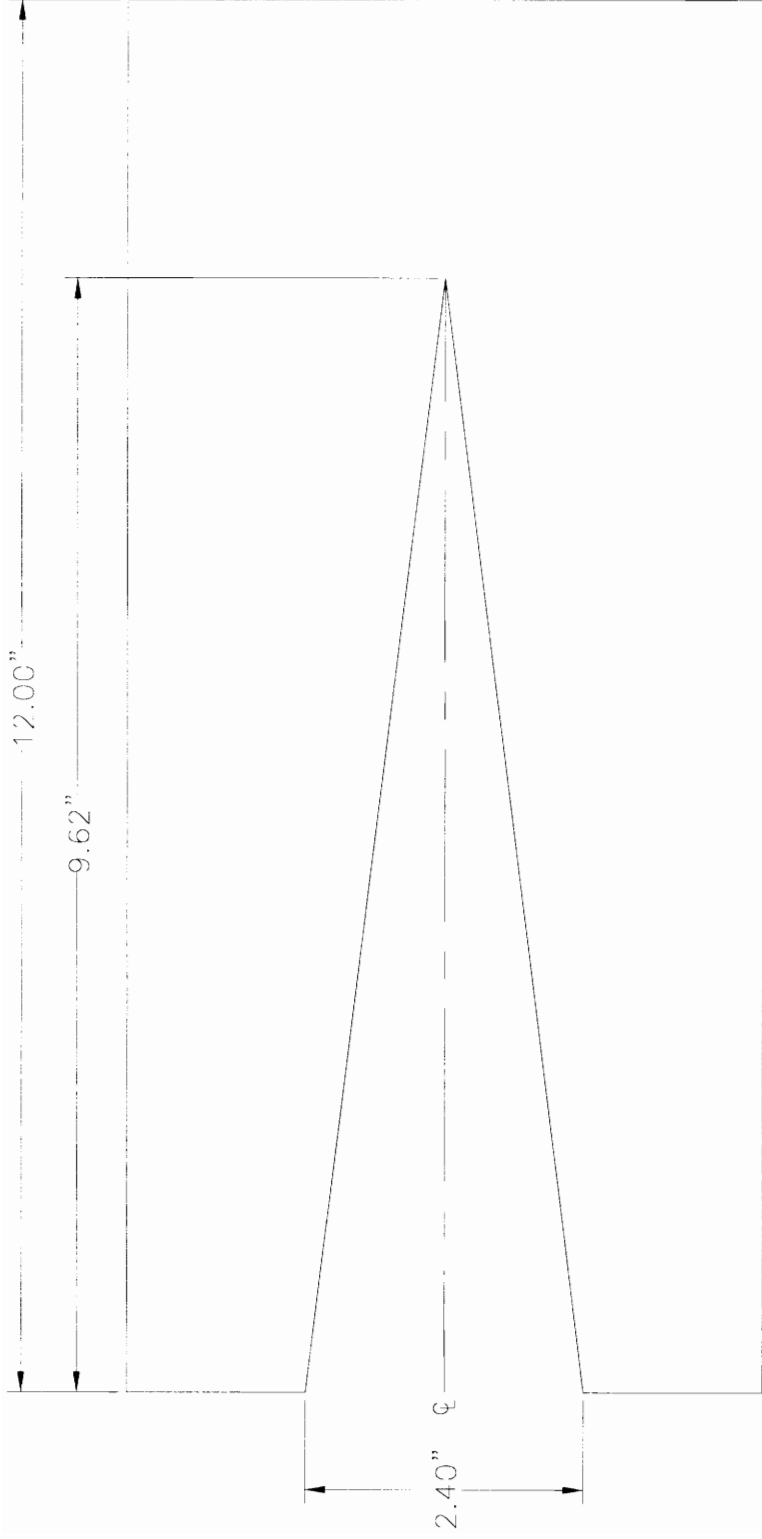
FIGURE 4--4





CONTROL SURFACE PROFILE JIG -- ELEVATOR

FIGURE 4-5



CONTROL SURFACE PROFILE JIG -- FLAP

FIGURE 4--6

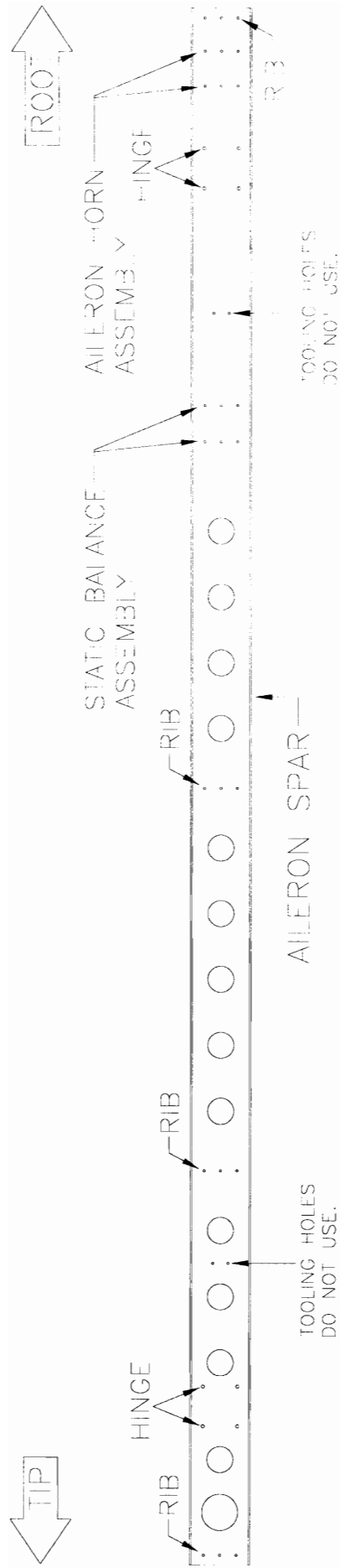


FIGURE 4-7

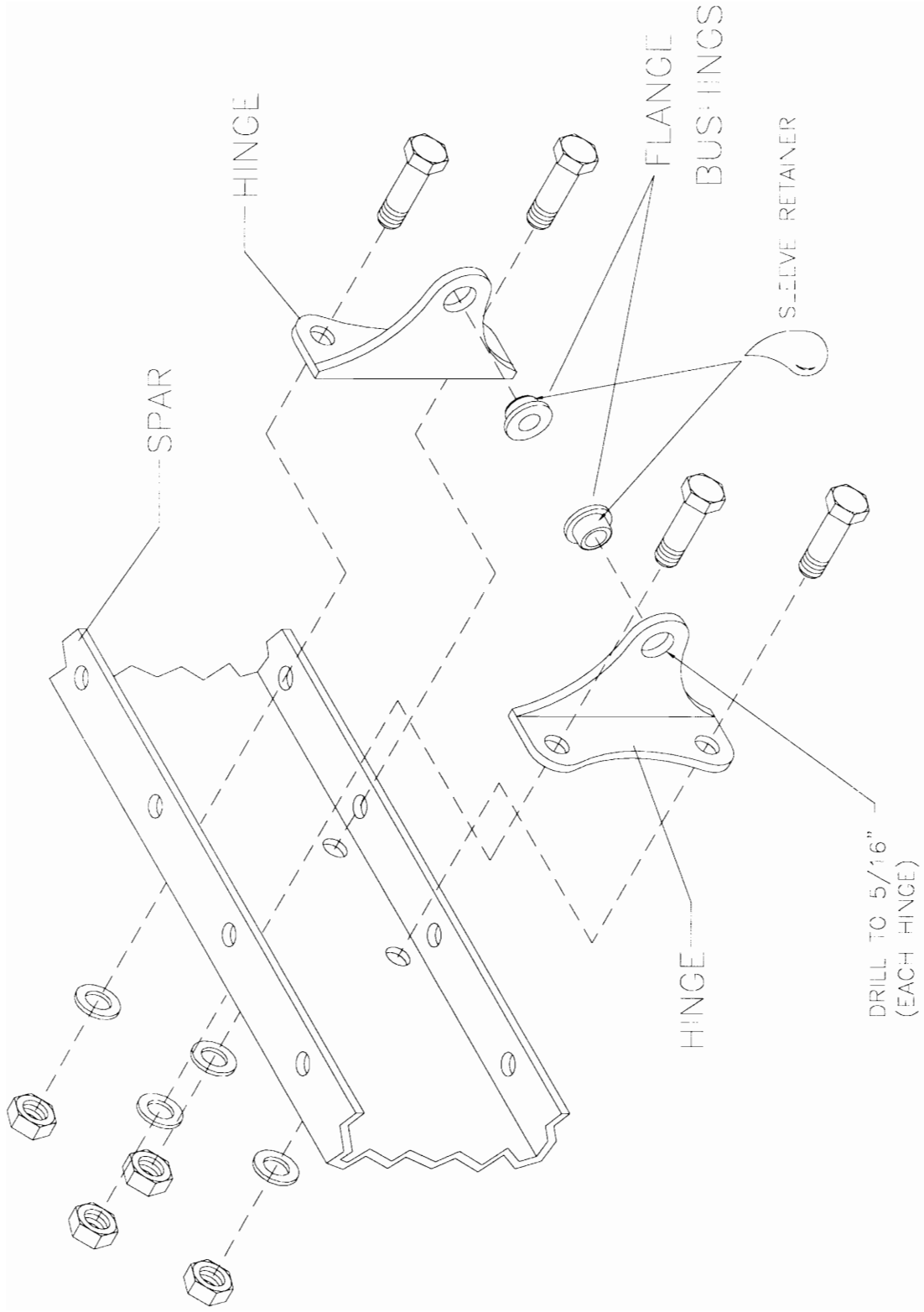
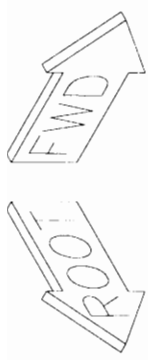


FIGURE 4-8



LEFT AILERON HORN  
ASSEMBLY SHOWN

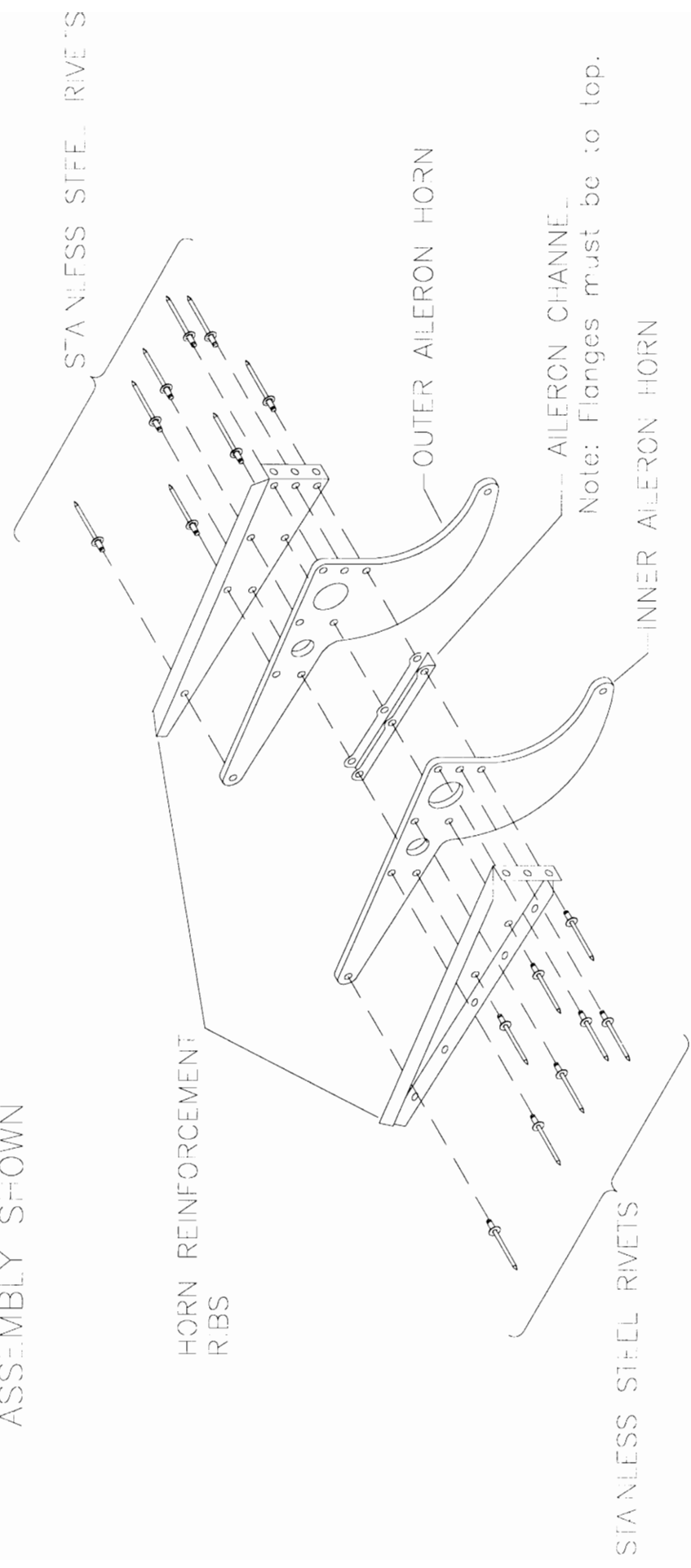
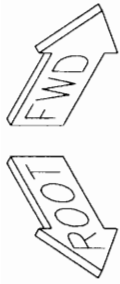


FIGURE 4--9



LEFT STATIC BALANCE ARM  
ASSEMBLY SHOWN

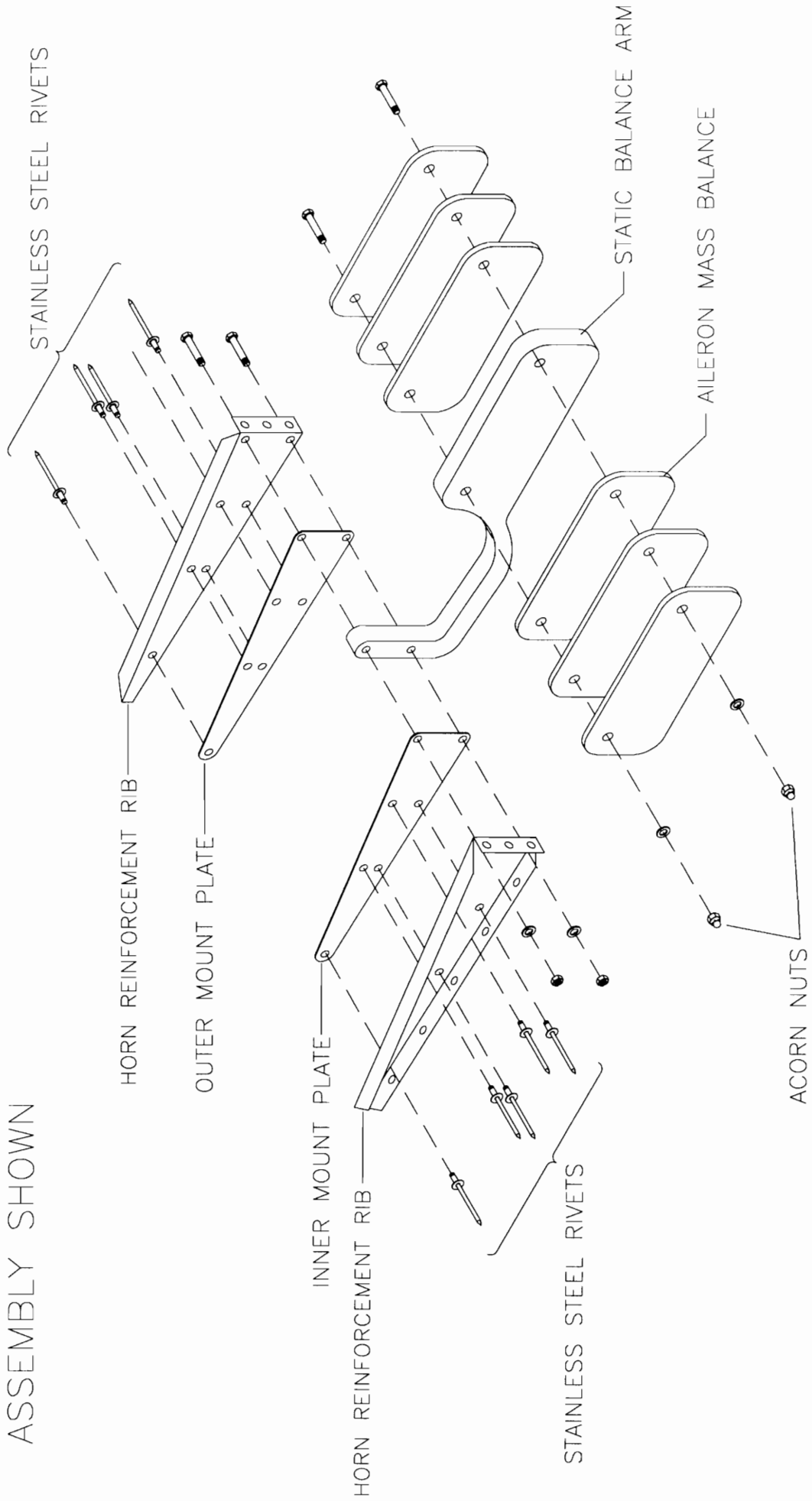
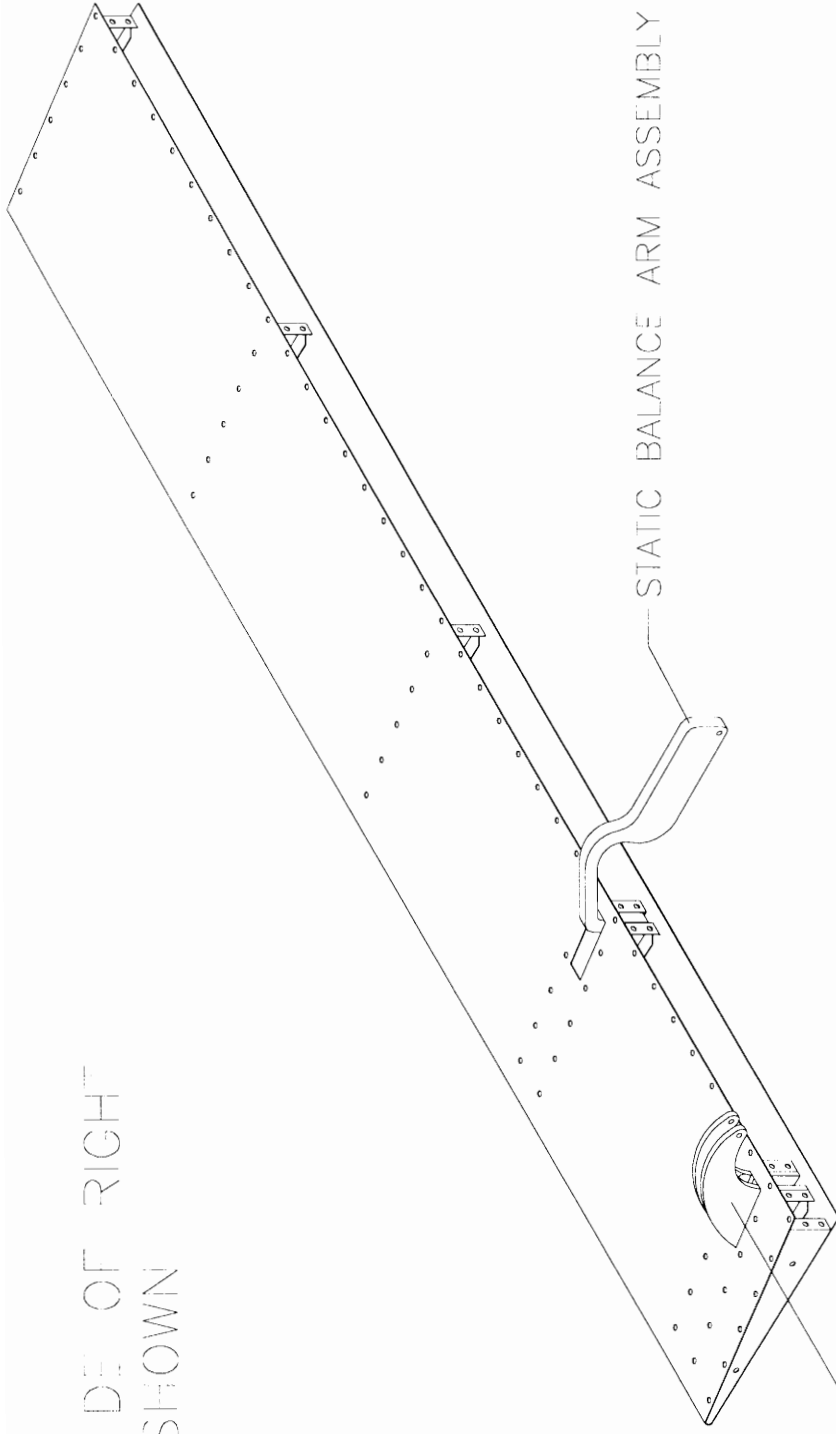


FIGURE 4-10

BOTTOM SIDE OF RIGHT  
AILERON SHOWN



AILERON HORN ASSEMBLY

STATIC BALANCE ARM ASSEMBLY

FIGURE 4-11

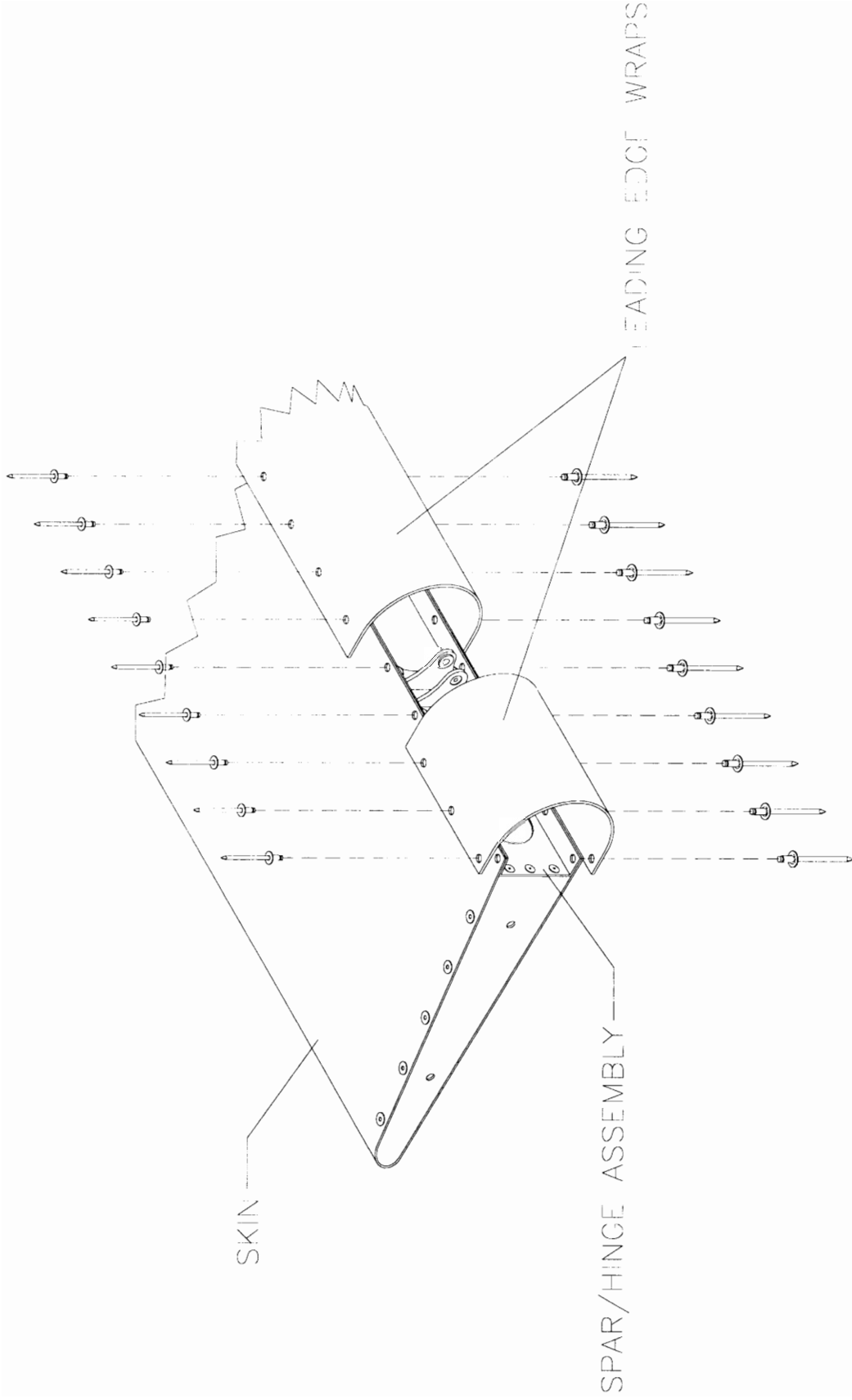


FIGURE 4-12



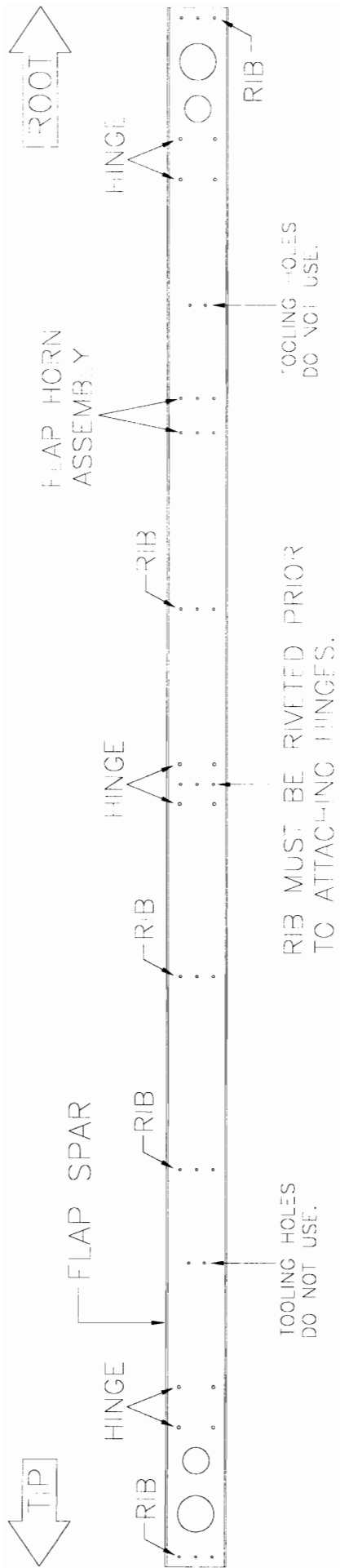


FIGURE 4-13

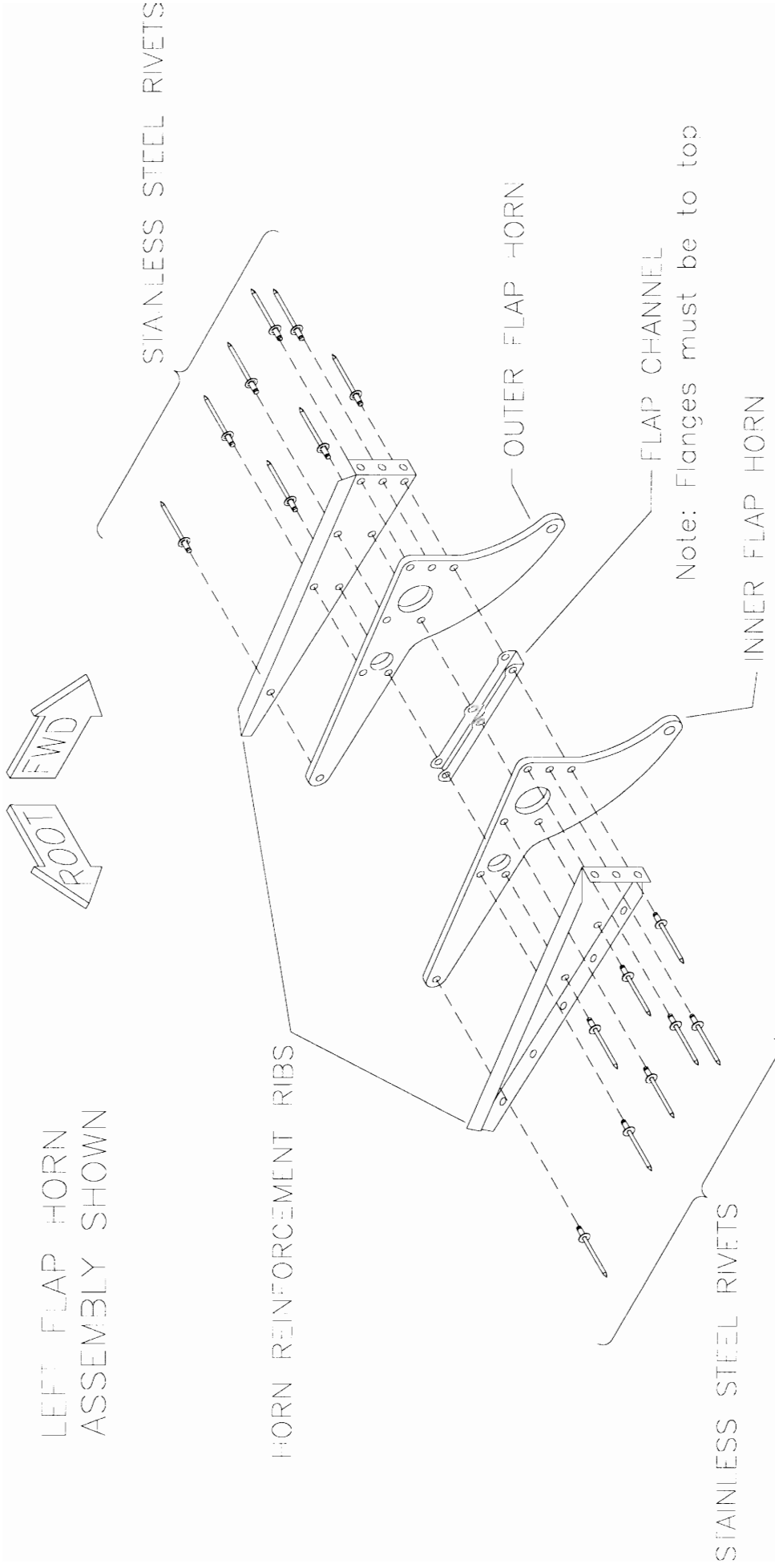


FIGURE 4-14

BOTTOM SIDE OF RIGHT  
FLAP SHOWN

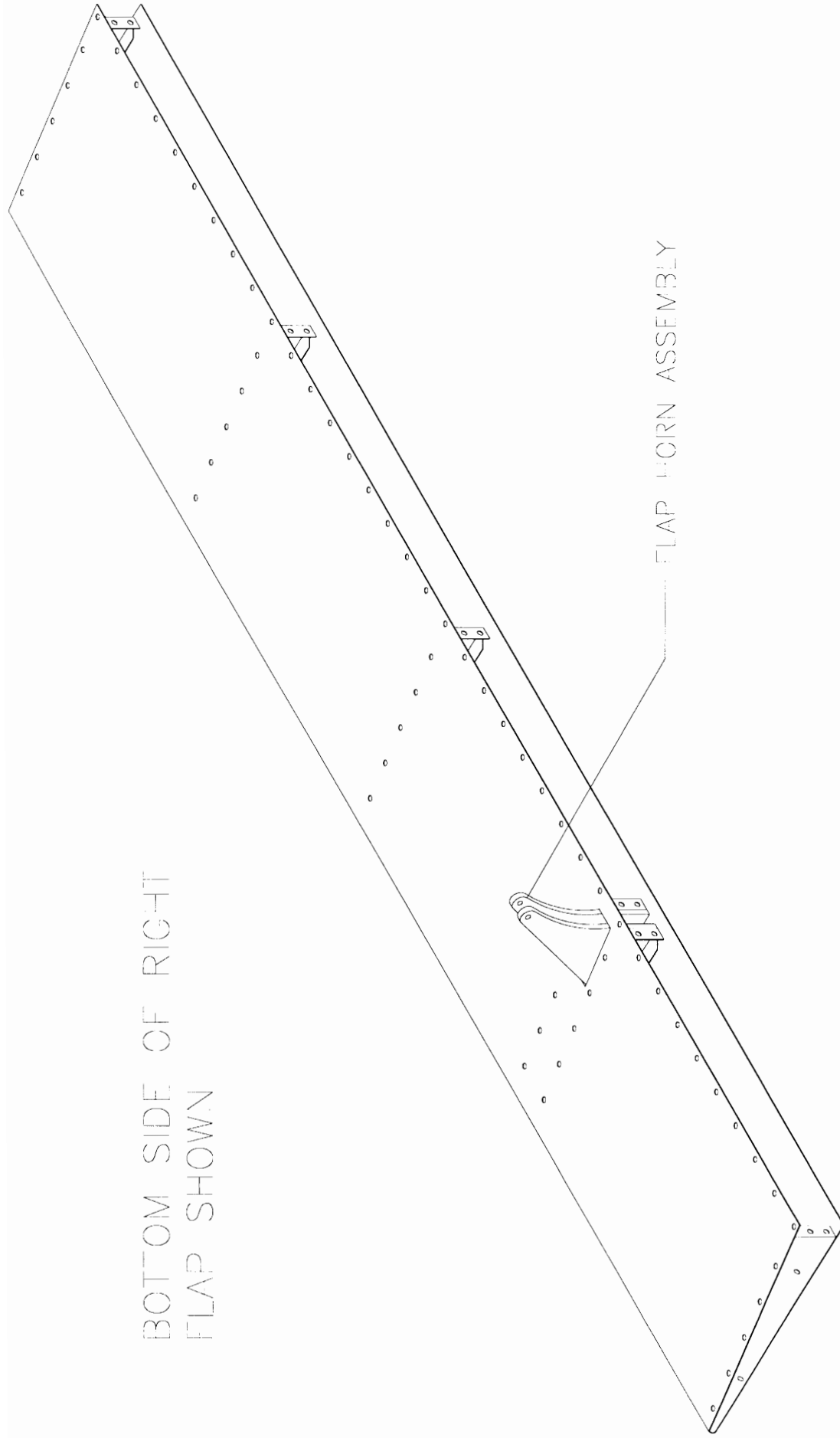


FIGURE 4-15

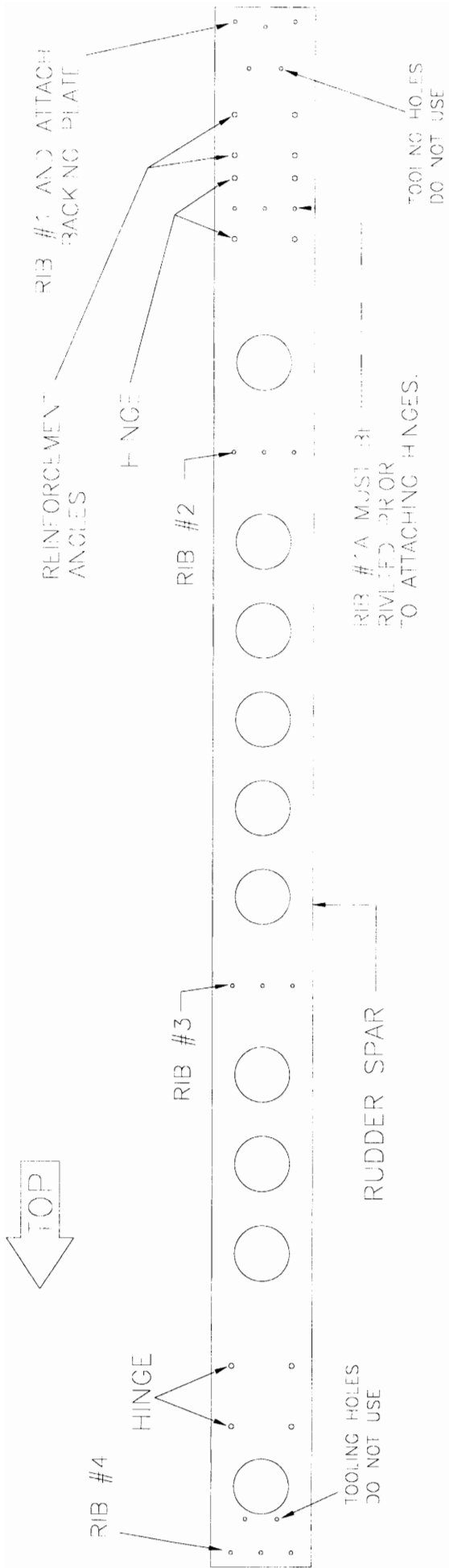


FIGURE 4-16

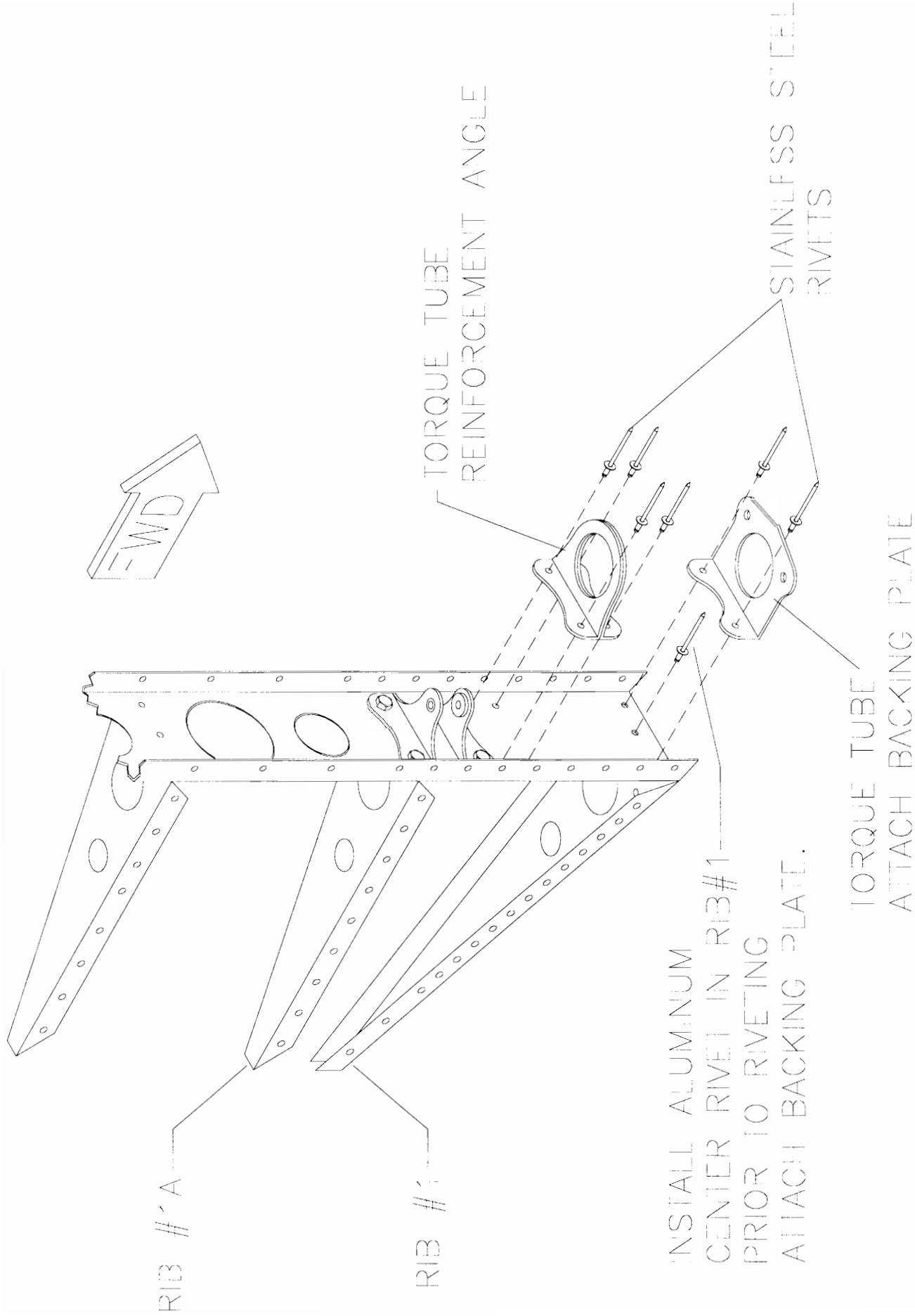


FIGURE 4-17

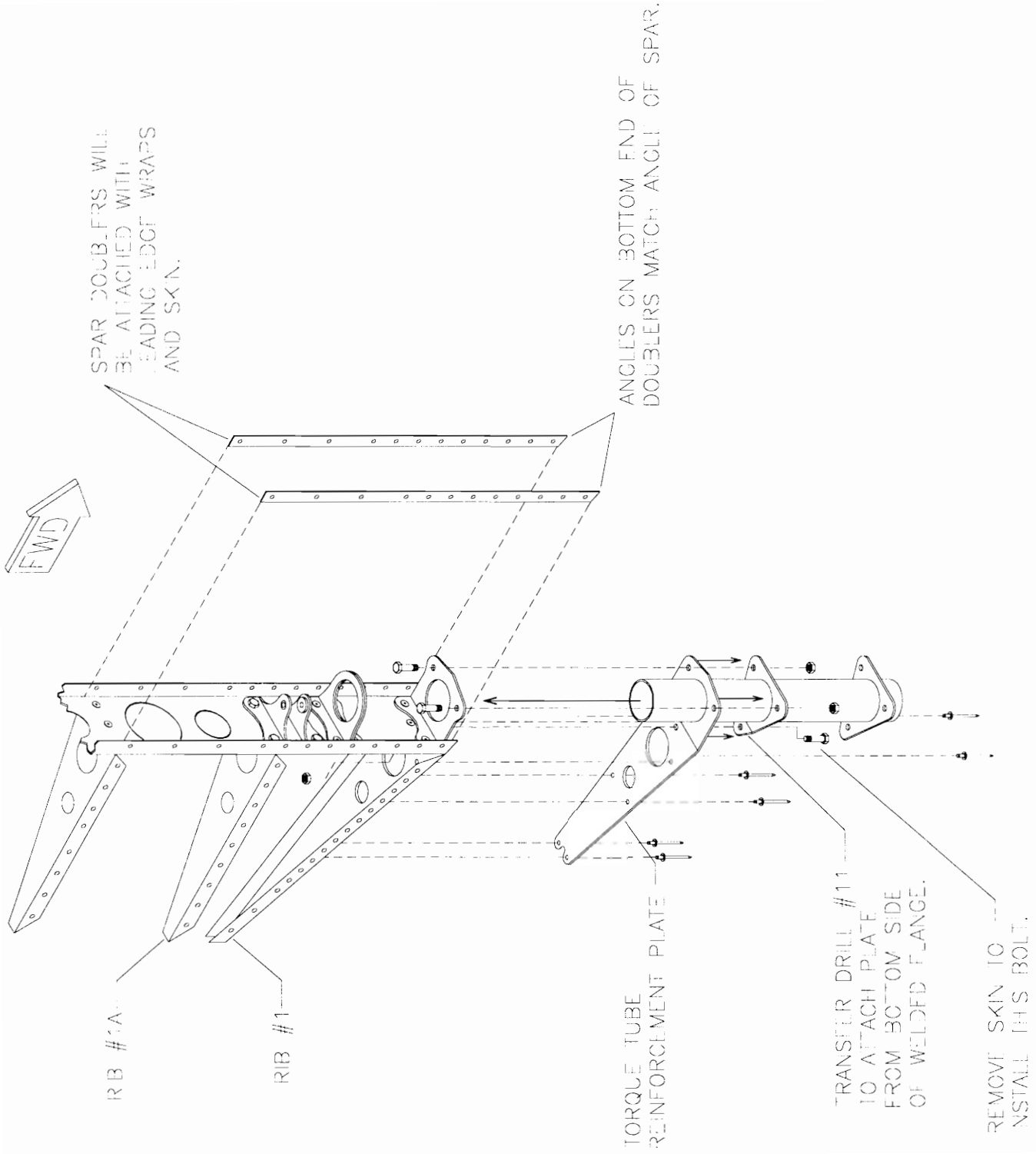
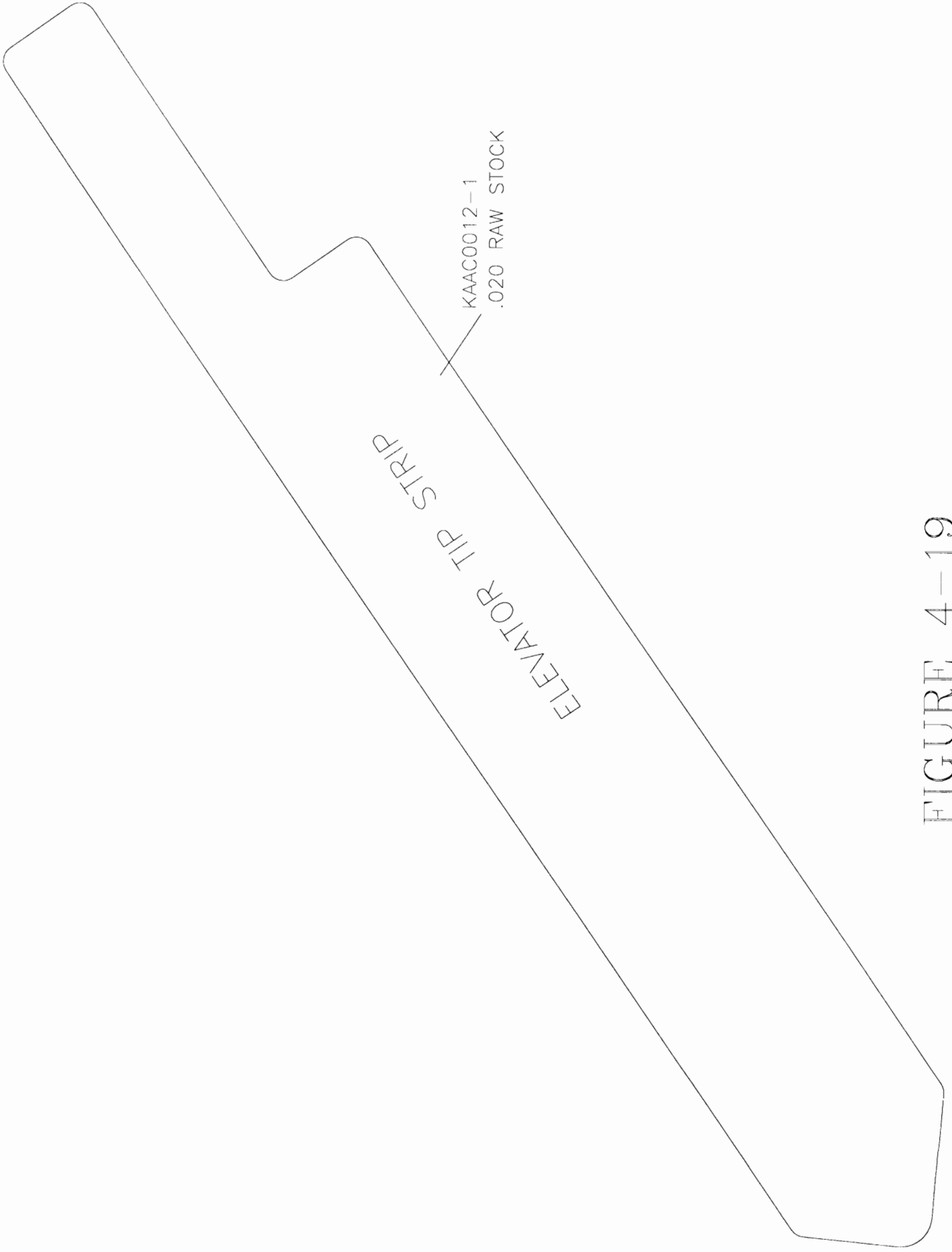


FIGURE 4-18



ELEVATOR TIP STRIP

KAAC0012-1  
.020 RAW STOCK

FIGURE 4-19

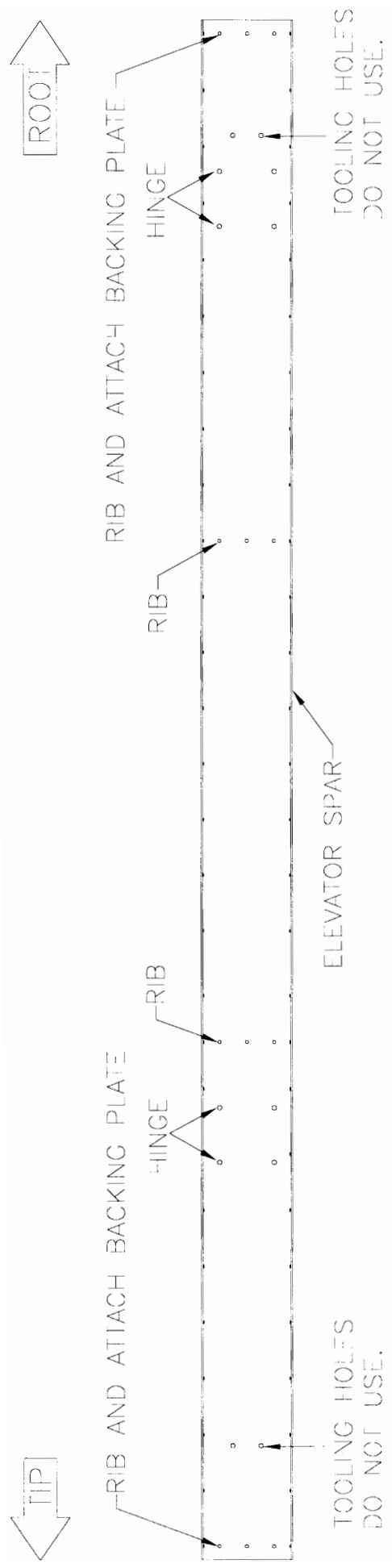
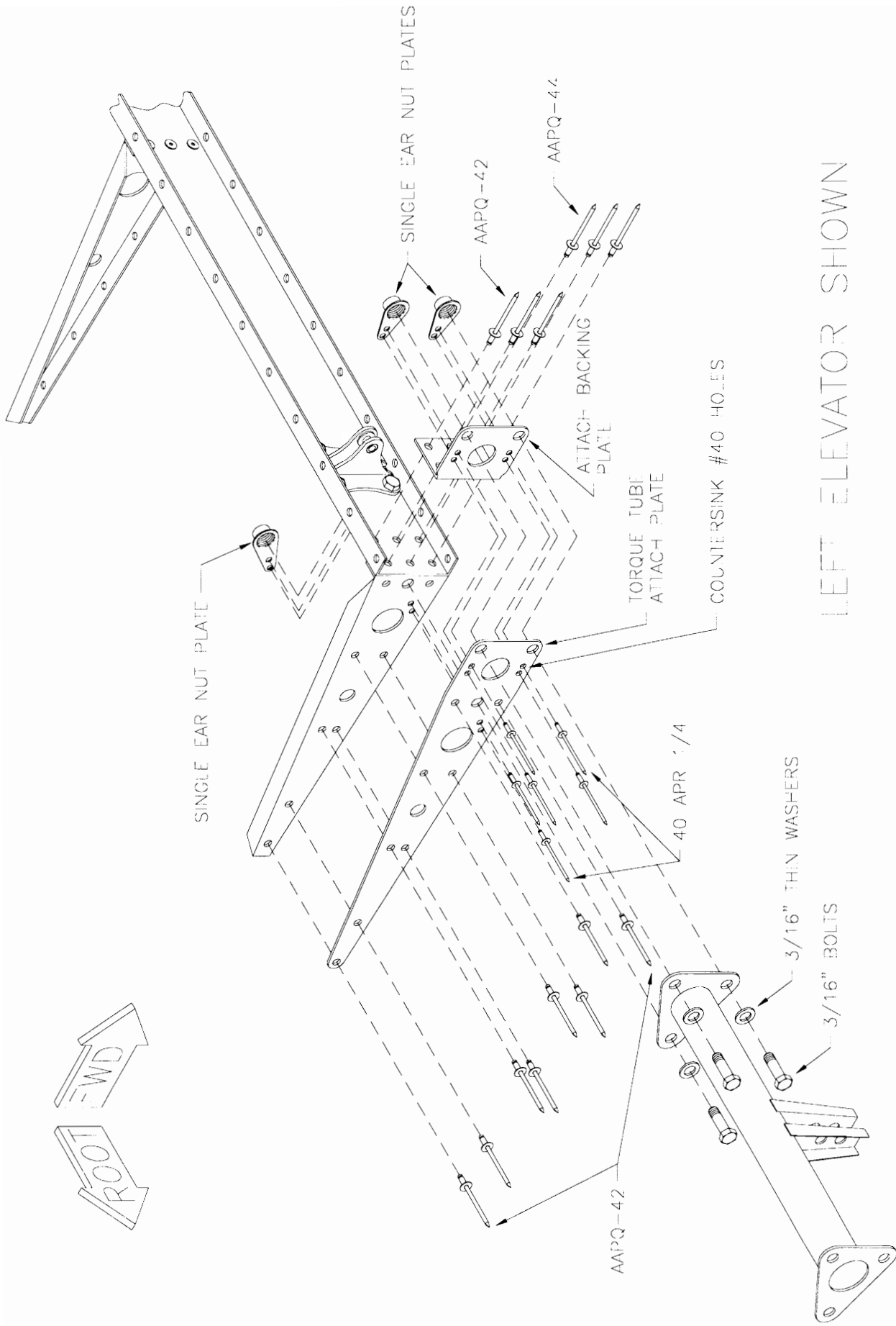


FIGURE 4--20





LEFT ELEVATOR SHOWN

FIGURE 4-21

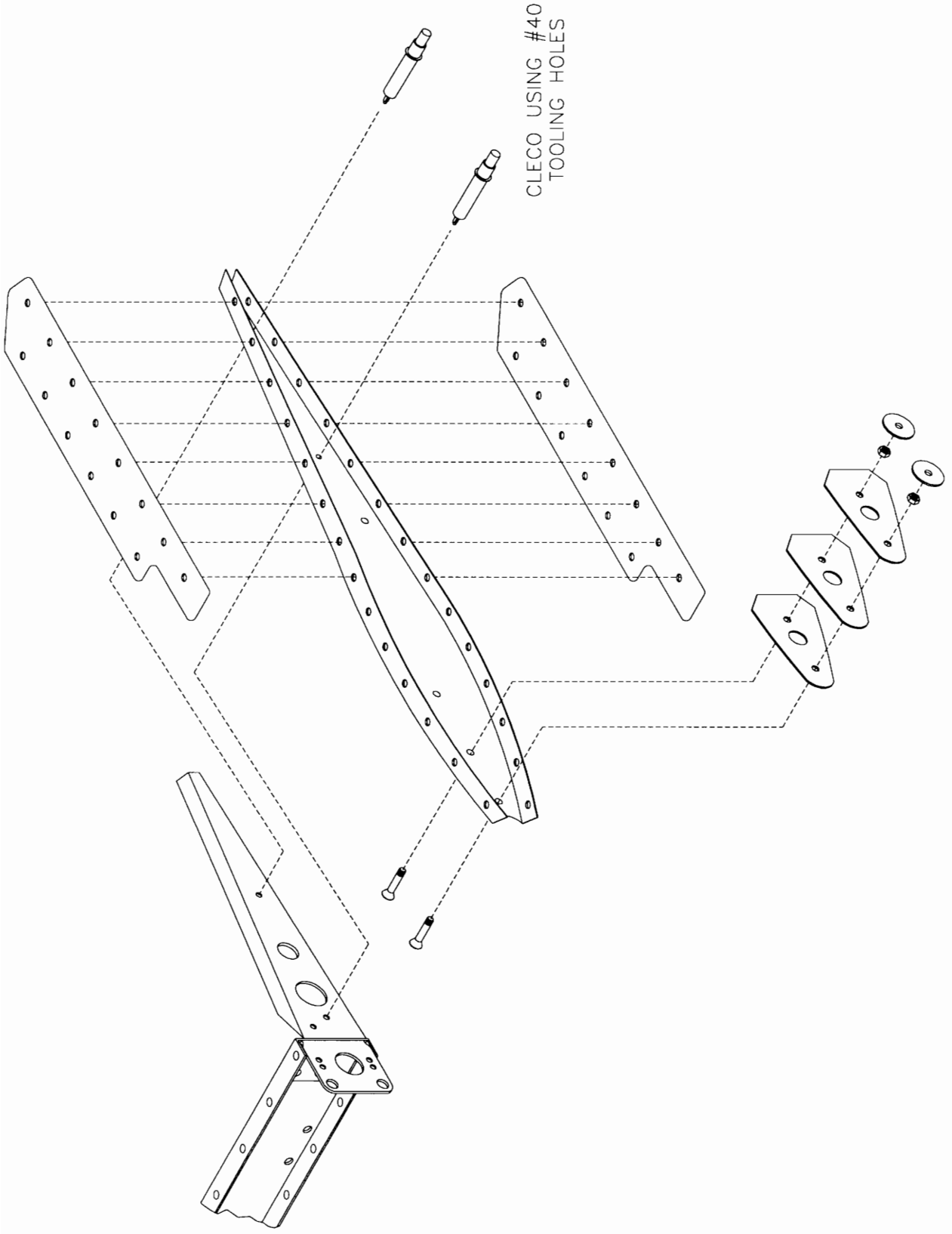
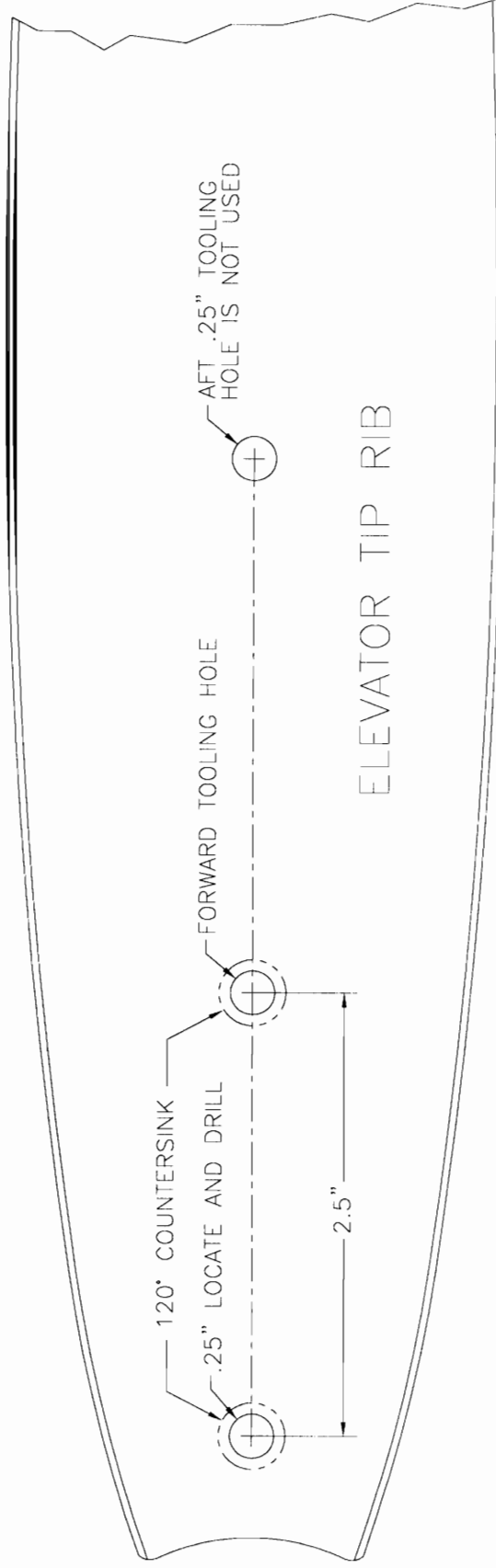
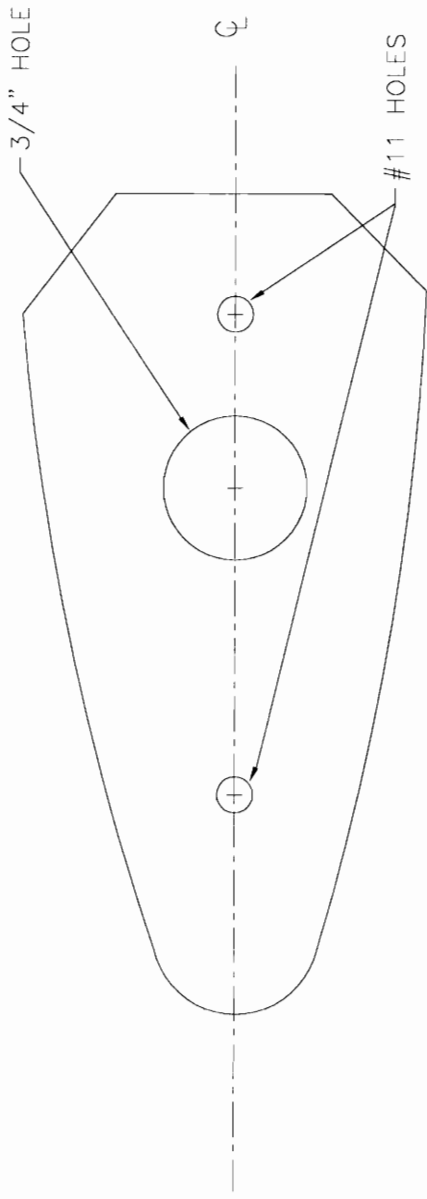


FIGURE 4-22

MASS BALANCE TEMPLATE



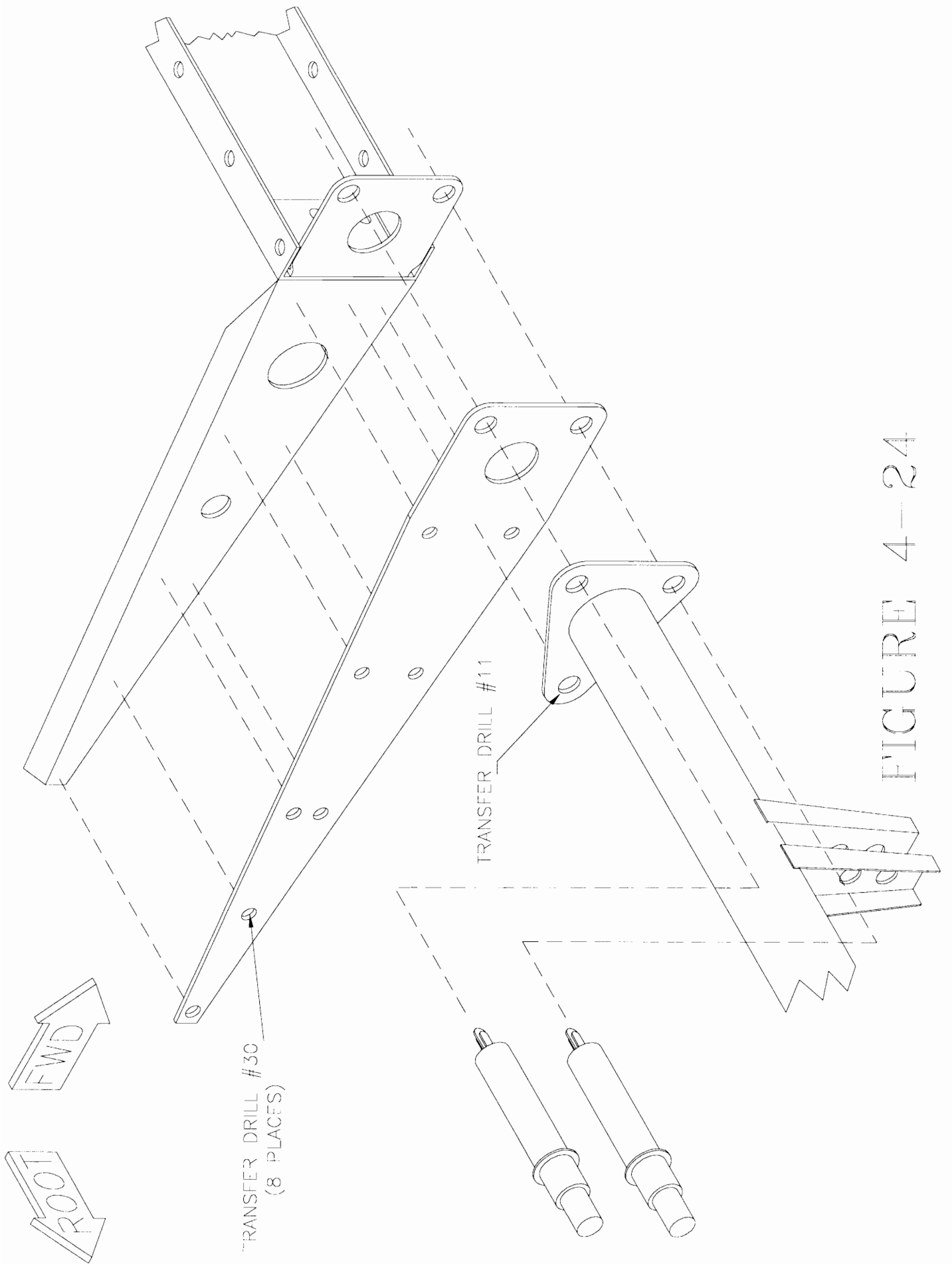
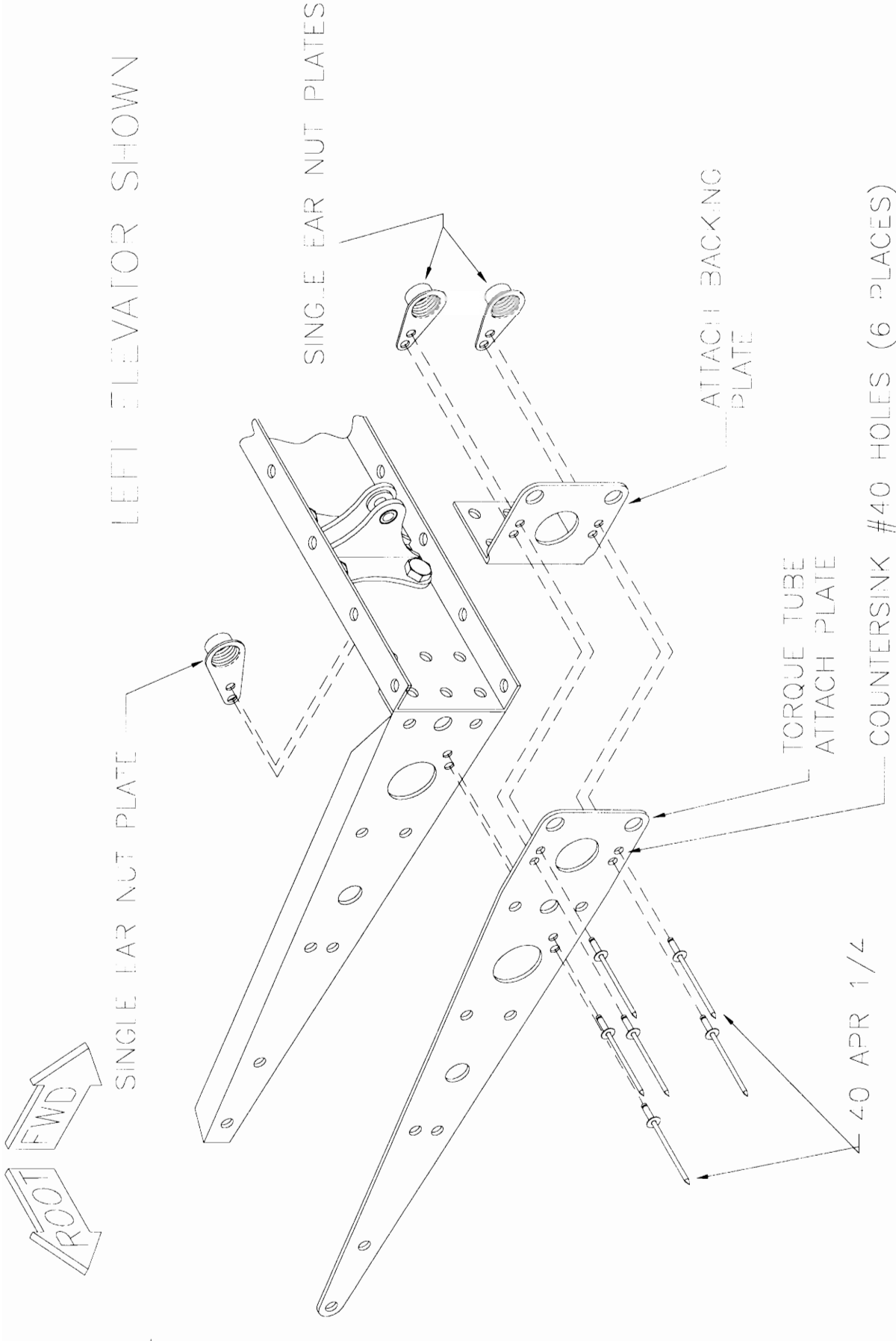


FIGURE 4-24



LEFT ELEVATOR SHOWN

SINGLE EAR NUT PLATE

SINGLE EAR NUT PLATES

ATTACH BACKING PLATE

TORQUE TUBE ATTACH PLATE

COUNTERSINK #40 HOLES (6 PLACES)

40 APR 1/4

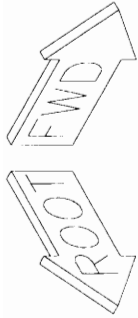


FIGURE 4--25

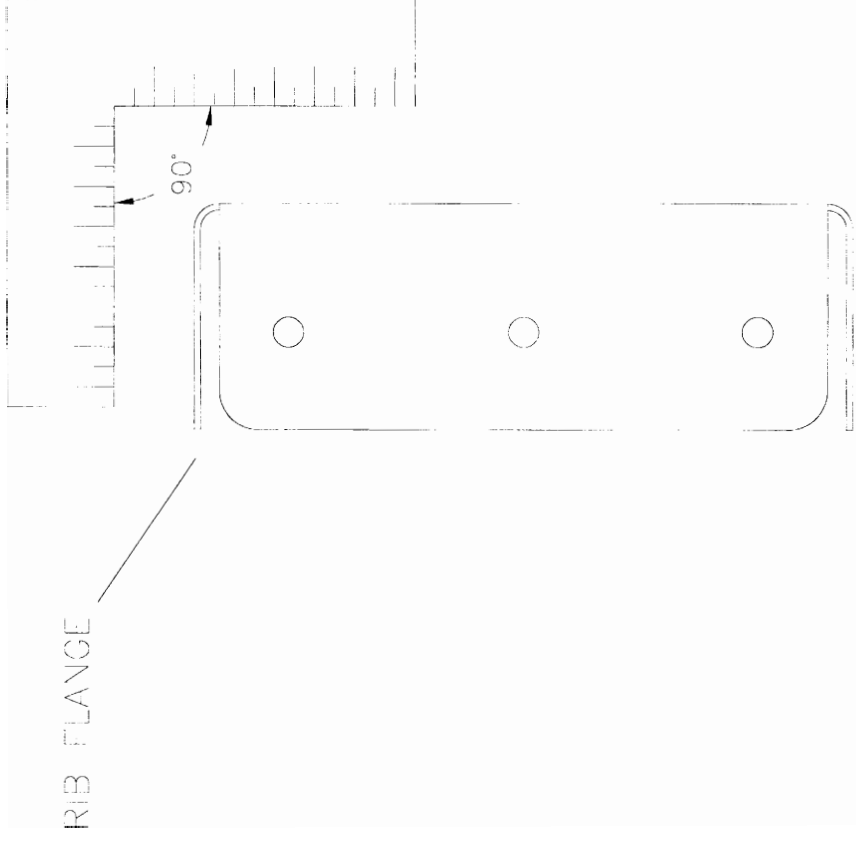
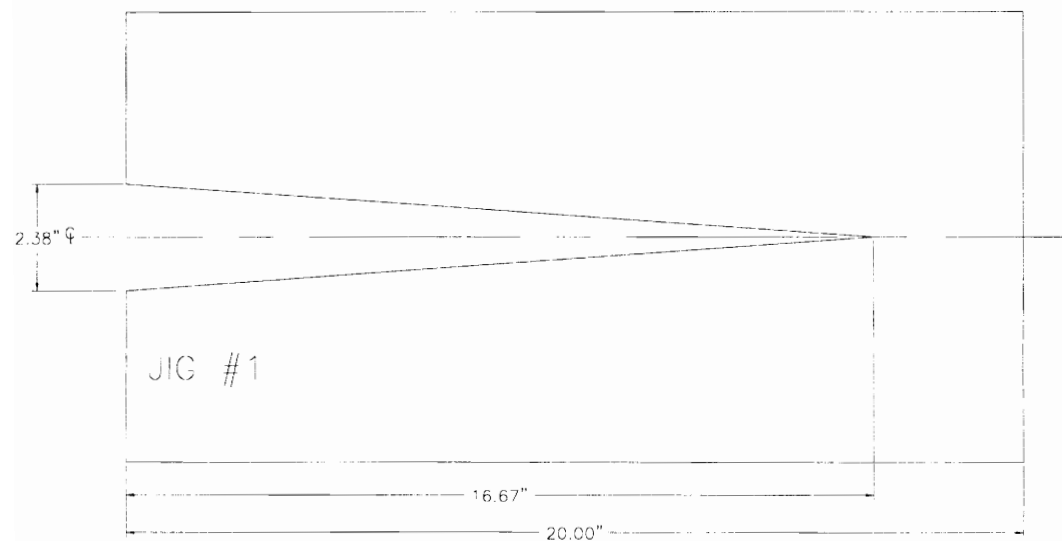
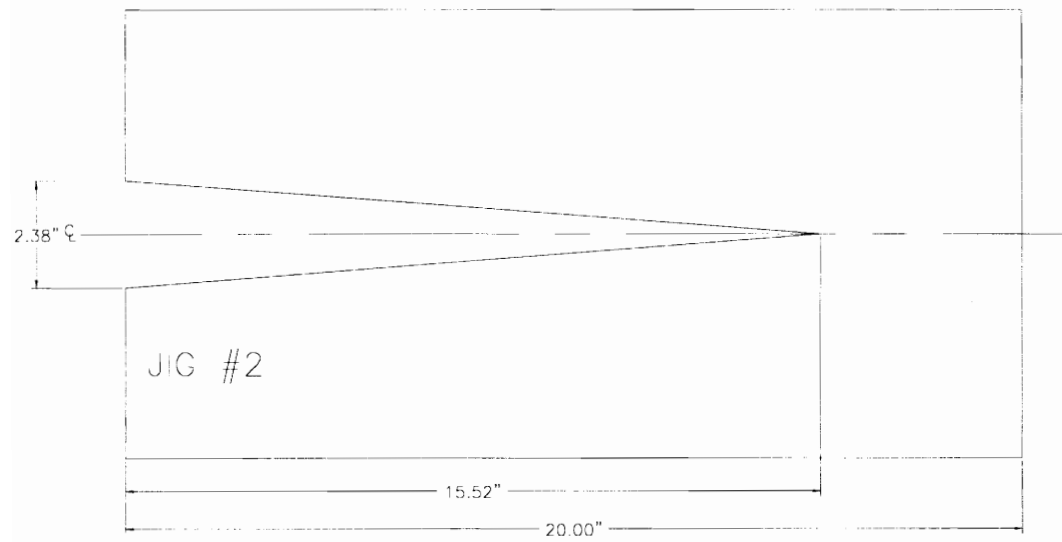
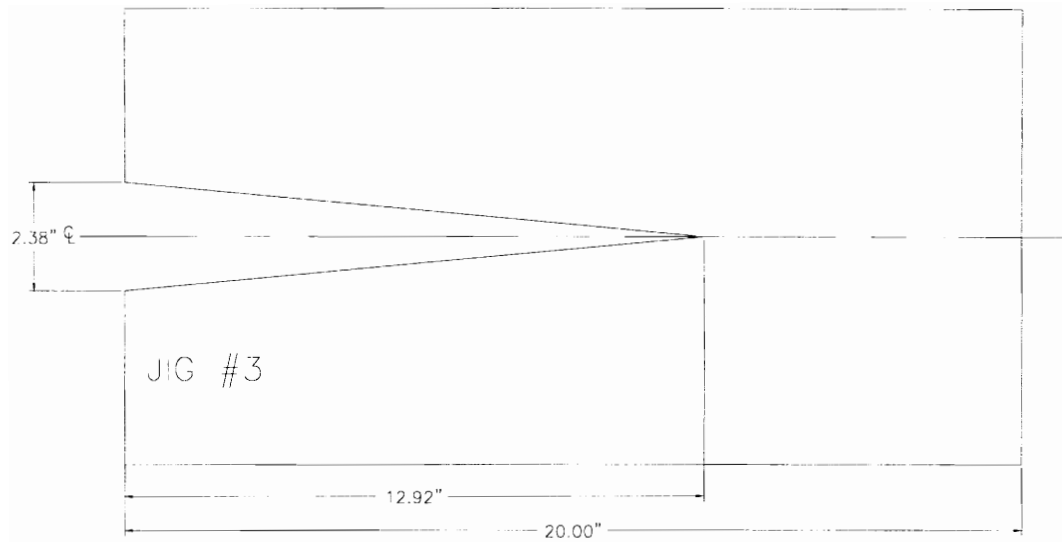
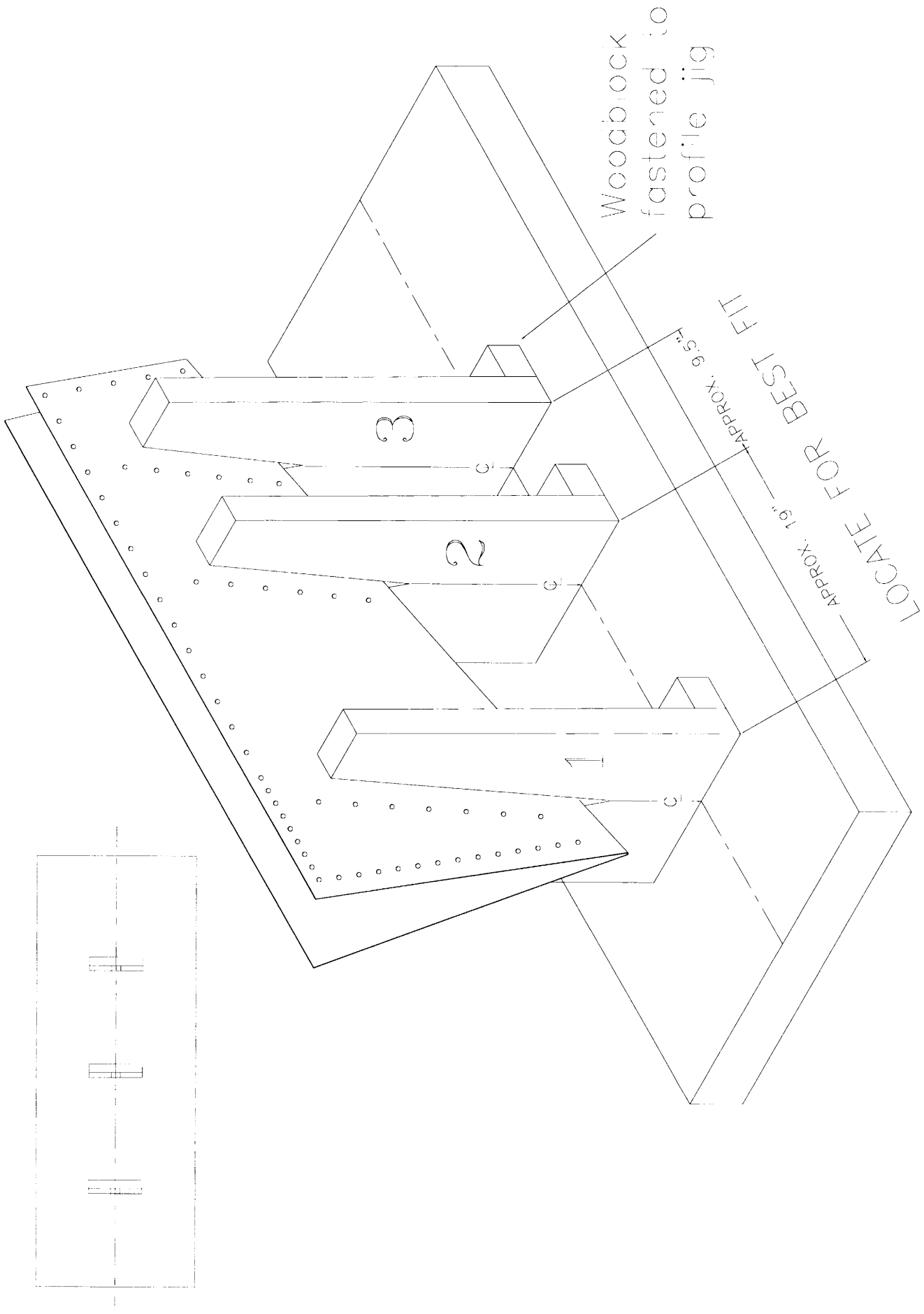


FIGURE 4 - 26







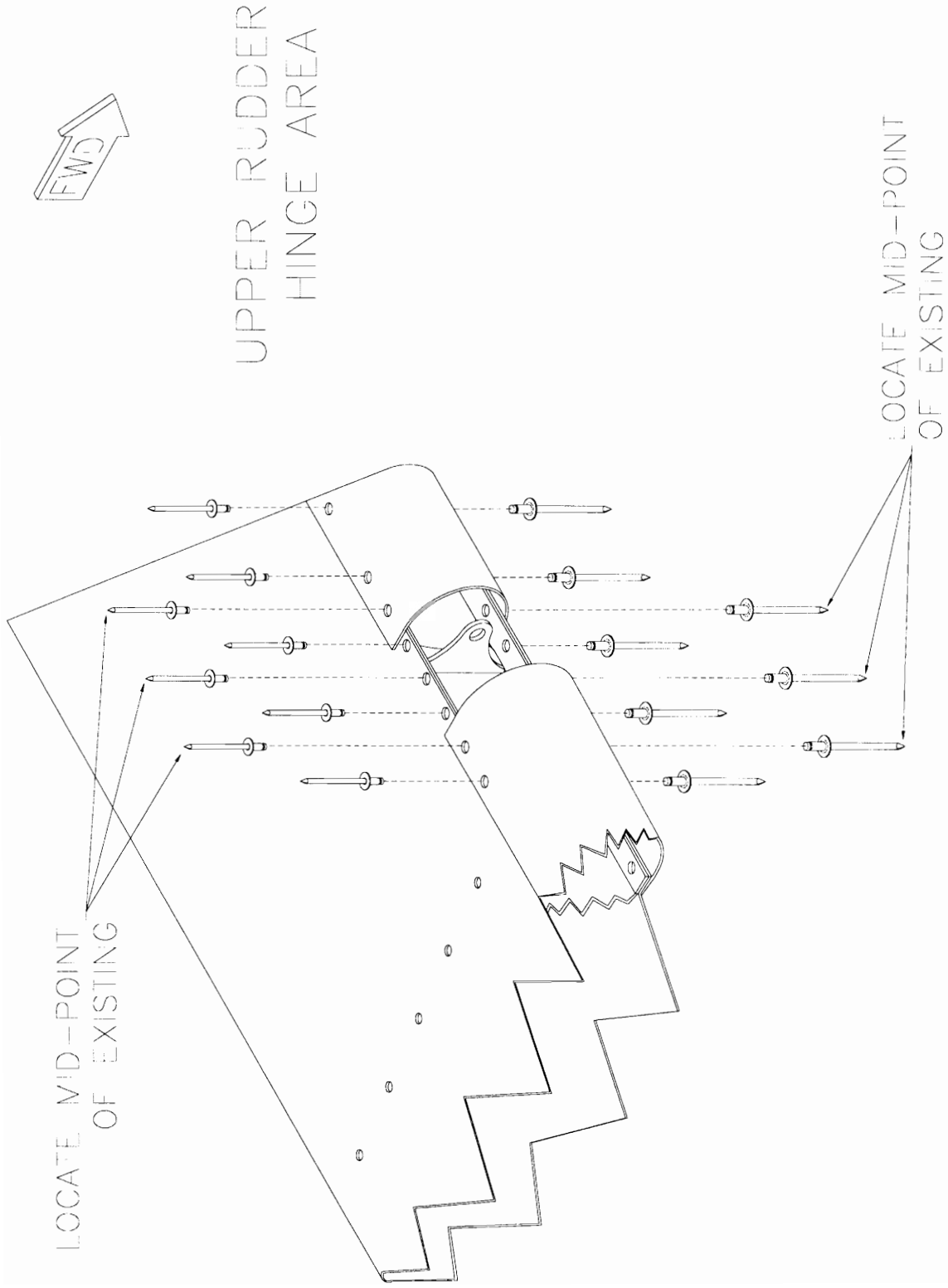


FIGURE 4-29

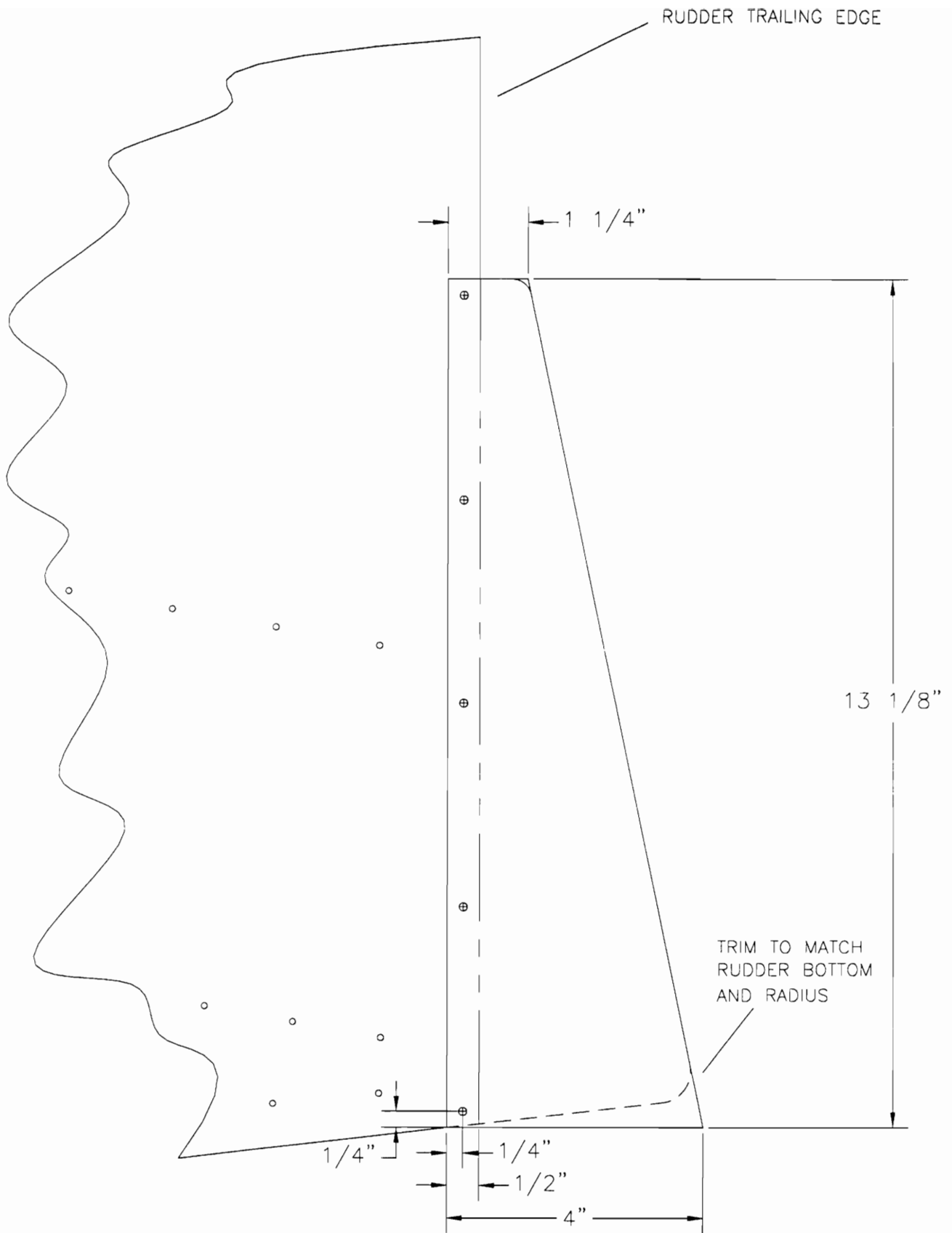


FIGURE 4-30

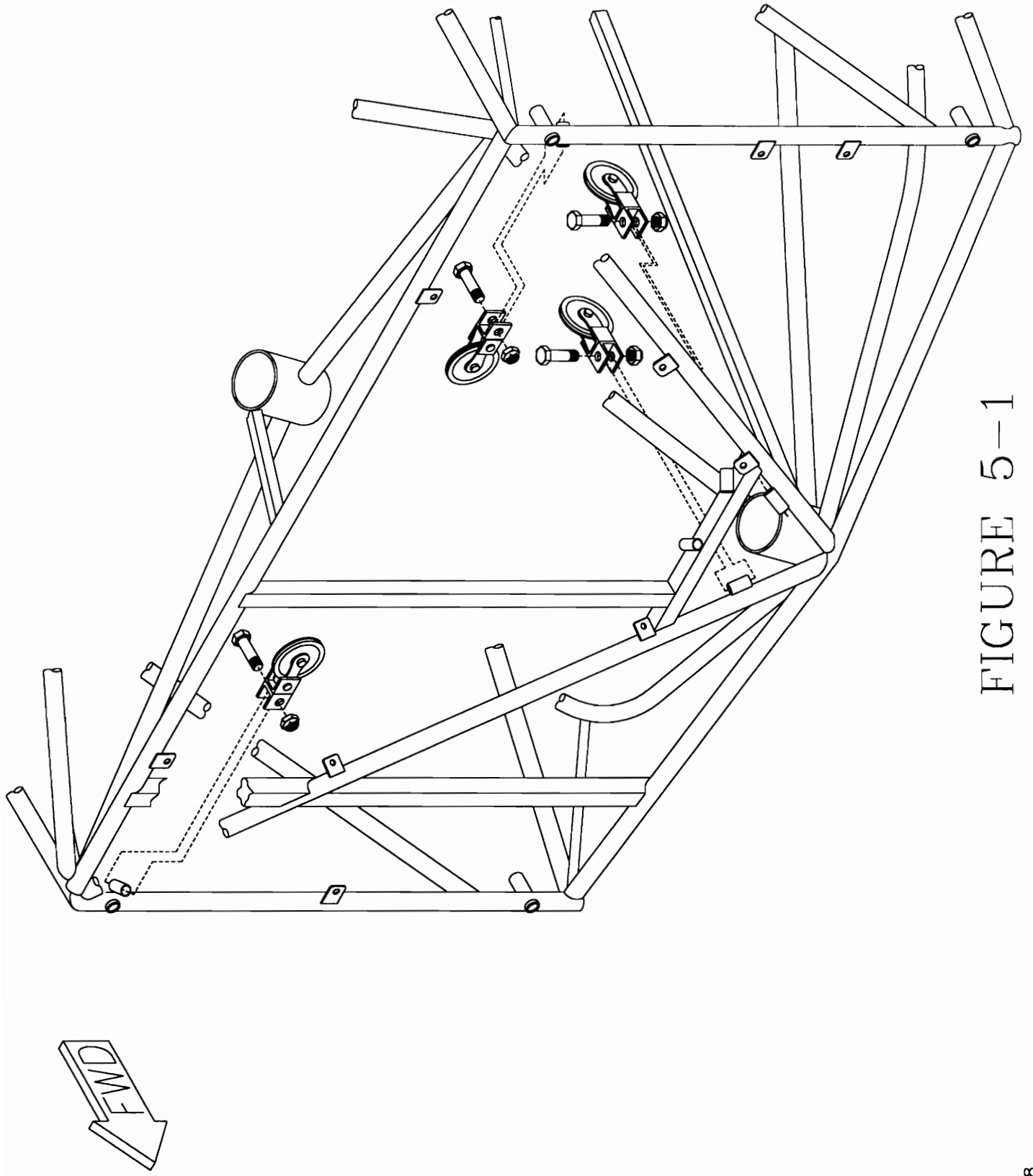


FIGURE 5-1

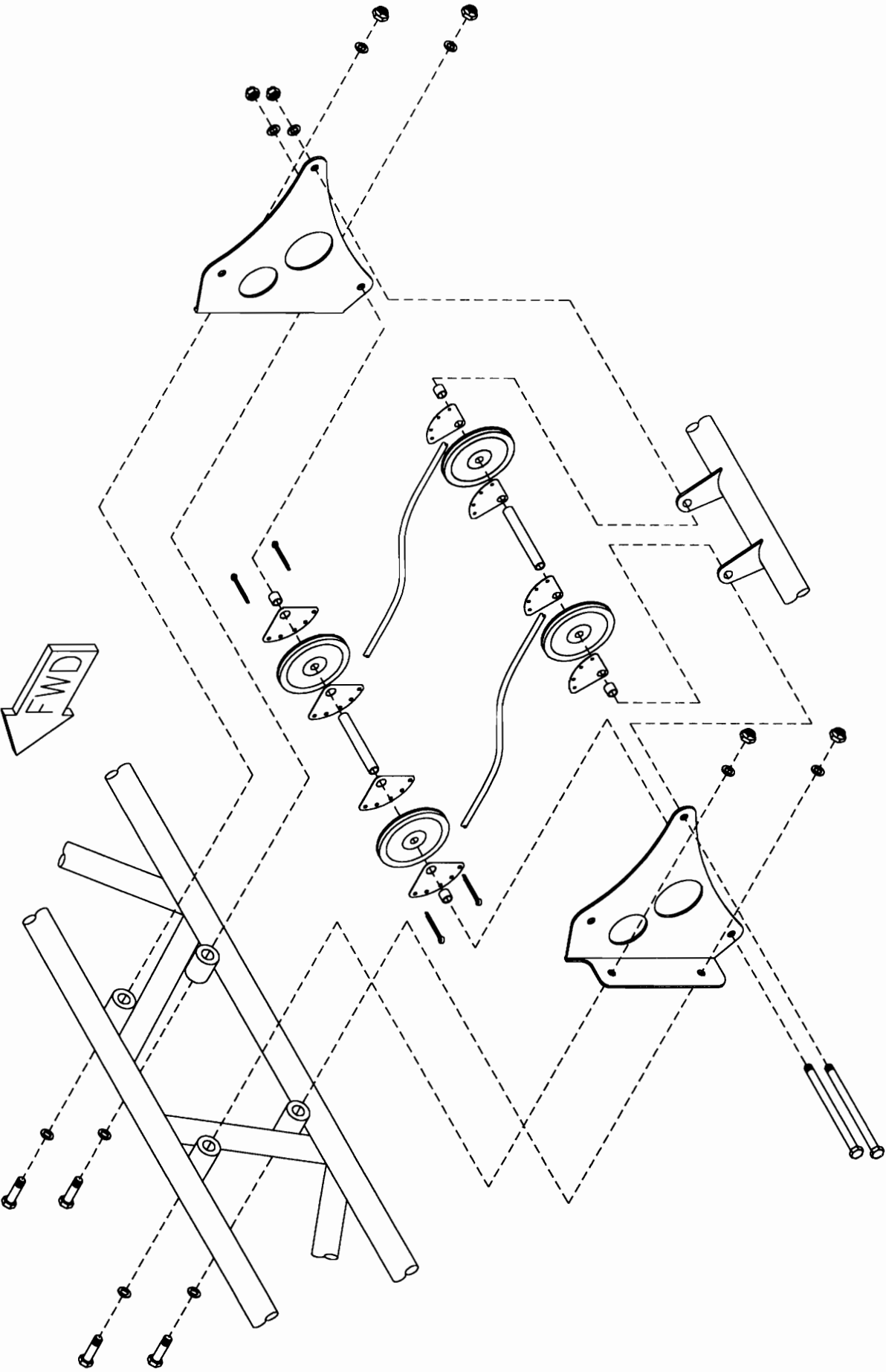


FIGURE 5-2

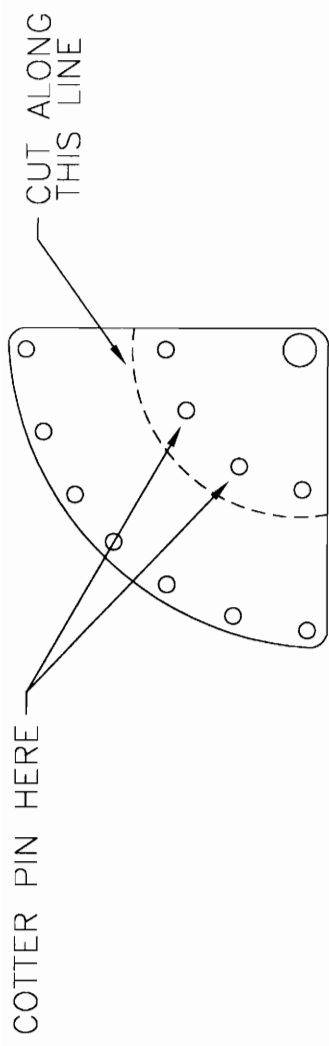


FIGURE 5-3

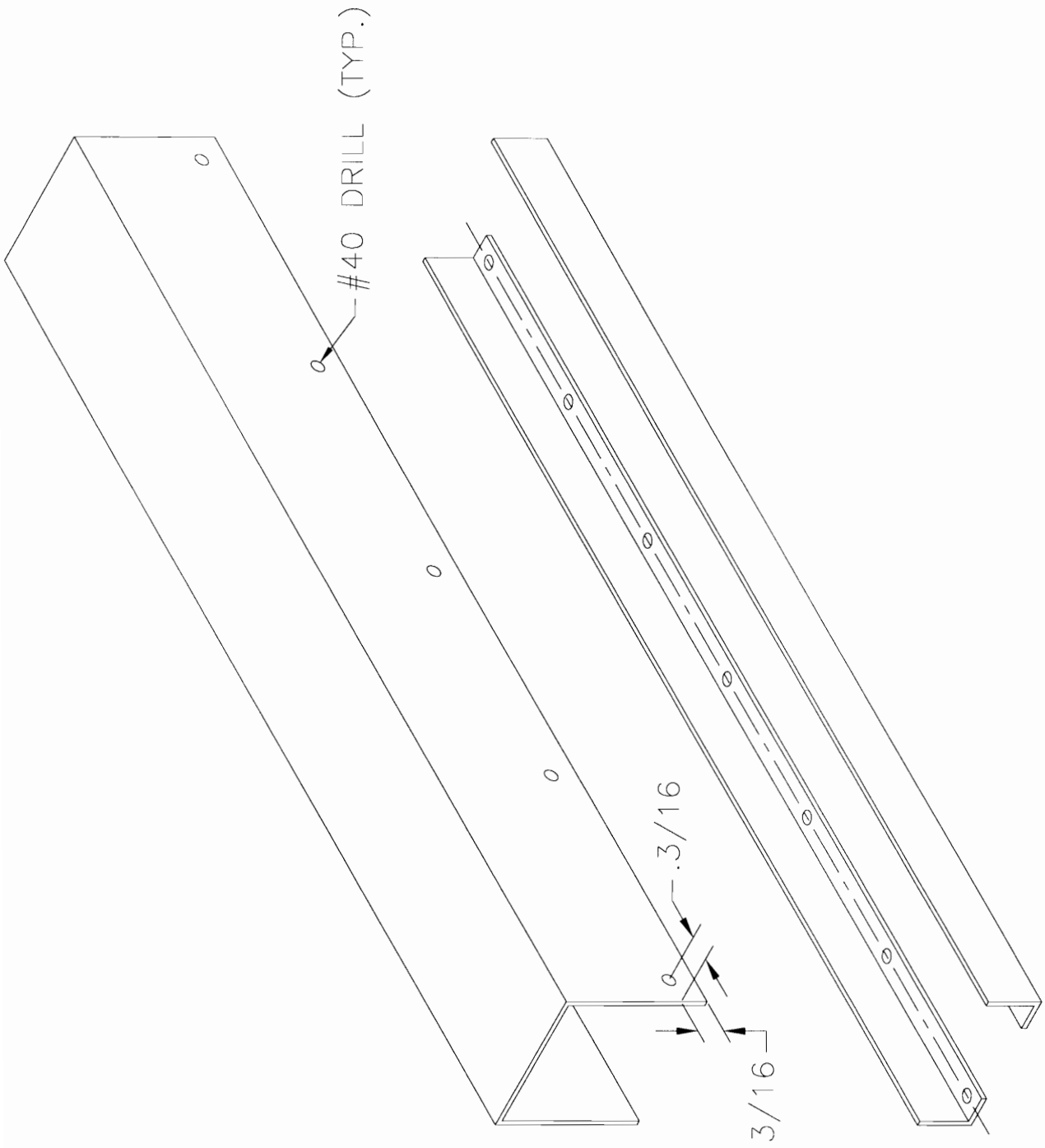


FIGURE 5-3A

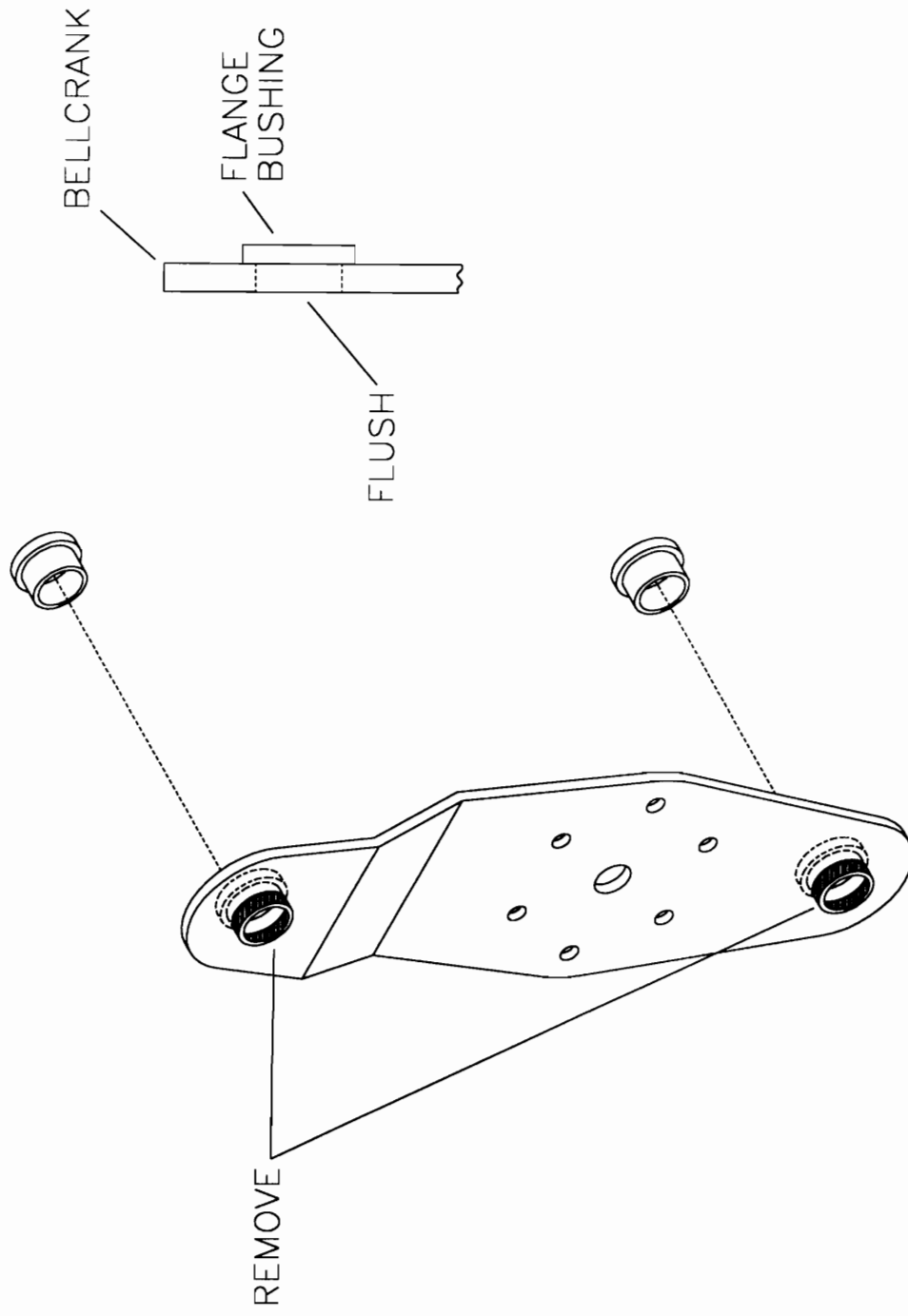
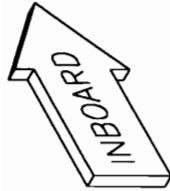
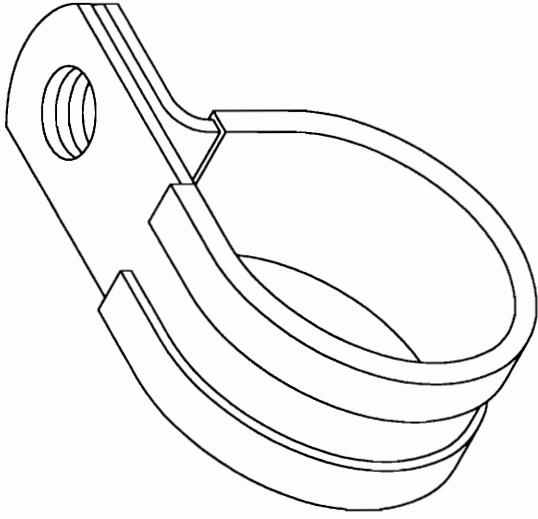
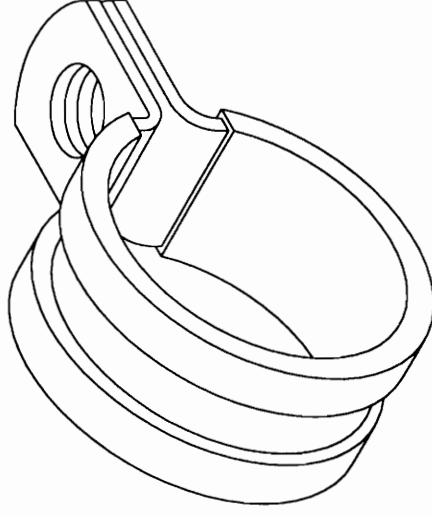


FIGURE 5-4

BEFORE



AFTER





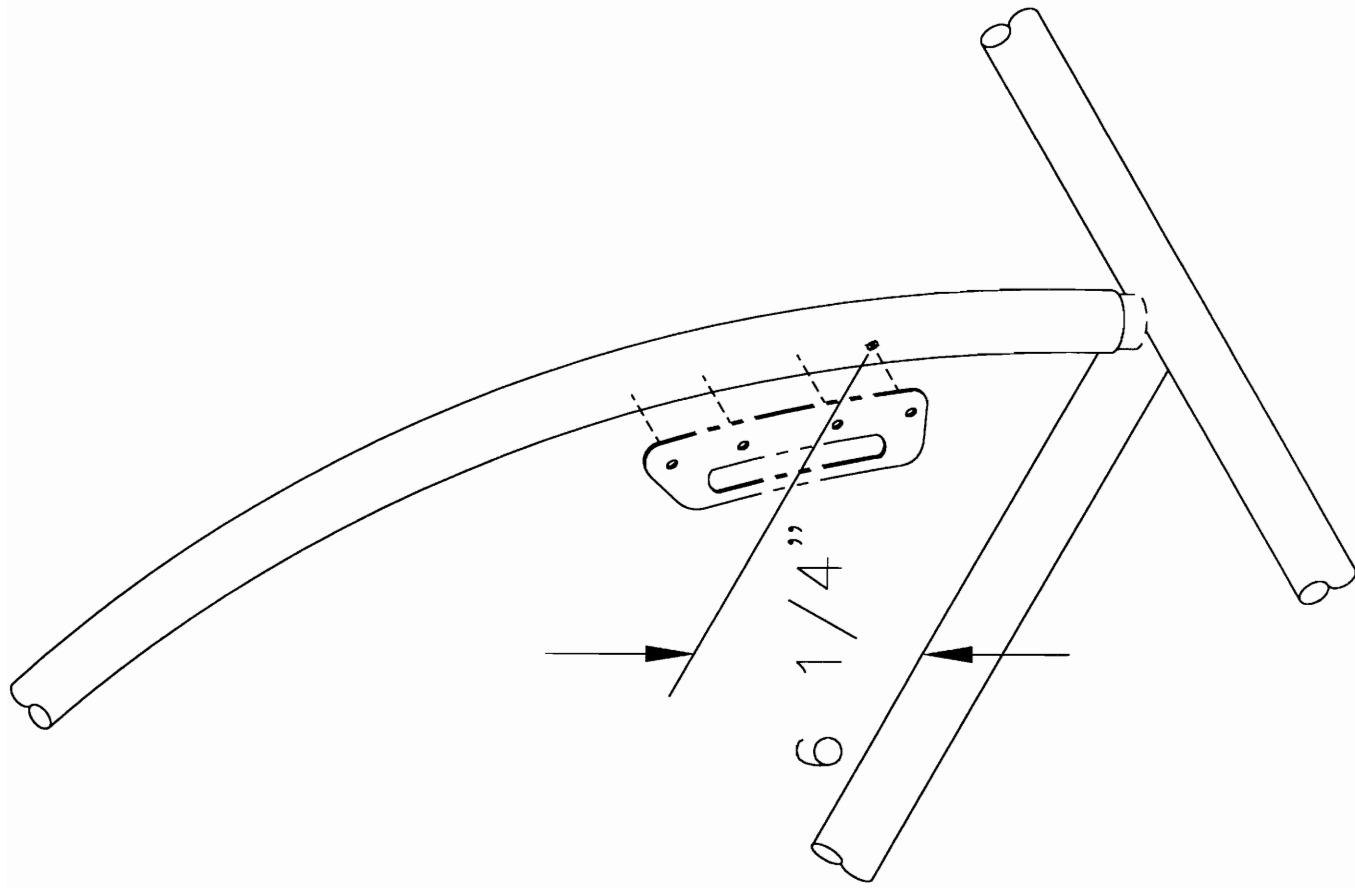
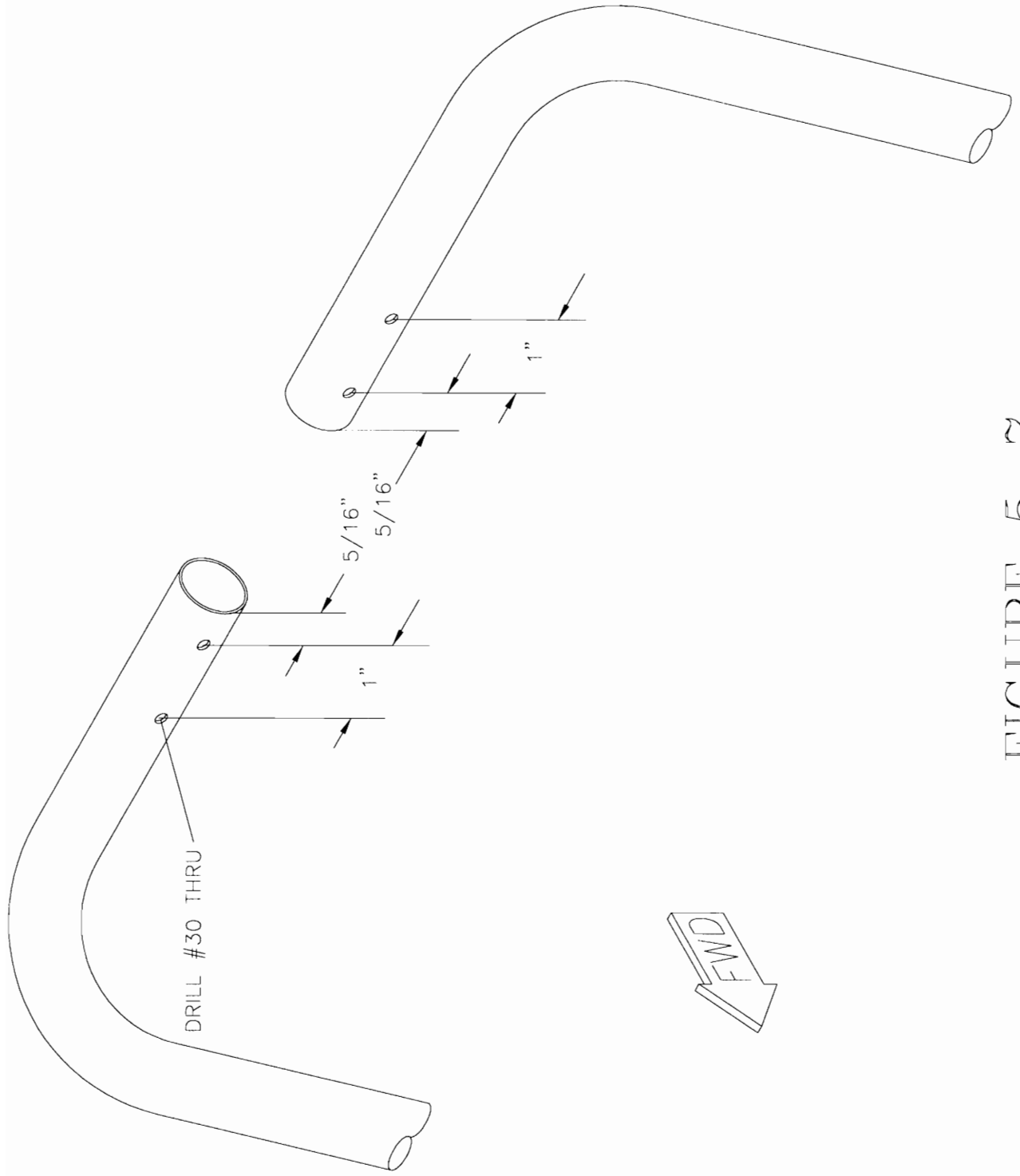


FIGURE 5-6



END

FIGURE 5-7

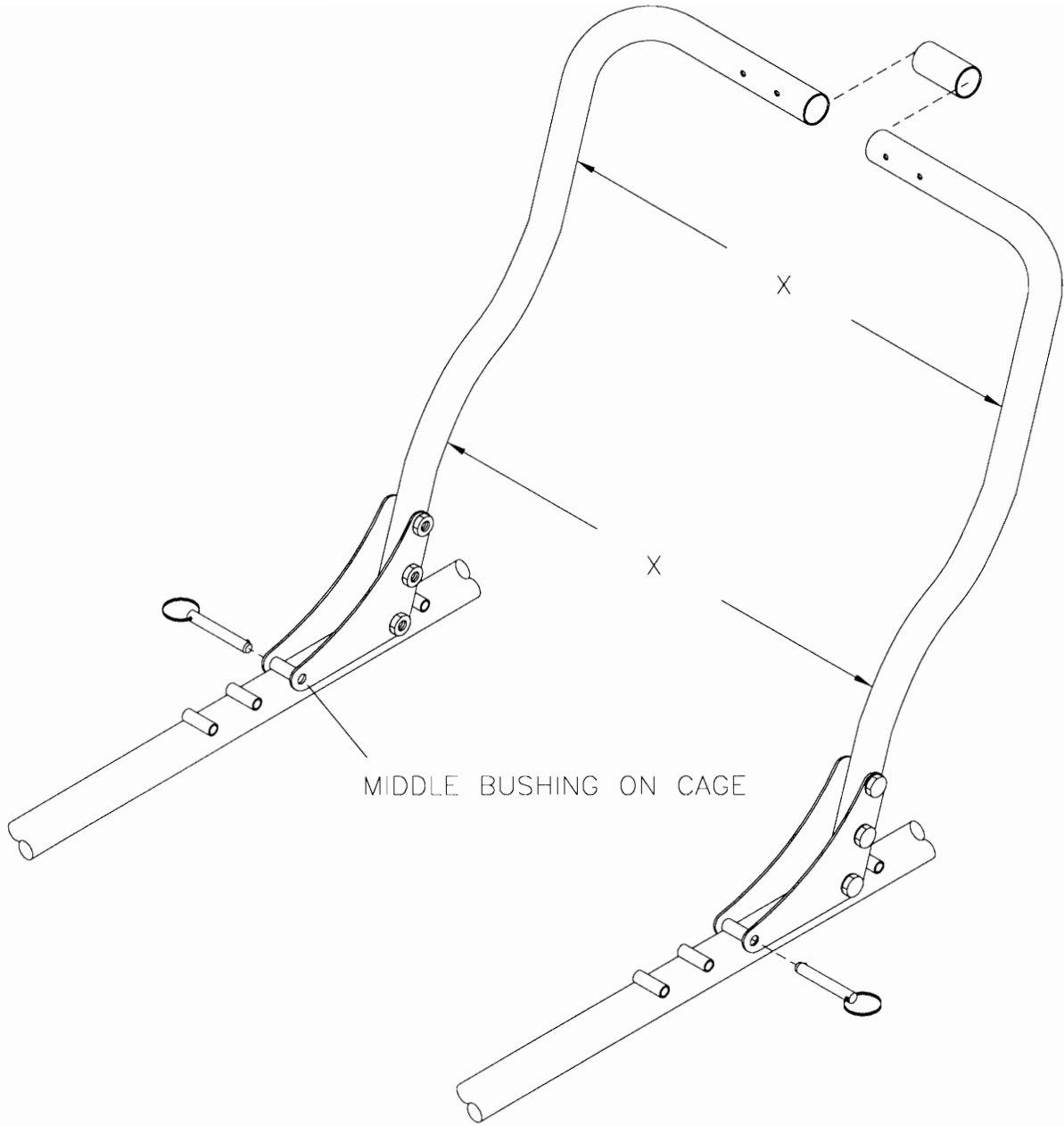


FIGURE 5-8

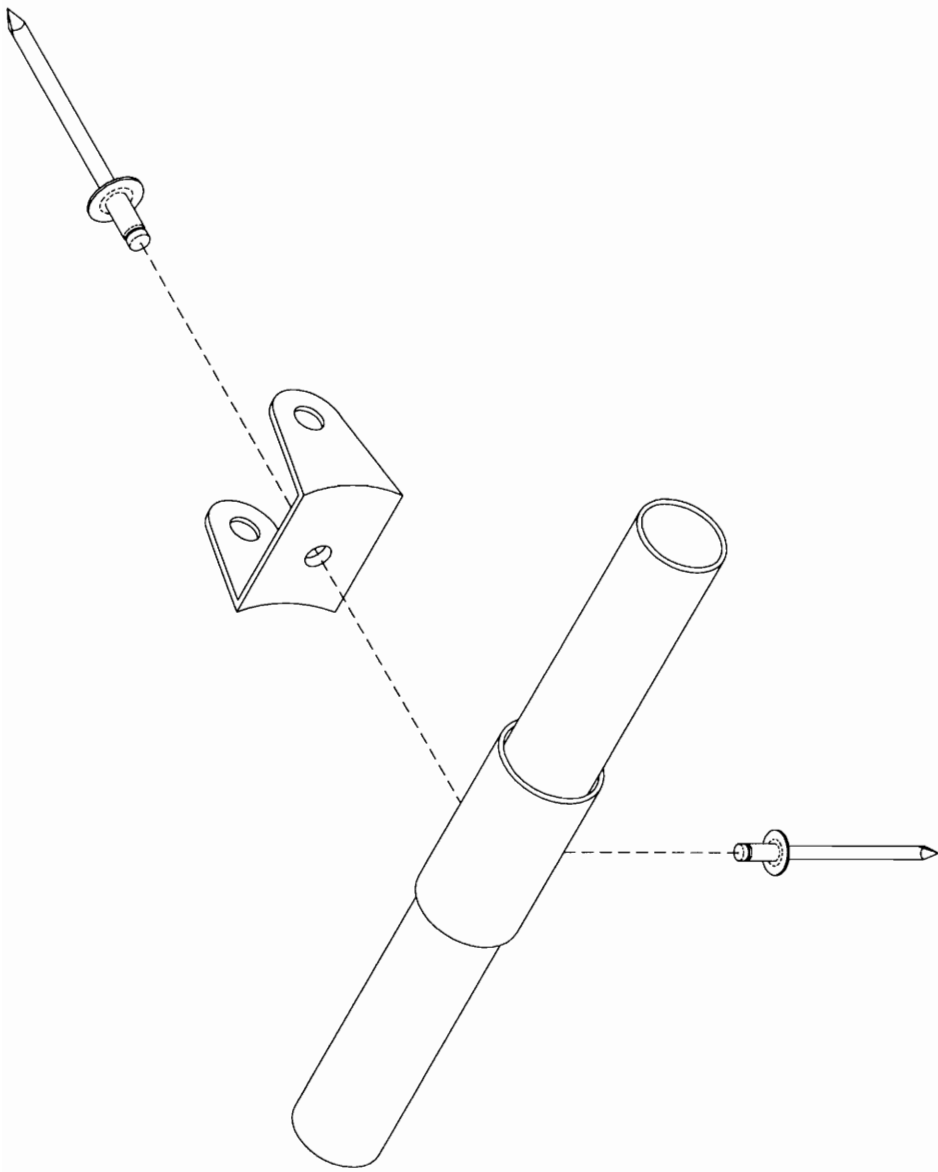


FIGURE 5-9

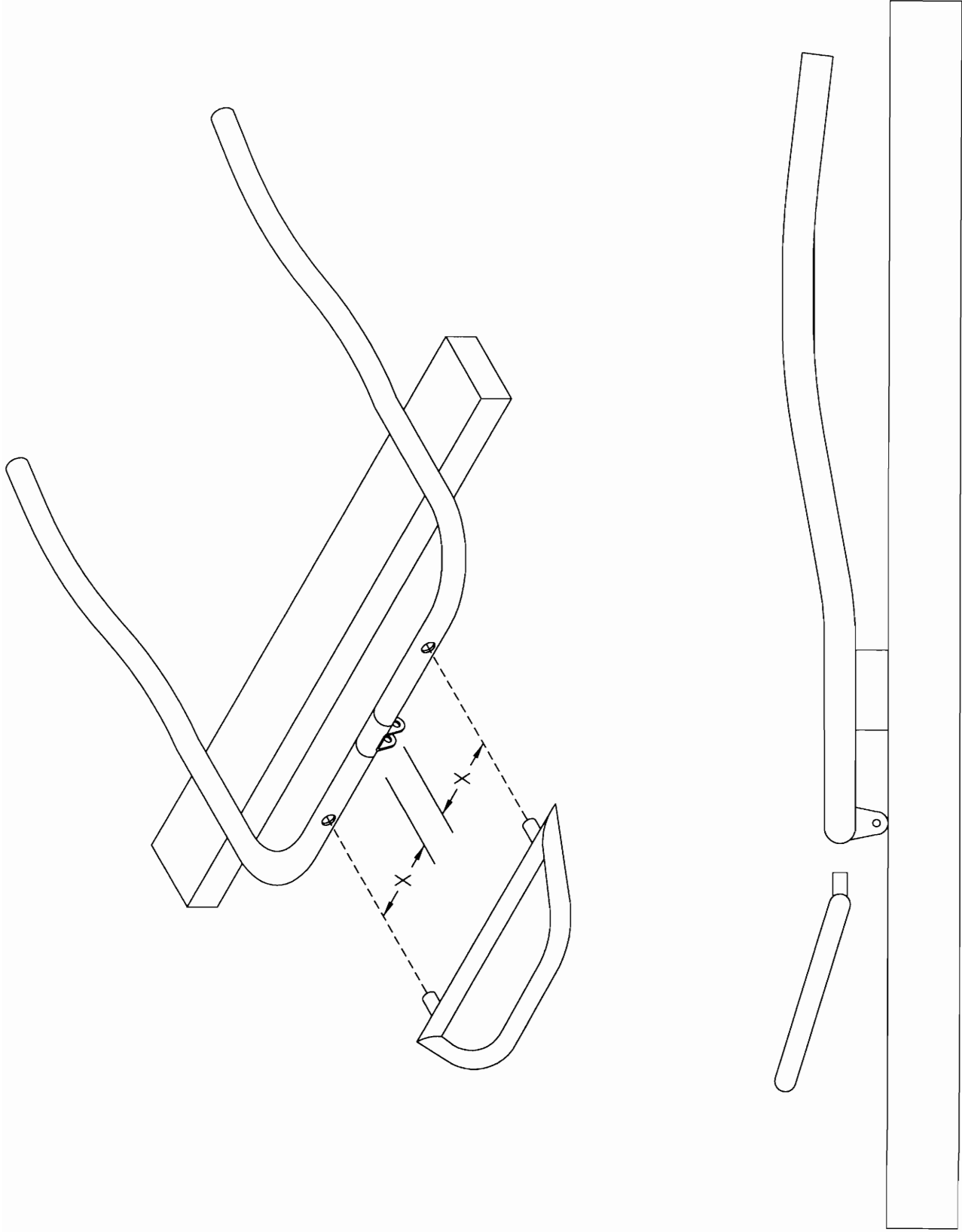


FIGURE 5-10

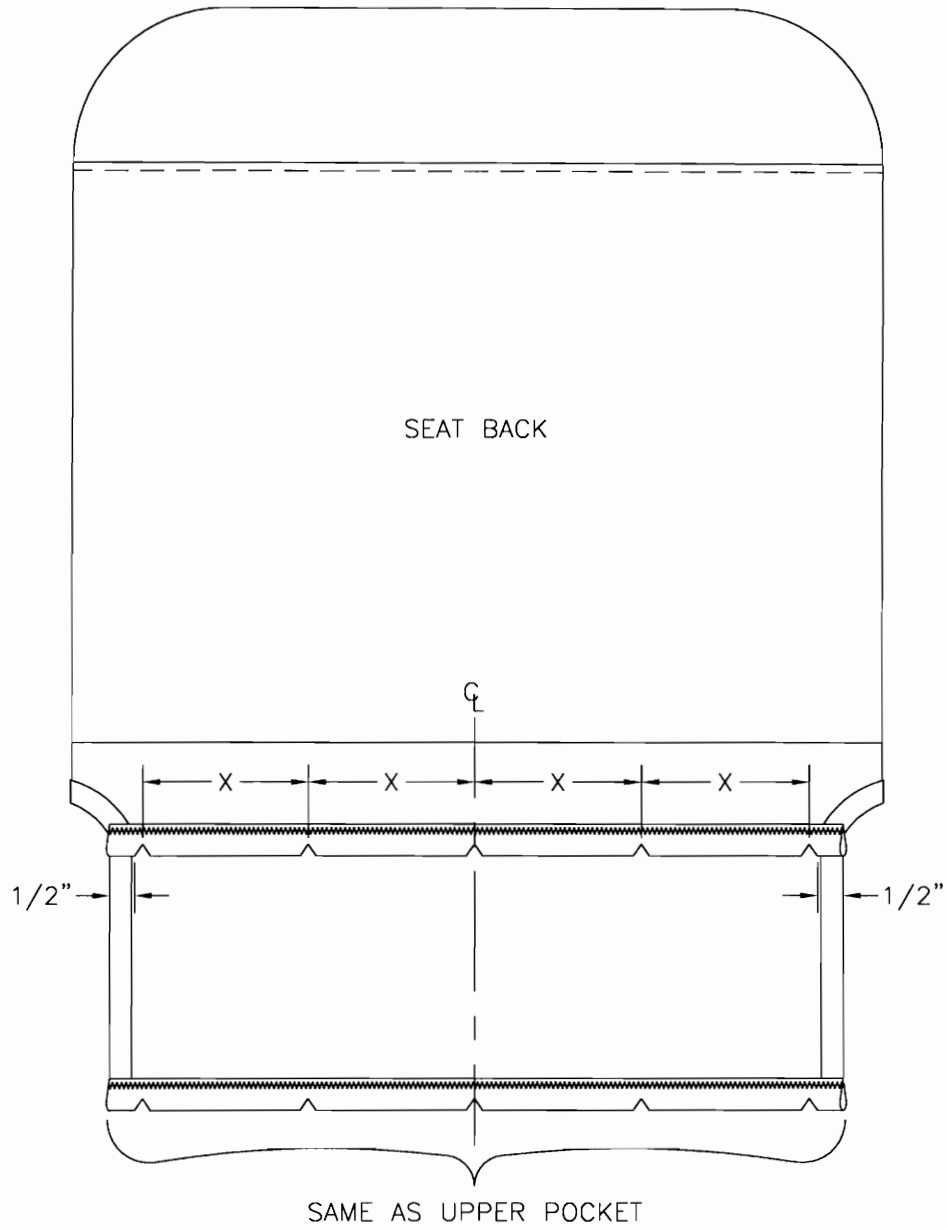


FIGURE 5-11

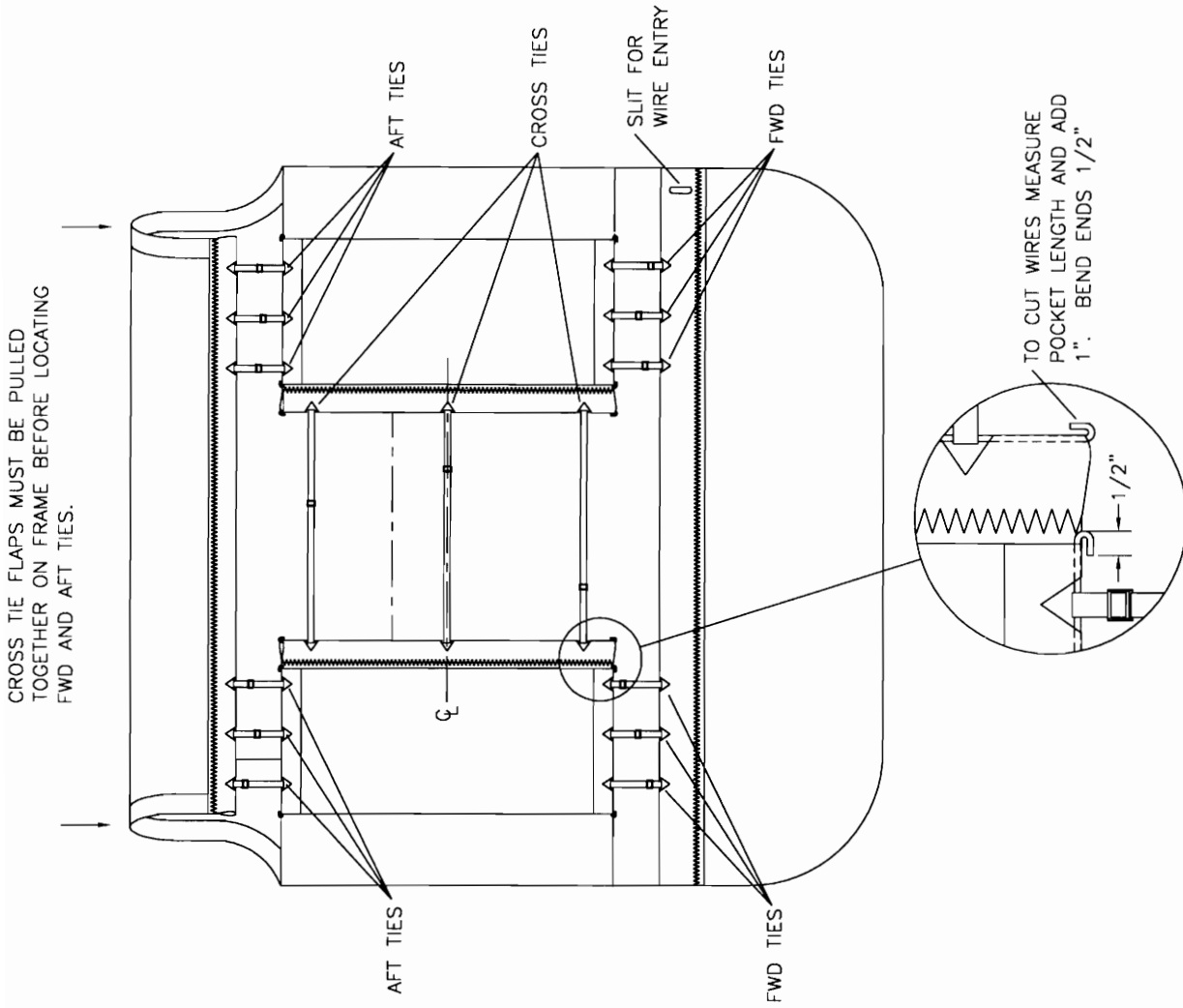


FIGURE 5-12

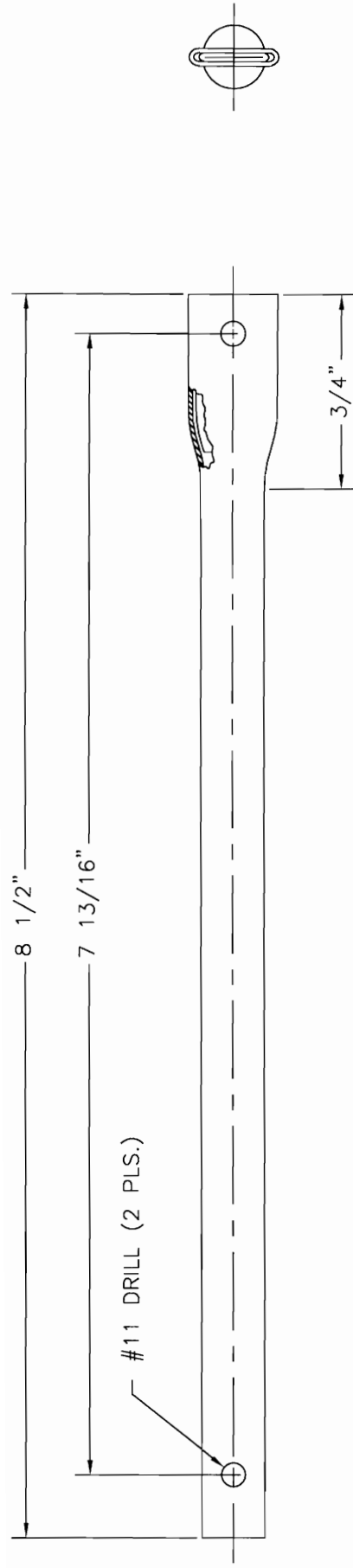


FIGURE 5-13



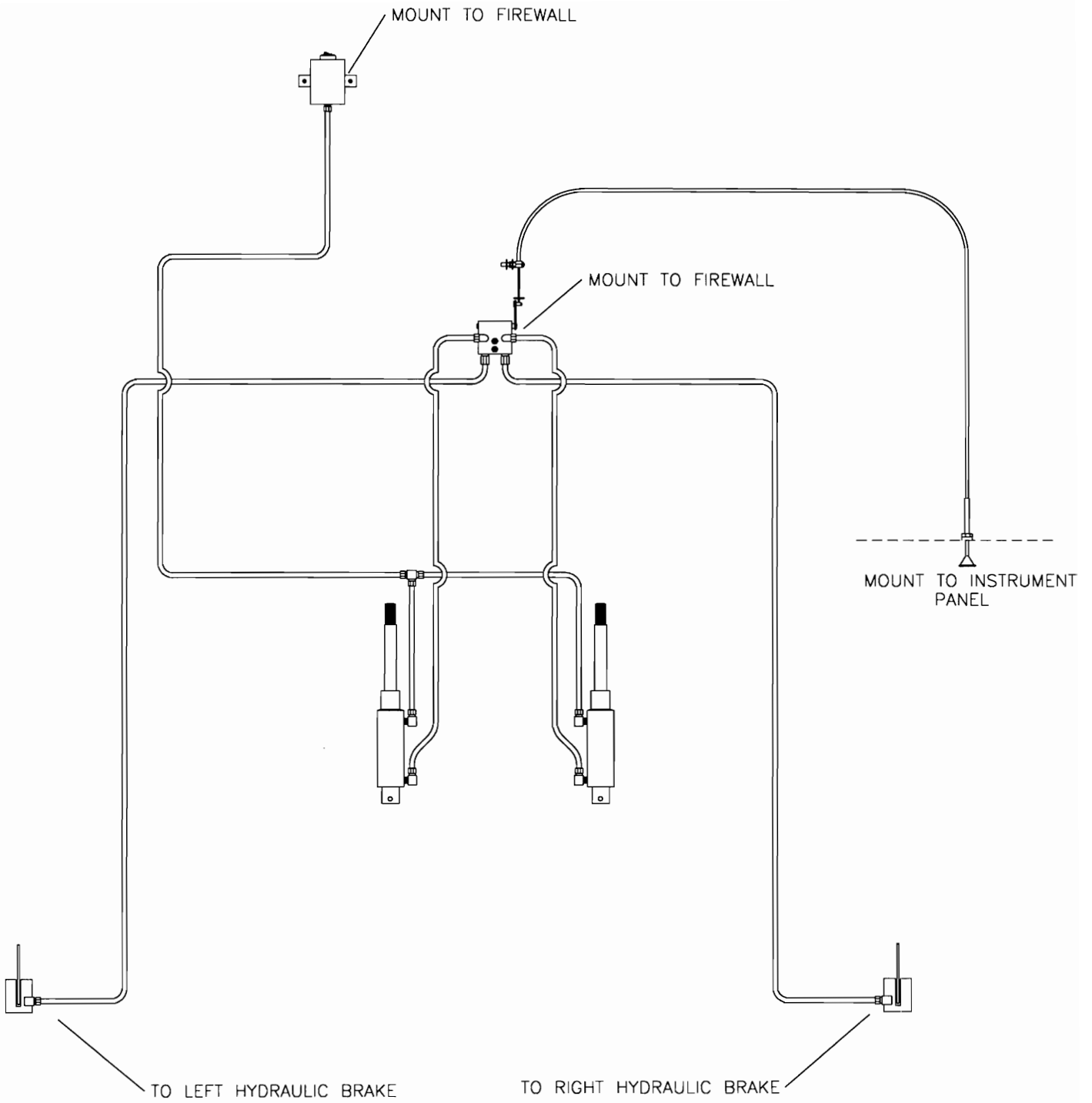


FIGURE 5-14

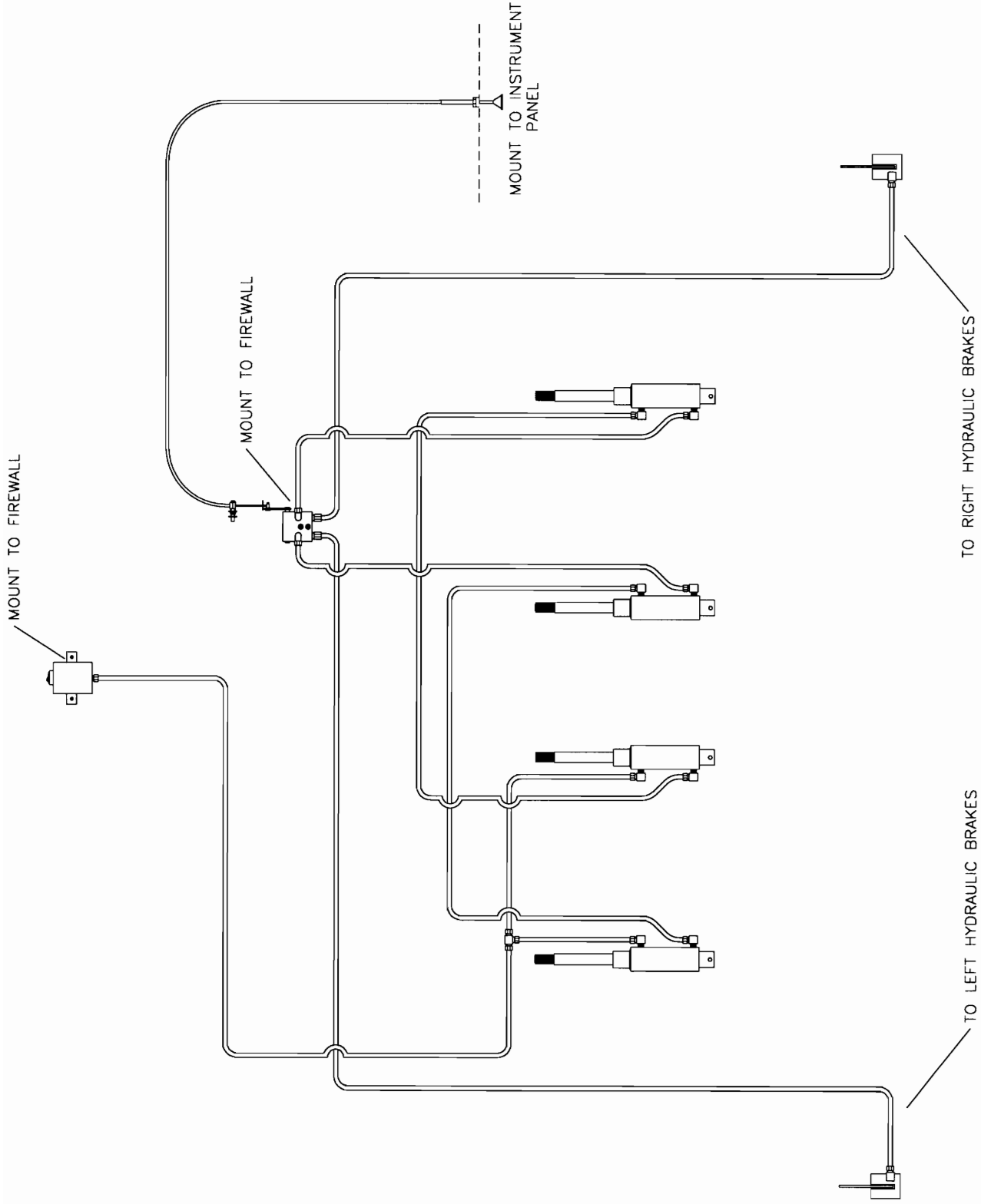


FIGURE 5-15

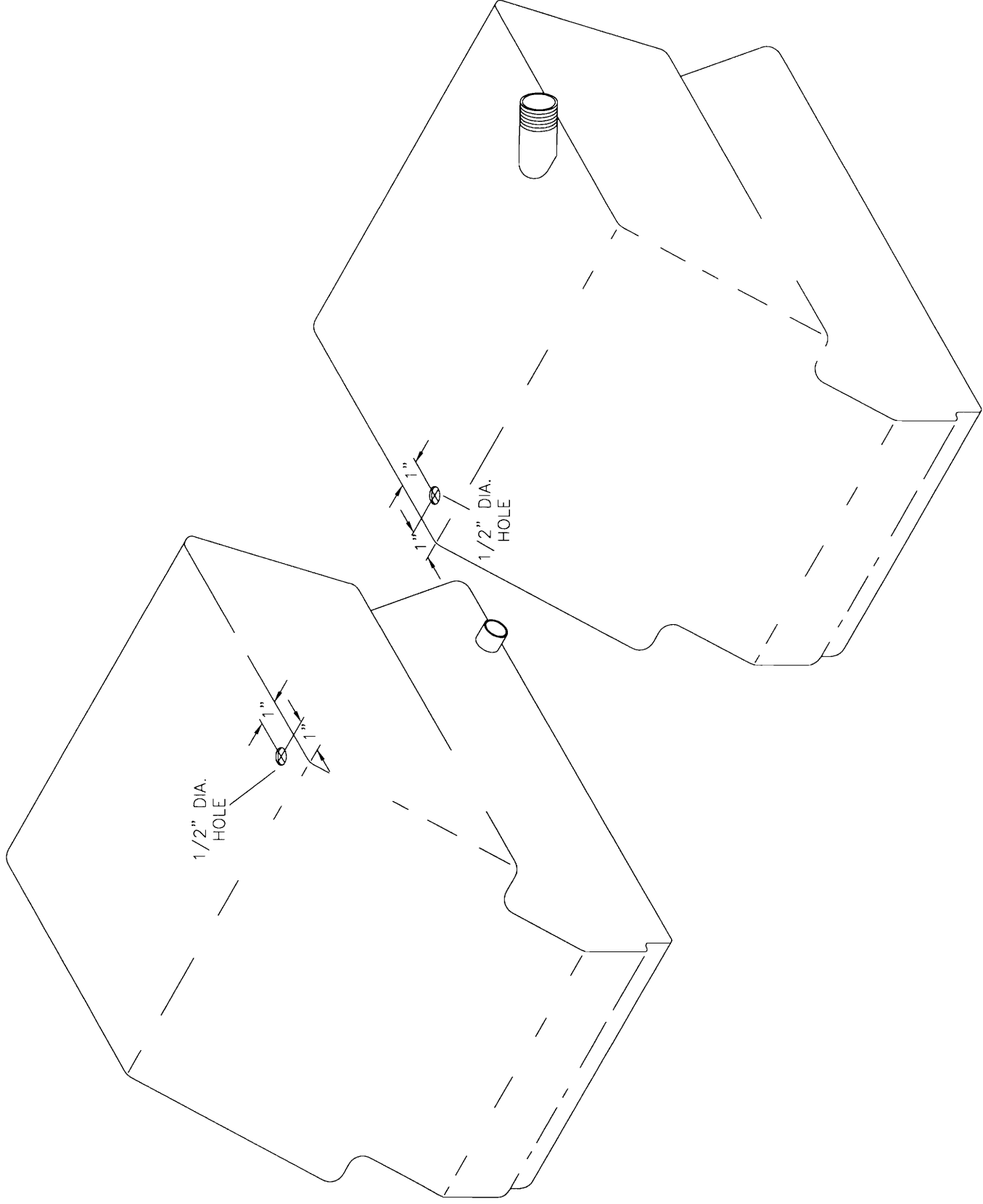
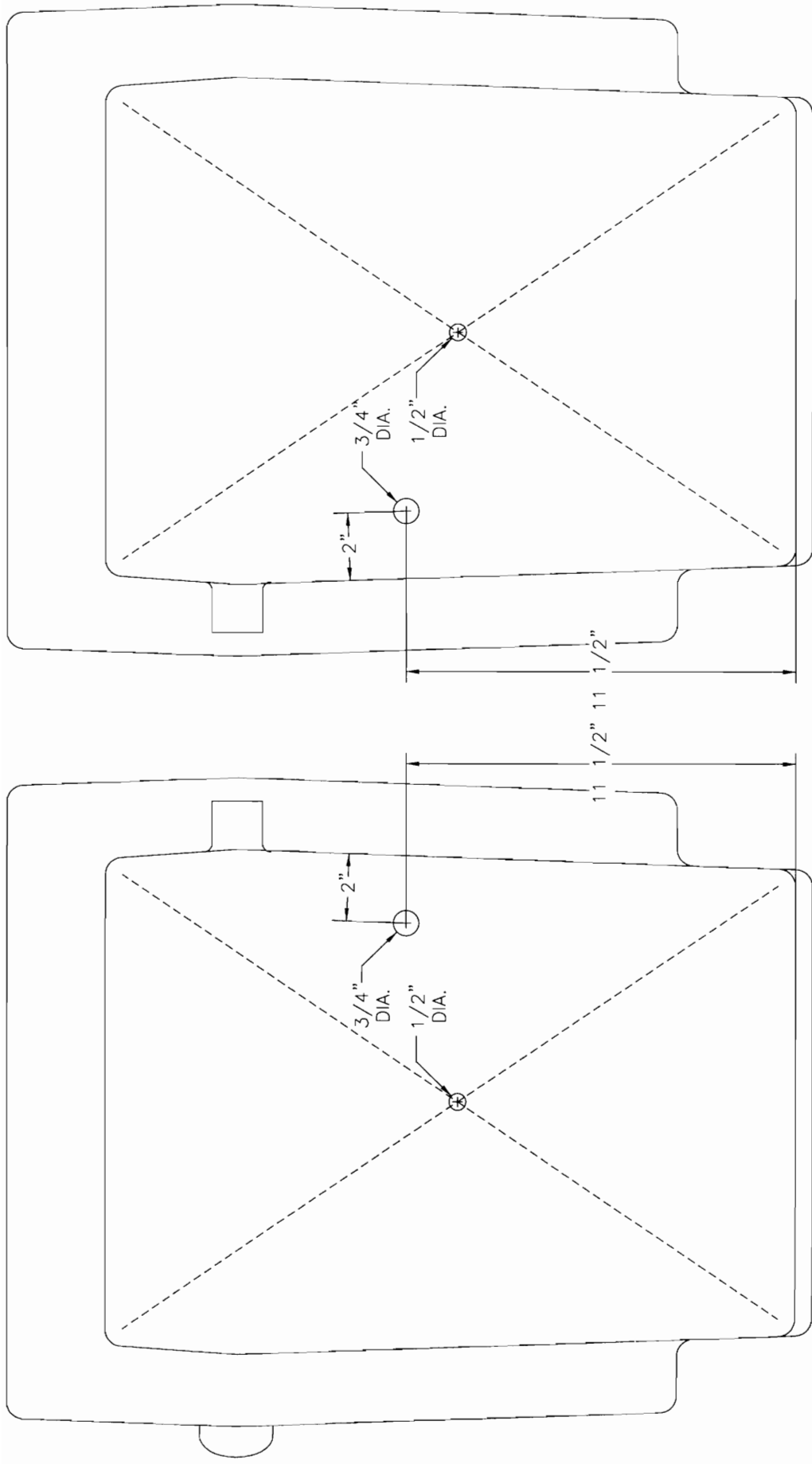
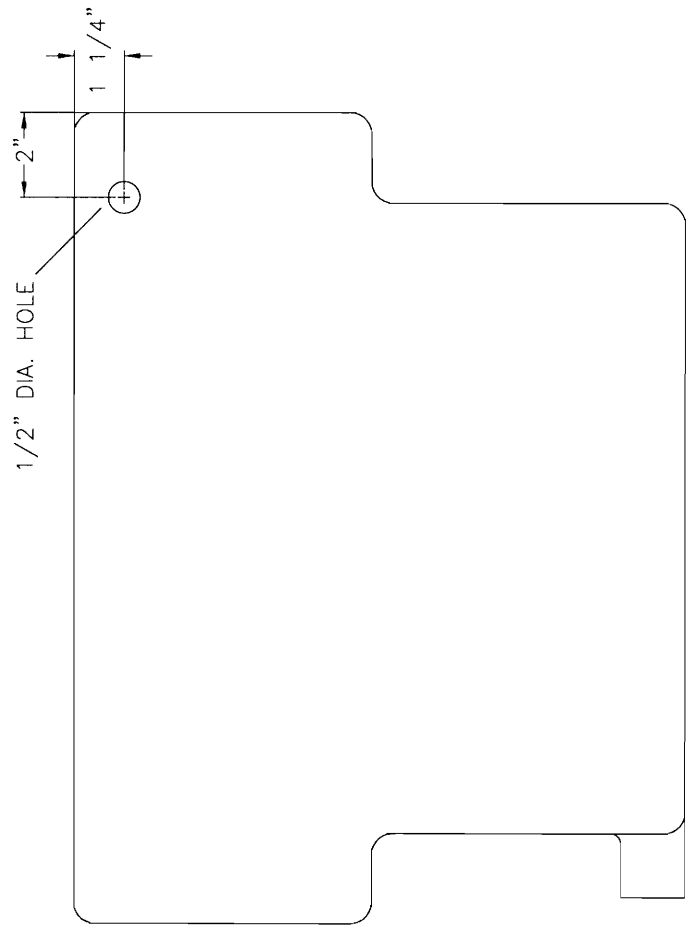


FIGURE 5-16



FUEL TANKS VIEWED FROM BOTTOM SIDE

FIGURE 5-17



RIGHT FUEL TANK ONLY  
VIEWED FROM AFT SIDE

FIGURE 5-18

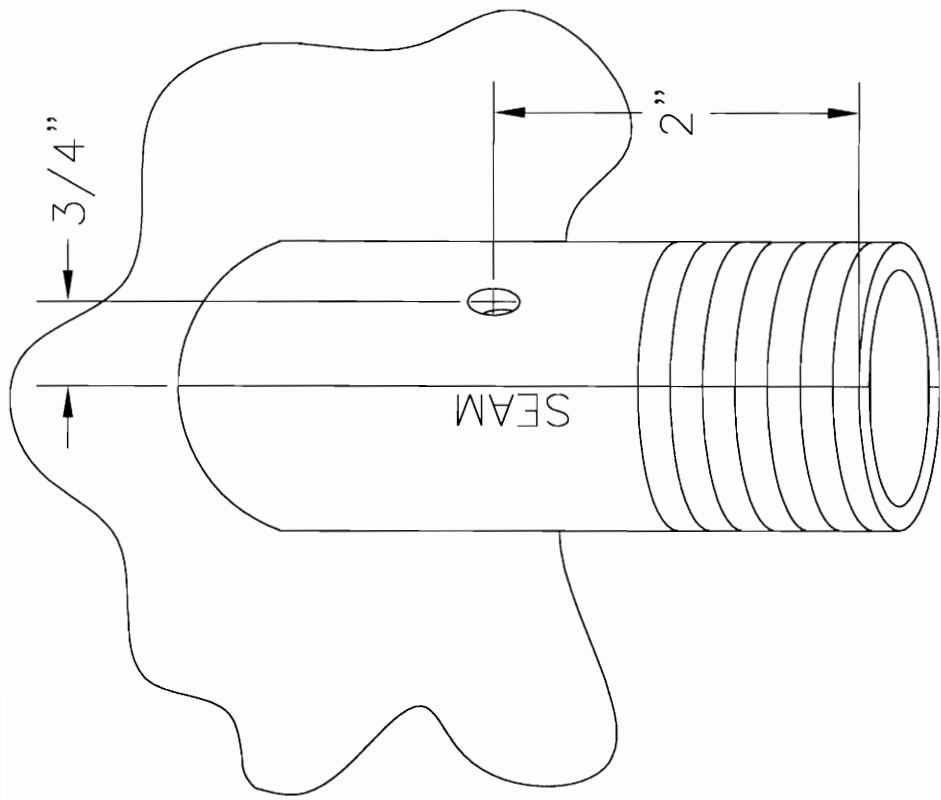


FIGURE 5-19

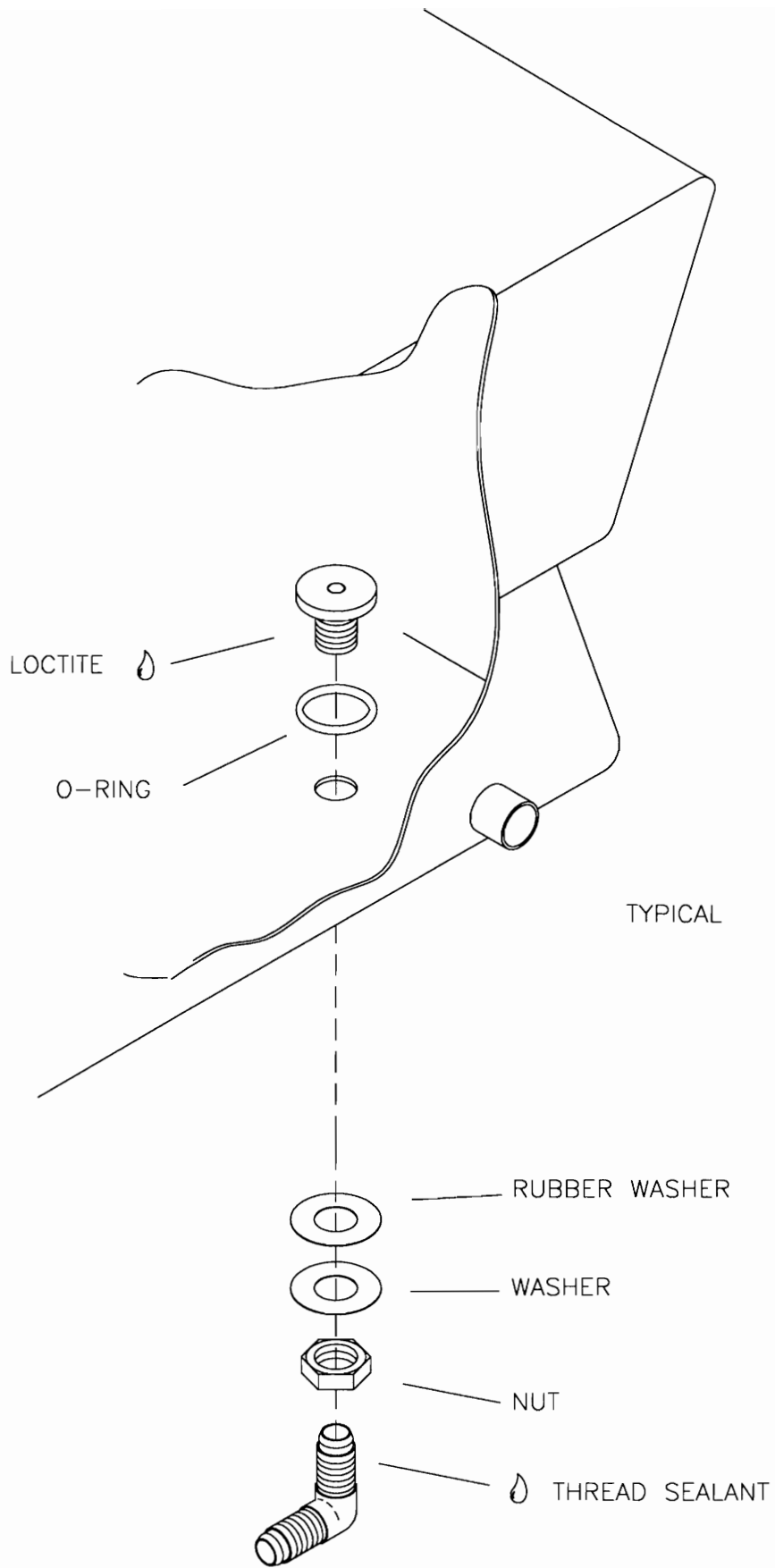


FIGURE 5-20

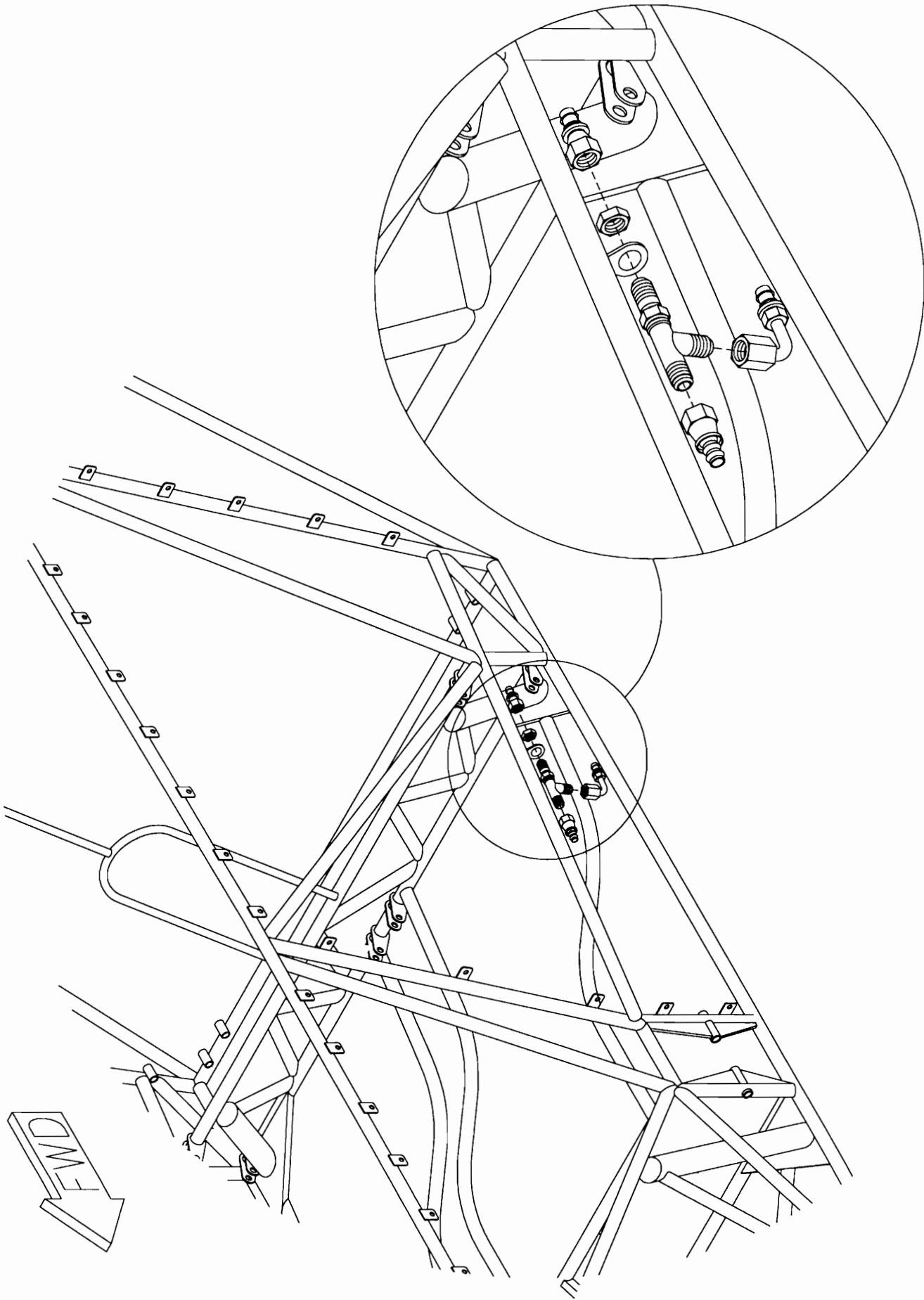


FIGURE 5-21



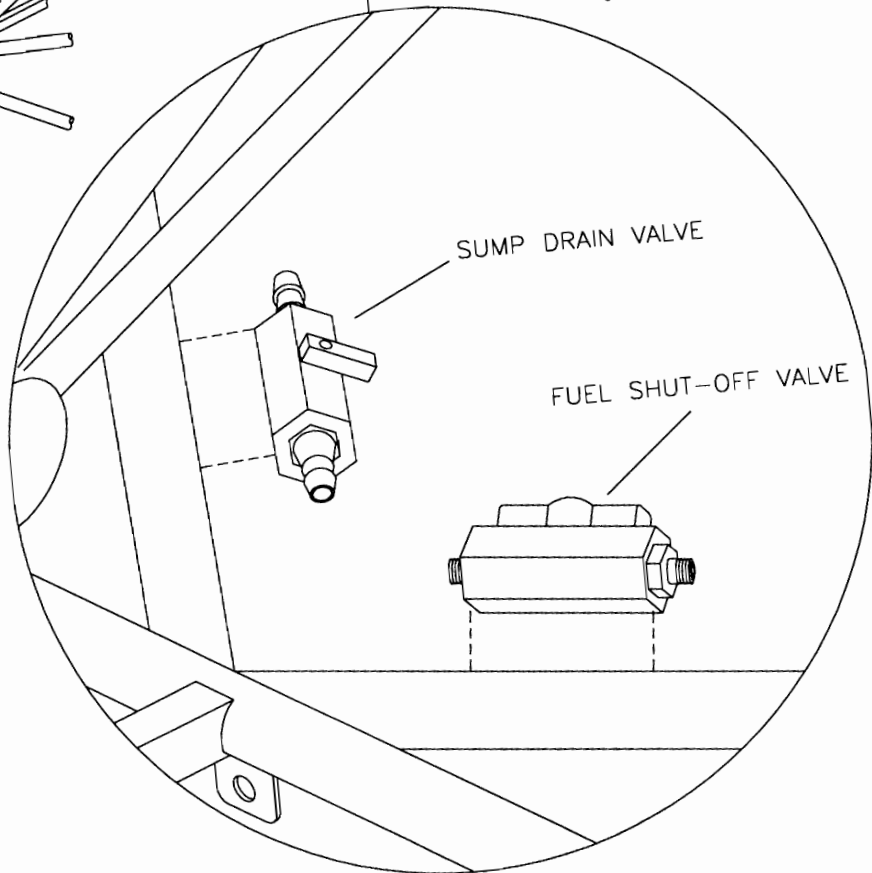
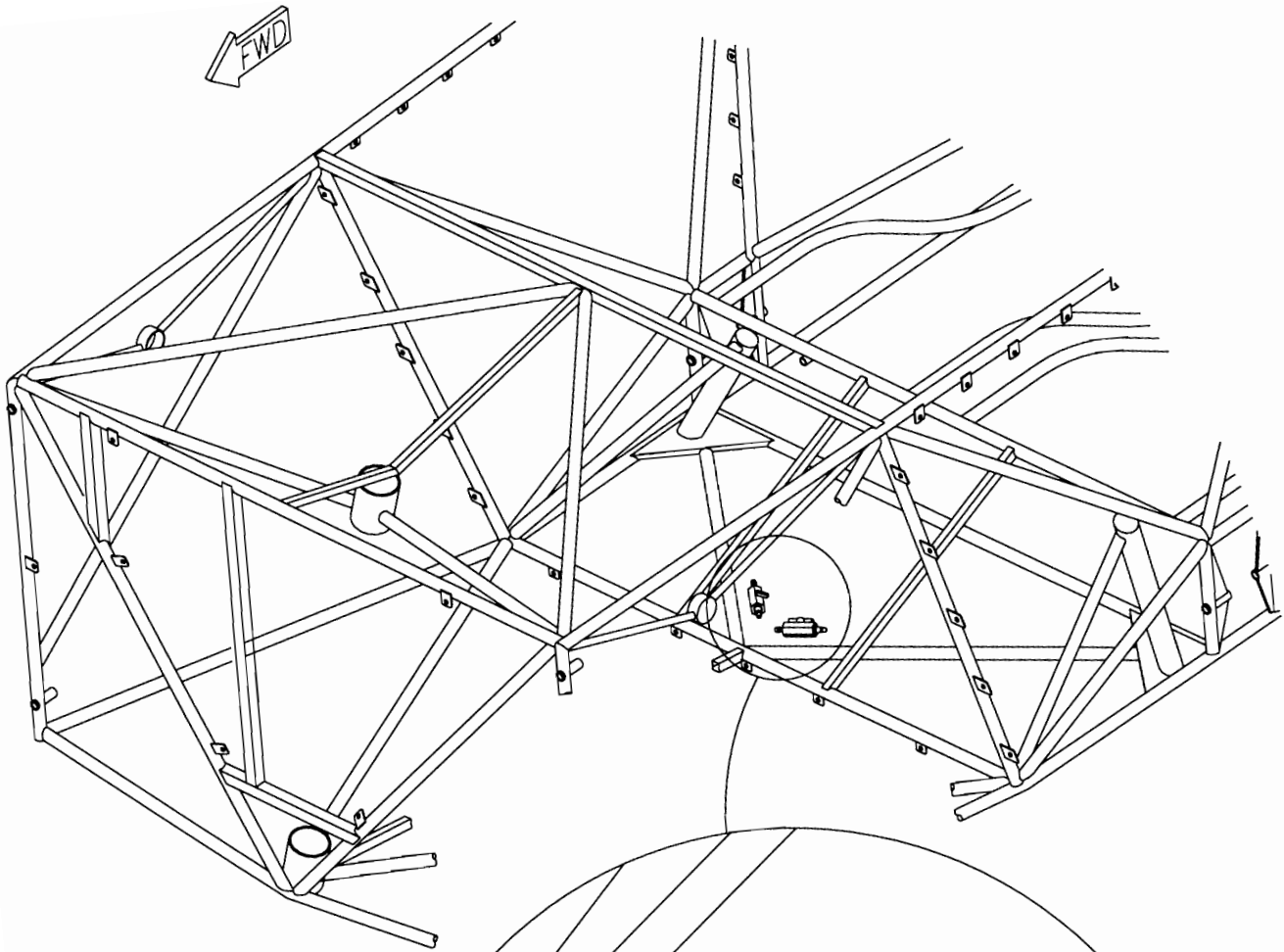


FIGURE 5-22

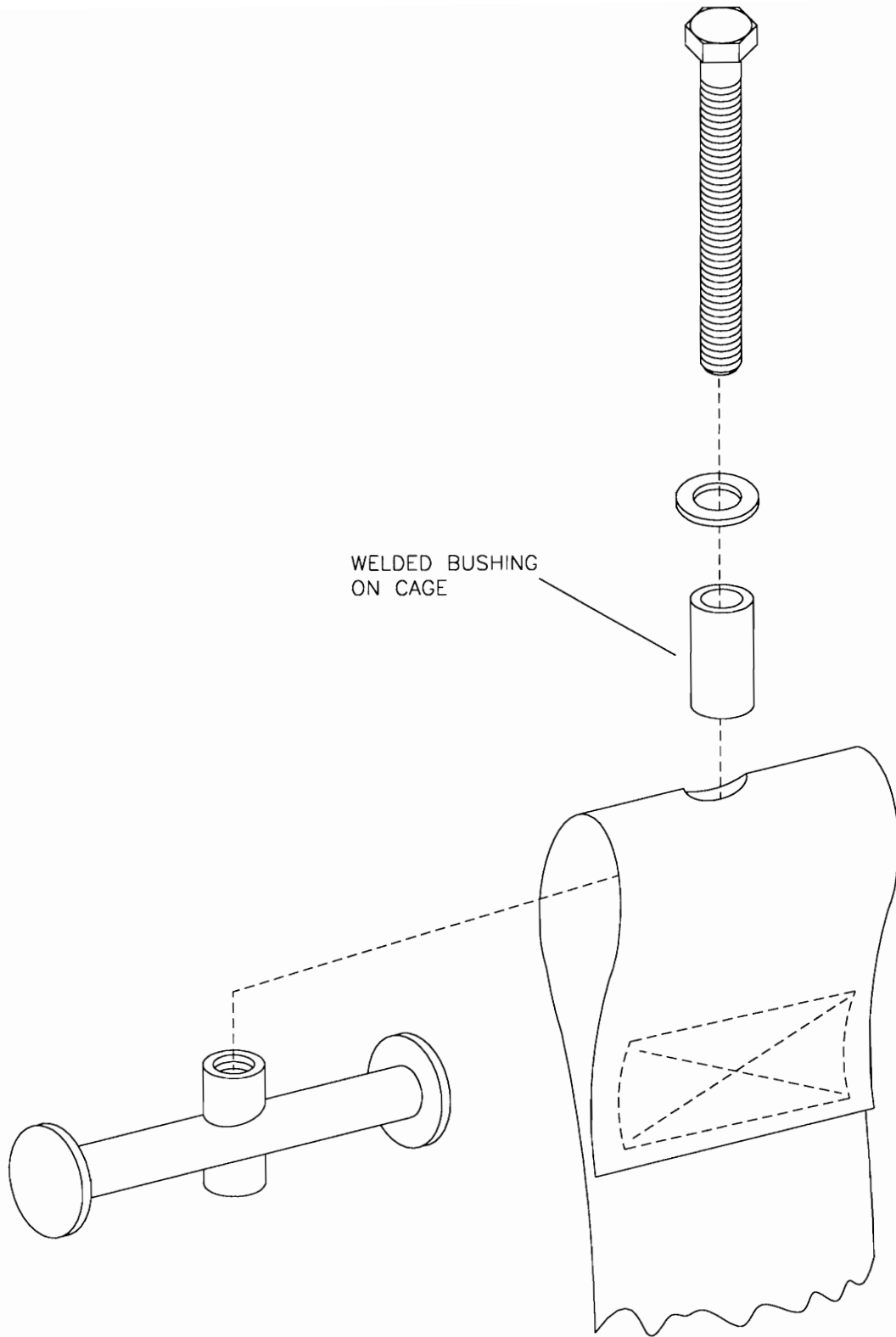


FIGURE 5-23

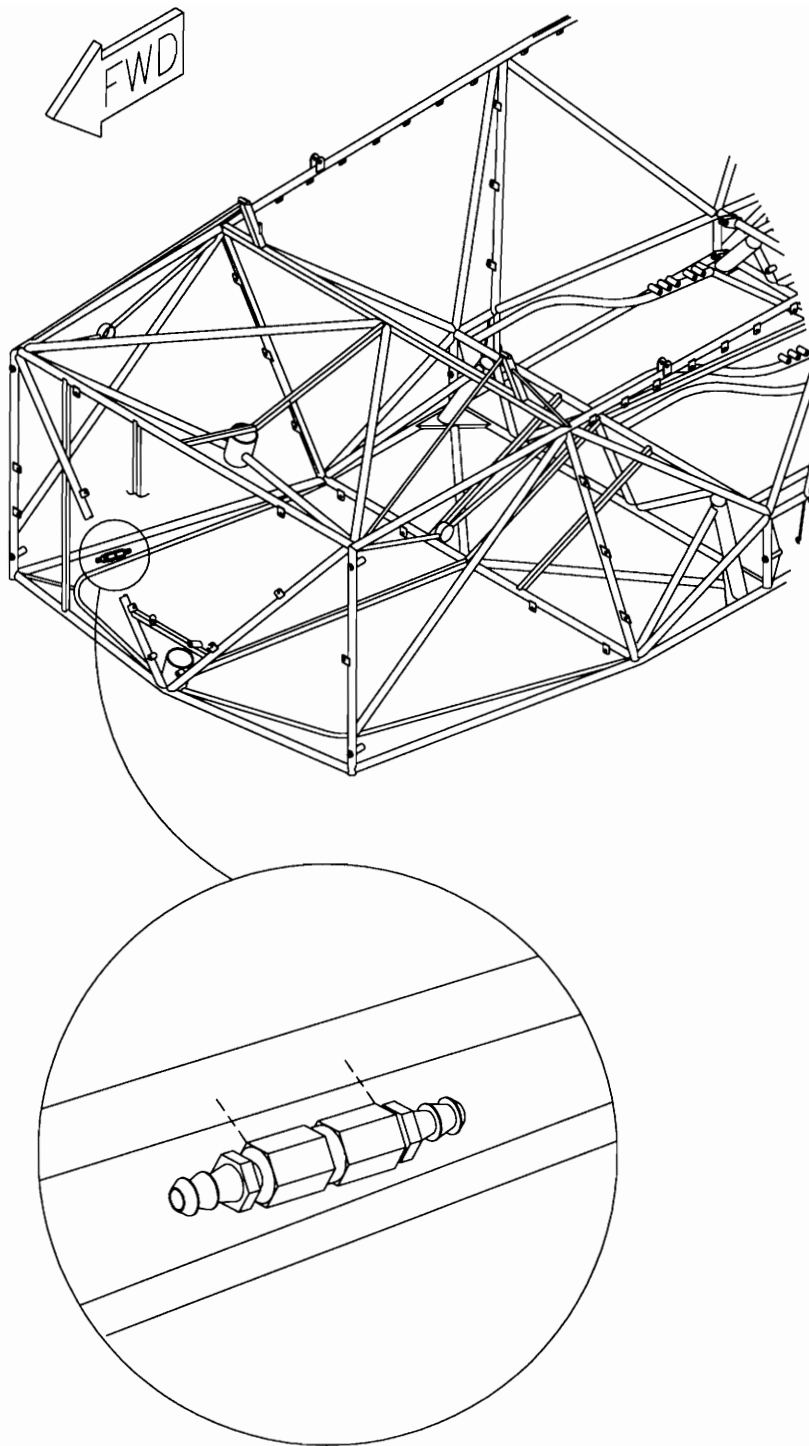


FIGURE 5-24

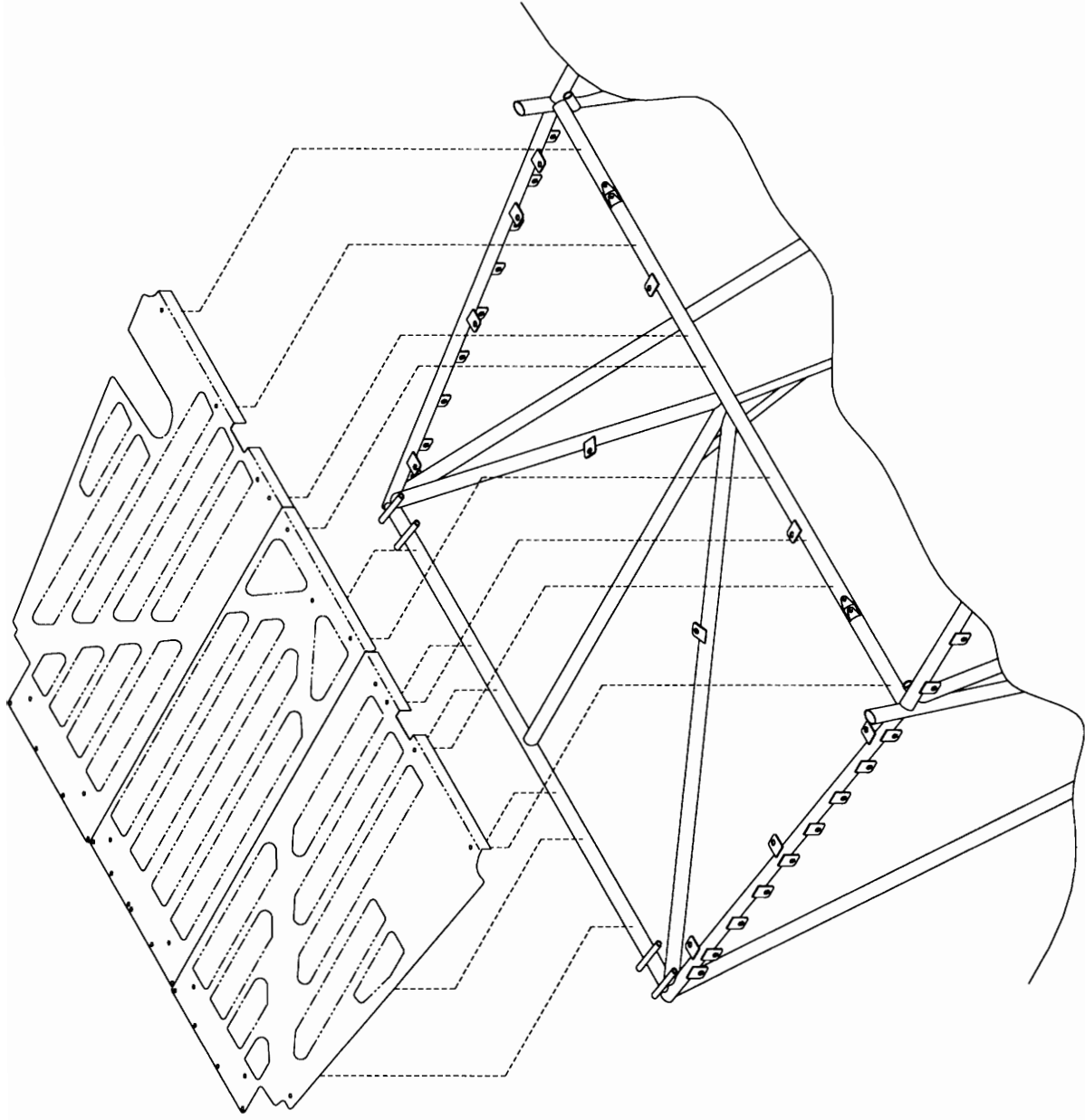


FIGURE 5-25

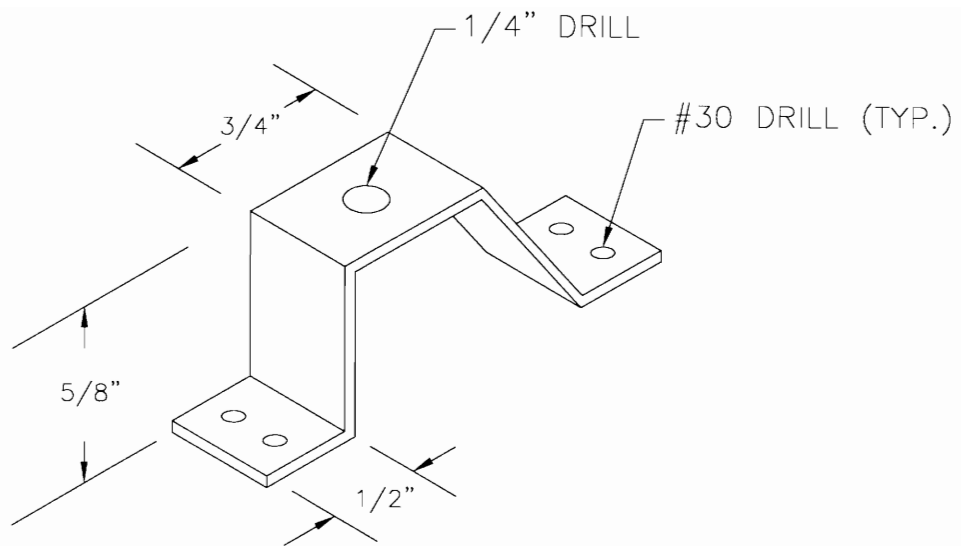


FIGURE 5-26

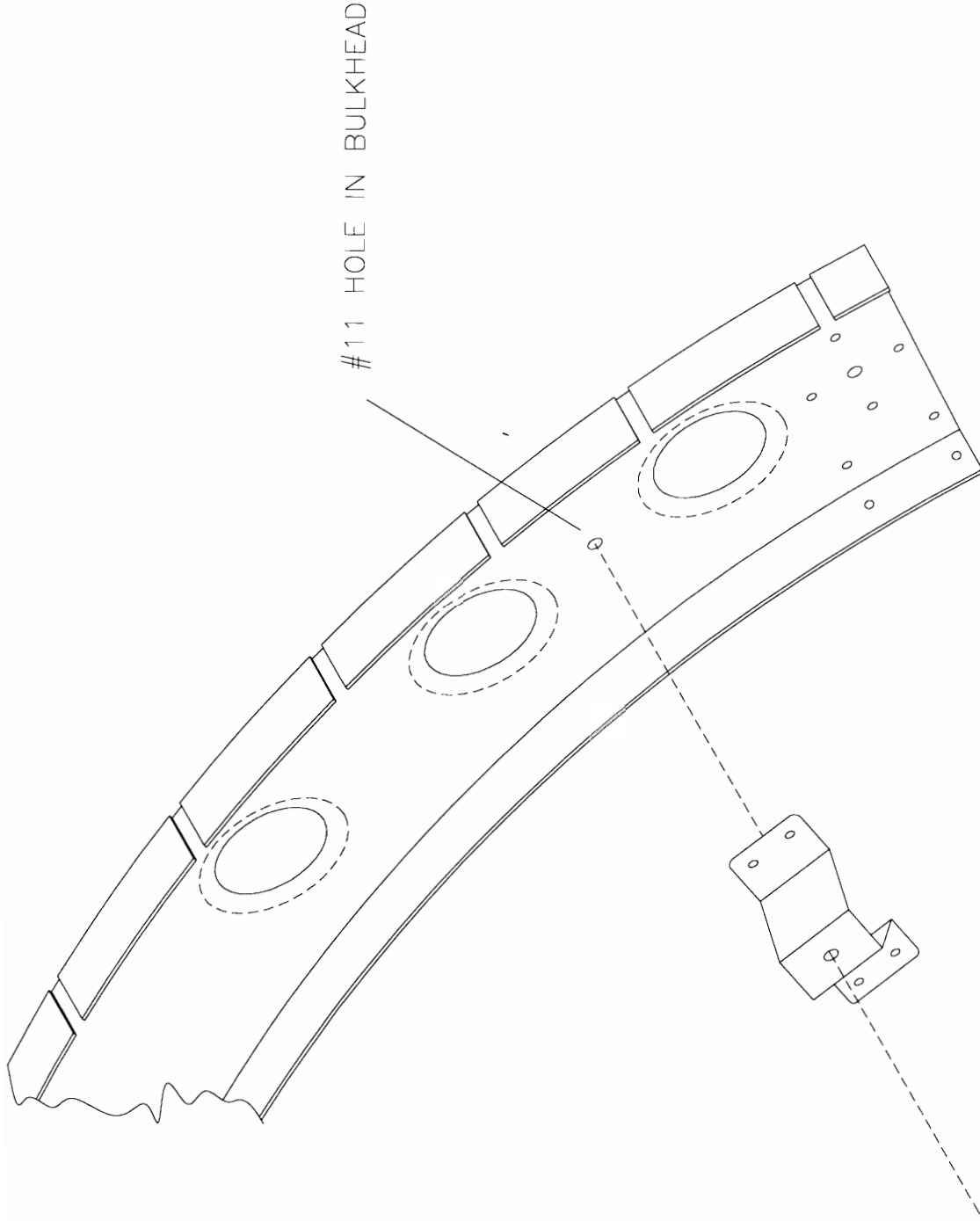


FIGURE 5-27

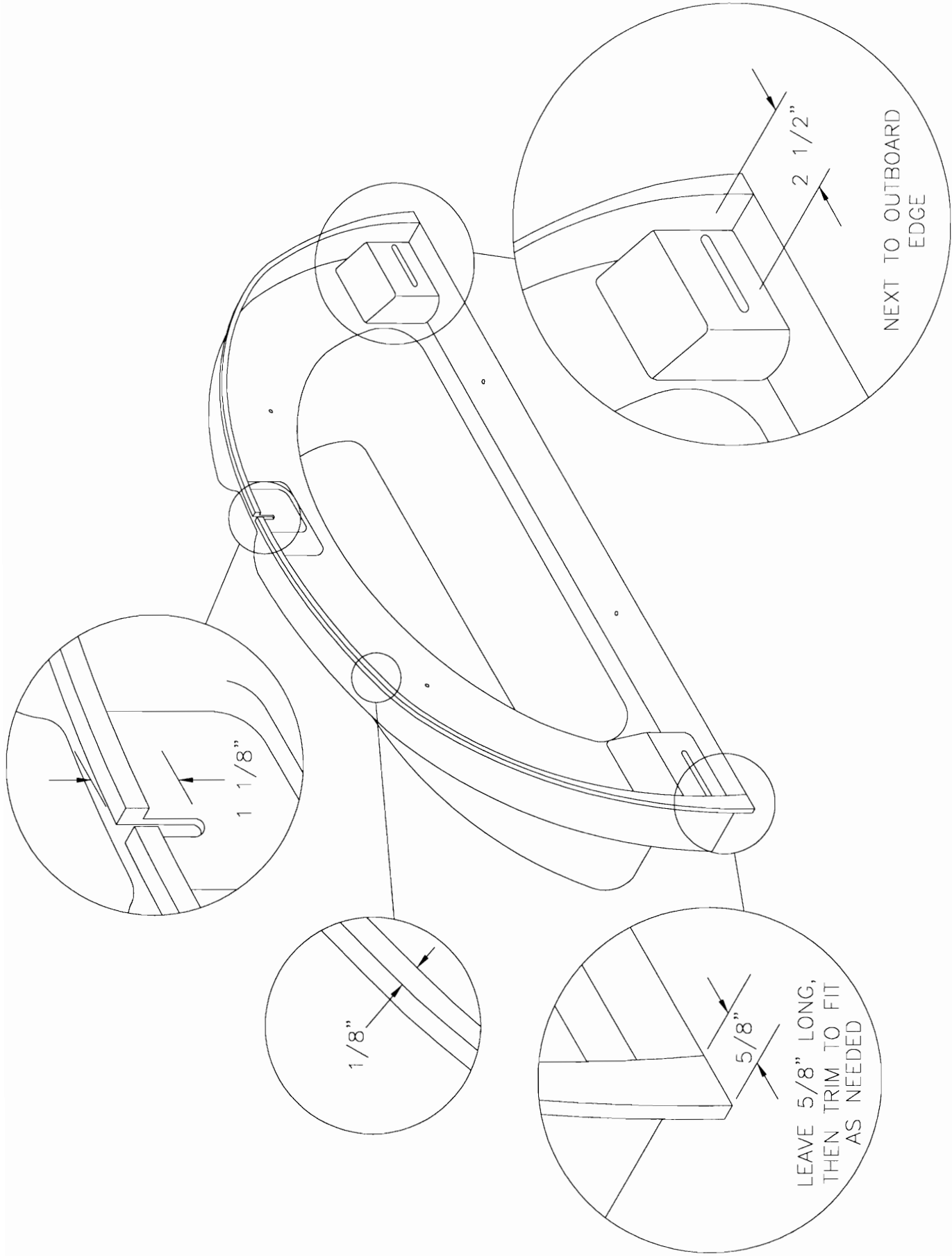


FIGURE 5-28

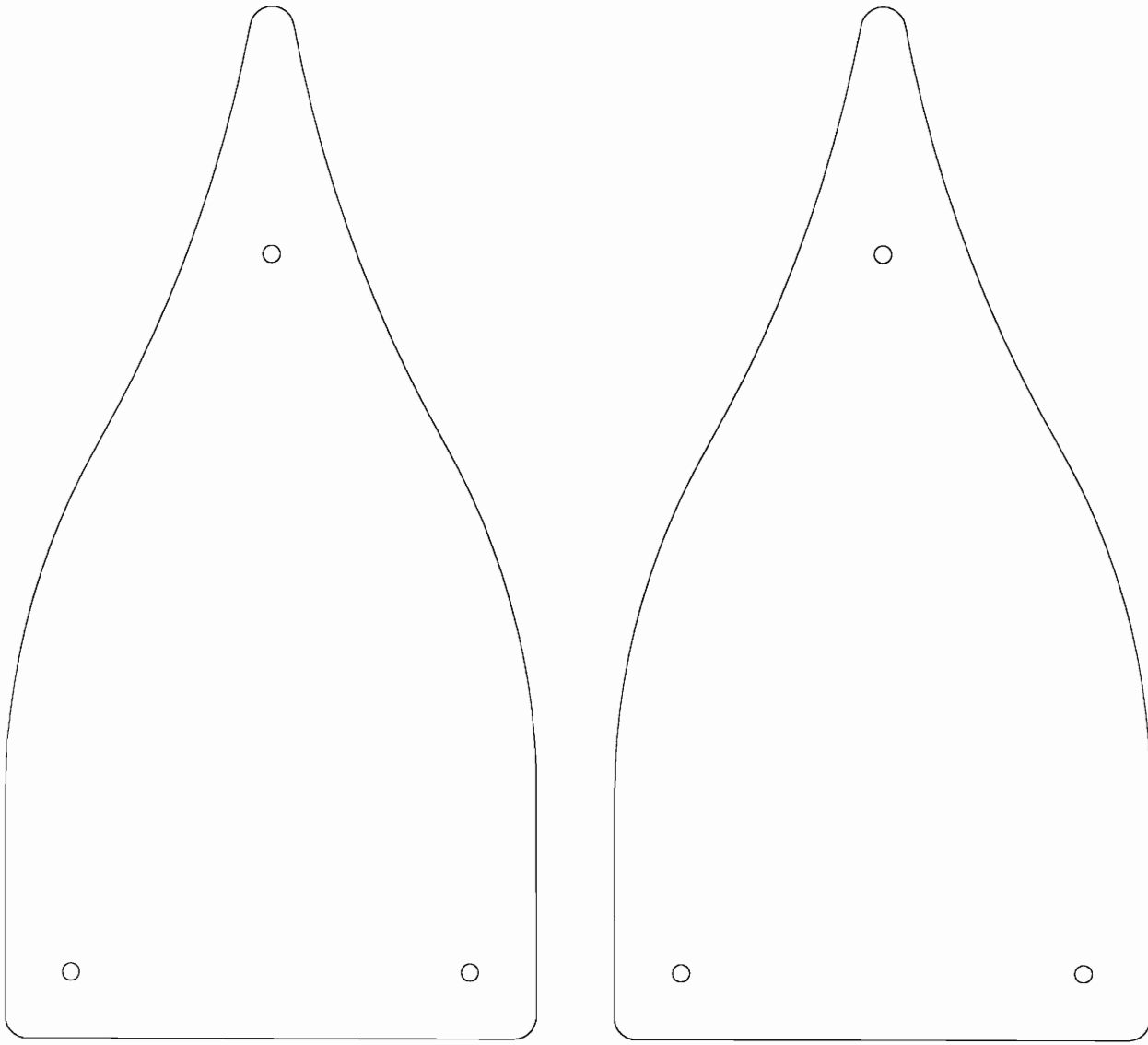


FIGURE 5-29



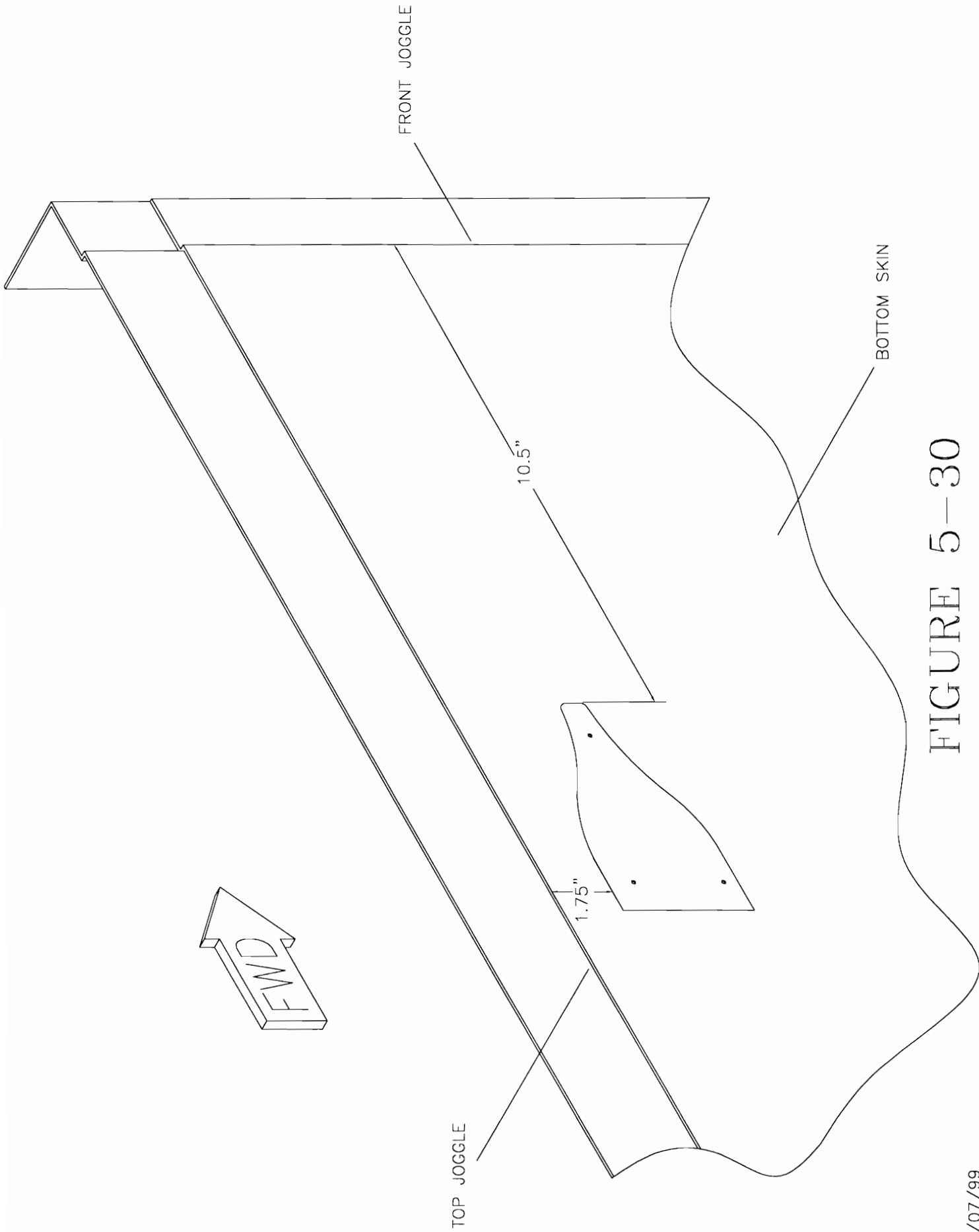


FIGURE 5-30

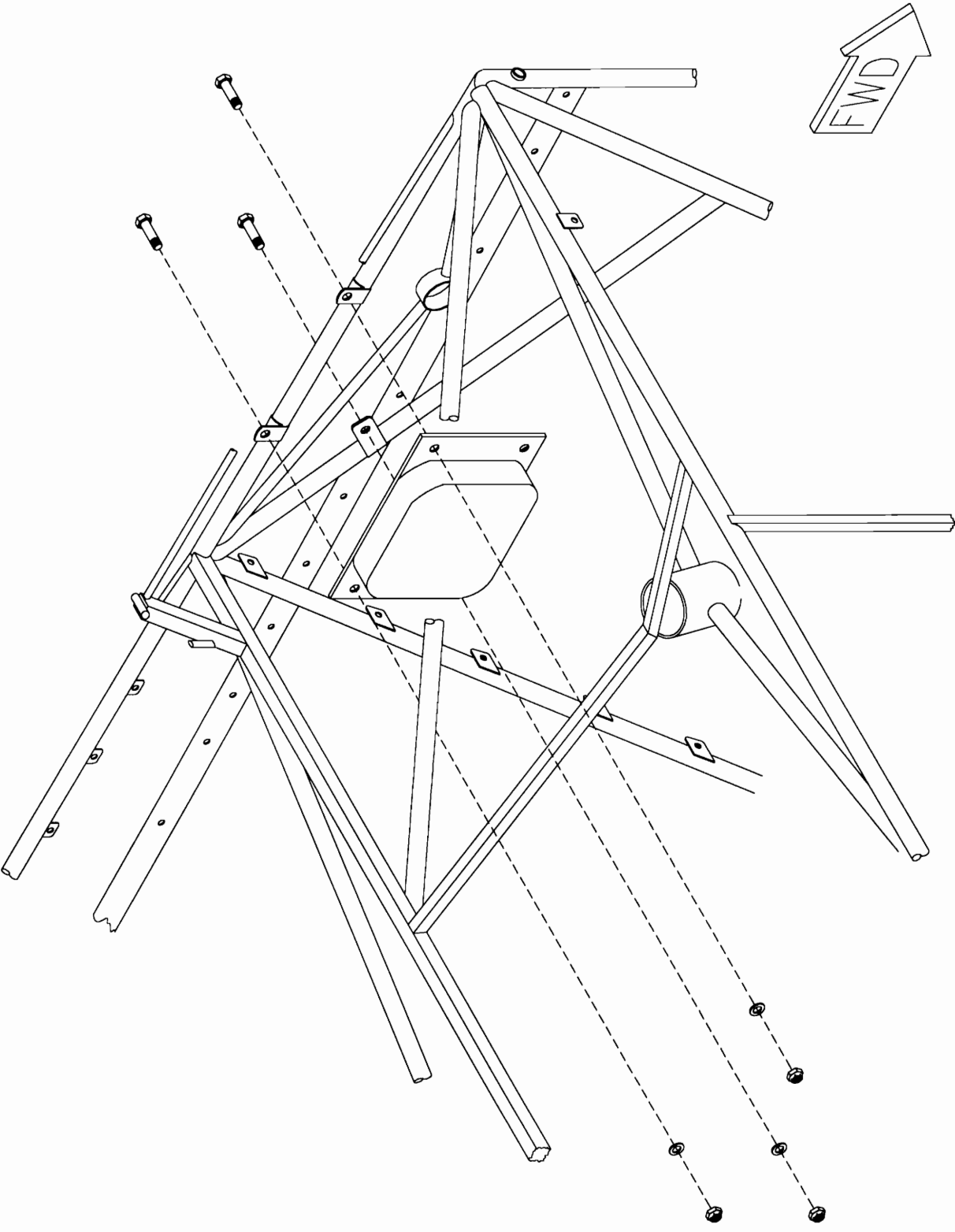


FIGURE 5-31

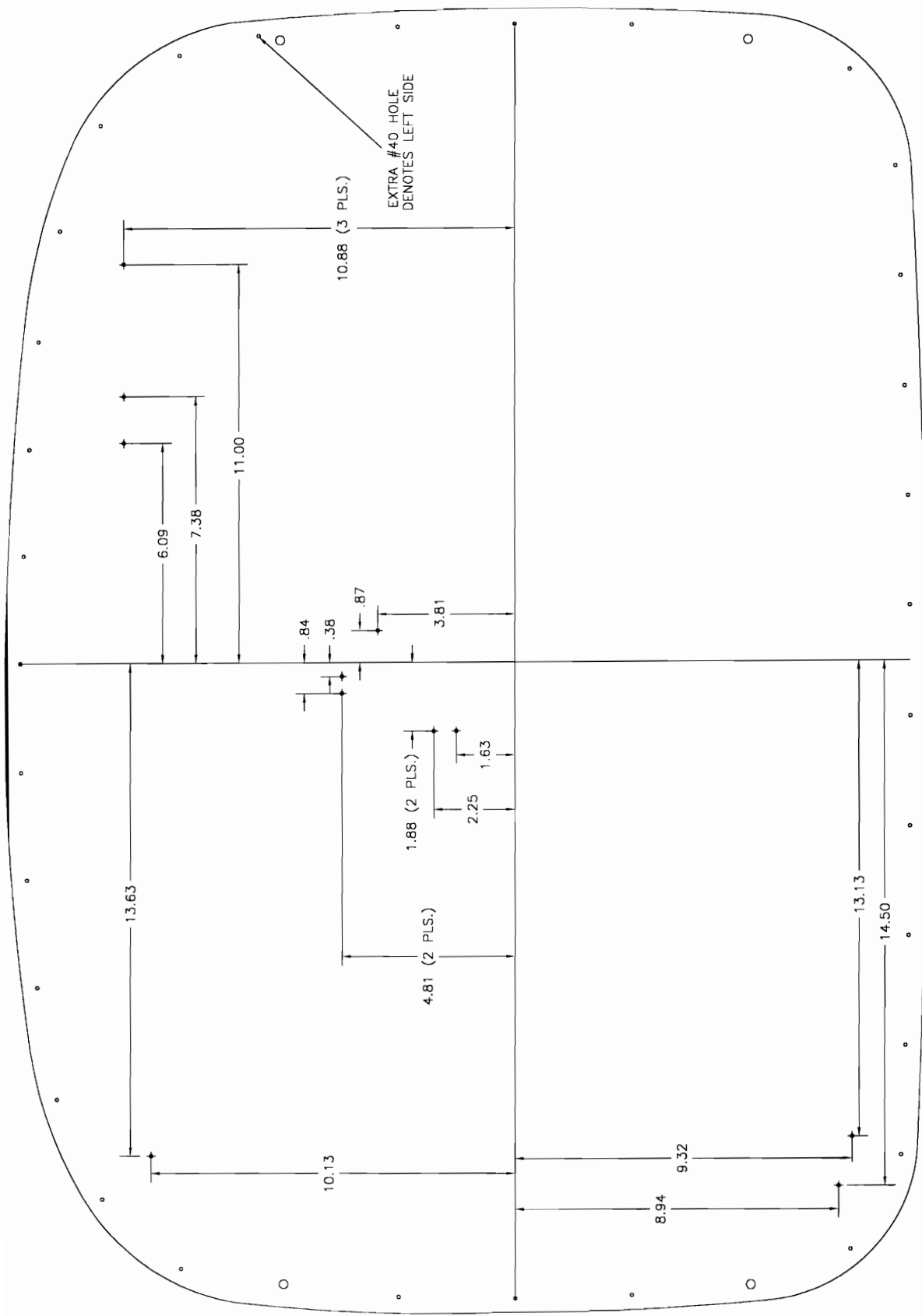


FIGURE 6-1

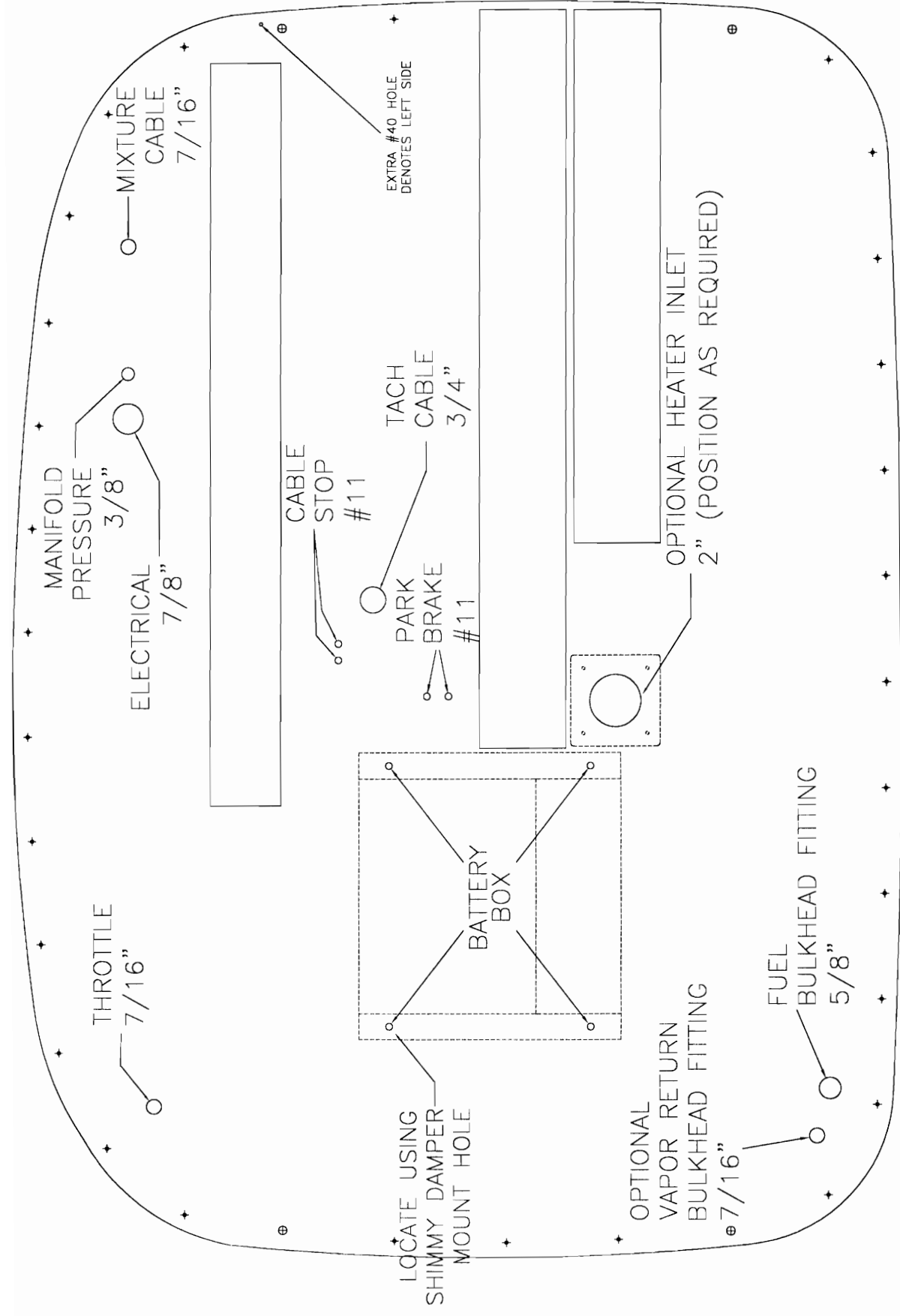


FIGURE 6-2

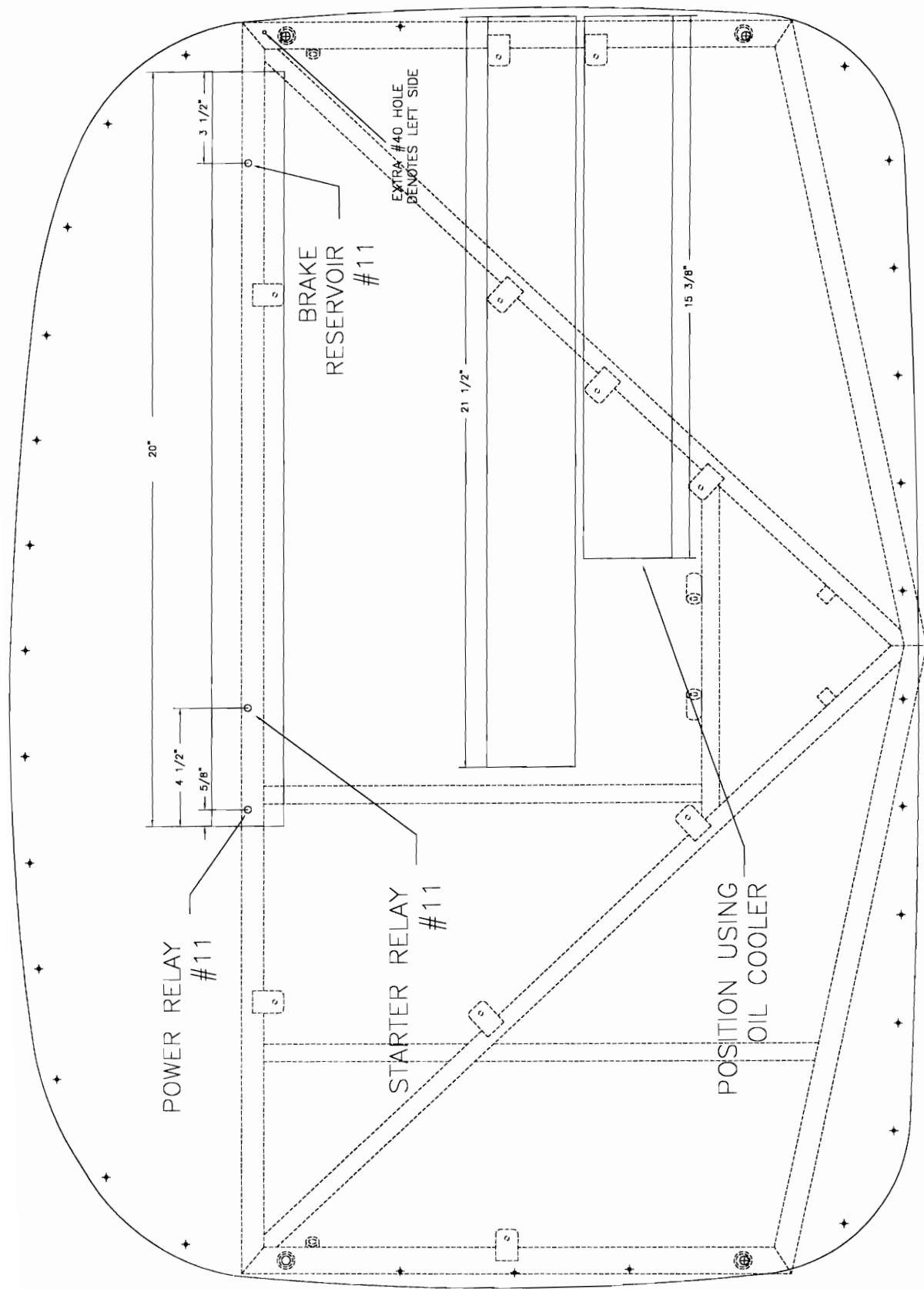


FIGURE 6-3

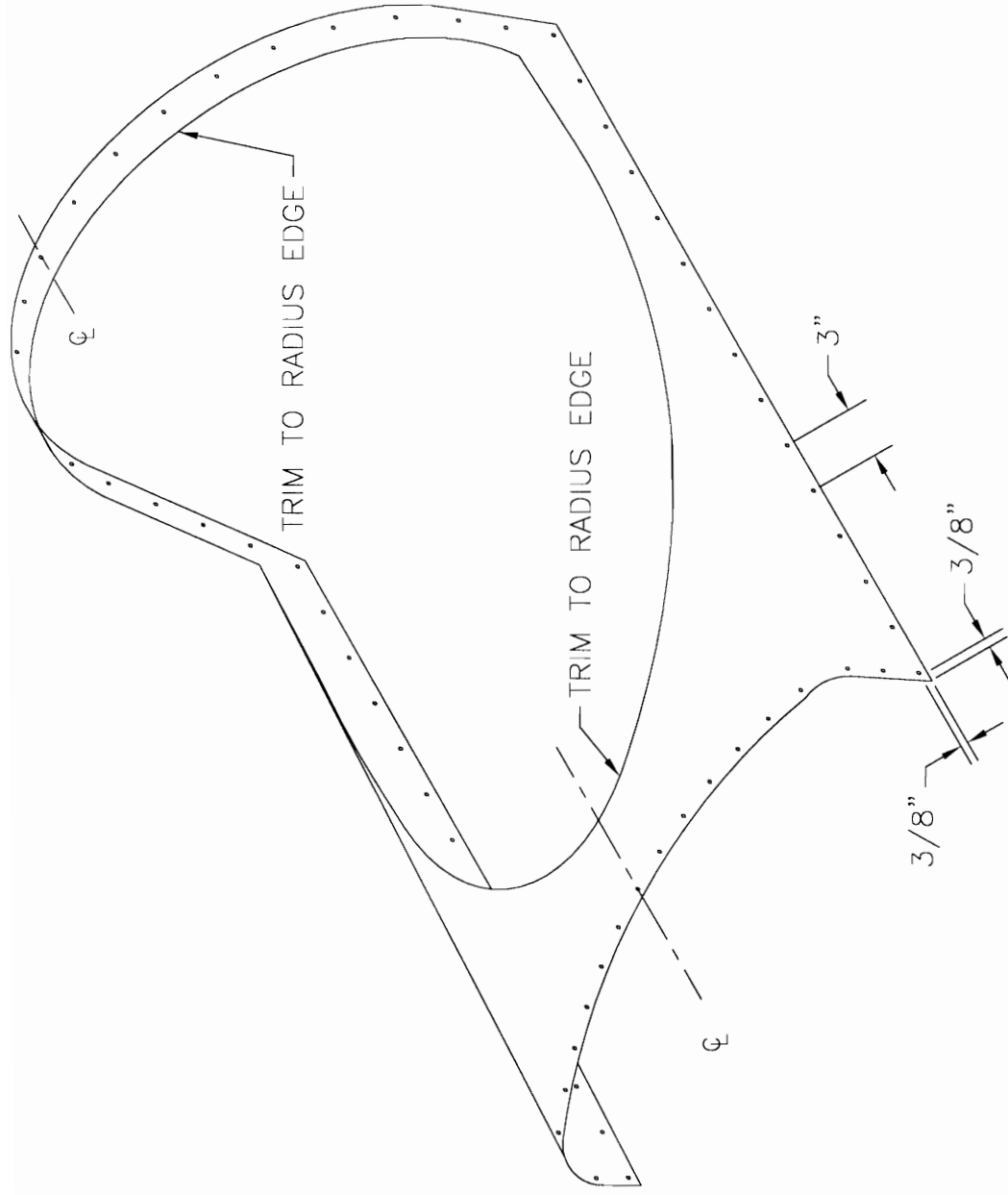


FIGURE 7--1

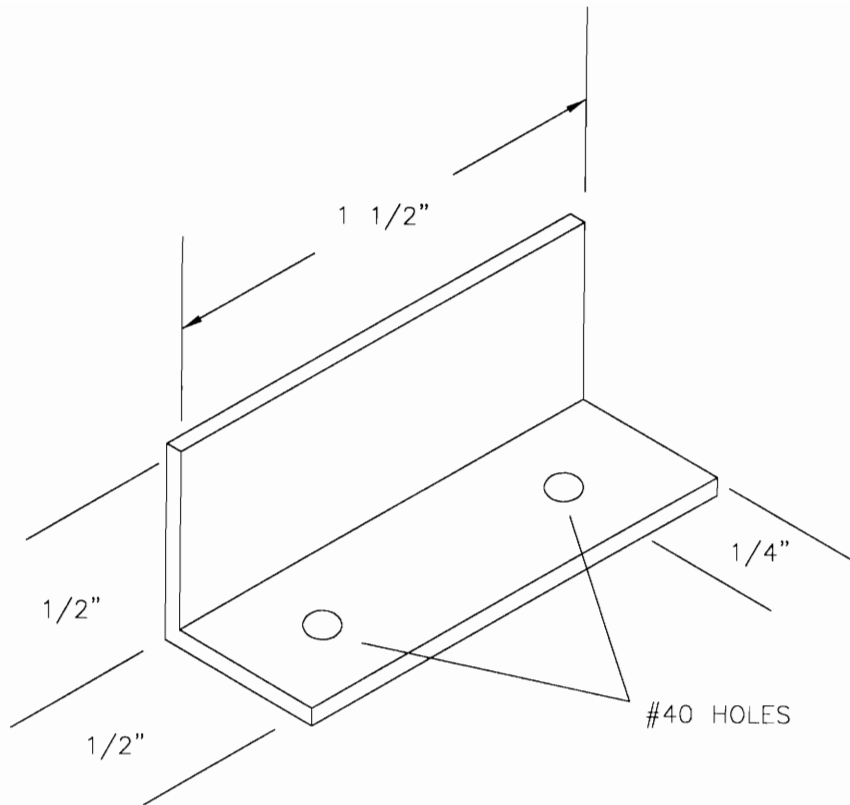


FIGURE 7-2

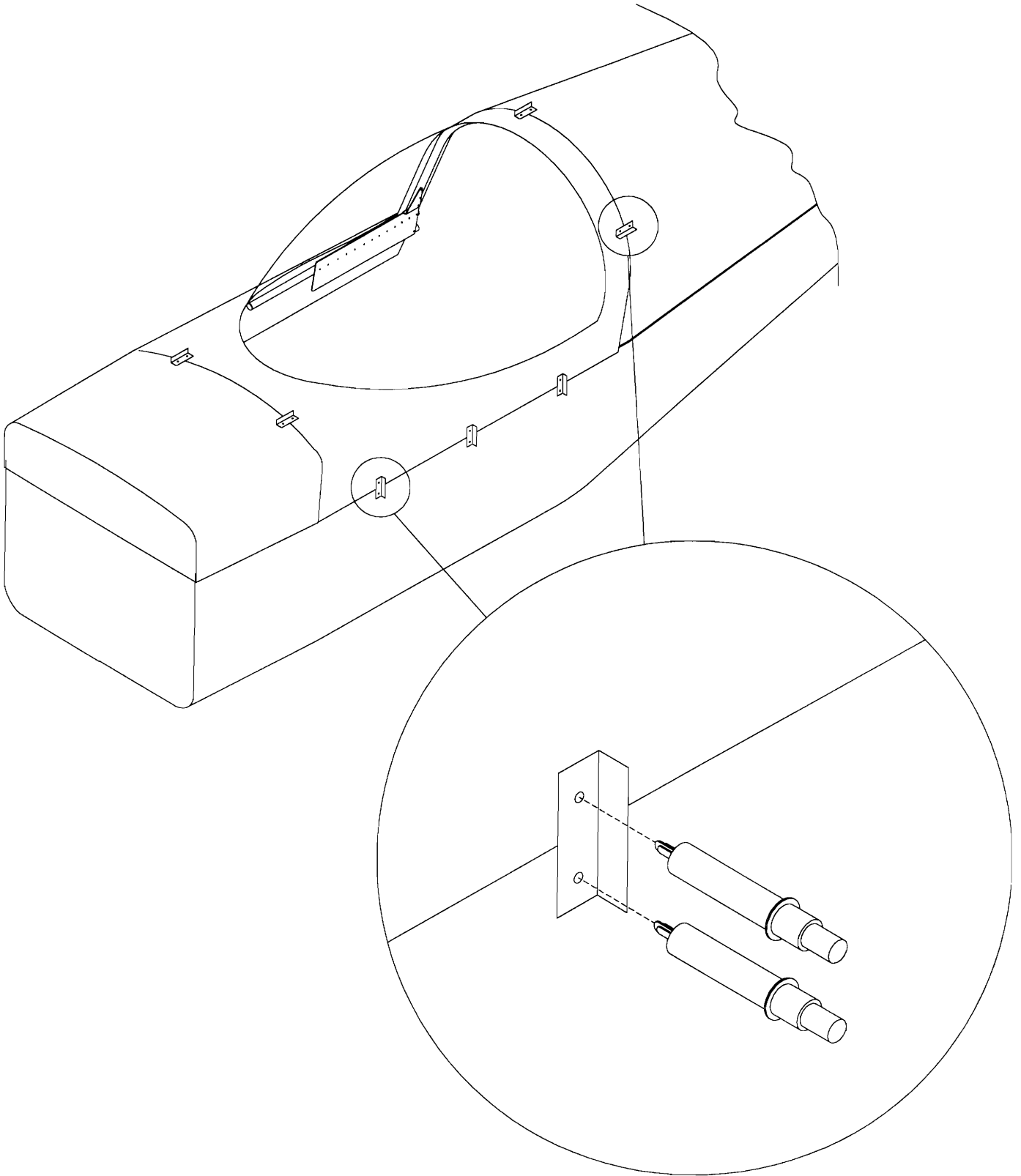


FIGURE 7-3



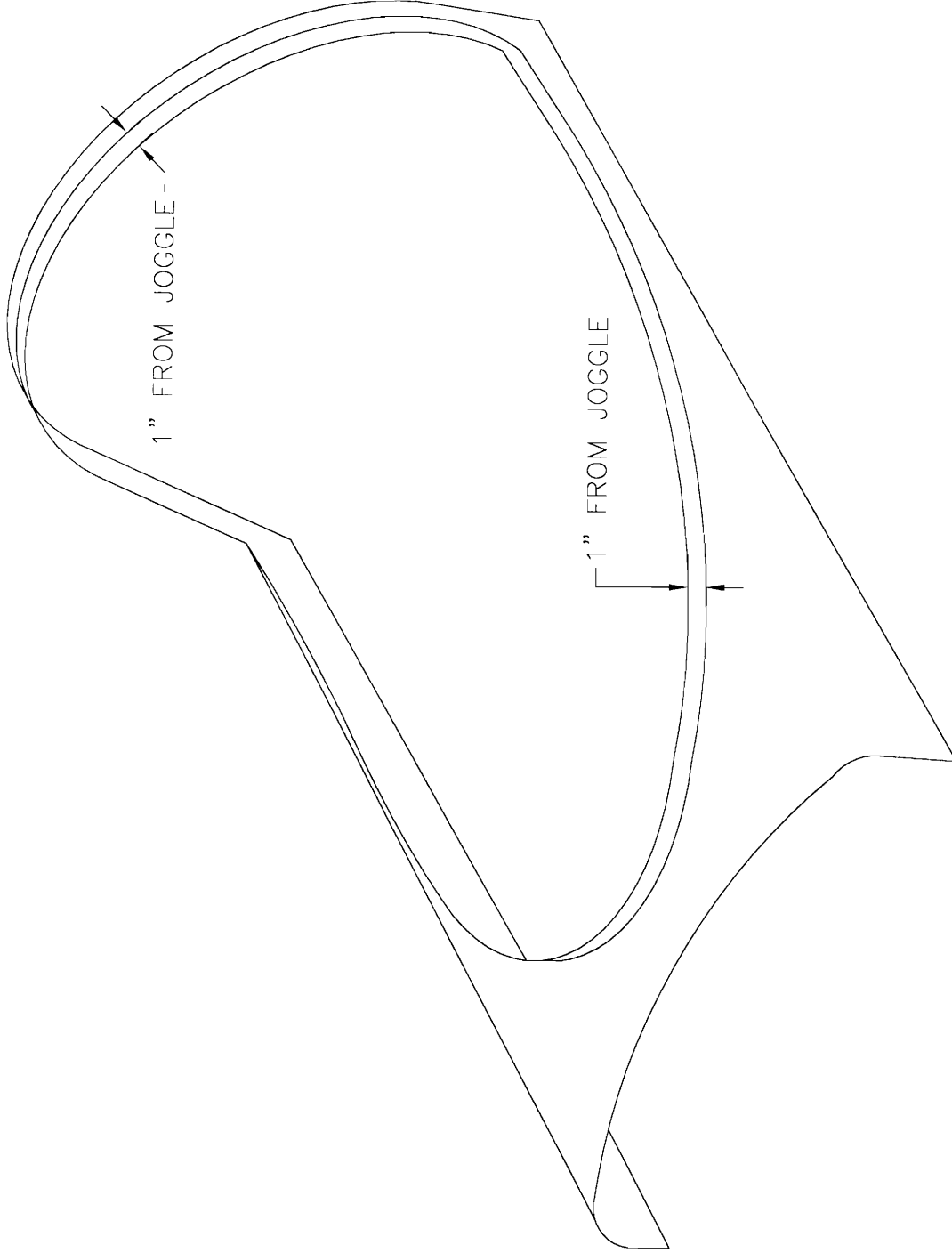


FIGURE 7-4

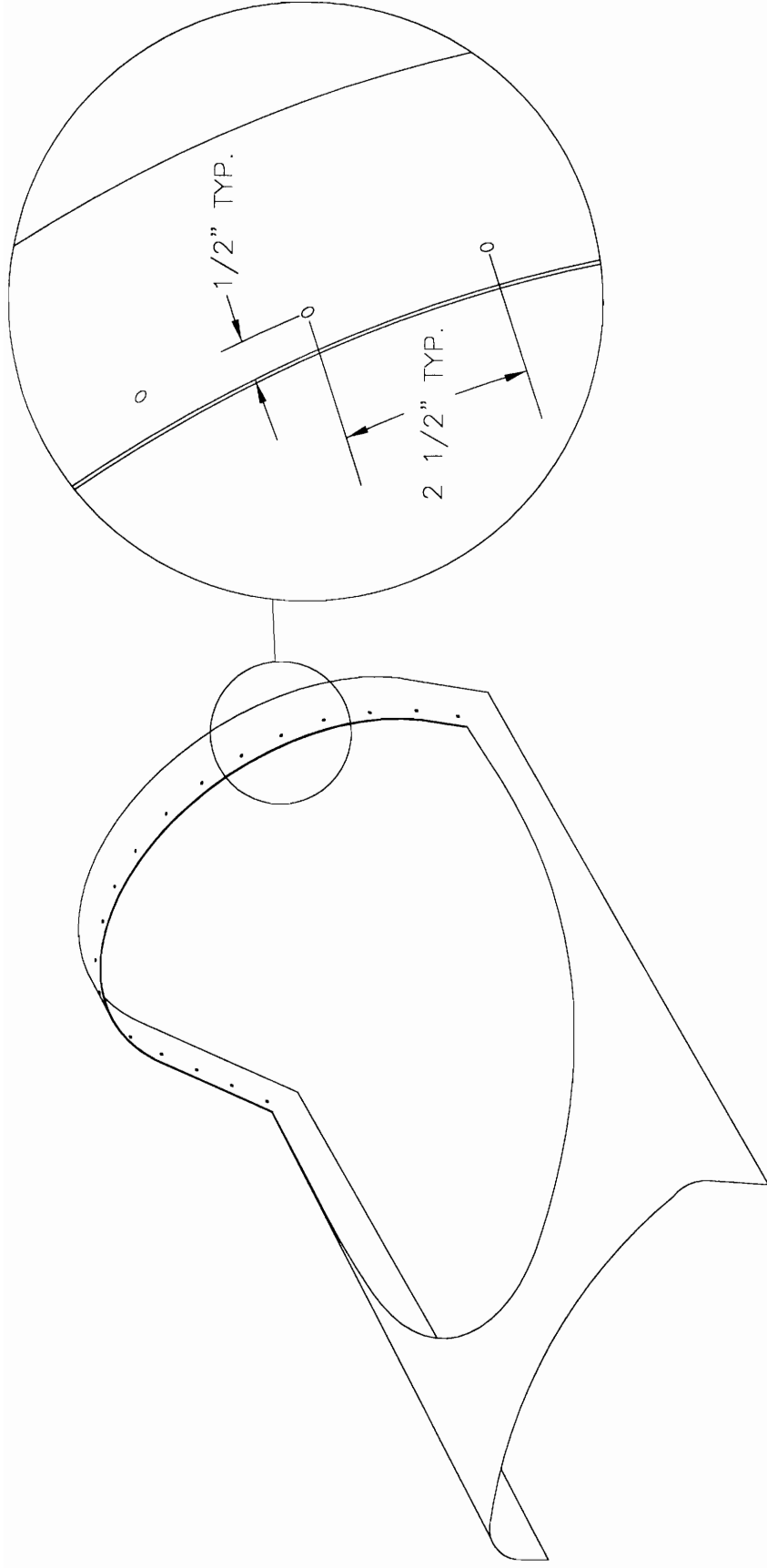


FIGURE 7-5

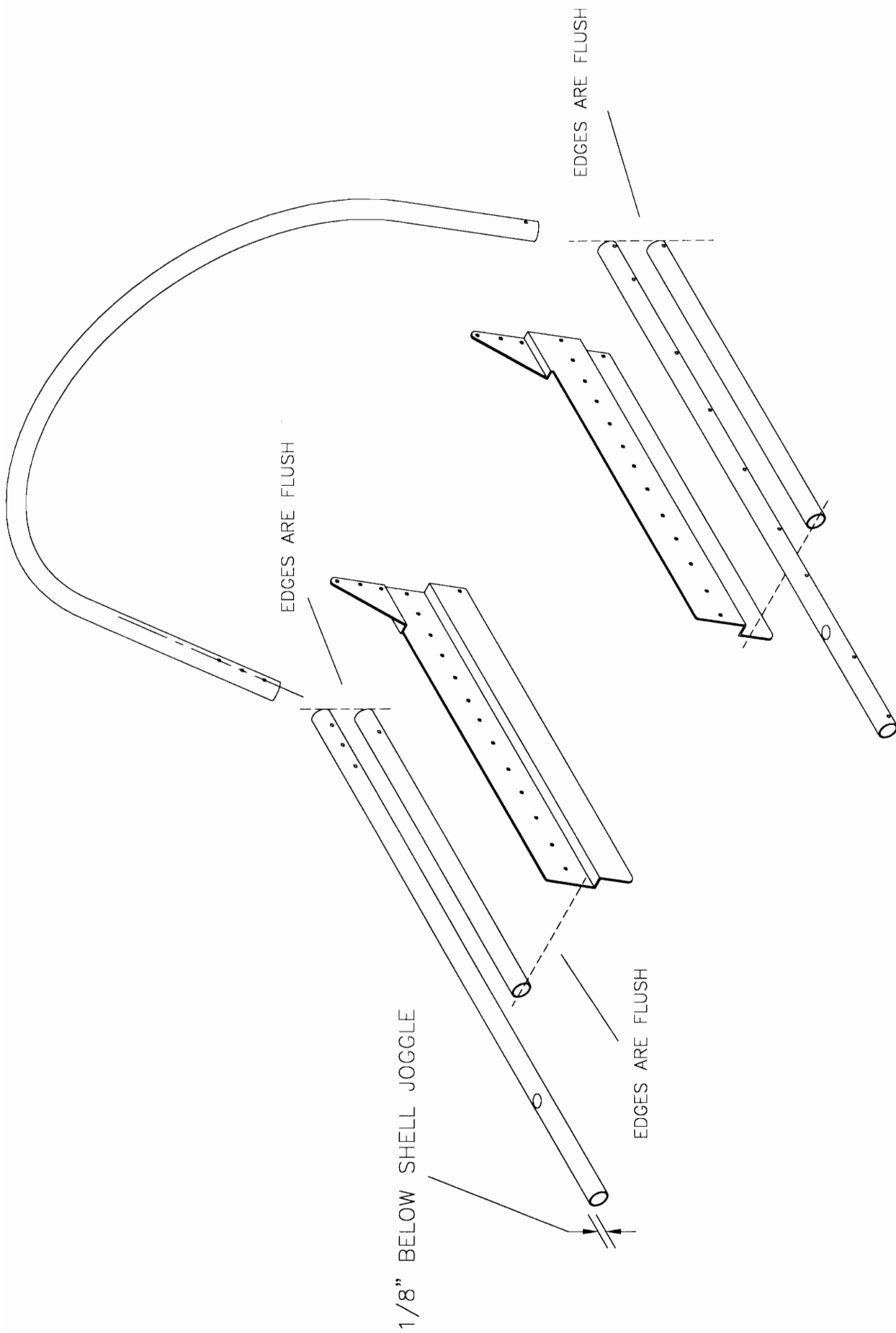


FIGURE 7-6

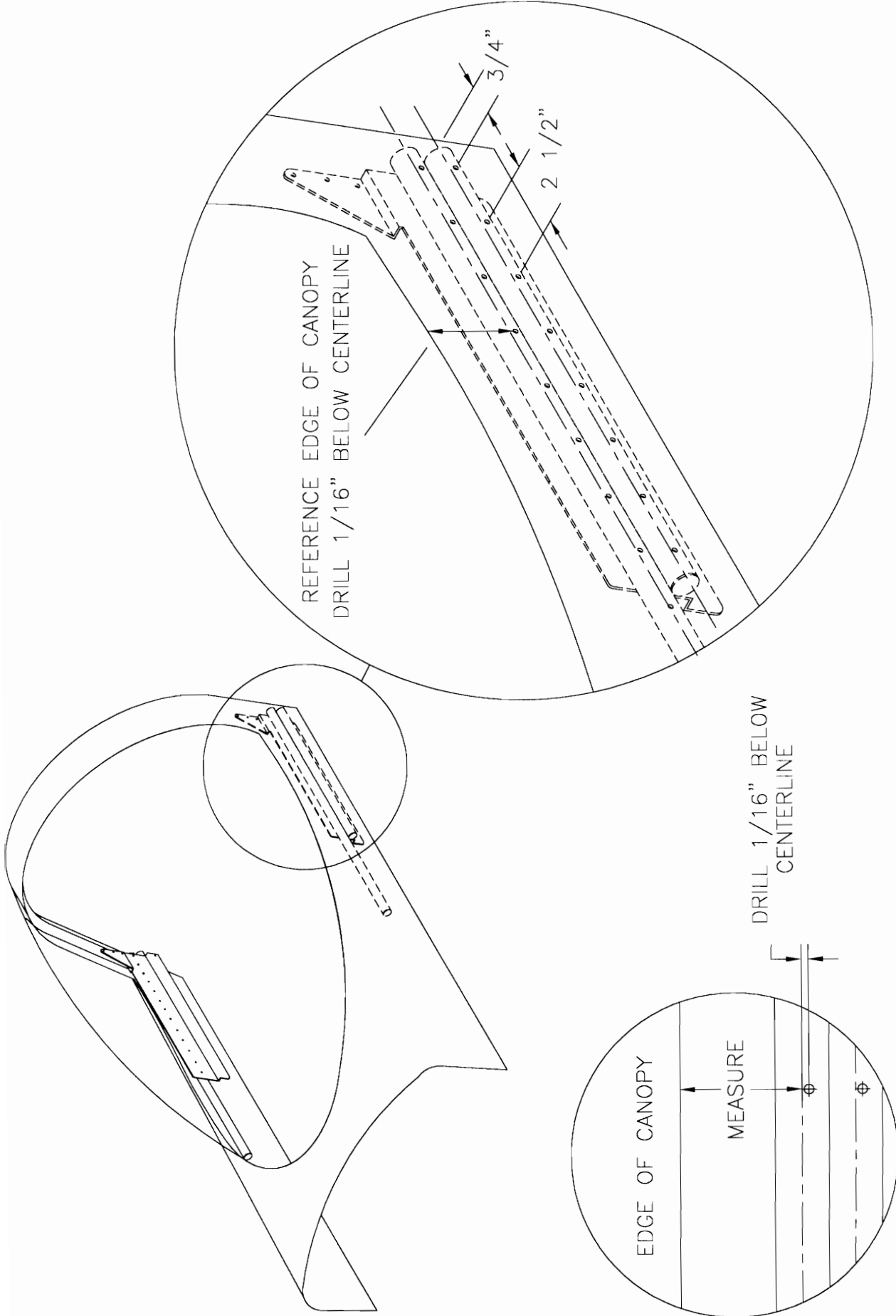


FIGURE 7-7

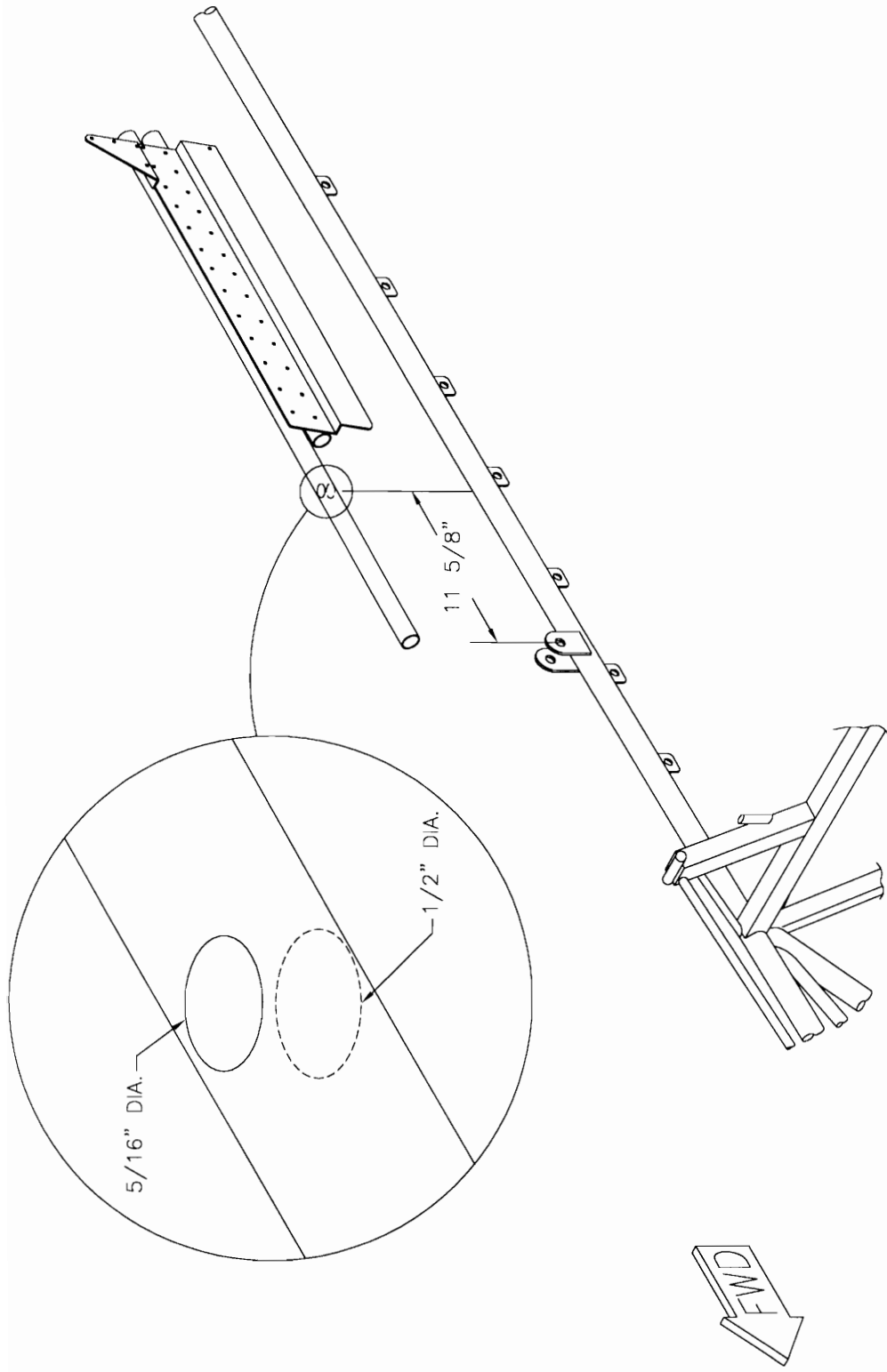


FIGURE 7-8

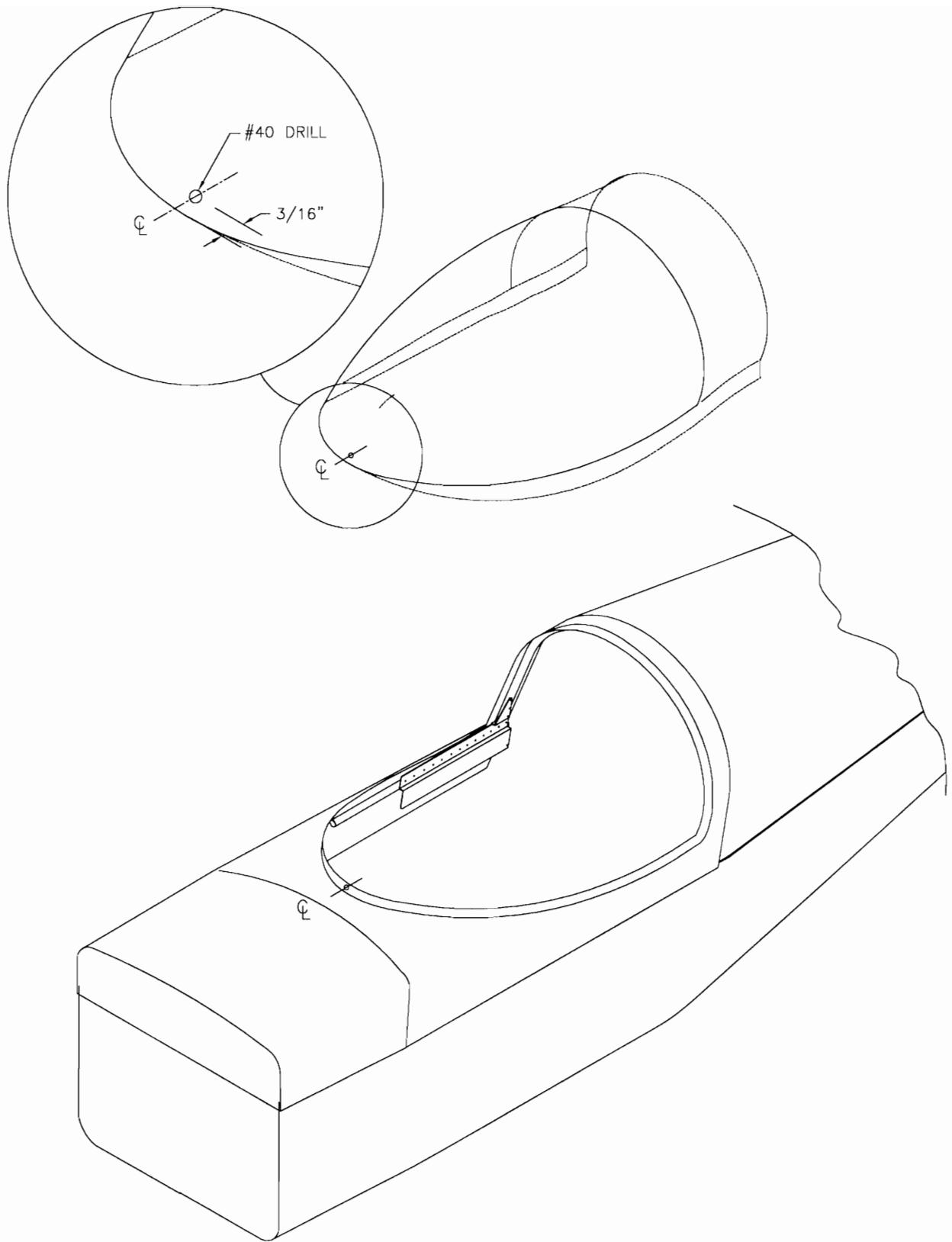


FIGURE 7-9

CUT OUT  $3/16$ " LARGER THAN ATTACH ANGLES

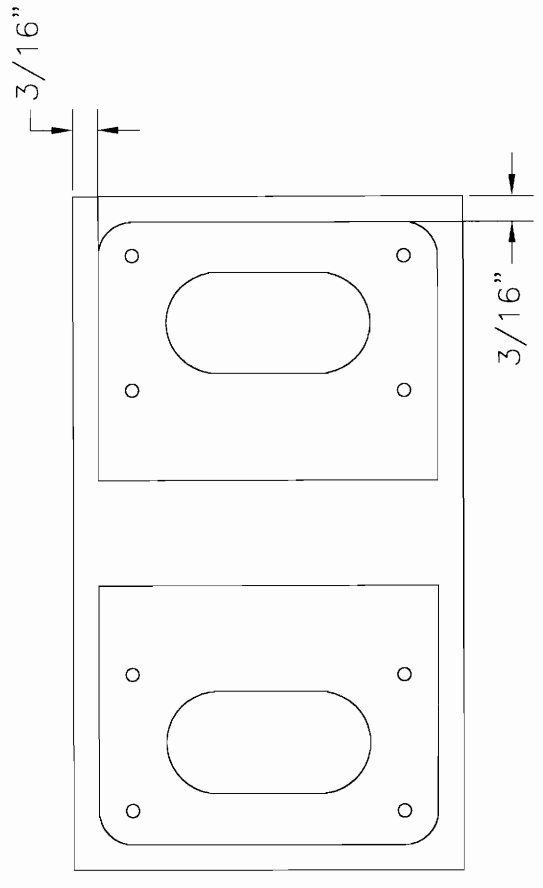
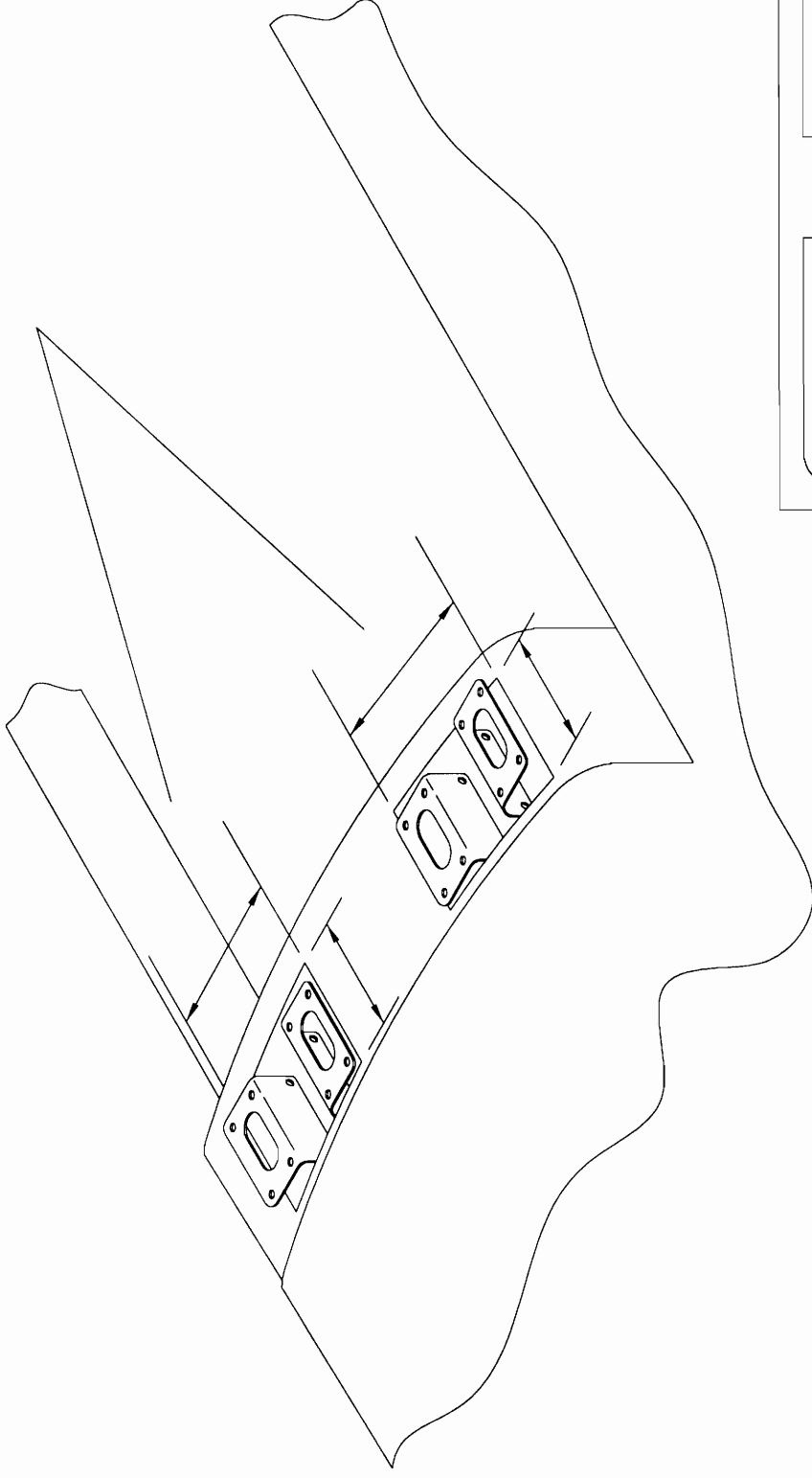


FIGURE 7-10

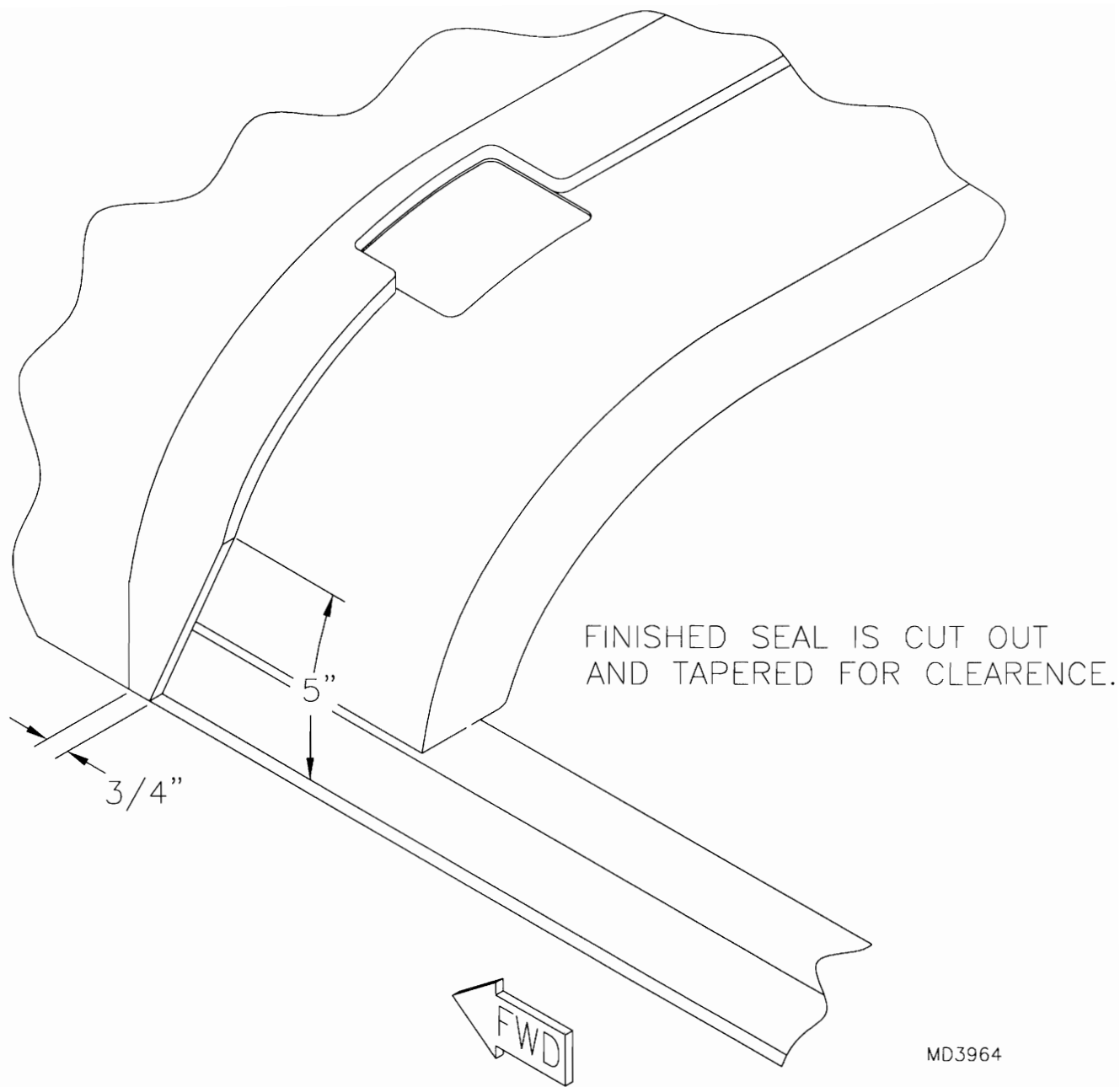


FIGURE 7-11



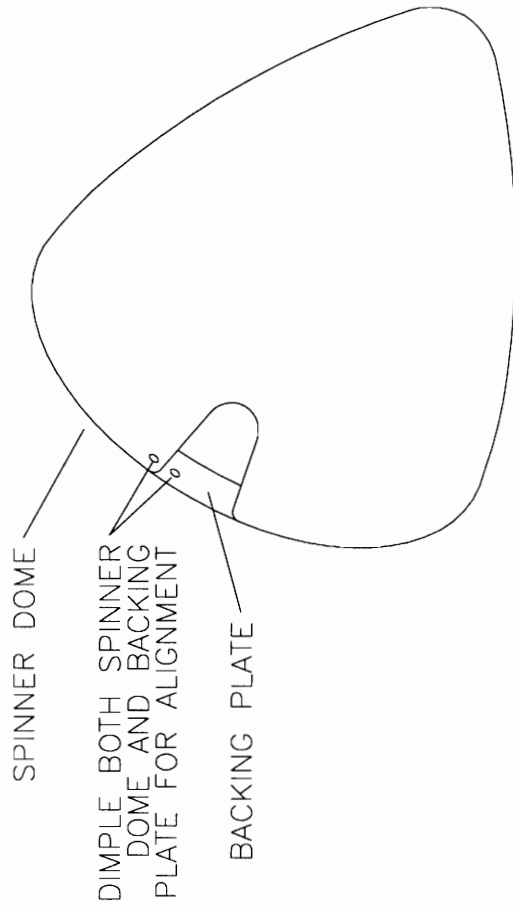
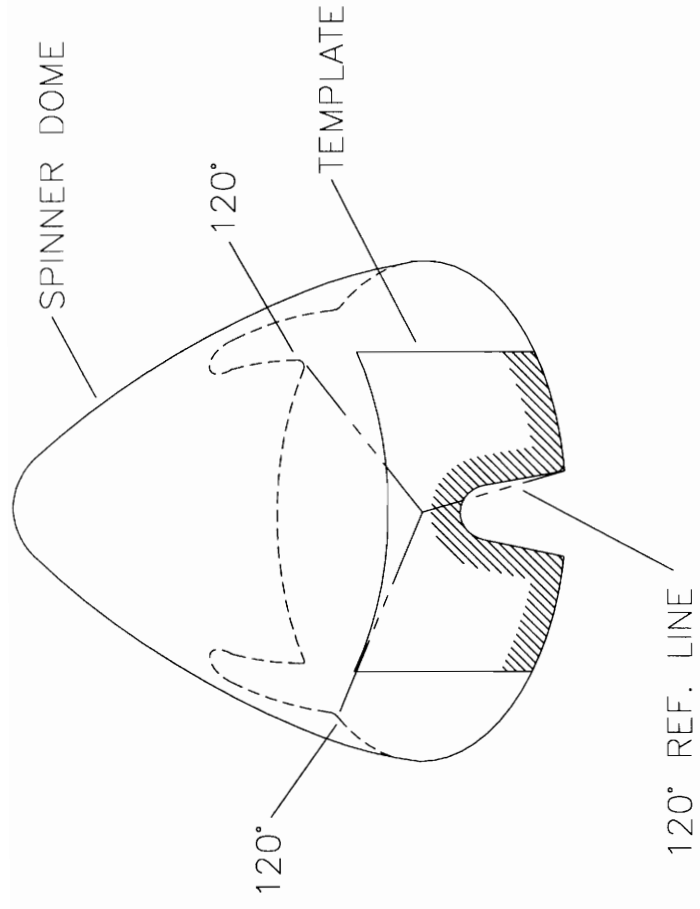
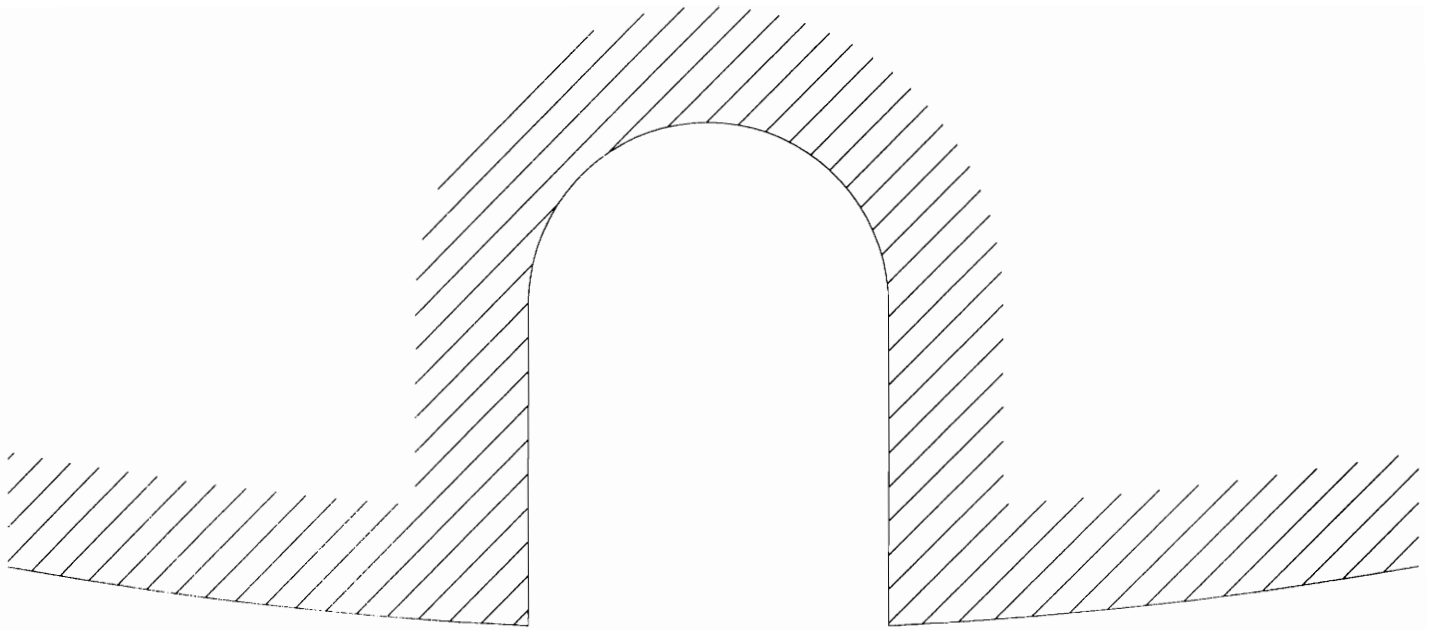
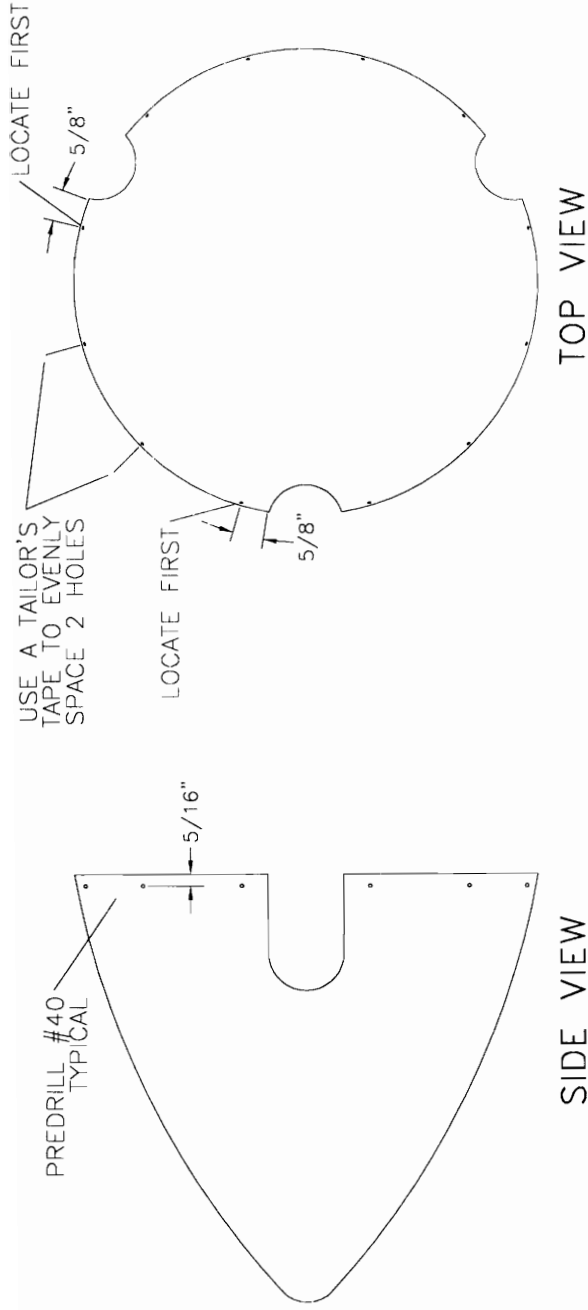


FIGURE 8-1



WARP DRIVE  
13" SPINNER TEMPLATE

FIGURE 8-2



THREE BLADE PROP HOLE SPACING

FIGURE 8-3

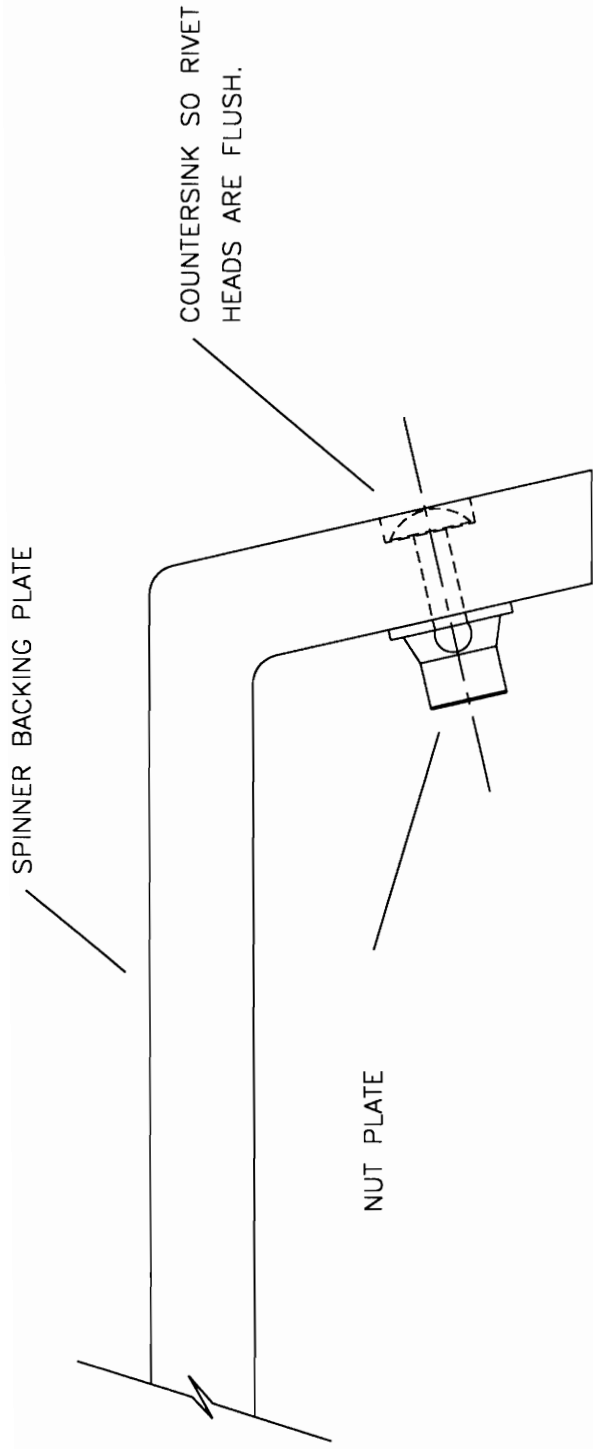


FIGURE 8-4



FIGURE 8-5

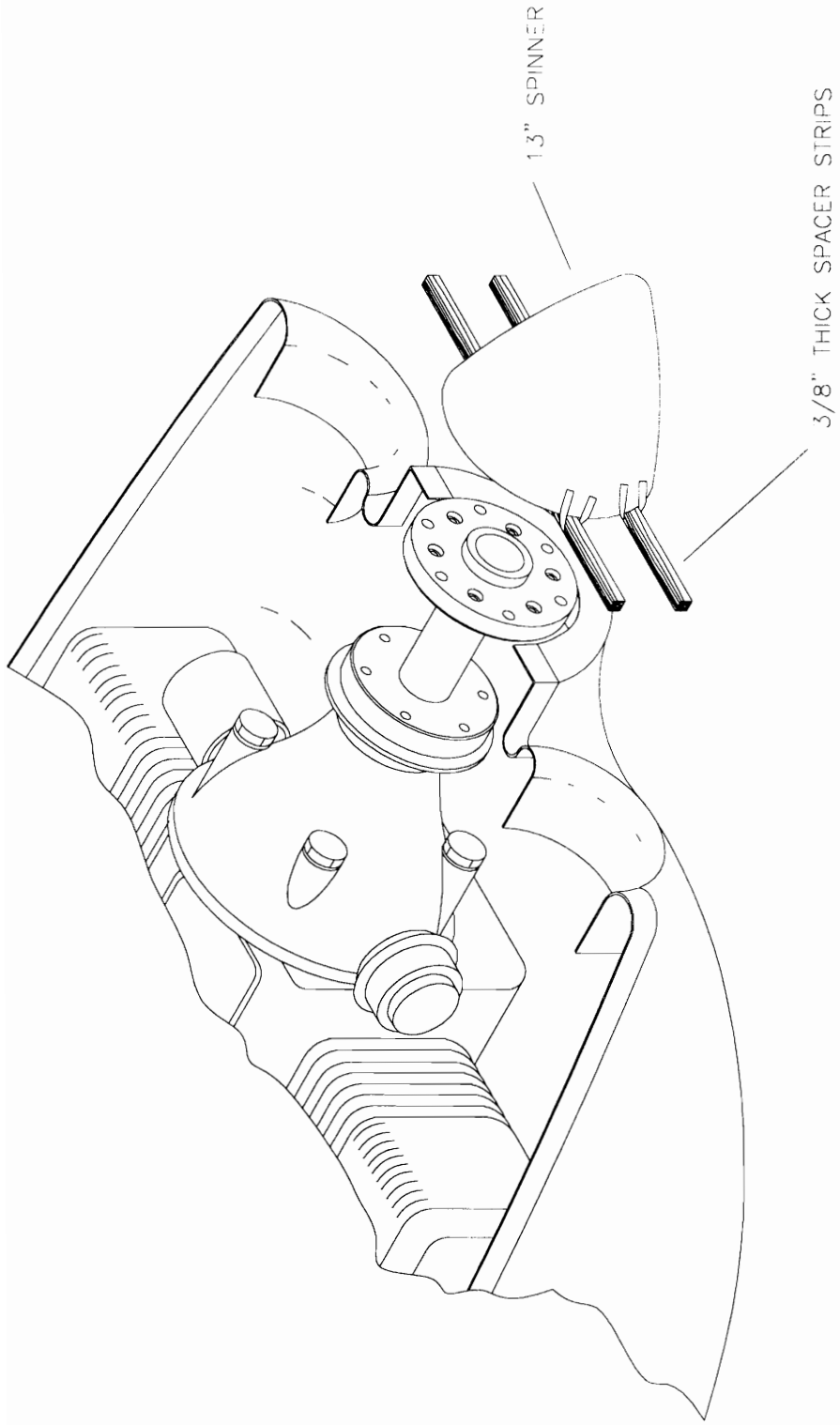


FIGURE 8-6

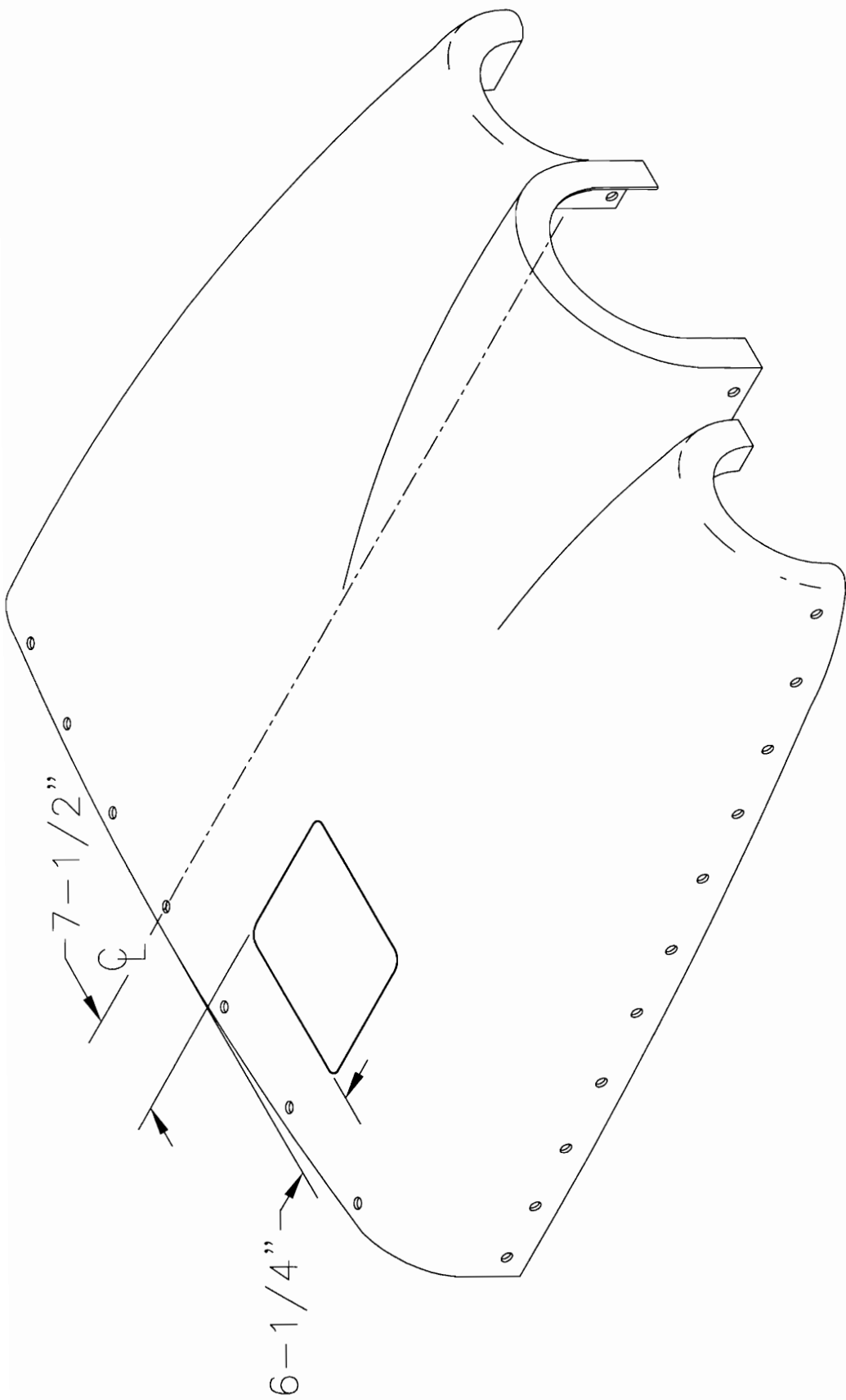


FIGURE 8-6A  
 S-16 SHEKARI

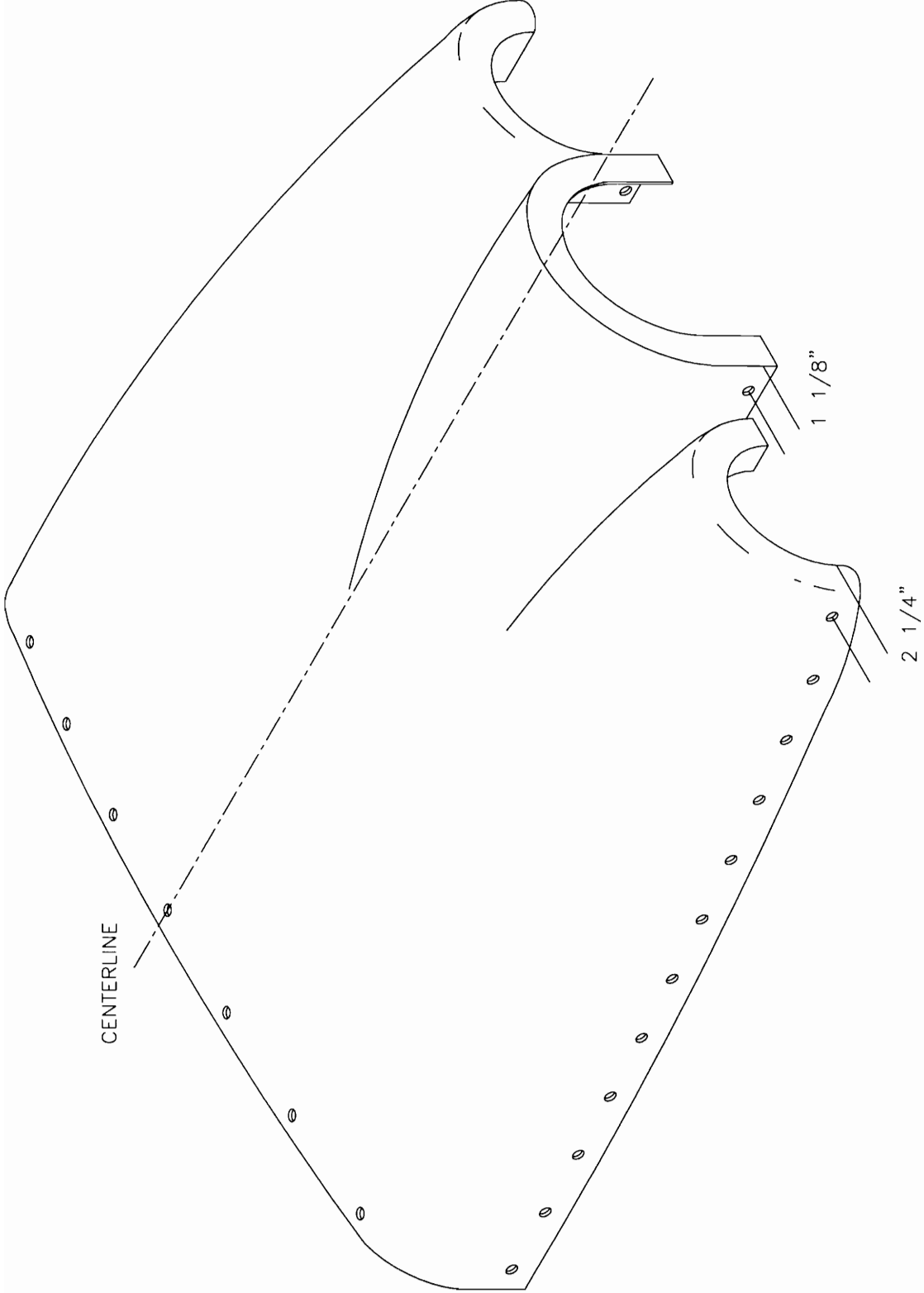


FIGURE 8--7



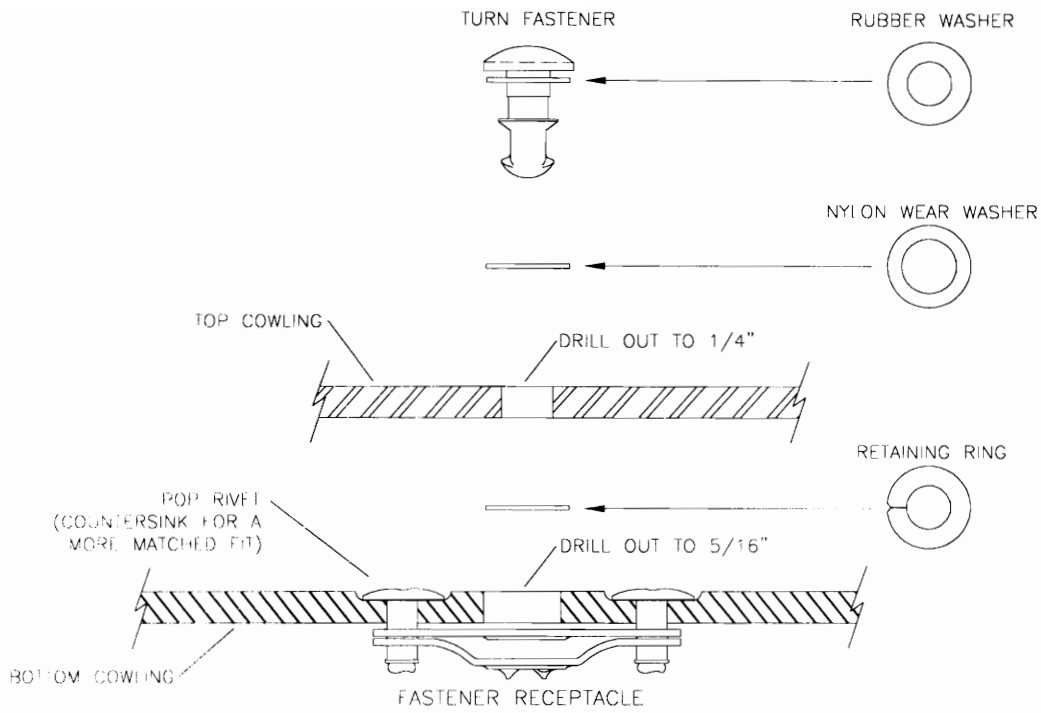
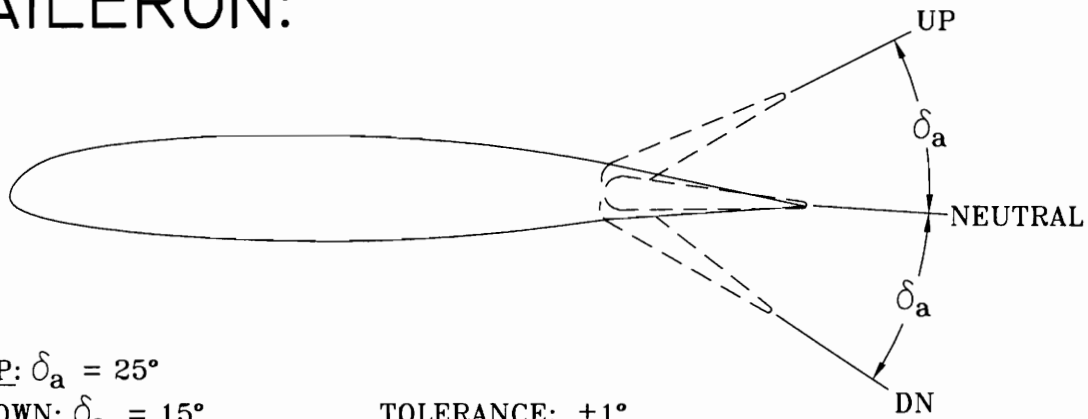
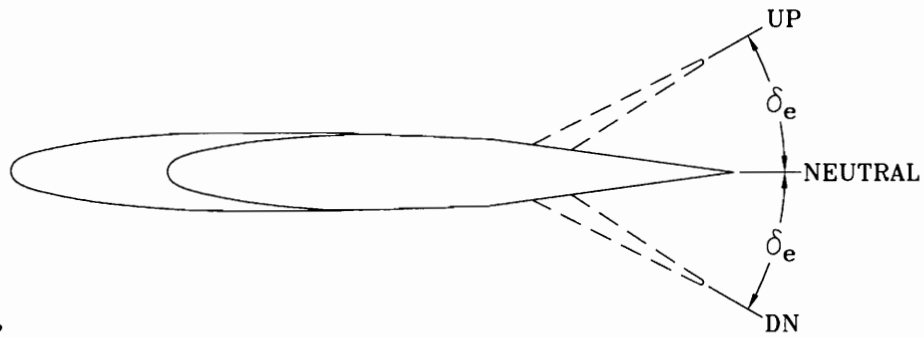


FIGURE 8-8

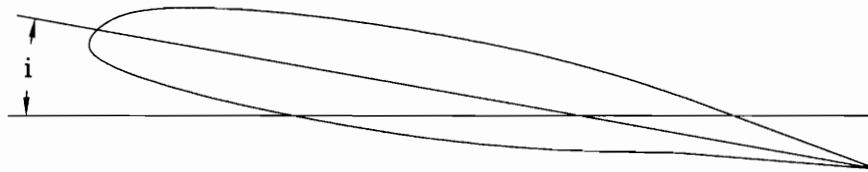
# AILERON:



# ELEVATOR:



# WING INCIDENCE:



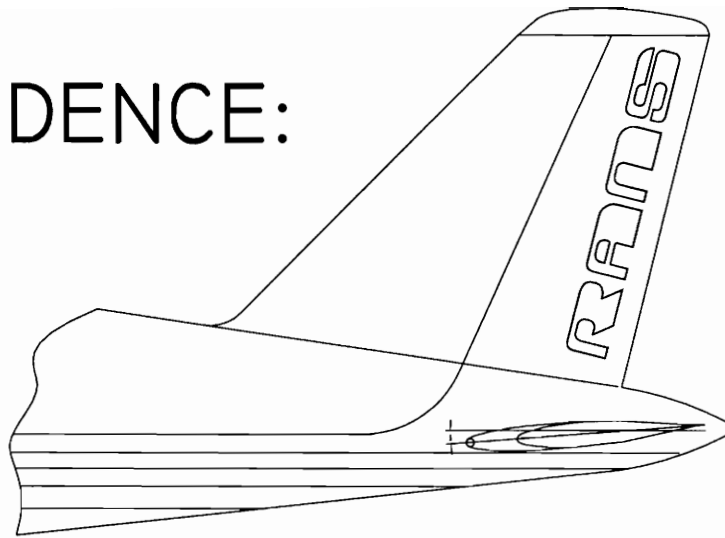
$i = 1^\circ$

FIGURE 9-1  
(PAGE 1 OF 5)  
S-16 SHEKARI

MD3450

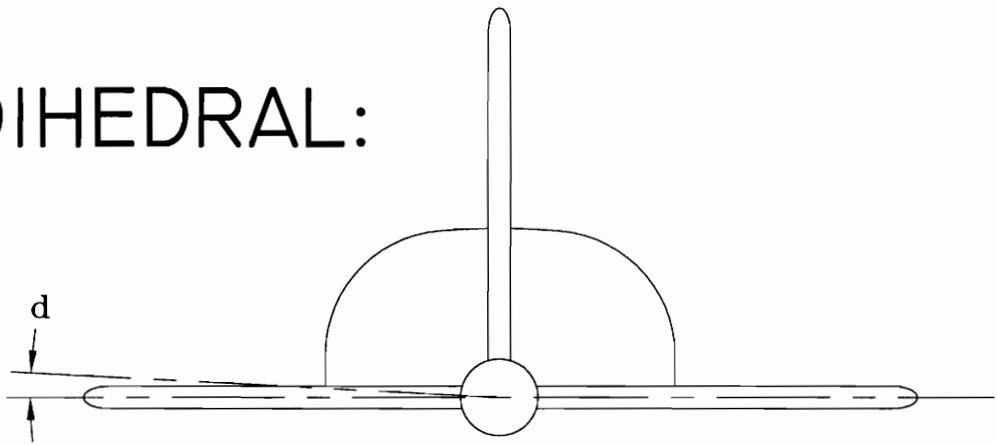
# TAIL INCIDENCE:

$i: = -1^\circ$



# TAIL DIHEDRAL:

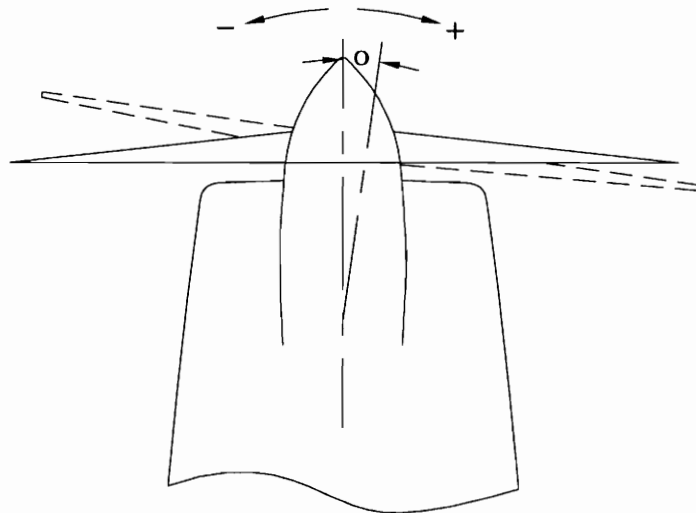
$d: = 0^\circ$



# ENGINE (TRACTOR):

$= +0.0^\circ$  (IO-240B)

FIGURE 9-1  
(PAGE 2 OF 5)  
S-16 SHEKARI



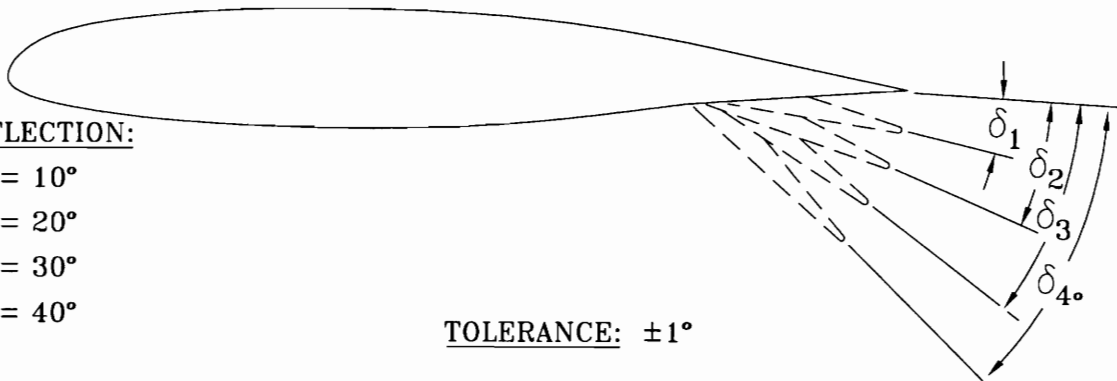
MD3450

# FLAP:

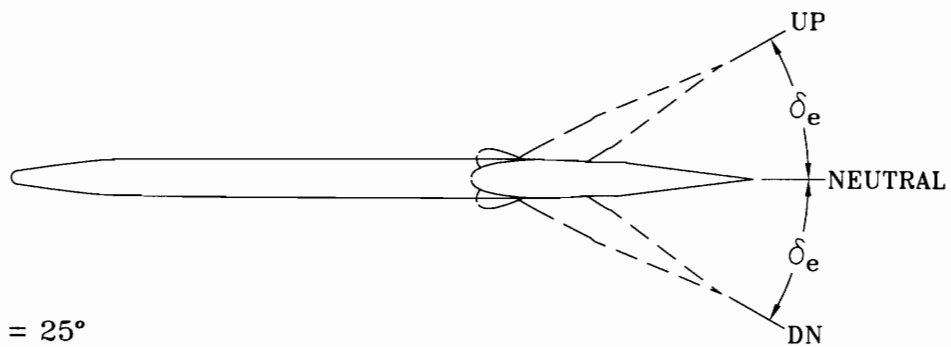
DEFLECTION:

- $\delta_1 = 10^\circ$
- $\delta_2 = 20^\circ$
- $\delta_3 = 30^\circ$
- $\delta_4 = 40^\circ$

TOLERANCE:  $\pm 1^\circ$

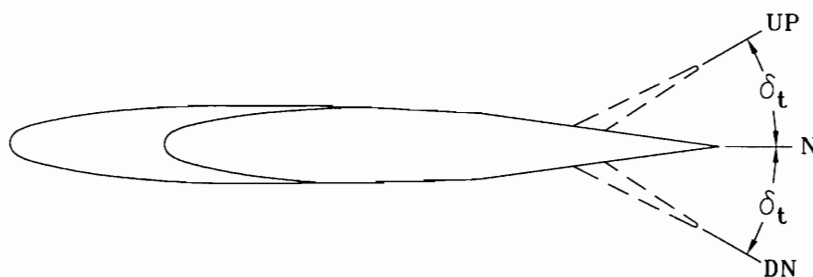


# RUDDER:



- RIGHT:  $\delta_r = 25^\circ$
- LEFT:  $\delta_r = 25^\circ$

# TRIM TAB:

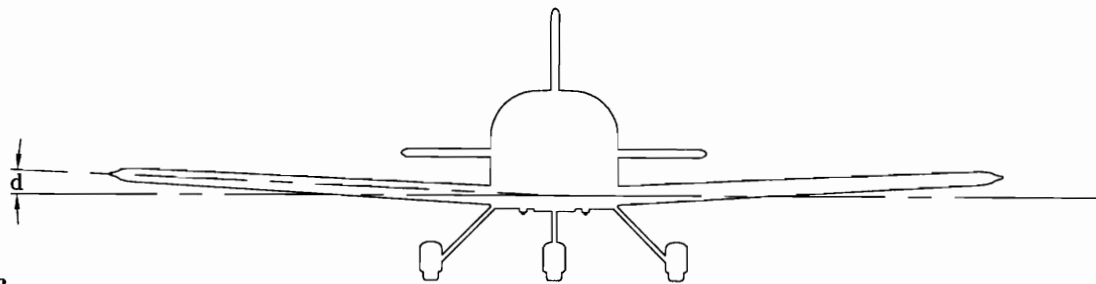


- UP:  $\delta_t = N/A$
- DOWN:  $\delta_t = N/A$

MD3450

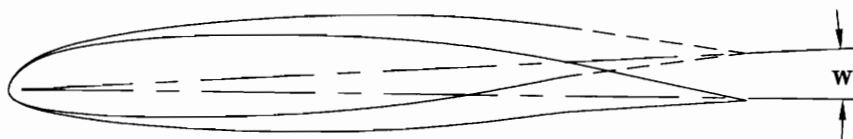
FIGURE 9-1  
(PAGE 3 OF 5)  
S-16 SHEKARI

# WING DIHEDRAL:



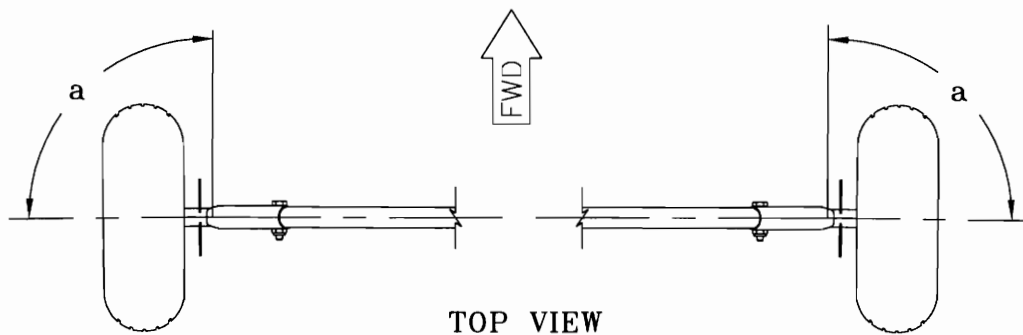
$d = 3^\circ$

# WING WASH-OUT:



$w = 0.5^\circ$

# LANDING GEAR TOE-OUT:



$a = 90.0^\circ$

MD3450

FIGURE 9-1  
(PAGE 4 OF 5)  
S-16 SHEKARI

# THRUST LINE

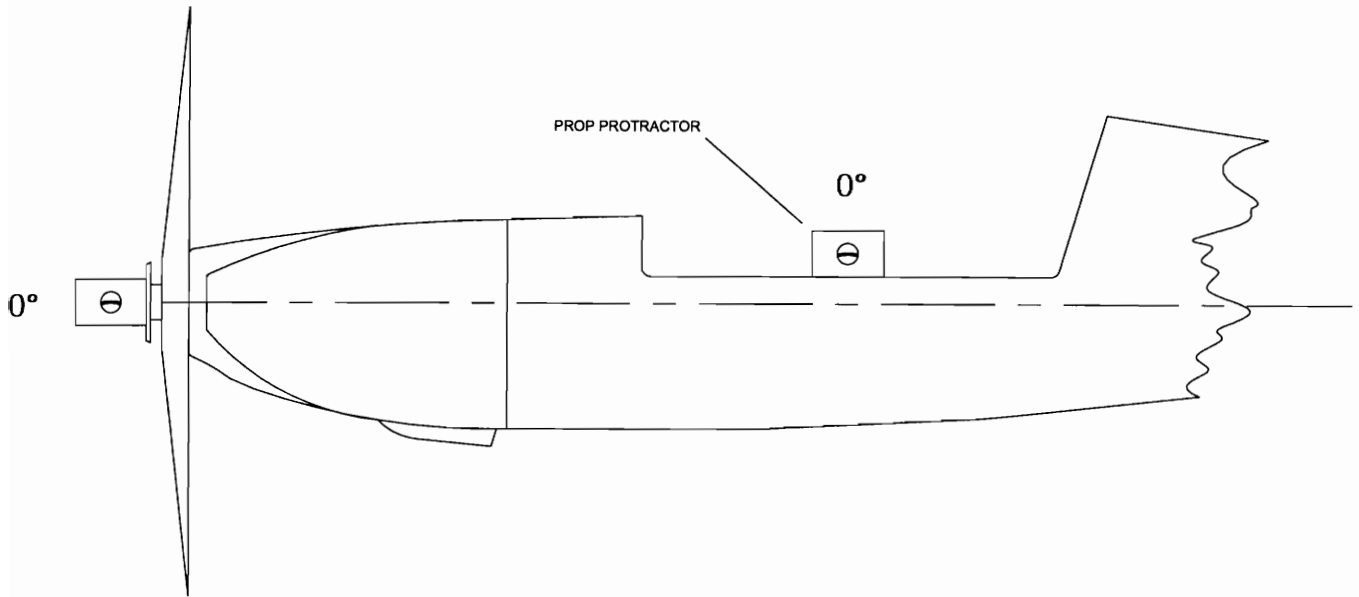


FIGURE 9-1  
(PAGE 5 OF 5)  
S-16 SHEKARI

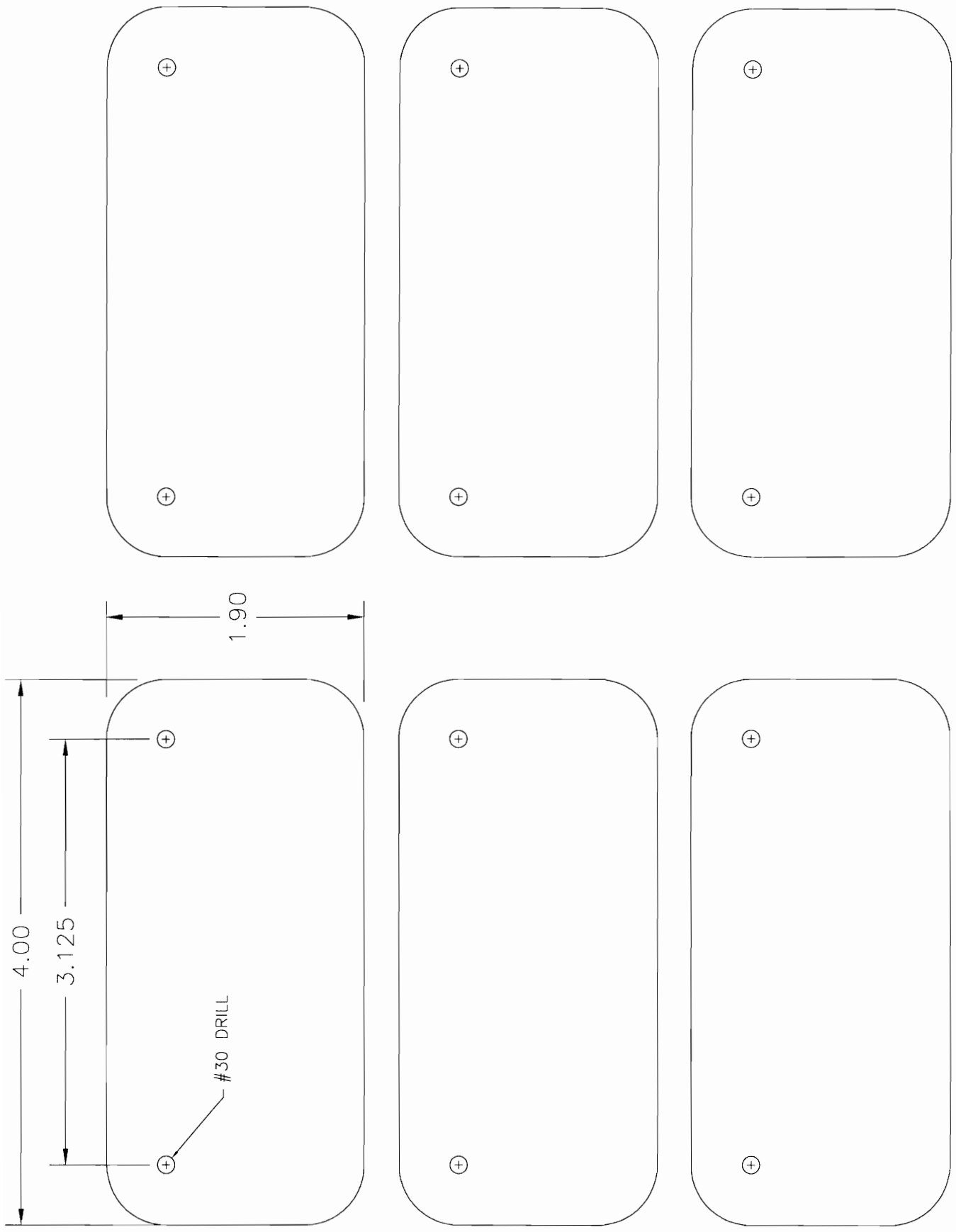


FIGURE 9-2

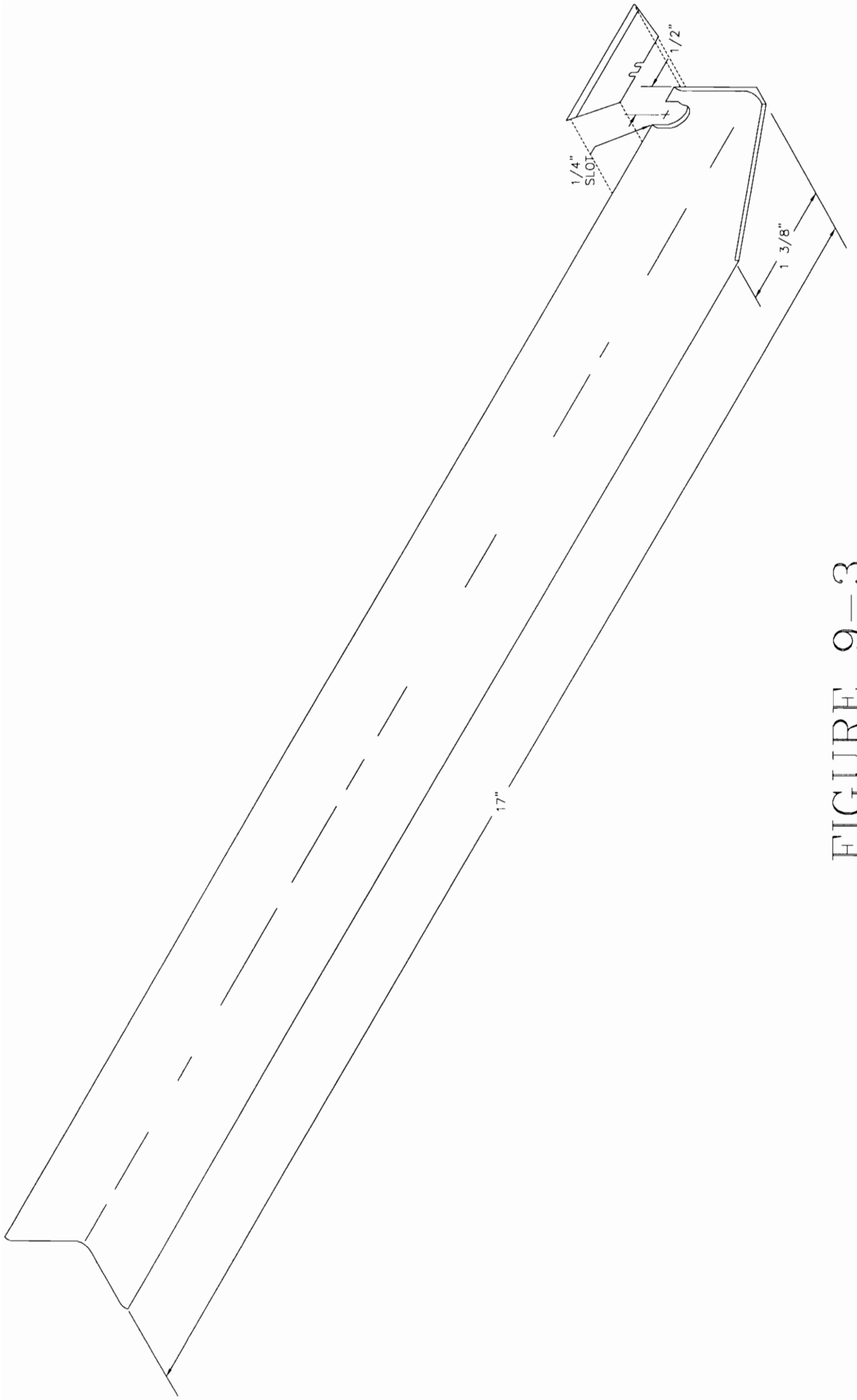


FIGURE 9-3



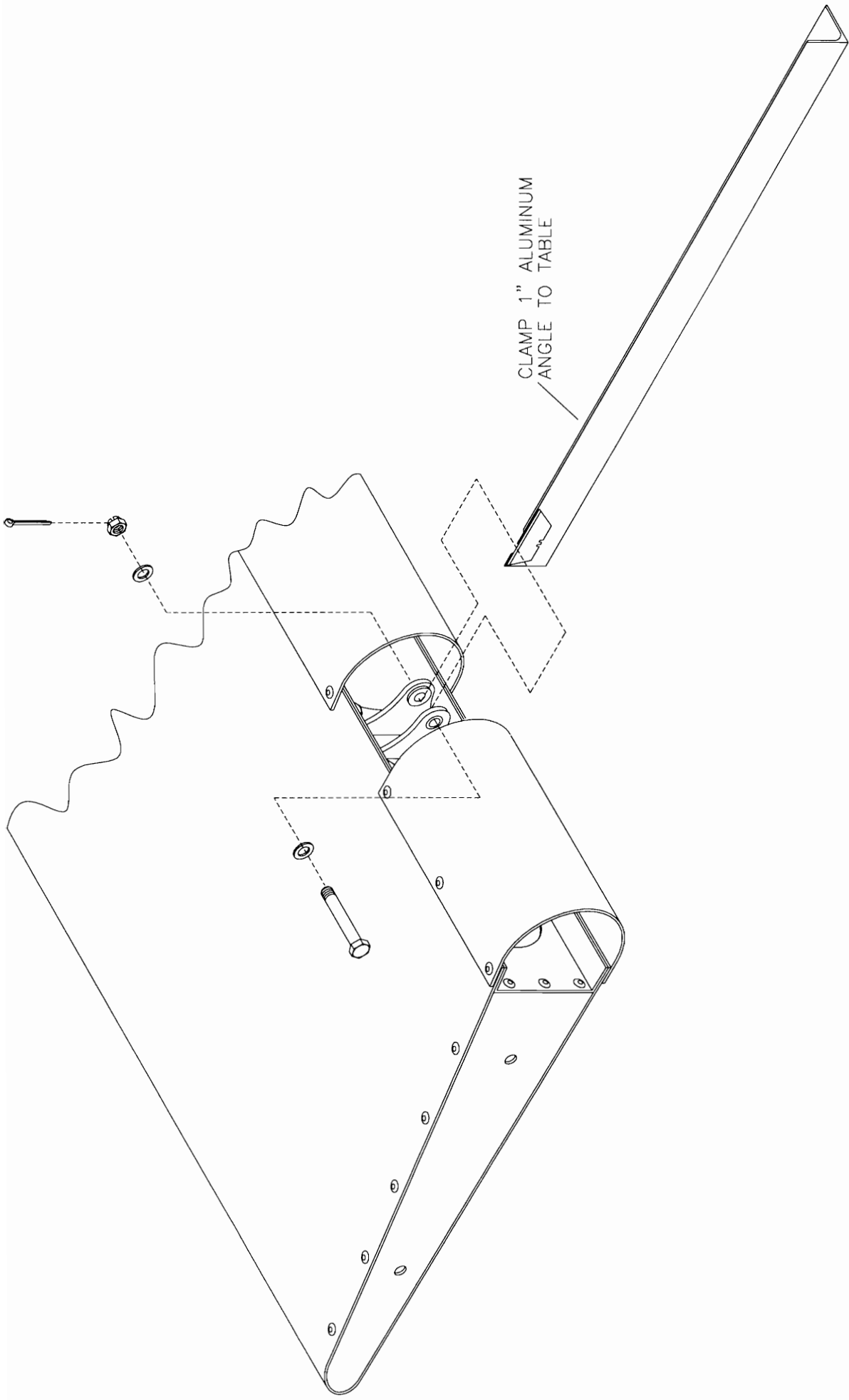


FIGURE 9-4

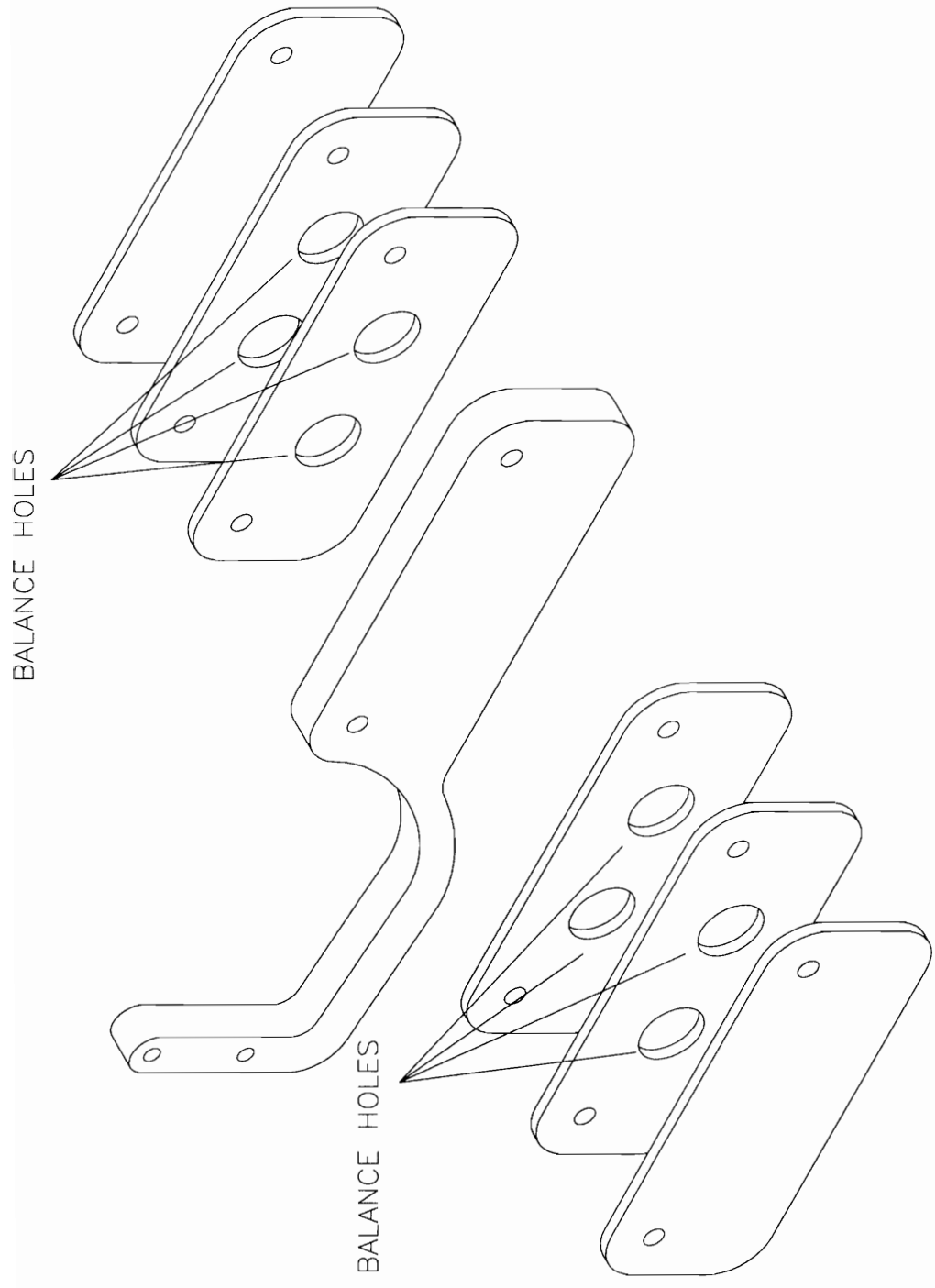
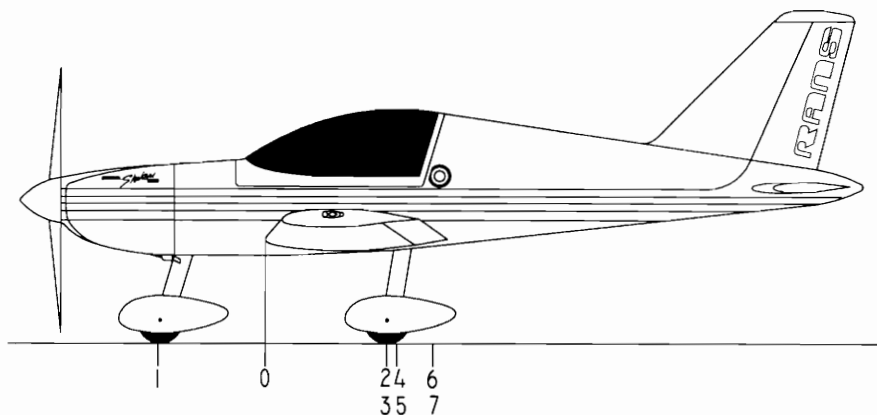


FIGURE 9-5



N _____	
DATE WEIGHED	
ENGINE TYPE	10-240B
C.G. CONDITION	
EMPTY WEIGHT	
GROSS WT.	1450 LBS.

### RAMS S-16 SHEKARI WEIGHT AND BALANCE

ACCEPTABLE C.G. 8.7" TO 15.6" FROM DATUM 0.  
DATUM = LEADING EDGE OF WING AT FUSELAGE INTERSECTION  
LEVEL ATTITUDE. (LEVEL REFERENCE TO  
CANOPY UPPER LONGERON)

#	ITEM	WEIGHT	ARM	MOMENT
1	NOSEWHEEL	353	-27.25	-9619
2	MAIN LH	301	24	7224
3	MAIN RH	293	24	7032
4	PILOT	185	24.5	4532
5	PASSENGER	185	24.5	4532
6	FUEL	100	45	4500
7	BAGGAGE	30	45	1350
TOTAL=		1447	TOTAL=	19551

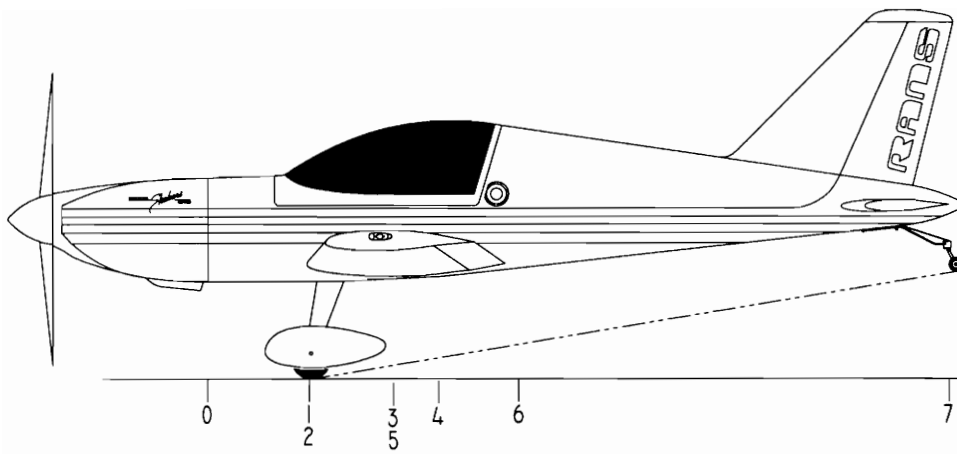
$$\frac{\text{TOTAL MOMENTS}}{\text{TOTAL WEIGHT}} = \text{C.G.} \quad \frac{19551}{1447} = 13.51 \quad \frac{(\text{CG}-1.7)}{44.3} * 100 = \% \text{MAC}$$

#	ITEM	WEIGHT	ARM	MOMENT
1	NOSEWHEEL		-27.25	(-)
2	MAIN LH		24	
3	MAIN RH		24	
4	PILOT		24.5	
5	PASSENGER		24.5	
6	FUEL		45	
7	BAGGAGE		45	
TOTAL=			TOTAL=	

$$\frac{\text{TOTAL MOMENTS}}{\text{TOTAL WEIGHT}} = \text{C.G.} \quad \text{_____} =$$

\* 50 LBS. MAXIMUM BAGGAGE

FIGURE 11-1



N _____	
DATE WEIGHED	
ENGINE TYPE	10-240B
C.G. CONDITION	
EMPTY WEIGHT	
GROSS WT.	1500 LBS.
ACRO WT.	1300 LBS.

### RANS S-16A SHEKARI WEIGHT AND BALANCE

ACCEPTABLE C.G. FOR AEROBATICS IS 32.2" TO 38.0" FROM DATUM O.  
ACCEPTABLE C.G. FOR NORMAL CAT. IS 32.2" TO 40.90" FROM DATUM O.

DATUM = FIREWALL  
LEVEL ATTITUDE. (LEVEL REFERENCE TO  
CANOPY UPPER LONGERON)

#	ITEM	WEIGHT	ARM	MOMENT
1	MAIN RH	467	26.8	12515
2	MAIN LH	462	26.8	12381
3	PILOT	170	48.5	8245
4	FUEL	170	69.1	11747
5	PASSENGER	170	48.5	8245
6	BAGGAGE*	50	69.1	3455
7	TAIL WHEEL	11	183	2013
TOTAL=		1500	TOTAL=	58601

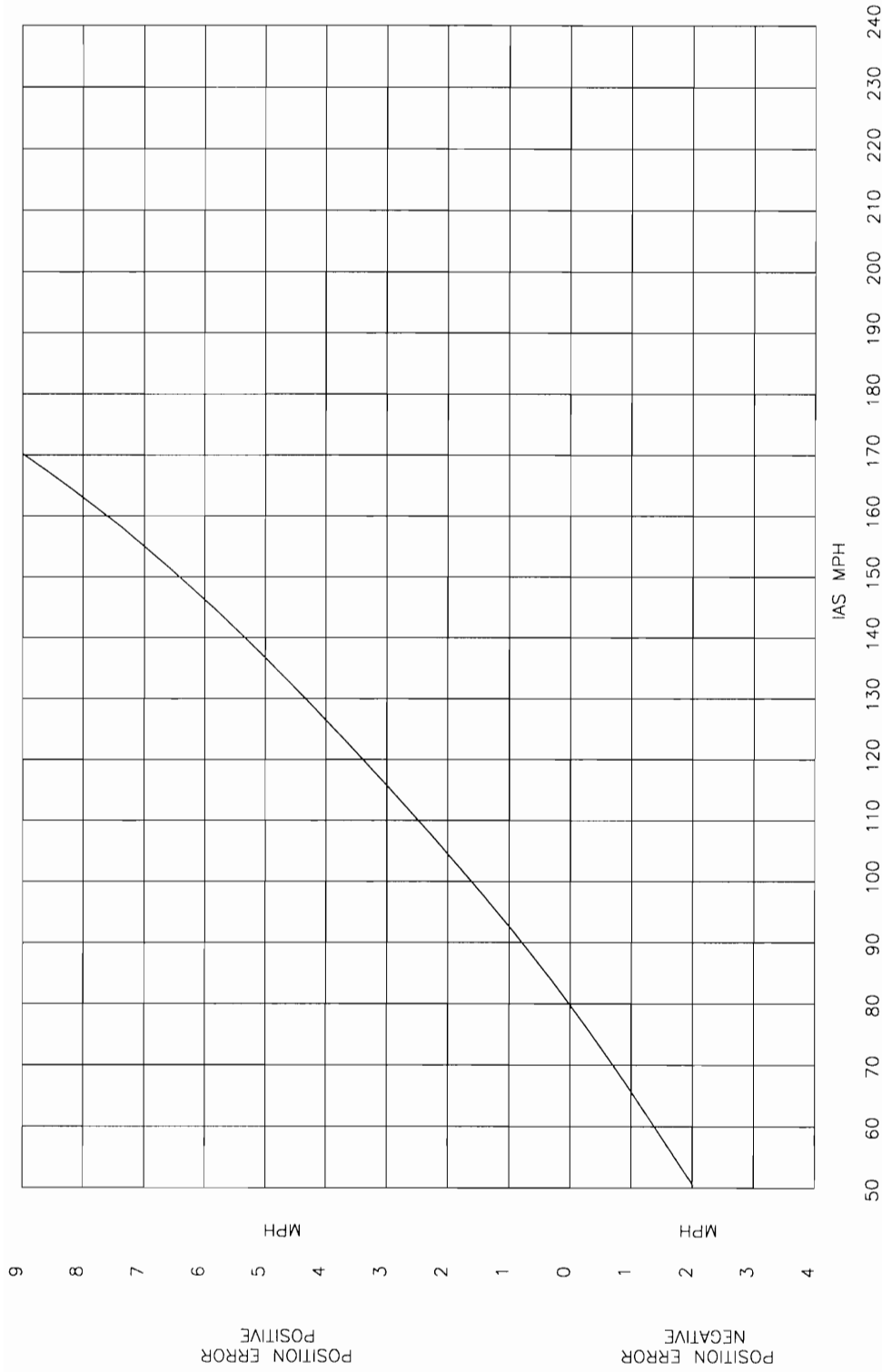
$$\frac{\text{TOTAL MOMENTS}}{\text{TOTAL WEIGHT}} = \text{C.G.} \quad \frac{58601}{1500} = 39.0"$$

#	ITEM	WEIGHT	ARM	MOMENT
1	MAIN RH		26.8	
2	MAIN LH		26.8	
3	PILOT		48.5	
4	FUEL		69.1	
5	PASSENGER		48.5	
6	BAGGAGE*		69.1	
7	TAIL WHEEL		183	
TOTAL=			TOTAL=	

$$\frac{\text{TOTAL MOMENTS}}{\text{TOTAL WEIGHT}} = \text{C.G.} \quad \text{_____} =$$

\* 50 LBS. MAXIMUM BAGGAGE

FIGURE 11-2



S-16 AIRSPEED INDICATOR CALIBRATION  
 ALL SPEEDS IN MPH  
 DEC 15 98

$CAS = IAS - POSITION ERROR$

EF-97 REV N/A  
 12/17/98

FIGURE 11-3

V<sub>n</sub> DIAGRAM BASED ON IAS

WHITE ARC	56 MPH – 80 MPH	FLAPS 40°
BLUE ARC	80MPH – 90 MPH	FLAPS 0° – 30°
GREEN ARC	64 MPH – 140 MPH	(CONSERVATIVE)
YELLOW ARC	140 MPH – 220 MPH	
RED LINE	220 MPH *	*(PENDING DIVE AND FLUTTER TESTING) *(CURRENTLY 180 MPH MAX. IAS ALLOWABLE)

OPERATING LIMITS

MAXIMUM TAKEOFF WEIGHT	1500 lbs.	
MAXIMUM AEROBATIC WEIGHT	1300 lbs.	GREEN ARC BASED ON ACRO WEIGHT OF 1300 lbs
MAXIMUM LANDING WEIGHT	1450 lbs.	

ERECT AND INVERTED SPINS – PROHIBITED

NEGATIVE G WITH 10–240 – NOT RECOMMENDED

FLIGHT WITH CANOPY OPEN – PROHIBITED